## REFERENCE DATA

## fou <br> RADIO EMGINEERS

82

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## REFERENCE DATA

## RADIO ENGINEERS

second edition

# Federal Telephone and Radio Corporation <br> an associate of 

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## Foreword

Widespread acceptance of the four printings of the first edition of Reference Data for Radio Engineers prompted this larger and improved second edition. Like its predecessor, it is presented by the Federal Telephone and Radio Corporation as an aid in the fields of research, development, production, operation, and education. In it will be found all the material that proved so useful in the first edition along with much additional data-some the result of helpful suggestions from readers, others stemming from rapid advances in the art, and still others now made possible by declassification of many war developments.

While the general arrangement remains unchanged, the present edition has been greatly enlarged and a subject index included. Chapters on transformers and room acoustics have been added. The material on radio propagation and radio noise has been revised. Because of their importance in television, in radar, and in laboratory technique, the data on cathode-ray tubes have been considerably expanded.

The section on electrical circuit formulas has been greatly enlarged; additions include formulas on $\mathrm{T}-\Pi$ and Y - $\Delta$ transformations, amplitude modulation, transients, and curves and numerous formulas on selective circuits. The attenuator section contains comprehensive design formulas and tables for various types of attenuators. The number of mathematical formulas also has been considerably increased.

As revised, the wave-guide chapter includes equations for both rectangular and cylindrical guides plus illustrations of field distribution patterns. Several methods of coupling to the $\mathrm{TE}_{0,1}$ mode are illustrated. A table of standard rectangular wave guides and connectors, giving useful frequency range and attenuation, has been added. Design curves for the gain and beam width of rectangular electromagnetic horn radiators are included, and a simple formula for the gain of a paraboloid reflector is given.

Many very helpful suggestions were received from the Armed Services.
Acknowledgment is made to Edward J. Content, consulting engineer, for his contribution of the chapter on room acoustics; its inclusion was made possible largely through the courtesy of the Western Electric Company in permitting the use of their engineering data. Acknowledgment also is due to I. E. Lempert, Allen B. Dumont Laboratories, Inc., for the descriptive material on cathode-ray tubes; and to Professor L. Brillouin of Harvard University for advice and suggestions on the wave-guide chapter.

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## General information

Conversion factors

| to convert | Into | multiply by | conversely multiply by |
| :---: | :---: | :---: | :---: |
| Acres | Square feel | $4.356 \times 10^{4}$ | $2.296 \times 10^{-6}$ |
| Acres | Square meters | 4,047 | $2.471 \times 10^{-4}$ |
| Ampere-hours | Coulomb | 3,600 | $2.778 \times 10^{-6}$ |
| Amperes per sq cm | Amperes per sq inch | 6.452 | 0.1550 |
| Ampere turns | Gilberts | 1.257 | 0.7958 |
| Ampere furns per cm | Ampere turns per inch | 2.540 | 0.3937 |
| Atmospheres | Mm of mercury @ $0^{\circ} \mathrm{C}$ | 760 | $1.316 \times 10^{-8}$ |
| Atmospheres | Feet of water (a) $4^{\circ} \mathrm{C}$ | 33.90 | $2.950 \times 10^{-8}$ |
| Atmospheres | Inches mercury @ $0^{\circ} \mathrm{C}$ | 29.92 | $3.342 \times 10^{-2}$ |
| Atmospheres | Kg per sq meter | $1.033 \times 10^{4}$ | $9.678 \times 10^{-5}$ |
| Atmospheres | Pounds per sq inch | 14.70 | $6.804 \times 10^{-2}$ |
| Bfu | Foot-pounds | 778.3 | $1.285 \times 10^{-8}$ |
| Btu | Joules | 1,054.8 | $9.480 \times 10^{-4}$ |
| Bto | Kilogram-calories | 0.2520 | 3.969 |
| Btu | Horsepower-hours | $3.929 \times 10^{-6}$ | 2,545 |
| Bushels | Cubic feet | 1.2445 | 0.8036 |
| Contigrade | Fahrenheit | $\left(C^{0} \times 9 / 5\right)+32$ | $\left(F^{\circ}-32\right) \times 5 / 9$ |
| Circular mils | Square centimefers | $5.067 \times 10^{-6}$ | $1.973 \times 10^{5}$ |
| Circular mils | Square mils | 0.7854 | 1.273 |
| Cubic feet | Cords | $7.8125 \times 10^{-3}$ | 128 |
| Cubic feet | Gallons (liq US) | 7.481 | 0.1337 |
| Cubic foer | Litors | 28.32 | $3.531 \times 10^{-2}$ |
| Cubic inches | Cubic centimeters | 16.39 | $6.102 \times 10^{-2}$ |
| Cubic inches | Cublc feet | $5.787 \times 10^{-4}$ | 1,728 |
| Cubic inches | Cubic meters | $1.639 \times 10^{-8}$ | $6.102 \times 10^{4}$ |
| Cubic inches | Gallons (liq US) | $4.329 \times 10^{-8}$ | 231 |
| Cubic meters | Cubic fedr | 35.31 | $2.832 \times 10^{-9}$ |
| Cubic meters | Cubic yards | 1.308 | 0.7646 |
| Degrees (angle) | Radians | $1.745 \times 10^{-2}$ | 57.30 |
| Dynes | Pounds | $2.248 \times 10^{-6}$ | $4.448 \times 10^{5}$ |
| Ergs | Foot-pounds | $7.367 \times 10^{-8}$ | $1.356 \times 10^{7}$ |
| Fothoms | Feet | 6 | 0.16666 |
| Feet | Centimeters | 30.48 | $3.281 \times 10^{-7}$ |
| Feot of water @ $4^{\circ} \mathrm{C}$ | Inches of mercury @ $0^{\circ} \mathrm{C}$ | 0.8826 | 1.133 |
| Feat of water @ $4^{\circ} \mathrm{C}$ | Kg per sq meter | 304.8 | $3.281 \times 10^{-8}$ |


| to convert | Into | multiply by | conversely multiply by |
| :---: | :---: | :---: | :---: |
| Feet of woter @ $4^{\circ} \mathrm{C}$ | Pounds per sq foot | 62.43 | $1.602 \times 10^{-2}$ |
| Foot-pounds | Horsepower-hours | $5.050 \times 10^{-7}$ | $1.98 \times 10^{6}$ |
| Foot-pounds | Kilogram-meters | 0.1383 | 7.233 |
| Foot-pounds | Kilowatt-hours | $3.766 \times 10^{-7}$ | $2.655 \times 10^{6}$ |
| Gallons | Cubic meters | $3.785 \times 10^{-2}$ | 264.2 |
| Gallons lliq USI | Gallons (liq Br Impl | 0.8327 | 1.201 |
| Gauss | Lines per sq inch | 6.452 | 0.1550 |
| Grams | Dynes | 980.7 | $1.020 \times 10^{-3}$ |
| Grams | Grains | 15.43 | $6.481 \times 10^{-2}$ |
| Grams | Ounces lavoirdupois) | $3.527 \times 10^{-2}$ | 28.35 |
| Grams | Poundals | $7.093 \times 10^{-2}$ | 14.10 |
| Grams per cm | Pounds par inch | $5.600 \times 10^{-8}$ | 178.6 |
| Grams per cu cm | Pounds per cu inch | $3.613 \times 10^{-2}$ | 27.68 |
| Grams per sq cm | Pounds per sq foot | 2.0481 | 0.4883 |
| Hectores | Acres | 2.471 | 0.4047 |
| Horsepower (boiler) | Bru per hour | $3.347 \times 10^{4}$ | $2.986 \times 10^{-6}$ |
| Horsepower (metrial ( 542.5 ff -lb per sec) | Btu per minute | 41.83 | $2.390 \times 10^{-2}$ |
| Horsepower (metric) ( $542.5 \mathrm{ft}-\mathrm{lb}$ per sed) | Foor-lb per minute | $3.255 \times 10^{4}$ | $3.072 \times 10^{-8}$ |
| Horsepower (metrial ( 542.5 ft -lb per sec) | Kg-calories per minute | 10.54 | $9.485 \times 10^{-2}$ $2357 \times 10^{-2}$ |
| Horsepower ( 550 ft -lb per sec) | Bfu per minute | 42.41 | $2.357 \times 10^{2}$ |
| Horsepower ( 550 ft -lb per sed) | Foor-lb per minute | $3.3 \times 10^{4}$ | $3.030 \times 10^{-8}$ |
| Horsepower (metric) ( 542.5 ff -lb per sec) | Horsepower ( 550 ft -lb per sed) | 0.9883 | 1.014 |
| Horsepower (550 ff-lb per sed) | Kg -calories per minute | 10.69 | $9.355 \times 10^{-2}$ |
| Inches | Centimeters | 2.540 | 0.3937 |
| Inches | Feet | $8.333 \times 10^{-2}$ | 12 |
| Inches | Miles | $1.578 \times 10^{-1}$ | $6.336 \times 10^{4}$ |
| Inches | Mils | 1,000 | 0.001 |
| Inches | Yards | $2.778 \times 10^{-2}$ | 36 |
| Inches of mercury @ $0^{\circ} \mathrm{C}$ | Lbs per sq inch | 0.4912 | 2.036 |
| Inches of water @ $4^{\circ} \mathrm{C}$ | Kg per sq metor | 25.40 | $3.937 \times 10^{-2}$ |
| Inches of water | Ounces per sq inch | 0.5781 | 1.729 |
| Inches of water | Pounds per sq foot | 5.204 | 0.1922 |
| Joules | Foot-pounds | 0.7376 | 1.356 |
| Joules | Ergs | $10^{7}$ | $10^{-7}$ |
| Kilogram-calories | Kilogram-mefers | 426.9 | $2.343 \times 10^{-8}$ |
| Kilogram-calories | Kilojoules | 4.186 | 0.2389 |
| Kilograms | Tons, long lavdp 2240 fb ) | $9.842 \times 10^{-4}$ | 1,016 |
| Kilograms | Tons, short lovdp 2000 lb ) | $1.102 \times 10^{-8}$ | 907.2 |
| Kilograms | Pounds lavoirdupais) | 2.205 | 0.4536 |
| Kg per sq meter | Pounds per sq foot | 0.2048 | 4.882 |
| Kilometers | Feot | 3,281 | $3.048 \times 10^{-6}$ |
| Kilowatt-hours | Bru | 3,413 | $2.930 \times 10^{-6}$ |
| Kilowatt-hours | Foot-pounds | $2.655 \times 10^{8}$ | $3.766 \times 10^{-7}$ |
| Kilowatt-hours | Joules | $3.6 \times 10^{8}$ | $2.778 \times 10^{-7}$ |
| Kilowatt-hours | Kilogram-calories | 860 | $1.163 \times 10^{-3}$ |
| Kilowatt-hours | Kilogram-meters | $3.671 \times 10^{5}$ | $2.724 \times 10^{-6}$ |
| Kilowott-hours | Pounds carbon oxydized | 0.235 | 4.26 |
| Kilowatt-hours | Pounds woter evaporated from and of $212^{\circ} \mathrm{F}$ | 3.53 | 0.283 |

## Greek alphabef

| name | capital | small | commonly used to deslgnate |
| :---: | :---: | :---: | :---: |
| ALPHA | A | $\alpha$ | Angles, coefficients, attenuation constant, absorption factor, area |
| BETA | B | $\beta$ | Angles, coefficients, phase constant |
| GAMMA | r | $\boldsymbol{\gamma}$ | Complex propagation constant icapl, specific gravity, angles, electrical conductivity, propagation constant |
| Delta | $\Delta$ | $\delta$ | Increment or decrement icap or small, determinant (capl, permittivity (cap), density, angles |
| EPSILON | E | - | Dielectric constant, permittivity, base of natural logarithms, oloctric intensity |
| ZETA | Z | 5 | Coordinates, coefficionts |
| ETA | H | 7 | Intrinsic impedance, officiency, surface charge density, hysterosis, coordinatos |
| THETA | $\theta$ | $\vartheta \theta$ | Angular phase displacement, time constant, reluctance, angles |
| IOTA | I | c | Unit vector |
| KAPPA | K | $\kappa$ | Susceptibility, coupling coefficient |
| LAMBDA | - | $\boldsymbol{\lambda}$ | Permeance (cap), wavelength, attenuation constans |
| MU | M | $\mu$ | Permeability, amplification factor, prefix micro |
| NU | $\mathbf{N}$ | * | Reluctivity, frequency |
| XI | z | $\xi$ | Coordinates |
| OMICRON | 0 | - |  |
| P1 | II | $\pi$ | 3.1416 |
| RHO | P | $p$ | Resistivity, volume charge density, coordinates |
| SIGMA | 2 | $\sigma$ \% | Summation (capl, surface charge density, complex propagation constont, oloctrical conductivity, loakago coofficient |
| taU | T | $\tau$ | Time constant, volume resistivity, time-phase displacement, transmission foctor, density |
| UPSIION | T | $v$ |  |
| PHI | \$ | $\phi \varphi$ | Scalar potential (eap), magnetic flux, angles |
| CHI | $\mathbf{X}$ | $\chi$ | Electric suscoptibility, angles |
| PS! | $\Psi$ | $\psi$ | Dielectric fux, phase diference, coordinates, angles |
| OMEGA | $\Omega$ | $\omega$ | Resistance in ohms (cap), solid angle (cap), angular velocity |
| Small lotter is used excopt where caplial is indicated. |  |  |  |



[^0]

## 18

Electromotive force series of the elements

| eloment | volts | Ion | element |  | volts | $10 n$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Lithium | 2.9595 |  | Tin |  | 0.136 |  |
| Rubidium | 2.9259 |  | Lead |  | 0.122 | Pb++ |
| Potassium | 2.9241 |  | Iron | - | 0.045 | $\mathrm{Fe}^{+++}$ |
| Strontium | 2.92 |  | Hydrogen | - | 0.000 |  |
| Barium | 2.90 |  | Antimony |  | -0.10 |  |
| Calcium | 2.87 |  | Bismuth |  | -0.226 |  |
| Sodium | - 2.7146 |  | Arsenic |  | -0.30 |  |
| Magnesium | 2.40 |  | Copper |  | $-0.344$ | $\mathrm{Cu}^{++}$ |
| Aluminum | 1.70 |  | Oxygen |  | -0.397 |  |
| Beryllium | 1.69 |  | Polonium |  | -0.40 |  |
| Uranium | 1.40 |  | Copper |  | -0.470 | $\mathrm{Cu}^{+}$ |
| Manganese | 1.10 |  | lodine |  | -0.5345 |  |
| Tellurium | 0.827 |  | Tellurium |  | -0.558 | Te ${ }^{++++}$ |
| Zinc | 0.7618 |  | Silver |  | -0.7978 | To |
| Chromium | 0.557 |  | Marcury |  | -0.7986 |  |
| Sulphur | 0.51 |  | Lead |  | -0.80 | Pb++++ |
| Gallium | 0.50 |  | Palladium |  | -0.820 |  |
| Iron | 0.441 | $\mathrm{Fe}^{++}$ | Platinum |  | -0.863 |  |
| Cadmium | 0.401 |  | Bromine |  | -1.0648 |  |
| Indium | 0.336 |  | Chlorine |  | $-1.3583$ |  |
| Thallium | 0.330 |  | Gold |  | $-1.360$ |  |
| Cobalt | 0.278 |  | Gold |  | $-1.50$ | $\mathrm{Au}^{+}$ |
| Nickel | 0.231 |  | Fluorine |  | -1.90 |  |

## Position of metals in the galvanic series

| Corroded and (anodic, or least noble) | Nickel \{active) Inconel (active) |
| :---: | :---: |
| Magnesium | Brasses |
| Magnesium alloys | Copper |
| Zinc | Bronzes |
| Aluminum 2S | Copper-nickel alloys Monel |
| Cadmium |  |
| Aluminum I7ST | Silver solder |
| Aluminum ITST | Nickel (passive) |
| Steel or Iron | Inconel (passive) |
| Cast Iron | Chromium-iron (passivel |
| Chromium-iron (active) | 18-8 Stainless (passive) |
| Ni-Resist | 18-8-3 Stainless (passive) |
| 18-8 Stainless (active) | Silver |
| 18-8-3 Stainless (active) | Graphite |
| Lead-tin solders | Platinum |
| Lead Tin | Protected end (eathodic, or mast noble) |

Nole: Groups of metals indicate they are closely stailor in properties.

## Atomic weights

| -lement | symbal | alomic number | atomic wolght | viement | symbol | efomic number | otomic wolght |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Aluminum | Al | 13 | 26.97 | Molybdenum | Mo | 42 | 95.95 |
| Antimony | Sb | 51 | 121.76 | Neodymium | Nd | 60 | 144.27 |
| Argon | A | 18 | 39.944 | Neon | Ne | 10 | 20.183 |
| Arsenic | As | 33 | 74.91 | Nickel | Ni | 28 | 58.69 |
| Barium | Bo | 56 | 137.36 | Nitrogen | N | 7 | 14.008 |
| Beryllium | Be | 4 | 9.02 | Osmium | Os | 76 | 190.2 |
| Bismuth | Bi | 83 | 209.00 | Oxygen | $\bigcirc$ | 8 | 16.0000 |
| Boron | B | 5 | 10.82 | Palladium | Pd | 46 | 106.7 |
| Bromine | Br | 35 | 79.916 | Phosphorus | P | 15 | 30.98 |
| Cadmium | Cd | 48 | 112.41 | Platinum | P4 | 78 | 195.23 |
| Calcium | Co | 20 | 40.08 | Potassium | K | 19 | 39.096 |
| Corbon | C | 6 | 12.010 | Praseodymium | Pr | 59 | 140.92 |
| Cerium | Co | 58 | 140.13 | Prolactinium | Pa | 91 | 231 |
| Cesium | Cs | 55 | 132.91 | Rodium | Ra | 88 | 226.05 |
| Chlorine | Cl | 17 | 35.457 | Radon | Rn | 86 | 222 |
| Chromium | Cr | 24 | 52.01 | Rhenium | Re | 75 | 186.31 |
| Cobalt | Co | 27 | 58.94 | Rhodium | Rh | 45 | 102.91 |
| Columbium | Cb | 41 | 92.91 | Rubidium | Rb | 37 | 85.48 |
| Copper | Cu | 29 | 63.57 | Ruthenium | Ru | 44 | 101.7 |
| Dysprosium | Dy | 66 | 162.46 | Samarium | Sm | 62 | 150.43 |
| Erbium | Er | 68 | 167.2 | Scandium | Sc | 21 | 45.10 |
| Europium | Eu | 63 | 152.0 | Selenium | Se | 34 | 78.96 |
| Fluorine | F | 9 | 19.00 | Silicon | Si | 14 | 28.06 |
| Gadolinium | Gd | 64 | 156.9 | Silver | Ag | 47 | 107.880 |
| Gallium | Ga | 31 | 69.72 | Sodium | Na | 11 | 22.997 |
| Germanium | Ge | 32 | 72.60 | Strontium | Sr | 38 | 87.63 |
| Gold | Au | 79 | 197.2 | Sulfur | S | 16 | 32.06 |
| Hafnium | Hf | 72 | 178.6 | Tantalum | Ta | 73 | 180.88 |
| Helium | He | 2 | 4.003 | Tellurium | Te | 52 | 127.61 |
| Holmium | Ho | 67 | 164.94 | Terbium | Tb | 65 | 159.2 |
| Hydrogen | H | 1 | 1.0080 | Thallium | TI | 81 | 204.39 |
| Indium | In | 49 | 114.76 | Thorium | Th | 90 | 232.12 |
| lodine | 1 | 53 | 126.92 | Thulium | Tm | 69 | 169.4 |
| Iridium | Ir | 77 | 193.1 | Tin | Sn | 50 | 118.70 |
| Iron | Fo | 26 | 55.85 | Titanium | Ti | 22 | 47.90 |
| Krypton | Kr | 36 | 83.7 | Tungsten | W | 74 | 183.92 |
| Lanthanum | Lo | 57 | 138.92 | Uranium | U | 92 | 238.07 |
| Lead | Pb | 82 | 207.21 | Vonodium | V | 23 | 50.95 |
| Lithium | 1 l | 3 | 6.940 | Xenon | Xe | 54 | 131.3 |
| Lutecium | tu | 71 | 174.99 | Ytterbium | Yb | 70 | 173.04 |
| Magnesium | Mg | 12 | 24.32 | Ytrium | Y | 39 | 88.92 |
| Manganese | Mn | 25 | 54.93 | Zinc | Zn | 30 | 65.38 |
| Mercury | Hg | 80 | 200.61 | Zirconium | Zr | 40 | 91.22 |

From the Journal of the American Chemical Society, 1943.


Exomple: Assume dry bulb reoding thermameter expased directly ta atmaspherel $1820^{\circ} \mathrm{C}$ and wet bulb reading $1 s 17^{\circ} \mathrm{C}$, or o difference of $3^{\circ} \mathrm{C}$. The relative humidity at $20^{\circ} \mathrm{C}$ ts then $74 \%$.


## Atmospheric pressure chart


altitude in feet obove sea-leve:

## Weather data

Compiled fram Climate and Man, Yearbook of Agriculure, U. S. Dept. of Agriculture, U. S.
Govt. Printing Offico, Woshington, D. C, 1941.

## Temperałure exfremes

United States

| Lowest temperature Highest temperature | $\begin{aligned} & -66^{\circ} \mathrm{F} \\ & 134^{\circ} \mathrm{F} \end{aligned}$ | Rivarside Range Station, Wyoming IFob. 9, 19331 Greenland Ranch, Death Valley, Colifornia Uuly 10, 1933 |
| :---: | :---: | :---: |
| Alaske |  |  |
| Lawest temperature Highest temperature | $\begin{array}{r} -78^{\circ} \mathrm{F} \\ 100^{\circ} \mathrm{F} \end{array}$ | Fort Yukan Uan. 14, 1934 fort Yukan |
| Werld |  |  |
| Lowest temperature <br> Highest temperature Lowest mean temperature lannual) Highest mean temperature lannuall | $\begin{array}{r} -90^{\circ} \mathrm{F} \\ 136^{\circ} \mathrm{F} \\ -14^{\circ} \mathrm{F} \\ 86^{\circ} \mathrm{F} \end{array}$ | Verkhayansk, Siberio Ifeb. 5 and 7, 1897 Azizio, Libya, Narth Africa ISopt. 13, 1922 Framhelm, Antarctica Massowo, Eritreo, Africa |

## Precipitation extremes

United States
Wettest state
Dryest state
Maximum recorded
Minlmum recorded

## World

Maximum recorded

Minimum recorded

Lavisiana-average annual rainfall 55.11 inches
Novada-avarage annual rainfall 8.81 inches
Now Smyrno, Flo., Oct. 10, 1924-23.22 inches in 24 hours
Bagdod, Calif., 1909-1913-3.93 inches in 5 years
Greenlond Ranch, Calif.- 1.35 inches annual average
Cherropunf1, India, Aug. 1841-241 Inches In 1 manth
(Average annual roinfall of Cherrapunfl is 426 inches
Bagut, Luzon, Philippines, July 14-15, 1911-46 Inches in 24 hours
Wadi Halfa, Anglo-Egyptian Sudan and Awan, Egypt are in the "rainless" areo; average annual rainfall is too small to be measured

World temperatures

| Ierrifory | $\underset{\substack{\text { of }}}{\substack{\text { maximum }}}$ | $\underset{{ }_{F}}{\operatorname{minimum}}$ | terifiory | $\mid \underset{o f}{\text { maxilmum }}$ | $\underset{\circ}{\text { minnman }}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| NORTH AMERICA | 100 |  | ASIA continued |  |  |
| Conada | 100 | -78 | Indio | 120 | -19 |
| Conal Zone | 97 | 63 | Jraq | 123 | 19 |
| Greenland | 86 | -46 | Molay Siates | 97 | -66 |
| Mexico | 118 | 11 | Philippine Islands | 101 | 58 |
| U. S. A. | 134 | -66 | Slom | 106 | 52 |
| West Indies | 102 | 45 | Tibet | 85 | -20 |
|  |  |  | Turkey | 111 | -22 |
| SOUTH AMERICA |  |  | U. S. S. R. | 109 | -90 |
| Argentina | 115 | -27 |  | 102 | - |
| Bolivio | 82 | 25 | AFRICA |  |  |
| Brazil | 108 | 21 | Algeria | 133 | 1 |
| Chile | 99 | 19 | Anglo-Egyptian Sudan | 126 | 28 |
| Venezuala | 102 | 45 | Angolo | 91 | 33 |
|  |  |  | Belgian Congo | 97 | 34 |
| EUROPE |  |  | Egypt | 124 | 31 |
| British Isies | 100 | 4 | Ethiopia | 111 | 32 |
| France | 107 | -14 | French Equatorial Africa | 118 | 46 |
| Germany | 100 | -16 | French West Africa | 122 | 41 |
| Icelond | 71 | -6 | Italian Somalliand | 93 | 61 |
| Iraly | 114 | 4 | libyo | 136 | 35 |
| Norway. | 95 | -26 | Moroceo | 119 | 5 |
| Spain | 124 | 10 | Rhodesio | 103 | 25 |
| Sweden | 92 | -49 | Tunisia | 122 | 28 |
| Turkey | 100 | 17 | Union of South Africa | 111 | 21 |
| U. S. S. ${ }^{\text {a }}$ | 110 | -61 | AUSTRALASIA |  |  |
|  |  |  | Australia | 127 | 19 |
| Arabla | 114 | 53 | Hawali | 91 | 51 |
| China | 111 | $-10$ | New Zealand | 94 | 23 |
| East Indies | 101 | 60 | Samoon Islands | 96 | 61 |
| French Indo-China | 113 | 33 | Solomon tsionds | 97 | 70 |


| iemitery | highed average |  |  |  | loweet everage |  |  |  | $\begin{aligned} & \text { yearly } \\ & \text { average } \\ & \text { inches } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | Jan inches | April Inches | duly Inches | Oct linches | Dan Inches | April inches | July Inches | Od inches |  |
| NORTH AMERICA |  |  |  |  |  |  |  |  |  |
| Alaska | 13.71 | 10.79 | 8.51 | 22.94 | . 15 | . 13 | . 93 | . 37 | 43.40 |
| Canada | 8.40 | 4.97 | 4.07 | 6.18 | . 48 | . 31 | 1.04 | . 73 | 26.85 |
| Conal Zone | 3.74 | 4.30 | 16.00 | 15.13 | .91 | 2.72 | 7.28 | 10.31 | 97.54 |
| Greenland | 3.46 | 2.44 | 3.27 | 6.28 | .35 | . 47 | . 91 | . 94 | 24.70 |
| Maxico | 1.53 | 1.53 | 13.44 | 5.80 | 04 | . 00 | .43 | . 35 | 29.82 |
| U. S. A. |  |  |  |  |  |  |  |  | 29.00 |
| West Indios | 4.45 | 6.65 | 5.80 | 6.89 | .92 | 1.18 | 1.53 | 5.44 | 49.77 |
| SOUTH AMERICA Argentina | 6.50 | 4.72 | 2.16 | 3.35 | . 16 | 28 | . 04 | . 20 | 16.05 |
| Bolivia | 6.34 | 1.77 | . 16 | 1.42 | 3.86 | 1.46 | . 16 | 1.30 | 24.18 |
| Brazil | 13.26 | 12.13 | 10.47 | 6.54 | 2.05 | 2.63 | . 01 | . 05 | 55.42 |
| Chile | 11.78 | 11.16 | 16.63 | 8.88 | . 00 | . 00 | . 03 | . 00 | 46.13 |
| Venezuela | 2.75 | 6.90 | 6.33 | 10.44 | . 02 | .61 | 1.87 | 3.46 | 40.01 |
| EUROPE |  |  |  |  |  |  |  |  |  |
| British lslos | 5.49 | 3.67 | 3.78 | 5.57 | 1.86 | 1.54 | 2.38 | 2.63 | 36.16 |
| France | 3.27 | 2.64 | 2.95 | 4.02 | 1.46 | 1.65 | . 55 | 2.32 | 27.48 |
| Germany | 1.88 | 2.79 | 5.02 | 2.97 | 1.16 | 1.34 | 2.92 | 1.82 | 26.64 |
| Iceland | 5.47 | 3.70 | 3.07 | 5.95 | 5.47 | 3.70 | 3.07 | 5.59 | 52.91 |
| Italy | 4.02 | 4.41 | 2.40 | 5.32 | 1.44 | 1.63 | . 08 | 2.10 | 29.74 |
| Norway | 8.54 | 4.13 | 5.79 | 8.94 | 1.06 | 1.34 | 1.73 | 2.48 | 40.51 |
| Spain | 2.83 | 3.70 | 2.05 | 3.58 | 1.34 | 1.54 | . 04 | 1.77 | 22.74 |
| Swoden | 1.52 | 1.07 | 2.67 | 2.20 | . 98 | . 78 | 1.80 | 1.60 | 18.12 |
| Turkey | 3.43 | 1.65 | 1.06 | 2.52 | 3.43 | 1.65 | 1.06 | 2.52 | 28.86 |
| U. S. S. R. | 1.46 | 1.61 | 3.50 | 2.07 | . 49 | . 63 | 20 | . 47 | 18.25 |
| ASIA |  |  |  |  |  |  |  |  |  |
| Arabia | 1.16 | . 40 | . 03 | . 09 | .32 | .18 | . 02 | .09 | 3.05 |
| China | 1.97 | 5.80 | 13.83 | 6.92 | . 15 | . 61 | 5.78 | . 67 | 50.63 |
| Easi Indies | 18.46 | 10.67 | 6.54 | 10.00 | 7.48 | 2.60 | .20 | . 79 | 78.02 |
| French Indo-China | 79 | 4.06 | 12.08 | 10.61 | . 52 | 2.07 | 9.24 | 3.67 | 65.64 |
| Indio | 3.29 | 33.07 | 99.52 | 13.83 | . 09 | . 06 | .47 | .00 | 75.18 |
| Iraq | 1.37 | . 93 | . 0 | . 08 | 1.17 | . 48 | .00 | 05 | 6.75 |
| Japan | 10.79 | 8.87 | 9.94 | 7.48 | 2.06 | 2.83 | 5.02 | 4.59 | 70.18 |
| Malay Stotes | 9.88 | 7.64 | 6.77 | 8.07 | 9.88 | 7.64 | 6.77 | 8.07 | 95.06 |
| Philipplas tslands | 2.23 | 1.44 | 17.28 | 10.72 | 82 | 1.28 | 14.98 | 6.71 | 83.31 |
| Siom | . 33 | 1.65 | 6.24 | 8.32 | . 33 | 1.65 | 6.24 | 8.32 | 52.36 |
| Turkey | 4.13 | 2.75 | 1.73 | 3.34 | 2.05 | 1.73 | . 21 | . 93 | 25.08 |
| U. S. S. R. | 1.79 | 2.05 | 3.61 | 4.91 | . 08 | .16 | .10 | . 06 | 11.85 |
| AFrica |  |  |  |  |  |  |  |  |  |
| Algeria | 4.02 | 2.06 | - $\quad 35$ | 3.41 | . 52 | . 11 | .00 | . 05 | 9.73 |
| Anglo-Egyptlon Sudan | .08 | 4.17 | - 7.87 | 4.29 | . 00 | 00 | . 00 | . 00 | 18.27 |
| Angole | 8.71 | 5.85 | . 00 | 3.80 | . 09 | . 63 | . 00 | . 09 | 23.46 |
| Belgion Congo | 9.01 | 6.51 | .13 | 2.77 | 3.69 | 1.81 | . 00 | 1.88 | 39.38 |
| Egypt | 2.09 | . 16 | 00 | . 28 | . 00 | . 00 | .00 | . 00 | 3.10 |
| Ethiopla | . 59 | 3.42 | 10.98 | 3.39 | . 28 | 3.11 | 8.23 | 79 | 49.17 |
| Franch Equatorial Africa | 9.84 | 13.42 | 6.33 | 13.58 | . 00 | . 34 | . 18 | . 86 | 57.55 |
| French West Africe | .10 | 1.61 | 8.02 | 1.87 | . 00 | .00 360 | . 18 | .00 242 | 19.51 |
| Italion Somalifand | . 00 | 3.66 | 1.67 | 2.42 | . 00 | 3.60 | 1.67 | 2.42 | 17.28 |
| libyo | 3.24 | . 48 | . 02 | 1.53 | 2.74 | . 18 | . 00 | . 67 | 13.17 1587 |
| Moroceo | 3.48 | 2.78 | . 07 | 2.47 | 1.31 | . 36 | . 00 | . 23 | 15.87 |
| Rhodesio | 8.40 | . 95 | . 04 | 1.20 | 5.81 | . 65 | . 00 | . 88 | 29.65 |
| Tunisio | 2.36 | 1.30 | . 08 | 1.54 | 2.36 | 1.30 | . 08 | 1.54 | 15.80 26.07 |
| Union of South Africa | 6.19 | 3.79 | 3.83 | 5.79 | . 06 | 23 | 27 | . 12 | 26.07 |
| AUSTRALASIA |  |  |  |  |  |  |  |  |  |
| Australio | 15.64 | 5.33 | 6.57 9.89 | 2.84 10.97 | . 3.54 | .85 2.06 | 1.04 | . 1.90 | 82.31 |
| Howall | 11.77 | 13.06 | 9.89 | 10.97 | 3.54 2.67 | 2.06 278 | 1.04 2.99 | 1.97 3.13 | 82.43 43.20 |
| Now Zealand | 3.34 18.90 | 3.80 11.26 | 5.55 2.60 | 4.19 7.05 | 2.67 | 11.78 | 2.99 2.60 | 3.13 7.05 | 118.24 |
| Samoan islands Solomon litands | 18.90 | 11.26 8.24 | 2.60 6.26 | 7.01 | 13.94 | 11.26 8.24 | 6.26 | 7.91 | 115.37 |

Principal power supplies in foreign countries

| merifiory | de volis | ace velts | frequen\% |
| :---: | :---: | :---: | :---: |
| NORTH AMERICA |  |  |  |
| Alosho |  | 110,220 | 60 |
| British Honduras | 110,220 |  |  |
| Conado | 110 | *110, 150, 115, 220 | 60, 25 |
| Costa Rica | 110 | *110 |  |
| Cuba | 110, 220 | * 110,220 | 60 |
| Dominicon Republic | 110 | -110, 220 | 60 |
| Guatemala | 220, 125 | -110,220 | 60, 50 |
| Haiti |  | 110, 220 | 60, 50 |
| Honduras | 110, 220 | * 110,220 |  |
| Mexico | 110, 220 | -110, 125, 115, 220, 230 | 60, 50 |
| Newfoundiand |  | 110,115 | 50, 60 |
| Nicarogua | 110 | * 110 |  |
| Panama Republic) |  | 110, 220 | 60, 50 |
| Panama Conal Zonel |  | 110 | 25 |
| Puerto flico | 110,220 | - 110 | 60 |
| Salvador | 110,220 | * 110 | 60 |
| Virgin lalonds | 110,200 |  |  |
| WEST MNDIES |  |  |  |
| Bahamas is. |  | 115 | 60 |
| Barbados |  | 110 | 50 |
| Bermuda |  | 110 | 60 |
| Curacao |  | 127 | 50 |
| Jamalca |  | 110 | 40, 60 |
| Nortinique | 110 | *110 | 50 |
| Trinidad |  | 110,220 | 60 |
| SOUTH AMERICA |  |  |  |
| Argentina | - 220 | 220, 225 | $50,60,43$ |
| Bolivia | 110 | * 110,220 , 120 | $50,60$ |
| Brozil | 110, 120, 220 | 110, 115, 120, 125, 20, 220 | 50, 60 |
| Chile | 220, 110 | +220 | 50, 60 |
| Colombla |  | *110, 220, 150 | $\begin{aligned} & 60,50 \\ & 60,50 \end{aligned}$ |
| Ecuodor | - 220 | 220. | 50 |
| Peru | 220, 110 | +220, 110 | 60,50 |
| Uruguay | 220 | +220 |  |
| Venezuela | 110,200 | *110 | 50,60 |
| EUROPE |  |  |  |
| Albania | 220 , 110,150 | ${ }^{*} 220,125,150$ | 50 |
| Austrio | 220, 110, 150 | *220, 120, 127, 110 | $50$ |
| Azores | 220 | $220$ | $50$ |
| Beloium | $220,110,120$ 200,120 | -220, 127, 110, 115, 135 | 50,40 |
| Bulgario | 220,120 +220 | -220, 120, 150 | $50$ |
| Cyprus [Br] | +220 $220,120,150,110$ | 110 $+220,110,115,127$ | $\begin{aligned} & 50 \\ & 50,42 \end{aligned}$ |
| Czechosiovalio <br> Denmark | 220,1110 $220,150,110$ | -220, 120, 127 | 50 |
| Estonia | ${ }^{-220,110}$ | 220, 127 | 50 |
| Finlond | * 120, 220, 110 | 220, 120, 115, 110 | 50 |
| France | 110, 220, 120, 125 | *110, 115, 120, 125, 220, 220 | 50, 25 |
| Germony | $220,110,120,250$ | *220, 127, 120, 110 | 50, 25 |
| Gibraltor | 440, 220 | -110 | 76 |
| Greece | - $2220,110,150$ | (127, 110, 220 | 50 |
| Hungary | 220, 110, 120 | * $100,105,110,220,120$ | 42, 50 |
| Iceland |  | 220 |  |
| Irish Free State | +220 | -220, 200 | $50$ |
| Ifoly | 110, 125, 150, 220, 250, 160 | ${ }^{*} 150,125,120,110,115,260,220$, | 42, 50, 45 |
| Lotvia | 220, 110 | -220, 120 | 50 |
| Lithuanla | 220, 110 | +220 | 50 |
| Malta |  | 105 | 100 |
| Monaco |  | 110 | 42 |
| Netherlands | 220 | 220, 120, 127 | 50 |
| Norway | 220 | *220, 230, 130, 127, 110, 120, 150 | 50 |
| Poland | 220, 110 | -220, 120, 110 | 50 |
| Portugal | 220, 150, 125 | +220, 110, 125 | 50,42 |
| Rumonio | -220, 110, 105, 120 | 120, 220, 110, 115, 105 | 50,42 |
| Russia | 220, 110, 120, 115, 250 | +120, 110, 220 | 50 |
| Spain | - $110,120,115,105$ | * 120, 125, 150, 110, 115, 220, 130 | 50 |
| Swedon | 220, 110, 120, 115, 250 | *220, 127, 110, 125 | $50,20,25$ |
| Switzerlond | 220, 120, 110, 150 | +120, 220, 145, 150, 110, 120 | 50,40 |
| Turkey | 110, 220 | -220, 110 | 50 |

Principal power supplies in foreign countries
continued

| ferriliory | de volls | ae volis | frequency |
| :---: | :---: | :---: | :---: |
| EUROPE continuad United Kingdorn Jugoslavila | $\begin{aligned} & 230,220,240 \\ & 110,120 \end{aligned}$ | -230, 240, others <br> *120, 220, 150 | $\begin{aligned} & 50,25,40 \\ & 50,42 \end{aligned}$ |
| ASIA |  |  |  |
| Arabia |  | 230 , | 50 |
| British Moloya |  |  |  |
| Fed. Malay States |  | 230 | 50, 60, 40 |
| Non-Fed. Maloy States | 230 |  |  |
| Strails Setflements | - 230 | 230 | 50 |
| North Borneo |  | 110 | 60 |
| Ceylon | $220$ | 230 | 50, 60 |
| China | $220,110$ | * 110, 200, 220 | $50,60,25$ |
| Howall |  | 110, 220 | 60, 25 |
| India | 220, 110, 225, 230, 250 | 230, 220, 110, others | 50, 25 |
| Fronch Indo-Ching | 110, 120, 220, 240 | *120, 220, 110, 115, 240 |  |
| Iran (Persia) | 220, 110 | 220 | 50 |
| Iraq | -220, 200 | 220, 230 |  |
| Japon | 100 | * 100,110 | 50,60 |
| Manchurla |  | 110 | 60,50, 25 |
| Palestine |  | 220 | 50 |
| Phitlippine lslands |  | 220 , 115 | 60 |
| Syrio Slam |  | 110, 115, 220 | 50 50 |
| Turkey | 220, 110 | -220, 110 | 50 |
| AFRICA |  |  |  |
| Angola (Port.) |  | 110 | 50 |
| Algerio | 220 | * $115,110,127$ | 50 |
| Belglon Congo |  | 220 | 60 |
| British West Africa | -220 | 230 | 50 |
| British East Africa | - 220 | -240, 230, 110, 100 | 50, 60, 100 |
| Conary lslands | 110 | *127, 110 | 50 |
| Egypt | 220 | 200, 110, 220 | 50, 40 |
| Italion Africo |  |  |  |
| Cyrenalca | 150 | *110, 150 | 50 |
| Eritrea |  | 127 | 50 |
| Libya (Tripolil |  | 125, 110, 270 | 50, 42, 45 |
| Somallland | 120 | - 230 |  |
| Morocco (Fr.) | 110 | 115, 110 | 50 |
| Morocco (Spanish) | 200 | * $127,110,115$ | 50 |
| Madagascar (FrJ Senegal If.J | 230 | 120 | 50 50 |
| Tunisio | 110 | * $110,115,220$ | 50 |
| Union of South Africo | 220, 230, 240, 110 | -220, 230, 240 | 60 |
| OCEANIA |  |  |  |
| Australia |  |  |  |
| New South Woles | -240 | -240 | 50 |
| Queensland | 220, 240 | - 240 | 50 |
| South Ausiralia | 200, 230, 220 | -200, 230, 240 | 60 |
| West Australia | -220, 110, 250 | 250 | 40 |
| Tosmonia | 230 | - 240 | 50 |
| New Zealand | 230 $240,110,250$ | +230 | 60 |
| Flif islands | 240, 110, 250 | 120 |  |
| Somoo |  | 110 | 60 |

Note: Where both ac and dc are avaliable, an asterisk mindicates the type of supoly and voltage predominating. Where approximately equal quantities of ac and dc are available, an asterisk precedes each of the principol voltages., Voltages and frequencies are listed in order of proference.
The electrical authorities of Great Britioin have adopied"a plan of unifying electrical distribotion systems. The standord potential for both ac and de supplies will be 230 volts. Systems ueving other voltages will be changed over. The standard ae frequency will be 50 cyelos

Caution: The listings in these tobles represent types of electrical supplies most generally used in perticular countries. For power supply charactoristics of particular citios of forsign countries, refer to the country section of World Electricat Markets, a pudlication of the U. S. Department of Commerce, Bureav of Foraign and Domestic Commarce, Washingion, D. C. In cases where definfe information relative to specific locations is nocessary, the Electrical Division of the above-named Bureau should be consulted.


Radio frequency classifications

| frequency in kilocycies | designations* | abbrevialions | wavelength In mefors $\dagger$ | wavelength in feet $\dagger$ |
| :---: | :---: | :---: | :---: | :---: |
| 10- 30 | Very Low | VLF | 30,000 -10,000 | 98,424 $-32,808$ |
| 30- 300 | Low | LF | 10,000-1,000 | 32,808 - 3,281 |
| 300- 3,000 | Medium | MF | 1,000 - 100 | 3,281-328 |
| 3,000- 30,000 | High | HF | $100-10$ | 328 - 32.8 |
| 30,000-300,000 | Vory High | VHF | 10-1 | 32.8 - $\quad 3.28$ |
| 300,000-3,000,000 | Ultra High | UHF | 1- 0.1 | 3.28- 0.33 |
| 3,000,000-30,000,000 | Super High | SHE | $0.1-0.01$ | 0.33-0.03 |

* Official FCC designation, March 2, 1943.
$\dagger$ Based on the established practice of considering the velocity of propagation in air as 300,000 kilometers per second instead of the true velocity of propagation of 299,796 kilomelers per second.


## Wavelength vs frequency charf



Conversion factors for wavelength vs frequency charf

| for frequencies from |  |
| ---: | ---: |
| $30-$ | 300 kilocycles |
| $300-$ | 3,000 kilocycles |
| $3,000-$ | 30,000 kilocycles |
| $30,000-$ | 300,000 kilocycles |
| $300,000-$ | $3,000,000$ kilocycles |
| $3,000,000-30,000,000$ kilocycles |  |


| multiply $f$ by |
| :---: |
| 0.1 |
| 1.0 |
| 10.0 |
| 100.0 |
| $1,000.0$ |
| $10,000.0$ |

multiply $\lambda$ by
| multiply $\boldsymbol{\lambda}$ by

## Wavelength vs frequency formulas

Wavelength in meters, $\boldsymbol{\lambda}_{\boldsymbol{m}}=\frac{300,000}{\text { frequency in kilocycles }}$
Wavelength in feet, $\lambda_{f t}=\frac{300,000 \times 3.28}{\text { frequency in kilocycles }}$

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## Frequency folerances

## Cairo revision 1938

| frequency bands (waveiengths) | column 1 | column 2 |
| :---: | :---: | :---: |
| A. From 10 to $550 \mathrm{kc}(30,000$ to 545 mefers): |  |  |
| a. Fixed stations | $0.1 \%$ | 0.1\% |
| b. Land stations | 0.1\% | $0.1 \%$ |
| c. Mobile stations using frequencies other than those of bands indicoted under (d) | 0.5\% | $0.1 \%$ |
| d. Mobile stations using frequencies of the bands $110-160 \mathrm{kc}$ (2,727 to 1,875 mefers), $365-515 \mathrm{kc}(822$ to 583 mefers) $\dagger$ <br> e. Aircraff stations | $\begin{aligned} & 0.5 \% \%^{*} \\ & 0.5 \% \end{aligned}$ | $\begin{aligned} & 0.3 \%{ }^{*} \\ & 0.3 \% \end{aligned}$ |
| e. Broadcasting stations |  | 20 cycles |
| B. From 550 to $1,500 \mathrm{kc}(545$ to 200 meters): <br> a. Broodcasting stations | 50 cycles | 20 cycles |

a. Broodcasting stations
b. Land stations
c. Mobile stations using the frequency of 1,364 kc 1220 meters)
$0.1 \%$
$0.05 \%$
C. From 1,500 to $6,000 \mathrm{kc}(200$ to 50 meters):
a. Fixed stations
b. Land stations
1
c. Mobile stations using frequencies other than those of bands indicated in (d):
1,560 to $4,000 \mathrm{kc}$ (192.3 to 75 meters) 4,000 to 6,000 kc 175 to 50 meters)
d. Mabile stations using frequencies within the bands: 4,115 to $4,165 \mathrm{kc}(72.90$ to 72.03 meters) $\}$ 5,500 to $5,550 \mathrm{kc}(54.55$ to 54.05 metersl $\}$
e. Aircraft stations
f. Broadcasting: between 1,500 and 1,600 kc (200 and 187.5 meters) between 1,600 and 6,000 kc (187.5 and 50 meters)
D. From 6,000 to $30,000 \mathrm{kc}$ ( 50 to 10 meters):
a. Fixed stations
b. Land stations
c. Mobile stations using frequencies other than those of bands indicated under (d)
d. Mobile stations using frequencies within the bands: 6,200 to $6,250 \mathrm{kc}$ ( 48,39 to 48 mefers) 8,230 to $8,330 \mathrm{kc}(36.45$ to 36.01 meters) 11,000 to $11,100 \mathrm{kc}$ [ 27.27 to 27.03 meters) 12,340 to $12,500 \mathrm{kc}(24.31$ to 24 meters) 16,460 to $16,660 \mathrm{kc}(18.23$ to 18.01 meters) 22,000 to $22,200 \mathrm{kc}$ (13.64 to 13.51 meters)
e. Aircraft stations
f. Broadcosting stations

Column 1: Transmlitters in service now and until January 1, 1944, after which date they will conform to the tolerances indicated in column 2.
Column 2: Now transmitters Installed beginning January 1, 1940.

* See preamble, under 3.
tit is recognized that a great number of spark tronsmitters and simple self-oscillator trans-
mitters exist in this service which are not oble to meet these requirements.

The frequency tolerance is the maximum permissible separation beween the actual frequency of an emission and the frequency which this emission should have (frequency notified or frequency chosen by the operator).
This separation results from the following errors:
a. Error made when the station was calibrated; this error presents a semipermanent character.
b. Error made during use of the station lerror variable from one transmission to another and resulting from actual operating conditions: ambient temperature, voltage of supply, antenna, skill of the operator, et ceteral. This error, which is usually small in other services, is particularly important in the case of mobile stations.
c. Error due to slow variations of the frequency of the transmitter during a transmission.

Note: In the case of transmissions without a carrier wave, the preceding definition applies to the frequency of the corrier wave before its suppression.
In the case of ship stations, the reference frequency is the frequency on which the transmission begins, and the figures appearing in the present table, marked by an asterisk, refer only to frequency separations observed during a ten-minute period of transmission.

In the frequency tolerance, modulation is not considered.
Note 1: The odministrations shall endeovor to profit by the progress of the art in order to reduce frequency tolerances progressively.
Note 2: It shall be understood that ship stations working in shared bands must observe the tolerances applicable to land stations and must conform to article 7, paragraph 21 (2) (a). [No. 186.]
Note 3: Radiotelephone stations with less than 25 watts power, employed by maritime beacons for communications with beacons isolated at sea, shall be comparable, with reference to frequency stability, to mobile stations indicated in C obove.
Note 4: Ships equipped with a transmitter, the power of which is under 100 watts, working in the band of $1560-4000 \mathrm{kc}$ (192.3-75 meters), shall not be subject to the stipulations of column I.

[^1]
## Frequency-band widths occupied by the emissions Cairo rovision, 1938*

The frequency bands necessary for the various types of transmission, at the present state of technical development, are indicated below. This table is based solely upon amplitude modulafion. For frequency or phase modulation, the band widths necessary for the various transmissions are many times greater.

| type of transmission | total width of the band in cycles for transmission with two sidebands |
| :---: | :---: |
| AO Continuous waves, no signaling |  |
| A1 Telegraphy, pure, continuous wave Morse code <br> 1 Baudol code Stop-start printer <br> Scanning-type printer | Numerically equal to the telegraph speed in bauds for the fundamental frequency, 3 times this width for the 3d harmonic, ofc. <br> [for a code of 8 time olements (dots or blanks) per letter and 48 time elements per word, the speod in bauds shall be equal to 0.8 times the speed in words per minute.] <br> $300-1,000$, for speeds of 50 words per minute, according to the conditions of operation and the number of lines scanned lfor example, 7 or 121. IHarmonics are not considered in the above values.) |
| A2 Telegraphy modulated to musical frequency | Figures appearing under $\mathbf{A l}$, plus twice the highest modulation frequency. |
| A3 Commercial radiotelephony <br> Broadcasting | ```Twice the number indicated by the C.C.I.F. Opinions (about 6,000 to 8,000). 1 15,000 to 20,000 .``` |
| A4 Facsimile | Approximately the ratio between the number of picture components ${ }^{2}$ to be transmitted and the number of seconds necessary for the transmission. |
| A5 Tolovision | Approximately the product of the number of picture components ${ }^{2}$ multiplied by the number of pictures transmittod per second. |

It is recognized that the band width may be wider for multiple-channol radiotelephony and secref radiotelephony.
Two picture components, one black and one white, constitute a cycle: thus, the modulation
frequency equals one half the number of components transmithed per second.

- See Footnole under Frequency Tolerances, Treaty Serfos No. 948, Telecommunication:


## Tolerances for the intensity of harmonics

## of fixed, Iand, and broadcasting stations ${ }^{1}$ <br> Cairo revision, 1938*

| frequency bands | telerences |
| :---: | :---: |
| Frequency under $3,000 \mathrm{kc}$ (wavelength <br> above 100 meters) | The field intensity produced by any harmonic must be <br> under $300 \mu \mathrm{v} / \mathrm{m}$ at 5 kilometers from the trans- <br> mitting antenna. |
| Frequency above $3,000 \mathrm{kc}$ (wavelength <br> under 100 meters) | The power of a harmonic in the antenna must be <br> 40 db under the power of the fundamental, but in <br> no case may it be above 200 milliwats. ${ }^{2}$ |

[^2]Classifieation of emissions Cairo revislon, 1938*

1. Emissions shall be classified below according to the purpose for which they are used, assuming their modulation or their possible keying to be only in amplitude.

## a.' Continuous waves:

Type AO. Waves the successive oscillations of which are identical under fixed conditions. ${ }^{1}$
Type A1. Telegraphy on pure continuous waves. A continuous wave which is keyed according to a telegraph code.
Type A2. Modulated telegraphy. A carrier wave modulated at one or more audible frequencies, the audible frequency or frequencies or their combination with the carrier wave being keyed according to a telegraph code. Type A3. Telephony. Waves resulting from the modulation of a carrier wave by frequencies corresponding to the voice, to music, or to other sounds.
Type A4. Facsimile. Waves resulting from the modulation of a carrier wave by frequencies produced at the time of the scanning of a fixed image with a view to its reproduction in a permanent form.
Type A5. Television. Waves resulting from the modulation of a carrier wave by frequencies produced at the time of the scanning of fixed or moving objects. ${ }^{2}$

Note: The band widths to which these emissions correspond are indicated under Frequency-Band Widths Occupied by the Emissions.

## b. Damped waves:

Type B. Waves composed of successive series of oscillations the amplitude of which, after attaining a maximum, decreases gradually, the wave trains being keyed according to a telegraph code.
2. In the above classification, the presence of a carrier wave is assumed in all cases. However, such carrier wave may or may not be transmitted.
This classification does not contemplate exclusion of the use, by the administrations concerned, under specified conditions, of types of waves not included in the foregoing definitions.
3. Waves shall be indicated first by their frequency in kilocycles per second (kc) or in megacycles per second (Mc). Following this indication, there shall be given, in parentheses, the approximate length in meters. In the present Regulations, the approximate value of the wavelength in meters is the quotient of the number 300,000 divided by the frequency expressed in kilocycles per second.

[^3]
## Relation between decibels and power, voltage, and current ratios

The decibel, abbreviated db , is a unit used to express the ratio between two amounts of power, $P_{1}$ and $P_{2}$, existing at two points.

By definition the number of $\mathrm{db}=10 \log _{10} \frac{P_{1}}{P_{2}}$
It is also used to express voltage and current ratios.
The number of $\mathrm{db}=20 \log _{10} \frac{V_{1}}{V_{2}}=20 \log _{10} \frac{I_{1}}{I_{2}}$
Strictly, it can be used to express voltage and current ratios only when the two points at which the voltages or currents in question have identical impedances.

| pewer refle | voltage and current refto | decibels | power ratio | $\qquad$ | declbeld |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1.0233 | 1.0116 | 0.1 | 19.953 | 4.4688 | 13.0 |
| 1.0471 | 1.0233 | 0.2 | 25.119 | 5.0119 | 14.0 |
| 1.0715 | 1.0351 | 0.3 | 31.623 | 5.6234 | 15.0 |
| 1.0965 | 1.0471 | 0.4 | 39.811 | 6.3096 | 16.0 |
| 1.1220 | 1.0593 | 0.5 | 50.119 | 7.0795 | 17.0 |
| 1.1482 | 1.0715 | 0.6 | 63.096 | 7.9433 | 18.0 |
| 1.1749 | 1.0839 | 0.7 | 79.433 | 8.9125 | 19.0 |
| 1.2023 | 1.0965 | 0.8 | 100.00 | 10.0000 | 20.0 |
| 12303 | 1.1092 | 0.9 | 158.49 | 12.589 | 22.0 |
| 1.2589 | 1.1220 | 1.0 | 251.19 | 15.849 | 24.0 |
| 1.3183 | 1.1482 | 1.2 | 398.11 | 19.953 | 26.0 |
| 1.3804 | 1.1749 | 1.4 | 630.96 | 25.119 | 28.0 |
| 1.4454 | 1.2023 | 1.6 | 1000.0 | 31.623 | 30.0 |
| 1.5136 | 12303 | 1.8 | 1584.9 | 39.811 | 32.0 |
| 1.5849 | 1.2589 | 2.0 | 2511.9 | 50.119 | 34.0 |
| 1.6595 | 1.2882 | 22 | 3981.1 | 63.096 | 36.0 |
| 1.7378 | 1.3183 | 2.4 | 6309.6 | 79.433 | 38.0 |
| 1.8197 | 1.3490 | 2.6 | 104 | 100.000 | 40.0 |
| 1.9055 | 1.3804 | 2.8 | $104 \times 1.5849$ | 12589 | 42.0 |
| 1.9953 | 1.4125 | 3.0 | $104 \times 2.5119$ | 158.49 | 44.0 |
| 2.2387 | 1.4962 | 3.5 | $10^{4} \times 3.9811$ | 199.53 | 46.0 |
| 2.5119 | 1.5849 | 4.0 | $10^{4} \times 6.3096$ | 251.19 | 48.0 |
| 2.8184 | 1.6788 | 4.5 | ${ }^{10}{ }^{\circ}$ | 316.23 | 50.0 |
| 3.1623 | 1.7783 | 5.0 | $10^{5} \times 1.5849$ | 398.11 | 52.0 |
| 3.5481 | 1.8836 | 5.5 | $10^{5} \times 2.5119$ | 501.19 | 54.0 |
| 3.9811 | 1.9953 | 6.0 | $10^{5} \times 3.9811$ | 630.96 | 56.0 |
| 5.0119 | 2.2387 | 7.0 | $10^{6} \times 6.3096$ | 794.33 | 58.0 |
| 6.3096 | 2.5119 | 8.0 | 105 | 1,000.00 | 60.0 |
| 7.9433 | 2.8184 | 9.0 | $10^{7}$ |  |  |
| 10.0000 | 3.1623 | 10.0 | $10^{0}$ | 10,000.0 | 80.0 |
| 12.589 | 3.5481 3.9811 | 11.0 | 100 | 31,623 100,000 | 900 1000 |
| 15.849 | 3.9811 | 120 | $10^{\circ}$ | 100,000 | 1000 |

To convert
Decibelh to nopors multiply by 0.1151
Nepers to decibels multiply by 8.686
Where the power ratio is loss than unity, It is usual to Invert the fraction
and express the answer as o decibel lasm

## Engineering and material data

Copper-wire łable-standard annealed copper
American wire gauge (B \& S)*

| $\begin{aligned} & \text { gauge } \\ & \text { no } \end{aligned}$ | diam. chor, mils | cross section |  | ohmess per 1,000 fi of $20^{\circ} \mathrm{C}$ ( $68^{\circ} \mathrm{F}$ ) | $\begin{aligned} & \mathrm{lb} \text { per } \\ & 1,000 \mathrm{ft} \end{aligned}$ | ${ }^{\text {f per lb }}$ | $\left\lvert\, \begin{gathered} H_{\text {per ohm }}^{\text {of }} 20^{\circ} \mathrm{C} \\ \left(68^{\circ} \mathrm{F}\right) \end{gathered}\right.$ | $\begin{aligned} & \text { ohms par bib } \\ & \text { of } 20^{\circ} \mathrm{C} \\ & \left(68^{\circ} \mathrm{F}\right) \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | eircular mils | square inches |  |  |  |  |  |
| 0000 | 460.0 | 211,600 | 0.1662 | 0.04901 | 640.5 | 1.561 | 20,400 | 0.00007652 |
| 000 | 409.6 | 167,800 | 0.1318 | 0.06180 | 507.9 | 1.968 | 16,180 | 0.0001217 |
| $\infty$ | 364.8 | 133,100 | 0.1045 | 0.07793 | 402.8 | 2.482 | 12,830 | 0.0001935 |
| 0 | 324.9 | 105,500 | 0.08289 | 0.09827 | 319.5 | 3.130 | 10,180 | 0.0003076 |
| 1 | 289.3 | 83,690 | 0.06573 | 0.1239 | 253.3 | 3.947 | 8,070 | 0.0004891 |
| 2 | 257.6 | 66,370 | 0.05213 | 0.1563 | 200.9 | 4.977 | 6,400 | 0.0007778 |
| 3 | 229.4 | 52,440 | 0.04134 | 0.1970 | 159.3 | 6.276 | 5,075 | 0.001237 |
| 4 | 204.3 | 41,740 | 0.03278 | 0.2485 | 126.4 | 7.914 | 4,025 | 0.001966 |
| 5 | 181.9 | 33,100 | 0.02600 | 0.3133 | 100.2 | 9.980 | 3,192 | 0.003127 |
| 6 | 162.0 | 26,250 | 0.02062 | 0.3951 | 79.46 | 12.58 | 2,531 | 0.004972 |
| 7 | 14.3 | 20,820 | 0.01635 | 0.4982 | 63.02 | 15.87 | 2,007 | 0.007905 |
| 8 | 128.5 | 16,510 | 0.01297 | 0.6282 | 49.98 | 20.01 | 1,592 | 0.01257 |
|  | 114.4 | 13,090 | 0.01028 | 0.7921 | 39.63 | 25.23 | 1,262 | 0.01999 |
| 10 | 101.9 | 10,380 | 0.008155 | 0.9989 | 31.43 | 31.82 | 1,001 | 0.03178 |
| 11 | 90.74 | 8,234 | 0.006467 | 1.260 | 24.92 | 40.12 | 794 | 0.05053 |
| 12 | 80.81 | 6,530 | 0.005129 | 1.588 | 19.77 | 50.59 | 629.6 | 0.08035 |
| 13 | 71.96 | 5,178 | 0.004067 | 2.003 | 15.68 | 63.80 | 499.3 | 0.1278 |
| 14 | 64.08 | 4,107 | 0.003225 | 2.525 | 12.43 | 80.44 | 396.0 | 0.2032 |
| 15 | 57.07 | 3,257 | 0.002558 | 3.184 | 9.858 | 101.4 | 314.0 | 0.3230 |
| 16 | 50.82 | 2,583 | 0.002028 | 4.016 | 7.818 | 127.9 | 249.0 | 0.5136 |
| 17 | 45.26 | 2,048] | 0.001609 | 5.064 | 6.200 | 161.3 | 197.5 | 0.8167 |
| 18 | 40.30 | 1,624 | 0.001276 | 6.385 | 4.917 | 203.4 | 156.6 | 1.299 |
| 19 | 35.89 | 1,288 | 0.001012 | 8.051 | 3.899 | 256.5 | 124.2 | 2.065 |
| 20 | 31.96 | 1,022 | 0.0008023 | 10.15 | 3.092 | 323.4 | 98.50 | 3.283 |
| 21 | 28.46 | 810.1 | 0.0006363 | 12.80 | 2.452 | 407.8 | 78.11 | 5.221 |
| 22 | 25.35 | 842.4 | 0.0005046 | 16.14 | 1.945 | 514.2 | 61.95 | 8.301 |
| 23 | 22.57 | 509.5 | 0.0004002 | 20.36 | 1.542 | 648.4 | 49.13 | 13.20 |
| 24 | 20.10 | 404.0 | 0.0003173 | 25.67 | 1.223 | 817.7 | 38.96 | 20.99 |
| 25 | 17.90 | 320.4 | 0.0002517 | 32.37 | 0.9699 | 1,031.0 | 30.90 | 33.37 |
| 26 | 15.94 | 254.1 | 0.0001996 | 40.81 | 0.7692 | 1,300 | 24.50 | 53.06 |
| 27 | 14.20 | 201.5 | 0.0001583 | 51.47 | 0.6100 | 1,639 | 19.43 | 84.37 |
| 28 | 12.64 | 159.8 | 0.0001255 | 64.90 | 0.4837 | 2,067 | 15.41 | 134.2 |
| 29 | 11.26 | 126.7 | 0.00009953 | 81.83 | 0.3836 | 2,607 | 12.22 | 213.3 |
| 30 | 10.03 | 100.5 | 0.00007894 | 103.2 | 0.3042 | 3,287 | 9.691 | 339.2 |
| 31 | 8.928 | 79.70 | 0.00006260 | 130.1 | 0.2413 | 4,145 | 7.685 | 539.3 |
| 32 | 7.950 | 63.21 | 0.00004964 | 164.1 | 0.1913 | 5,227 | 6.095 | 857.6 |
| 33 | 7.080 | 50.13 | 0.00003937 | 206.9 | 0.1517 | 6,591 | 4.833 | 1,364 |
| 34 | 6.305 | 39.75 | 0.00003122 | 260.9 | 0.1203 | 8,310 | 3.833 | 2,168 |
| 35 | 5.615 | 31.52 | 0.00002476 | 329.0 | 0.09542 | 10,480 | 3.040 | 3,448 |
| 36 | 5.000 | 25.00 | 0.00001964 | 414.8 | 0.07588 | 13,210 | 2.411 | 5,482 |
| 37 | 4.453 | 19.83 | 0.00001557 | 523.1 | 0.05001 | 16,660 | 1.912 | 8,717 |
| 38 | 3.965 | 15.72 | 0.00001235 | 659.6 | 0.04759 | 21,010 | 1.516 | 13,860 |
| 39 | 3.531 | 12.47 | 0.000009793 | 831.8 | 0.03774 | 26,500 | 1.202 | 22,040 |
| 40 | 3.145 | 9.888 | 0.00000776 | 1,049.0 | 0.02993 | 33,410 | 0.9534 | 35,040 |

## Temperature coefficient of resistance:

The resistance of a conductor at temperature ${ }^{\circ} \mathrm{C}$ is given by
$R_{t}=R_{20}\left[1+0_{50}(t-201]\right.$
where $\mathrm{R}_{20}$ is the resistance of $20^{\circ} \mathrm{C}$ and $\mathrm{on}_{0}$ is the temperature coefficient of resistance of $20^{\circ} \mathrm{C}$.
For copper, $0_{0}=0.00393$. That is, the resistance of a copper conductor increases approxi-
matoly $4 / 10$ of 1 percent per degree contigrade rise in tomperature.

* For additional data on wire, seo poges 36, 37, 38, 60 , and 126 .

Copper-wire table-English and metric units $\dagger$

| Anver wire AWG (BAS) | $\begin{aligned} & \text { Burn } \\ & \text { wire } \\ & \text { gouge } \\ & \text { BWG } \end{aligned}$ | $\begin{aligned} & \text { Impertion } \\ & \text { or Britith } \\ & \text { std } \\ & \text { SWGG } \\ & \text { (NBS) } \end{aligned}$ | English units |  |  | melric ovilis ' |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | diam <br> In inches | weight <br> lbs per wire mile | restistance ohrms per wire mille $20^{\circ} \mathrm{C}$ $\left(68^{\circ} \mathrm{F}\right)$ | $\begin{aligned} & \text { diam } \\ & \text { In } \\ & \text { mm } \end{aligned}$ | $\begin{aligned} & \text { voloht } \\ & \text { kg per } \\ & \text { wire } \\ & \mathrm{km} \end{aligned}$ | $\begin{aligned} & \text { resistance } \\ & \text { ohms per } \\ & \text { wire } \mathrm{km} \\ & 20^{\circ} \mathrm{C} \\ & \left(68^{\circ} \mathrm{F}\right) \\ & \hline \end{aligned}$ |
|  | - | - | . 1968 | 618 | 1.415 | 5.0 | 174.0 | . 879 |
|  | - | - | . 1940 | 600 | 1.458 | 4.928 | 169.1 | . 905 |
|  |  | 6 | . 1920 | 589.2 | 1.485 | 4.875 | 168.2 | . 922 |
| - | - | - | . 1855 | 550 | 1.590 | 4.713 | 155.2 | . 987 |
| 5 |  |  | . 1819 | 528.9 | 1.654 | 4.620 | 149.1 | 1.028 |
| 5 | 7 |  | . 1800 | 517.8 | 1.690 | 4.575 | 146.1 | 1.049 |
|  | - | - | . 1771 | 500 | 1.749 | 4.5 | 141.2 | 1.086 |
|  |  | 7 | . 1762 | 495.1 | 1.769 | 4.447 | 140.0 | 1.098 |
| - | - | - | . 1679 | 450 | 1.945 | 4.260 | 127.1 | 1.208 |
|  | 8 |  | . 1650 | 435.1 | 2.011 | 4.190 | 123.0 | 1.249 |
| 6 |  |  | . 1620 | 419.5 | 2086 | 4.115 | 118.3 | 1.296 |
|  |  | 8 | . 1600 | 409.2 | 2.139 | 4.062 | 115.3 | 1.328 |
|  | - | - | . 1582 | 400 | 2.187 | 4.018 | 113.0 | 1.358 |
| - | - | - | . 1575 | $\overline{395} .3$ | 2.213 | 4.0 | 111.7 | 1.373 |
|  | 9 |  | . 1480 | 350.1 | 2.500 | 3.760 | 98.85 | 1.552 |
| 7 |  |  | . 1443 | 332.7 | 2.630 | 3.665 | 93.78 | 1.634 |
|  |  | 9 | . 1440 | 331.4 | 2641 | 3.658 | 93.40 | 1.641 |
|  | - | - | . 1378 | 302.5 | 2.892 | 3.5 | 85.30 | 1.795 |
| - | - | - | . 1370 | 300 | 2916 | 3.480 | 84.55 | 1.812 |
|  | 10 |  | . 1341 | . 287.0 | 3.050 | 3.405 | 80.95 | 1.893 |
| 8 |  |  | . 1285 | 263.8 | 3.317 | 3.264 | 74.37 | 2.061 |
|  |  | 10 | . 1280 | 261.9 | 3.342 | 3.252 | 73.75 | 2.077 |
| - | - | - | . 1251 | 250 | 3.500 | 3.180 | 70.50 | 2.173 |
|  | - | - | . 1181 | 27.8 | 3.930 | 3.0 | 62.85 | 2440 |
| 9 |  |  | . 1144 | 209.2 | 4.182 | 2.906 | 58.98 | 2.599 |
| - | - | - | . 1120 | 200 | 4.374 | 2845 | 56.45 | 2.718 |
|  | 12 |  | . 1090 | 189.9 | 4.609 | 2.768 | 53.50 | 2862 |
|  |  | 12 | . 1040 | 172.9 | 5.063 | 2.640 | 48.70 | 3.144 |
| * 10 |  |  | . 1019 | 165.9 | 5.274 | 2.588 | 46.77 | 3.277 |
| - | - | - | . 0984 | 154.5 | 5.670 | 2.5 | 43.55 | 3.520 |
| - | - | - | . 0970 | 150 | 5.832 | 2.460 | 42.30 | 3.620 |
|  | \$14 |  | . 0830 | 110.1 | 7.949 | 2108 | 31.03 | 4.930 |
| * 12 |  |  | . 0808 | 104.4 | 8.388 | 2.053 | 29.42 | 5.211 |
|  |  | 14 | . 0801 | 102.3 | 8.556 | 2037 | 28.82 | 5.315 |
| - | - | - | . 0788 | 99.10 | 8.830 | 2.0 | 27.93 | 5.480 |
| *13 |  |  | . 0720 | 82.74 | 10.58 | 1.828 | 23.33 | 6.571 |
| +14 |  |  | . 0641 | 65.63 | 13.33 | 1.628 | 18.50 | 8.285 |
|  |  |  | . 0508 | 41.28 | 21.20 | 1.291 | 11.63 | 13.17 |
| *17 |  |  | . 0453 | 32.74 | 26.74 | 1.150 | 9.23 | 16.61 |
| *18 |  |  | . 0403 | 25.98 | 33.71 | 1.024 | 7.32 | 20.95 |
| *19 |  |  | . 0359 | 20.58 | 42.51 | . 912 | 5.802 | 26.42 |
| -22 |  |  | . 0253 | 10.27 | 85.24 | . 644 | 2.894 | 52.96 |
| - 24 |  |  | . 0201 | 8.46 | 135.5 | . 511 | 1.820 | 84.21 |
| - 26 |  |  | . 0159 | 4.06 | 215.5 | . 405 | 1.145 | 133.9 |
| - 27 |  |  | . 0142 | 3.22 | 271.7 | . 361 | . 908 | 168.9 |
| - 28 |  |  | . 0126 | 2.56 | 342.7 | . 321 | . 720 | 212.9 |

. When used in cable, weight and resistance of wire should be increased about $3 \%$ to allow
for increase due to twist.

+ For additional dato on wire, see pages 35, 37, 38, 60, and 126.

Solid copperweld wire-mechanical and electrical properties

| size | diem | cross section area |  | welght |  |  | $\begin{array}{r} \text { resis } \\ \text { ohms/i00 } \end{array}$ | ne: <br> fin $68^{\circ} \mathrm{F}$ | breaking load, pounds |  | $\begin{aligned} & \text { attenuction-db } \\ & \text { per mille* } \end{aligned}$ |  |  |  | characteristie impedance" |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | circula mils | square Inch | $\begin{aligned} & \text { Per } \\ & 1000 \\ & \text { foet } \\ & \hline \end{aligned}$ | per <br> mille | $\begin{aligned} & \text { peur } \\ & \text { peund } \end{aligned}$ | 40\% | 30\% | $\begin{aligned} & 40 \% \\ & \text { conduct } \end{aligned}$ | $\begin{gathered} 30 \% \\ \text { conduct } \end{gathered}$ |  |  |  | cond | 40\% | $30 \%$ |
| 4 | . 2043 | 41,740 | . 03278 | 115.8 | 611.6 | 8.63 | 0.6337 | 0.8447 | 3,541 | 3,934 | - | - | - | - | - |  |
| 5 | . 1819 | 33,100 | . 02600 | 91.86 | 485.0 | 10.89 | 0.7990 | 1.065 | 2,938 | 3,250 | - | - | - | - | - |  |
| 6 | .1620 | 26,250 | . 02062 | 72.85 | 384.6 | 13.73 | 1.008 | 1.343 | 2,433 | 2,680 | . 078 | . 086 | . 103 | . 109 | 650 | 686 |
| 7 | .1443 | 20,820 | . 01635 | 57.77 | 305.0 | 17.31 | 1.270 | 1.694 | 2,011 | 2,207 | . 093 | . 100 | . 122 | . 127 | 685 | 732 |
| 8 | . 1285 | 16,510 | . 01297 | 45.81 | 241.9 | 21.83 | 1,602 | 2.136 | 1,660 | 1,815 | . 111 | . 118 | . 144 | . 149 | 727 | 787 |
| 9 | . 1144 | 13,090 | . 01028 | 36.33 | 191.8 | 27.52 | 2.020 | 2.693 | 1,368 | 1,491 | . 132 | . 138 | . 169 | . 174 | 776 | 852 |
| 10 | . 1019 | 10,380 | . 008155 | 28.81 | 152.1 | 34.70 | 2.547 | 3.396 | 1,130 | 1,231 | . 156 | . 161 | . 196 | . 200 | 834 | 920 |
| 11 | . 0907 | 8,234 | . 006467 | 22.85 | 120.6 | 43.76 | 3.212 | 4.28 | 896 | 975 | . 183 | . 188 | . 228 | . 233 | 910 | 1,013 |
| 12 | . 0808 | 6,530 | . 005129 | 18.12 | 95.68 | 55.19 | 4.05 | 5.40 | 711 | 770 | 216 | .220 | . 262 | . 266 | 1,000 | 1,120 |
| 13 | . 0720 | 5,178 | . 004067 | 14.37 | 75.88 | 69.59 | 5.11 | 6.81 | 490 | 530 |  |  |  |  | ,100 | 1,120 |
| 14 | .0641 | 4,107 | . 003225 | 11.40 | 60.17 | 87.75 | 6.44 | 8.59 | 400 | 440 |  |  |  |  |  |  |
| 15 | . 0571 | 3,257 | . 002558 | 9.038 | 47.72 | 110.6 | 8.12 | 10.83 | 300 | 330 |  |  |  |  |  |  |
| 16 | . 0508 | 2,583 | . 002028 | 7.167 | 37.84 | 139.5 | 10.24 | 13.65 | 250 | 270 |  |  |  |  |  |  |
| 17 | . 0453 | 2,048 | . 001609 | 5.684 | 30.01 | 175.9. | - 12.91 | 17.22 | 185 | 205 |  |  |  |  |  |  |
| 18 | . 0403 | 1,624 | . 001276 | 4.507 | 23.80 | 221.9 | 16.28 | 21.71 | 153 | 170 |  |  |  |  |  |  |
| 19 | . 0359 | 1,288 | . 001012 | 3.575 | 18.87 | 279.8 | 20.53 | 27.37 | 122 | 135 |  |  |  |  |  |  |
| 20 | . 0320 | 1,022 | . 0008023 | 2.835 | 14.97 | 352.8 | 25.89 | 34.52 | 100 | 110 |  |  |  |  |  |  |
| 21 | . 0285 | 810.1 | . 0006363 | 2248 | 11.87 | 444.8 | 32.65 | 43.52 | 73.2 | 81.1 |  |  |  |  |  |  |
| 22 | . 0253 | 642.5 | . 0005046 | 1.783 | 9.413 | 560.9 | 41.17 | 54.88 | 58.0 | 64.3 |  |  |  |  |  |  |
| 23 | . 0222 | 509.5 | . 0004002 | 1.414 | 7.465 | 707.3 | 51.92 | 69.21 | 48.0 | 51.0 |  |  |  |  |  |  |
| 24 | . 0201 | 404.0 | . 00003173 | 1.121 | 5.920 | 891.9 | 65.46 | 87.27 | 36.5 | 40.4 |  |  |  |  |  |  |
| 25 | . 0179 | 320.4 | . 00002517 | 0.889 | 4.695 | 1,125 | 82.55 | 110.0 | 28.9 | 32.1 |  |  |  |  |  |  |
| 26 | . 0159 | 254.1 | . 0001996 | 0.705 | 3.723 | 1,418 | 104.1 | 138.8 | 23.0 | 25.4 |  |  |  |  |  |  |
| 27 | . 0142 | 201.5 | .0001583 | 0.559 | 2.953 | 1,788 | 131.3 | 175.0 | 18.2 | 20.1 |  |  |  |  |  |  |
| 28 | . 0126 | 159.8 | . 0001255 | 0.443 | 2.342 | 2,255 | 165.5 | 220.6 | 14.4 | 15.9 |  |  |  |  |  |  |
| 29 | . 0113 | 126.7 | . 0000995 | 0.352 | 1.857 | 2,843 | 208.7 | 278.2 | 11.4 | 12.6 |  |  |  |  |  |  |
| 30 | . 0100 | 100.5 | . 0000769 | 0.279 | 1.473 | 3,586 | 263.2 | 350.8 | 9.08 | 10.0 |  |  |  |  |  |  |
| 31 | . 0089 | 79.70 | . 0000626 | 0.221 | 1.168 | 4,521 | 331.9 | 442.4 | 7.20 | 7.95 |  |  |  |  |  |  |
| 32 | . 0080 | 63.21 | . 0000496 | 0.175 | 0.926 | 5,701 | 418.5 | 557.8 | 5.71 | 6.30 |  |  |  |  |  |  |
| 33 | . 0071 | 50.13 | . 00000394 | 0.139 | 0.734 | 7,189 | 527.7 | 703.4 | 4.53 | 5.00 |  |  |  |  |  |  |
| 34 | . 0063 | 39.75 | . 00000312 | 0.110 | 0.582 | 9,065 | 665.4 | 1887.0 | 3.59 | 3.97 |  |  |  |  |  |  |
| 35 | . 0056 | 31.52 | . 0000248 | 0.087 | 0.462 | 11,430 | 839.0 | 1,119 | 2.85 | 3.14 |  |  |  |  |  |  |
| 36 | . 0050 | 25.00 | . 00000196 | 0.069 | 0.366 | 14,410 | 1,058 | 1,410 | 2.26 | 2.49 |  |  |  |  |  |  |
| 37 | . 0045 | 19.83 | . 0000156 | 0.055 | 0.290 | 18,180 | 1,334 | 1,778 | 1.79 | 1.98 |  |  |  |  |  |  |
| 38 | . 0040 | 15.72 | . 0000123 | 0.044 | 0.230 | 22,920 | 1,682 | 2,243 | 1.42 | 1.57 |  |  |  |  |  |  |
| 39 | . 0035 | 12.47 | . 00000979 | 0.035 | 0.183 | 28,900 | 2,121 | 2,828 | 1.13 | 1.24 |  |  |  |  |  |  |
| 40 | . 0031 | 9.89 | . 00000777 | 0.027 | 0.145 | 36,440 | 2,675 | 3,566 | 0.893 | 0.986 |  |  |  |  |  |  |

due to a shortage of facilities for making these smaller sizes.

- DP Insulators, 12 -Inch Wire Spocing, 1000 eycles
for additional information on wire, see poges $35,36,38,60$, and 126.

Standard stranded copper conductors

## American wire gauge

| circular mits | $\begin{aligned} & \text { size } \\ & \text { AWG } \end{aligned}$ | number of wires | Individual wire diam Inches | cable diom inches | area square inches | woight <br> Ibs per 1000 ft | weight <br> Ibs per mile | *maximum resistance ohms / 1000 f of $20^{\circ} \mathrm{C}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 211,600 | 4/0 | 19 | . 1055 | . 528 | 0.1662 | 653.3 | 3,450 | 0.05093 |
| 167,800 | 3/0 | 19 | . 0940 | . 470 | 0.1318 | 518.1 | 2,736 | 0.06422 |
| 133,100 | 2/0 | 19 | . 0837 | . 419 | 0.1045 | 410.9 | 2,170 | 0.08097 |
| 105,500 | 1/0 | 19 | . 0745 | . 373 | 0.08286 | 325.7 | 1.720 | 0.1022 |
| 83,690 | 1 | 19 | . 0664 | . 332 | 0.06573 | 258.4 | 1,364 | 0.1288 |
| 66,370 | 2 | 7 | . 0974 | . 292 | 0.05213 | 204.9 | 1,082 | 0.1624 |
| 52,640 | 3 | 7 | . 0867 | . 260 | 0.04134 | 162.5 | 858.0 | 0.2048 |
| 41,740 | 4 | 7 | . 0772 | . 232 | 0.03278 | 128.9 | 680.5 | 0.2582 |
| 33,100 | 5 | 7 | . 0688 | . 206 | 0.02600 | 102.2 | 539.6 | 0.3256 |
| 26,250 | 6 | 7 | . 0612 | . 184 | 0.02362 | 81.05 | 427.9 | 0.4105 |
| 20,820 | 7 | 7 | . 0545 | . 164 | 0.01635 | 64.28 | 339.4 | 0.5176 |
| 16,510 | 8 | 7 | . 0486 | . 146 | 0.01297 | 50.98 | 269.1 | 0.6528 |
| 13,090 | 9 | 7 | . 0432 | . 130 | 0.01028 | 40.42 | 213.4 | 0.8233 |
| 10,380 | 10 | 7 | . 0385 | . 116 | 0.008152 | 32.05 | 169.2 | 1.038 |
| 6,530 | 12 | 7 | . 0305 | . 0915 | 0.005129 | 20.16 | 106.5 | 1.650 |
| 4,107 | 14 | 7 | . 0242 | . 0726 | 0.003226 | 12.68 | 66.95 | 2.624 |
| 2,583 | 16 | 7 | . 0192 | . 0576 | 0.002029 | 7.975 | 42.11 | 4.172 |
| 1,624 | 18 | 7 | . 0152 | . 0456 | 0.001275 | 5.014 | 26.47 | 6.636 |
| 1,022 | 20 | 7 | . 0121 | . 0363 | 0.008027 | 3.155 | 18.66 | 10.54 |

* The resistance values in this table are trade maxima for saft ar annealed capper wire and are higher than the average values far cammercial cable. The fallawing values for the canductivity and resistivity of capper of $20^{\circ}$ centigrade were used:
Canductivify in terms af International Annealed Capper Standard $\quad 98.16 \%$
Resistivity in pounds per mile-ahm
The resistance of hard drawn capper is slightly greater than the values given, being about $2 \%$ to $3 \%$ greater for sizes from 4/0 to 20 AWG.


## Machine screw head styles

Method of length measurement


Standard machine screw dafa including hole sizes


All dimensions in Inches
Clearance drill sizes ore practical values for use of the engineer or technician doing his own shop work.
t Tap drill sizes are for use in hand tapping matertal such as brass or soff stoel. For coppor
aluminum, or Norway iron, the drill should be a slze or two lorger diameter than shown. For cost íron and bakelite, or for very thin material, the top drill should be a size or two smaller diameter than shown.

| meforlal | dielectric consfant |  |  | electrical properties* power facter |  |  | diolectic strength kv/mm $\dagger$ | resistivity ohms- cm $25^{\circ} \mathrm{C}$ | physical propertios |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $60 \sim$ | $10^{6} \sim$ | 109~ | 60~ | 10¢ | $10^{8} \sim$ |  |  | $\begin{aligned} & \text { expansion } \\ & \text { per }{ }^{\circ} \mathrm{C} \end{aligned}$ | point |
| Aniline Formaldahyde Resin | 3.6 | 3.5 | 3.4 | . 003 | . 007 | . 004 | 16-25 | $>1012$ | $5.4 \times 10^{-6}$ | $260^{\circ} \mathrm{F}$ |
| Cosein | 3.6 | 6.2 | 3.4 | . 0 | . 052 | . 04 | 16-28 | Poor | $5.4 \times 10^{-6}$ | $200^{\circ} \mathrm{F}$ |
| Cellulose Acetate (plostic) | 4.6 | 3.9 | 3.4 | . 007 | . 039 | . 039 | 10-14 | $10^{10}$ | $6-15 \times 10^{-8}$ | 100-190 ${ }^{\circ} \mathrm{F}$ |
| Cellulose Acetobutyrate | 3.6 | 3.2 | 3.0 | . 004 | . 017 | . 019 | 10-16 | $10^{10}$ | $11-17 \times 10^{-6}$ | 110-1800 F |
| Ebonlte | 3.0 | 2.8 | 2.8 | . 008 | . 006 | . 004 | 18 | $2 \times 10^{31}$ | $7 \times 10^{-6}$ | $140^{\circ} \mathrm{F}$ |
| Ethyl Cellulose 707 | 4.0 | 3.4 | 3.2 | . 005 | . 028 | . 024 | 16-28 | $1{ }^{104}$ | $3.4 \times 10^{-4}$ | $120^{\circ} \mathrm{F}$ |
| Gloss, Coming 707 | 4.0 | 4.0 | 4.0 | . 0006 | . 0008 | . 0012 |  | $1.5 \times 10^{11}$ of $250^{\circ} \mathrm{C}$ | $31 \times 10^{-7}$ | $1400^{\circ} \mathrm{F}$ |
| Gloss, Corning 774 | 5.6 | 5.2 | 5.0 | . 0136 | . 0048 | . 008 |  | $1.4 \times 10^{\circ}$ of $250^{\circ} \mathrm{C}$ | $33 \times 10^{-7}$ | $1500^{\circ} \mathrm{F}$ |
| Glass, Corning 790 | 3.9 5 | 3.9 | 3.9 | . 0006 | . 0006 | . 0008 |  | $5.2 \times 10^{\circ}$ af $250^{\circ} \mathrm{C}$ | $8 \times 10^{-7}$ | $2600^{\circ} \mathrm{F}$ |
| Glass, Corning 7052 Halowax | 5.2 3.8 | 5.1 3.7 | 5.1 3.4 | .008 .002 | . 0024 | .0036 |  | $1 \times 10^{\circ}$ af $250^{\circ} \mathrm{C}$ | $47 \times 10^{-7}$ | $1300^{\circ} \mathrm{F}$ |
| Halowax | 3.8 | 3.7 6.0 | 3.4 | . 002 | .0014 .0018 | . 105 |  | $10^{12} 10^{46}$ |  | $190^{\circ} \mathrm{F}$ |
| Molomine Formaldehyde Resin | 7.5 | 4.5 | 4.5 | . 08 | . 08 | . 03 | 18 |  | $3.5 \times 10^{-6}$ | $260^{\circ} \mathrm{F}$ |
| Methyl Mothocrylato-a luclie HM119 | 3.3 | 2.6 | 2.6 | . 068 | . 015 | . 007 | 16 | $10^{15}$ | $11-14 \times 10^{-6}$ | $160^{\circ} \mathrm{F}$ |
| Mica b Plexiglos | 3.5 | 2.6 | 2.6 | . 084 | . 015 | . 007 | 16 | 1045 | $8 \times 10^{-6}$ | $160^{\circ} \mathrm{F}$ |
| Mico Mycalex 364 | 5.45 | 5.4 | 5.4 | . 005 | . 0003 | . 0003 |  | $5 \times 1014$ |  |  |
| Mycolax 364 Nylon | 7.1 3.6 | 7.0 3.6 | 7.0 3.6 | . 0064 | .0021 .020 | . 0022 | 14 |  | $8.9 .9 \times 10^{-0}$ | $6600^{\circ} \mathrm{F}$ |
| Porallin Oil | 2.2 | 2.2 | 2.2 | . 0001 | .0001 | . 00004 | 15 | $10 \times$ | $5.7 \times 10^{-6}$ | ${ }_{\text {liquid }} 160^{\circ} \mathrm{F}$ |
| Patroleum Wax IParafin Wax | 2.25 | 2.25 | 2.25 | . 0002 | . 0002 | .0002 | 8-12 | 1018 | $7.1 \times 10^{-1}$ | M.P. $132^{\circ} \mathrm{F}$ |
| Phenot Formaldehyde Resins a general purpose |  |  |  |  |  |  |  |  |  |  |
| it general purpose | 5.5 4.6 | 4.5 | 4.0 4.3 | . 018 | .014 .006 | .014 .012 | 14 | 104 | $3-4 \times 10^{-6}$ | $275{ }^{\circ} \mathrm{F}$ |
| c. cost | 8.0 | 8.0 | 8.0 | . 05 | . 05 | . 08 | 10 |  | $7.5-15 \times 10^{-6}$ | $212^{\circ} \mathrm{F}$ $140^{\circ} \mathrm{F}$ |
| Phenol Furfural Resins | 7.0 | 5.0 | 4.0 | . 20 | . 04 | . 05 |  |  | $7.015 \times 10$. | 140 |
| Polyethylene | 2.25 | 2.25 | 225 | . 0003 | . 0003 | . 0003 | 40 | $>1005$ | Varios | $220{ }^{\circ} \mathrm{F}$ |
| Polyisobutylene MW 100,000 | 2.20 | 2.22 | 2.22 | . 0003 | . 0003 | . 0004 |  | $10^{13}$ | Varles | $>0^{\circ} \mathrm{F}$ |
| Polystyrone MW 80,000 | 2.55 | 2.53 | 2.52 | . 0002 | . 0002 | . 0003 | 20-30 | 1017 | $7 \times 10^{-6}$ | $175{ }^{\circ} \mathrm{F}$ |
| Polyvinyl Carbazole | 2.95 | 2.95 | 2.95 | . 0017 | . 0005 | . 0006 | 31-40 |  | $4.5-5.5 \times 10^{-6}$ | $300^{\circ} \mathrm{F}$ |
| Polywinyl Chlor-Acelate Polyvinyl Chloride | 3.2 | 2.9 | 2.8 | . 009 | . 014 | . 009 |  |  |  | $180^{\circ} \mathrm{F}$ |
| Polyvinyl Chioride Polyvinylidine Chloride-Saran | 3.2 4.5 | 2.9 3.0 | 2.9 2.8 | ${ }^{.012}$ | .016 .046 | . 0008 |  | 1015 |  | $180^{\circ} \mathrm{F}$ |
| Quartz tfused) | 3.9 | 3.8 | 3.8 | . 0009 | . 0002 | . 0002 | 60 | 10 | $1.58 \times 10^{-7}$ | $175{ }^{\circ} \mathrm{F}$ $3000^{\circ} \mathrm{F}$ |
| Shallac | 3.9 | 3.5 | 3.1 | . 006 | . 031 | . 030 | $\omega$ | $10^{18}$ | $5.7 \times 10^{-7}$ | $3000 \%$ |
| Styroloy 22 | 2.4 | 2.4 | 2.4 | . 0010 | . 0012 | . 0043 | 30 | $10^{14}$ | $1.8 \times 10^{-1}$ | $150^{\circ} \mathrm{F}$ |
| Styromle | 2.9 | 2.75 | 2.73 | . 003 | . 0002 | .0002 |  |  | $7 \times 10^{-6}$ | $175^{\circ} \mathrm{F}$ |
| Styromic HT | 2.64 | 2.64 | 2.62 | . 0002 | . 0002 | .0002 |  |  | 8 $\times 10-1$ | $250^{\circ} \mathrm{F}$ |
| Urea formaldehyda Resins Wood-Áfrican Mahogany idryl | 6.6 2.4 | 5.6 2.1 | 5.0 2.1 | .032 .01 | . 023 | .05 | 15 | 104 | $2.6 \times 10^{-8}$ | $260^{\circ} \mathrm{F}$ |
| Wolse ldryt | 1.4 | 2.4 | 1.3 | . 048 | . 012 | . 013 . |  |  |  |  |

- Values given are average for the materials listed.
tTo convert Killovolis per millimeter to volis per mil, multiply by 25.4


## Plastics: trade names

| mrade $n$ | composifion | trade name | composition |
| :---: | :---: | :---: | :---: |
| Acryloid | Methacrylate Resin | Indur | Phenot formaldehyde |
| Alvor | Polyvinyl Acetal | Kodapak | Callulose Acetate |
| Amerith | Cellulose Nirrate | Kodapak II | Cellulose Acetobutyrate |
| Ameripol | Butadiene Copolymer | Koroseal | Modified Polyvinyl Chloride |
| Ameroid | Casain | Lectrofilm | Polyvinyl Carbazole Icon- |
| Bakelite | Phenol Formaldehyde |  | denser material; mica sub- |
| Bakelite | Urea Formaldehyde |  | stitute) |
| Bakelite | Cellulose Acetate | Loalin | Polystyrene |
| Bakelite | Polystyrene | Lucite | Methyl Methacrylate Resin |
| Beckamine | Urea formaldehyde Resins | Lumarith | Cellulose Acetate |
| Beotle | Urea Formaldehyde | Lumarith X | Collulose Acetare |
| Butacite | Polyvinyl Butyral | Lustron | Polystyrene |
| Butvar | Polyvinyl Butyral | luvican | Polyvinyl Carbazole |
| Cardolite | Phenol-aldehyde (coshew nut derivative) | Makalot Marblette | Phenol formaldehyde Phenol formaldehyde (cas |
| Cerex | Styrene Copolymer | Marbon B | Cyclized Rubber |
| Catalin | Phenol formaldehyde (cast) | Marbon C | Rubber Hydrochloride |
| Collophane | Regenerated Cellulose film | Melmac | Melamine formaldehyde |
| Celluloid | Cellulose Nitrate | Methocel | Methyl Cellulose |
| Cibanite Crysialite | Aniline Formaldehyde Acrylate and Methacrylate | Micabond | Glycerol Phtholic Anhydride, Mica |
| Crystalite | Acrylate and Methacrylate Resin | Micarta | Phenol formaldehyde llami- |
| Cumar Dilectene 100 | Cumarone-indene Resin <br> Aniline Formaldehyde Syn- | Monsanto | nation) Cellulose Nitrate |
| Dilectene 100 | thetic Resin | Monsanto | Polyvinyl Acetals |
| Dilecto | Ureo Formaldehyde (phenol formaldehyde) | Monsanto Monsanto | Collulose Acetate Phenol formaldehyde |
| Dilecto UF | Ureo Formaldehyde | Mycalex | Mica Bonded Class |
| Distrene | Polystyrene | Neoprene | Chloroprene Synthetic Rub- |
| Durez | Phenol formaldehyde |  | ber |
| Durito | Phenol formaldehyde | Nevidene | Cumarone-indene |
| Durite | Phenol Furfural | Nitron | Cellulose Nitrate |
| Erinofort | Cellulose Acetate | Nixonite | Cellulose Acetate |
| Erinoid | Cosein | Nixonoid | Cellulose Nirrate |
| Ethocel | Ethyl Cellulose | Nylon | Synthetic Polyamides and |
| Ethocel PG | Ethyl Collulose |  | Super Polyamides |
| Ethofoil | Ethyl Cellulose | Nypene | Polyterpene Resins |
| Ethomelt | Ethyl Cellulose thot pouring | Opolon | Phenol formaldehyde Phenol Formaldehyde llam |
| Ethomulsion | Ethyl Cellulose llacquer emulsion) | Panelyte | nate) <br> Phenol formaldehyde |
| Fibestos | Cellulose Acetate | Parlon | Chlorinated Rubber |
| Flamenol | Vinyl Chloride (plasticized) | Perspex | Methyl Methacrylic Ester |
| formico | Phenol formaldehyde (lamination) | Ploskon Plastacole | Urea Formaldehyde Cellulose Acetate |
| Formvar | Polyvinyi formal | Plexiglas | Methyl Methacrylate |
| Galalith | Cosein | Plexiglas | Acrylate and Methacrylate |
| Gelva | Polyvinyl Acetate |  | Resin |
| Gemstone | Phenol formaldehyde | Ploskon | Urea formaldehyde |
| Geon | Polyvinyl Chloride | Plastacele | Cellulose Acetate |
| Glyptal | Glycerol-phthalic Anhydride | Pliofim | Rubber Hydrochloride |
| Haveg | Phenol Formaldehyde Asbes. | Plioform | Rubber Derivative |
|  | tos | Pliolite | Rubber Derivative |
| Hercose AP | Cellulose Acetate Propionate | Polyfibre | Polystyrene |
| Heresite | Phenol Formaldehyde | Polythene | Polyethylene |

Plastics: trade names continued

| trade name | composition | frade name | composition |
| :---: | :---: | :---: | :---: |
| Protectoid | Cellulose Acetate | Styron | Polystyrene |
| Prystal | Phenol Formaldehyde | Super Styrex | Polystyrene |
| Pyralin | Cellulose Nitrote | Synthane | Phenol Formaldehyde |
| PVA | Polyvinyl Alctohol | Tenite | Cellulose Acetate |
| Pyralin | Cellulose Nirrore | Tenite II | Cellulose Acetobutyrate |
| Resinox | Phenol Formaldehyde | Textolite | Various |
| Resoglaz | Polystyrene | Textolite 1421 | Cross-linked Polystyrene |
| Rhodolene M | Polystyrene | Tornesit | Rubber Derivative |
| Rhodoid | Cellulose Acetate | Trolitul | Polystyrene |
| Ronilla L | Polystyrene | Vec | Polyvinylidene Chloride |
| Ronilla M | Polystyrene | Victron | Polystyrene |
| Saflex | Polyvinyl Butyral | Vinylite A | Polyvinyl Acetate |
| Saron | Polyvinylidene Chloride | Vinylite Q | Polyvinyl Chloride |
| Styraflex | Polystyrene | Vinylite V | Vinyl Chloride-Acetate Co- |
| Styramic | Polystyrene-Chlorinated Diphenyl | Vinylite X | polymer <br> Polyvinyl Butyral |
| Styramic HT | Polydichlorstyrene |  |  |

## Wind velocities and pressures

| Indicoted velocities miles per hour* $\mathbf{V}_{i}$ | actual velociftes miles per hour $V_{a}$ | cylindrical surfaces pressure lbs per sq fi projected areos $\mathbf{P}=0.0025 \mathbf{V}_{\boldsymbol{a}^{2}}$ | flat surfaces pressure lbs per square foof $P=0.0042 V_{a^{2}}$ |
| :---: | :---: | :---: | :---: |
| 10 | 9.6 | 0.23 | 0.4 |
| 20 | 17.8 | 0.8 | 1.3 |
| 30 | 25.7 | 1.7 | 2.8 |
| 40 | 33.3 | 2.8 | 4.7 |
| 50 | 40.8 | 4.2 | 7.0 |
| 60 | 48.0 | 5.8 | 9.7 |
| 70 | 55.2 | 7.6 | 12.8 |
| 80 | 62.2 | 9.7 | 16.2 |
| 90 | 69.2 | 12.0 | 20.1 |
| 100 | 76.2 | 14.5 | 24.3 |
| 110 | 83.2 | 17.3 | 29.1 |
| 120 | 90.2 | 20.3 | 34.2 |
| 125 | 93.7 | 21.9 | 36.9 |
| 130 | 97.2 | 23.6 | 39.7 |
| 140 | 104.2 | 27.2 | 45.6 |
| 150 | 111.2 | 30.9 | 51.9 |
| 160 | '118.2 | 34.9 | 58.6 |
| 170 | 125.2 | 39.2 | 65.7 |
| 175 | 128.7 | 41.4 | 69.5 |
| 180 | 132.2 | 43.7 | 73.5 |
| 190. | 139.2 | 48.5 | 81.5 |
| 200 | 146.2 | 53.5 | 89.8 |

[^4]
## Temperature chart of heated metals



Physical constants of various metals and alloys*

| material | relative resisiance | temp coefficient of resistivity at $20^{\circ} \mathrm{C}$ | speciffc gravity | coefficlent of thermal cond K wafts $/ \mathrm{cm}^{\circ} \mathrm{C}$ | melting polnt ${ }^{\circ} \mathrm{C}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Advance (55 Cu 45 Ni ) | soo | Constantan |  |  |  |
| Aluminum | 1.64 | . 004 | 2.7 | 2.03 | 660 |
| Antimony | 24.21 | . 0036 | 6.6 | 0.187 | 630 |
| Arsenic | 19.33 | . 0042 | 5.73 | - | sublimes |
| Bismuth | 69.8 | . 004 | 9.8 | 0.0755 | 270 |
| Brass (66 Cu 34 Zn ) | 3.9 | . 002 | 8.47 | 1.2 | 920 |
| Cadmium | 4.4 | . 0038 | 8.64 | 0.92 | 321 |
| Chromax 115 Cr 35 Ni balance Fol | 58.0 | . 00031 | 7.95 | 0.130 | 1380 |
| Cobalt | 5.6 | . 0033 | 8.71 | - | 1480 |
| Constantan (55Cu45Ni) | 28.45 | $\pm .0002$ | 8.9 | 0.218 | 1210 |
| Copper-annealed | 1.00 | . 00393 | 8.89 | 3.88 | 1083 |
| hard drawn | 1.03 | . 00382 | 8.89 | - | 1083 |
| Eureka ( 55 Cu 45 Ni ) | soo | Constantan |  |  |  |
| Gas carban | 2900 | -. 0005 | - | - | 3500 |
| Gold | 1.416 | . 0034 | 19.32 | 0.296 | 1063 |
| Ideal (55 Cu 45 Ni ) |  |  |  |  |  |
| Iron, pure | 5.6 | $.0052-.0062$ | 7.8 | 0.67 | 1535 |
| Kovar A 129 Ni 17 Co 0.3 Mn balance Fol | 28.4 | - | 8.2 | 0.193 | 1450 |
| Lead | 12.78 | . 0042 | 11.37 | 0.344 | 327 |
| Magnesium | 2.67 | . 004 | 1.74 | 1.58 | 651 |
| Mongonin 184 Cu 12 Mn $4 \mathrm{Ni})$ | 26 | $\pm .00002$ | 8.5 | 0.63 | 910 |
| Mercury | 55.6 | . 00089 | 13.55 | 0.063 | -38.87 |
| Molybdonum, drawn | 3.3 | . 0045 | 10.2 | 1.46 | 2630 |
| Monel metal 157 Ni 30 Cu 1.4 Fol Mn ) | 27.8 | . 002 | 8.8 | 0.25 | 1300-1350 |
| Nichromel 165 Ni 12 Cr 23 Fol | 65.0 | . 00017 | 8.25 | 0.132 | 1350 |
| Nickel | 5.05 | . 0047 | 8.85 | 0.6 | 1452 |
| Nickel silver 164 Cu $18 \mathrm{Zn} 18 \mathrm{Ni}$ | 16.0 | . 00026 | 8.72 | 0.33 | 1110 |
| Palladium | 6.2 | . 0038 | 12.16 | 0.7 | 1557 |
| Phosphor-bronze 14 Sn 0.5 P balance Cu | 5.45 | - | 8.9 | 0.82 | 1050 |
| Platinum | 6.16 | . 0038 | 21.4 | 0.695 | 1771 |
| Silver | 0.95 | . 004 | 10.5 | 4.19 | 960.5 |
| Steol, manganese (13 Mn 1 C 85 Fol | 41.1 | - | 7.81 | 0.113 | 1510 |
| Stool, SAE 1045 10.4-0.5 C balance Fol | 7.6-12.7 | - | 7.8 | 0.59 | 1480 |
| Steol, 18-8 stainloss 10.1 C 18 Cr 8 Ni balance Fol | 52.8 | - | 7.9 | 0.163 | 1410 |
| Tontalum | 9.0 | . 0033 | 16.6 | 0.545 | 2850 |
| Tin | 6.7 | . 0042 | 7.3 | 0.64 | 231.9 |
| Tophot A 180 Ni 20 Cr$)$ | 62.5 | . $02-.07$ | 8.4 | 0.136 | 1400 |
| Tungsten | 3.25 | . 0045 | 19.2 | 1.6 | 3370 |
| Zinc | 3.4 | . 0037 | 7.14 | 1.12 | 419 |
| Zirconium | 2.38 | . 0044 | 6.4 | - | 1860 |

[^5]
## Physical constants of various metals and alloys continued

## Definitions of physical constants in preceding table

The preceding table of relative resistances gives the ratio of the resistance of any material to the resistance of a piece of annealed copper of identical physical dimensions and temperature.

1. The resistance of any substance of uniform cross-section is proportional to the length and inversely proportional to the cross-sectioned area.
$R=\frac{\rho L}{A^{\prime}}$ where $\rho=$ resistivity, the proportionality constant,
$L=$ length, $A=$ cross-sectional area, $R=$ resistance in ohms.
If $L$ and $A$ are measured in centimeters, $\rho$ is in ohm-centimeters.
If $L$ is measured in feet, and $A$ in circular mils, $\rho$ is in ohm-circular mils per foot. Relative resistance $=\rho$ divided by the resistivity of copper $1.7241 \times 10^{-6}$ ohm-cm).
2. The temperature coefficient of resistivity gives the ratio of the change in resistivity due to a change in temperature of $1^{\circ} \mathrm{C}$ relative to the resistivity at $20^{\circ} \mathrm{C}$. The dimensions of this quantity are ohms per ${ }^{\circ} \mathrm{C}$ per ohm or $1 /{ }^{\circ} \mathrm{C}$.

The resistance at any temperature is:
$R=R_{0} \|+\alpha T, R_{0}=$ resistance at $0^{\circ}$ in ohms, $T=$ temperature in degrees centigrade, $\alpha=$ temperature coefficient of resistivity $1 /{ }^{\circ} \mathrm{C}$.
3. The specific gravity of a substance is defined as the ratio of the weight of a given volume of the substance to the weight of an equal volume of water.

In the cgs system, the specific gravity of a substance is exactly equal to the weight in grams of one cubic centimeter of the substance.
4. Coefficient of thermal conductivity is defined as the time rate of heat transfer through unit thickness, across unit area, for a unit difference in temperature. Expressing rate of heat transfer in watts, the coefficient of thermal conductivity
$K=\frac{W L}{A \Delta T}$
$W=$ watts, $L=$ thickness in $\mathrm{cm}, A=$ area in $3 q \mathrm{~cm}, \Delta T=$ temperature in ${ }^{\circ} \mathrm{C}$.
5. Specific heat is defined as the number of calories required to heat one gram of a substance one degree Centigrade.
$H=m s \Delta T$ or change in heat $m=$ mass in grams
$\Delta T=$ temperature change ${ }^{\circ} \mathrm{C} \quad s=$ specific heat in cal/gm $/{ }^{\circ} \mathrm{C}$

Thermocouples and their characteristics

| type | copper/constantan | Iron/constanlan | chromel/constonfan | chromel/alumel | $\left\lvert\, \begin{gathered} \text { plafinum/platinum } \\ \text { rhodium (10) } \end{gathered}\right.$ | $\left\lvert\, \begin{gathered}\text { platinum/platinum } \\ \text { rhodium (13) }\end{gathered}\right.$ | carbon/silicon carbid. |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Composition, percent | 100 Cu 54 Cu 46 Ni <br> 99.9 Cu 55 Cu <br>  60 Cu <br> 40 Ni  | 100 fe55 Cu 44 Ni <br>  <br>  <br> Si |  | $\|$$90 \mathrm{Ni} \mathrm{10Cr}$ 95 Ni 2 Al 2 Mn 1 Si <br> 89.6 Ni 8.9 Cr $97 \mathrm{Ni} 3 \mathrm{Al}+\mathrm{Si}$ <br> 89 Ni 10 Cr 94 Ni 2 Al 1 Si <br> 89 Ni 9.8 Cr 2.5 Mn 0.5 Fa <br> 1 Fa 0.2 Mn  | Pt 90Pt 10Rh | Pt 87Pt 13Rh | $C$ SIC |
| Range of application, ${ }^{\circ} \mathrm{C} \cdot$ | $1-250$ 10 +600 | $1-200$ 10 + 1050 | 10 to 1100 | 10 10 1100 | 10 to 1550 |  | \|ro 2000 |
| Resistivity, micro-ohm.C.M.\| | 11.7549 | 11049 | $170 \quad 49$ | $170 \quad 29.4$ | 11021 |  |  |
| Temperature coefficient of resistivity, ${ }^{\circ} \mathrm{C}$ | $\mid .0039$. 00001 | 1.005 . 00001 | 1.00035 . 0002 | . 00035 . 000125 | . 0030 . 0018 |  |  |
| Molting femperapure, ${ }^{\circ} \mathrm{C}$ | \|1085 1190 | 115351190 | $11400 \quad 1190$ | 11400 | 11755 . 1700 |  | 130002700 |
| EMF in my reference junction at $0^{\circ} \mathrm{C}$ | $100{ }^{\circ} \mathrm{C}$ 4.24 mv <br> 200 9.06 <br> 300 14.42 | $\begin{array}{\|cc\|}100^{\circ} \mathrm{C} & 5.28 \mathrm{mv} \\ 200 & 10.78 \\ 400 & 21.82 \\ 600 & 33.16 \\ 800 & 45.48 \\ 1000 & 58.16\end{array}$ | $100^{\circ} \mathrm{C}$ 6.3 mv <br> 200 13.3 <br> 400 28.5 <br> 600 44.3 | $\begin{array}{\|cc\|}100^{\circ} \mathrm{C} & 4.1 \mathrm{mv} \\ 200 & 8.13 \\ 400 & 16.39 \\ 600 & 24.90 \\ 800 & 33.31 \\ 1000 & 41.31 \\ 1200 & 48.85 \\ 1400 & 35.81\end{array}$ | $\begin{array}{\|cc\|}100^{\circ} \mathrm{C} & 0.643 \pi \mathrm{~V} \\ 200 & 1.436 \\ 400 & 3.251 \\ 600 & 5.222 \\ 800 & 7.330 \\ 1000 & 9.569 \\ 1200 & 11.924 \\ 1400 & 14.312 \\ 1600 & 16.674\end{array}$ | $100^{\circ} \mathrm{C}$ 0.646 mv <br> 200 1.464 <br> 400 3.398 <br> 600 5.561 <br> 800 7.927 <br> 1000 10.470 <br> 1200 13.181 <br> 1400 15.940 <br> 1600 18.680 | $\begin{array}{ll}1210^{\circ} \mathrm{C} & 353.6 \mathrm{mv} \\ 1300 & 385.2 \\ 1360 & 403.2 \\ 1450 & 424.9 \\ & \\ & \\ & \end{array}$ |
| Inlluence of temperature and gas atmosphere | Subject to oxidation and alteration above $400^{\circ} \mathrm{C}$ due Cu , above $600^{\circ}$ due constantan wire. Ni-plating of Cu fube gives profec. tion, in acid-containing gos. Contamina. tion of Cu affects calibration greatly. Resistance 10 oxid. atm. good. Reslstance to reducing atm. good. Requires prolection fumes. from acid | Oxidizing and reducing atmosphere hove little effect on accuracy. Bost used In dry afmosphero. Reslstance to oxida. tion good to $400^{\circ} \mathrm{C}$ Resistance to reduc. ing afmosphere good. Protact from oxygen, molsfure, sulphur. | Chromel attacked by sulphurous almosphere. Resistance to oxida. tlon good. Resistance to reducing atmos. phere poor. | Resistance to oxidizing atmosphare vary good. Resistance to reducing atmosphere poor. Afioctod by sulphur, reducing or sulphurous gas, $\mathrm{SO}_{2}$ and $\mathrm{H}_{2} \mathrm{~S}$. | Resistance to oxidizIng atmosphere vary good. Resisiance to reducing atmosphere poor. Susceptible to chemical alteration by As, Si, P vapor in reducing gas $\mathrm{CO}_{2}, \mathrm{H}_{2}$, $\mathrm{H}_{2} \mathrm{~S}, \mathrm{SO} 4$. Pt corrodes easily above $1000^{\circ}$. Used in gostight protecting tube. |  | Used as tube alement. Corbon sheoth chemically inert. |
| Particular applications | \|low temperature, industrial. Internal combustion angine. Used as a fube element for meosurements isfeam line. | Low temperafure, indusitial. Steal annealing, boiler flues, tube stills. Used in reducing or neutral atmosphere. |  | Used in oxidizing atmosphere. Industrial. Ceramic kilns, sube stills, electric furnaces. | ternational Standd 630 to $1065^{\circ} \mathrm{C}$. | Similar to Pt/PtRh 1101 but has higher omf. | Steel furnace and ladie temperafures. laboratory measurements. |

## Thermocouples and their characteristics

continued

## Characteristics of typical thermocouples




Compiled from "Temperature Measurement and Control" by R. L. Weber, pages 68-71.

Melting points of solder

| pure alloys |  | melifing points |  |
| :---: | :---: | :---: | :---: |
| $\begin{gathered} \text { percent } \\ \text { tin } \\ \hline \end{gathered}$ | $\begin{gathered} \text { percent } \\ \text { lead } \\ \hline \end{gathered}$ | degrees centigrade | degroes fahranheif |
| 100 |  | 232 | 450 |
| 90 | 10 | 213 | 415 |
| 80 | 20 | 196 | 385 |
| 70 | 30 | 186 | 367 |
| 65 | 35 | 181 | 358 |
| 60 | 40 | 188 | 370 |
| 50 | 50 | 212 | 414 |
| 40 | 60 | 238 | 460 |
| 30 | 70 | 257 | 496 |
| 20 | 80 | 290 | 554 |
| 10 | 90 | 302 | 576 |
|  | 100 | 327 | 620 |

Spark-gap break down voliages


Data for a voltage which is continuous or at a frequency low enough to permit complete deionization between cycles, between needle points or clean, smooth spherical surfaces in dustfree dry air. The following multiplying factors apply for atmospheric conditions other than those stated above:

| pressure |  | temperature ${ }^{\circ} \mathrm{C}$ |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| ${ }^{\prime} \mathrm{Hg}$ | mm Hg | $-40$ | $-20$ | 0 | 20 | 40 | 60 |
| 5 | 127 | 0.26 | 0.24 | 0.23 | 0.21 | 0.20 | 0.19 |
| 10 | 254 | 0.47 | 0.44 | 0.42 | 0.39 | 0.37 | 0.34 |
| 15 | 381 | 0.68 | 0.64 | 0.60 | 0.56 | 0.53 | 0.50 |
| 20 | 508 | 0.87 | 0.82 | 0.77 | 0.72 | 0.68 | 0.64 |
| 25 | 635 | 1.07 | 0.99 | 0.93 | 0.87 | 0.82 | 0.77 |
| 30 | 762 | 1.25 | 1.17 | 1.10 | 1.03 | 0.97 | 0.91 |
| 35 | 889 | 1.43 | 1.34 | 1.26 | 1.19 | 1.12 | 1.05 |
| 40 | 1016 | 1.61 | 1.51 | 1.42 | 1.33 | 1.25 | 1.17 |
| 45 | 1143 | 1.79 | 1.68 | 1.58 | 1.49 | 1.40 | 1.31 |
| 50 | 1270 | 1.96 | 1.84 | 1.73 | 1.63 | 1.53 | 1.44 |
| 55 | 1397 | 2.13 | 2.01 | 1.89 | 1.78 | 1.67 | 1.57 |
| 60 | 1524 | 2.30 | 2.17 | 2.04 | 1.92 | 1.80 | 1.69 |

Table I

|  | discharge in US gallons per minute |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| in feet | $1 / 2 *$ | 1 | 8/4* | 1 | $1 *$ | 1 | $11 /{ }^{*}$ | 1 | 11/2" | 1 | $2{ }^{\prime \prime}$ | 1 | $21 / 2{ }^{\prime \prime}$ | 1 | 3 ' |  | $31 / 2$ | $4 *$ | 1 | $8^{\prime \prime}$ | 1 | 6 * |
| 1 | . 19 |  | . 54 |  | 1.11 |  | 1.96 |  | 3.09 |  | 6.34 |  | 11.07 |  | 17.41 |  | 25.58 | 35.79 |  | - 62.57 |  | 98.72 |
| 2 | . 28 |  | . 77 |  | 1.59 |  | 2.76 |  | 4.36 |  | 8.96 |  | 15.61 |  | 24.62 |  | 36.15 | 50.56 |  | 88.39 |  | 139.31 |
| 4 | . 40 |  | 1.09 |  | 2.25 |  | 3.92 |  | 6.17 |  | 12.73 |  | 22.10 |  | 34.95 |  | 51.28 | 71.58 |  | 124.90 |  | 198.54 |
| 6 | . 48 |  | 1.33 |  | 2.75 |  | 4.78 |  | 7.55 |  | 15.49 |  | 27.02 |  | 42.63 |  | 62.69 | - 87.67 |  | 152.52 |  | 241.39 |
| 9 | . 59 |  | 1.63 |  | 3.36 |  | 5.86 |  | 9.26 |  | 19.09 |  | 33.27 |  | 52.36 |  | 76.98 | 107.48 |  | 187.35 |  | 295.43 |
| 12 | . 68 |  | 1.89 |  | 3.90 |  | 6.77 |  | 10.69 |  | 21.98 |  | 38.43 |  | 60.53 |  | 88.87 | 123.70 |  | 216.17 |  | 342.27 |
| 16 | . 79 |  | 2.17 |  | 4.48 |  | 7.82 |  | 12.37 |  | 25.34 |  | 44.31 |  | 69.77 |  | 102.56 | 142.91 |  | 249.80 |  | 395.11 |
| 20 | . 89 |  | 2.44 |  | 5.02 |  | 8.74 |  | 13.81 |  | 28.34 |  | 49.48 |  | 77.94 |  | 114.57 | 159.73 |  | 279.82 |  | 440.74 |
| 25 | . 98 |  | 2.73 |  | 5.61 |  | 9.78 |  | 15.50 |  | 31.70 |  | 55.36 |  | 87.19 |  | 127.30 | 178.94 |  | 312.24 |  | 493.59 |
| 30 | 1.08 |  | 2.98 |  | 6.14 |  | 10.71 |  | 16.93 |  | 34.59 |  | 60.65 |  | 95.47 |  | 139.31 | 195.75 |  | 342.27 |  | 540.42 |
| 40 | 1.25 |  | 3.46 |  | 7.10 |  | 12.37 |  | 19.58 |  | 40.23 |  | 70.01 |  | 110.49 |  | 162.13 | 225.78 |  | 395.11 |  | 624.49 |
| 50 | 1.39 |  | 3.86 |  | 7.94 |  | 13.81 |  | 21.86 |  | 44.92 |  | 78.30 |  | 122.50 |  | 180.14 | 252.20 |  | 441.95 |  | 697.75 |
| 75 | 1.71 |  | 4.72 |  | 9.73 |  | 16.93 |  | 26.78 |  | 54.88 |  | 95.96 |  | 150.12 |  | 220.97 | 309.84 |  | 541.62 |  | 855.07 |
| 100 | 1.98 |  | 5.46 |  | 11.23 |  | 19.58 |  | 30.81 |  | 63.41 |  | 110.72 |  | 174.14 |  | 235.80 | 357.88 |  | 625.69 |  | 987.17 |
| 150 | 2.44 |  | 6.71 |  | 13.81 |  | 23.90 |  | 37.83 |  | 77.94 |  | 139.19 |  | 213.77 |  | 314.65 | 439.54 |  | 765.00 |  | 1,214.15 |
| 200 | 2.80 |  | 7.71 |  | 15.85 |  | 27.62 |  | 43.59 |  | 89.59 |  | 156.12 |  | 248.19 |  | 361.48 | 505.60 |  | 883.89 |  | 1,394.29 |
| 250 | 3.13 |  | 8.65 |  | 17.77 |  | 30.81 |  | 48.88 |  | 100.52 |  | 175.34 |  | 276.22 |  | 404.72 | 565.64 | 1 | - 989.57 |  | 1,564.82 |
| 500 | 4.43 |  | 12.25 |  | 25.10 |  | 43.71 | , | 69.05 |  | 141.71 |  | 247.39 |  | 390.31 |  | 571.65 | 801.03 |  | 1,397.89 |  | 2,209.73 |

Discharge in gallons per minute through 1000 ff . plpe line of $1 / 2^{\prime \prime}$ to $6^{\prime \prime}$ bore with average number of bends and fittings. For other pipe lengths see Table II.

## Table II

| Length in feet | 50 | 100 | 150 | 200 | 300 | 400 | 500 | 750 | 1,000 | 1,250 | 1,500 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Factor | 4.47 | 3.16 | 2.58 | 2.237 | 1.827 | 1.580 | 1.414 | 1.154 | 1.0 | 0.895 | 0.817 |
| tength in feot | 1,750 | 2,000 | 2,500 | 3,000 | 4,000 | 5,000 | 7,500 | 10,000 | 5 ml . | 10 mi . | 50 ml . |
| Factor | 0.756 | 0.707 | 0.633 | 0.577 | 0.500 | 0.447 | 0.365 | 0.316 | 0.195 | 0.138 | 0.0616 |

Multiplication foctor to be applied to Table Ifor pipe lengths other than 1000 ft .
Exampler Required-approximate discharge of a line of piping $4^{*}$ bore, 5000 feel long,
under 30 foot head.
Approximate discharge for the 1000 foot line from Table $1=195.75$ gallons per minute. foctor from Table Il $=0.447$
$\therefore$ Approximote discharge $=19575 \times 0.447=87.5$

## 50

## Materials and finishes for tropical and marine use

Ordinary finishing of equipment fails in meeting satisfactorily conditions encountered in tropical and marine use. Under these conditions corrosive influences are greatly aggravated by prevailing higher relative humidities, and temperature cycling causes alternate condensation on, and evaporation of moisture from, finished surfaces. Useful equipment life under adverse atmospheric influences depends largely on proper choice of base materials and finishes applied. Especially important in tropical and marine applications is avoidance of electrical contact between dissimilar metals.

Dissimilar metals, widely separated in the galvanic series, should not be bolted, riveted, etc., without separation by insulating material at the faying surfaces. The only exception occurs when both surfaces have been coated with the same protective metal, e.g., electroplating, hot dipping, galvanizing, etc.
In addition to choice of deterioration-resistant materials, consideration must be given to weight, need for a conductive surface, availability of ovens, appearance, etc.

## A-order of preference:

## Base materials

1. Brass
2. Aluminum, anodized
3. Nickel silver
4. Steel, zinc phosphated
5. Phosphor-bronze
6. Steel, cadmium phosphated
7. Monel
8. Steel, phosphated
9. Stainless steel

## Finishes

1. Baked paint
2. Force dried paint
3. Air dried paint (pigmentless paint, e.g., varnish)
$B$-order of preference: (if $A$ is impracticable)

## Base materials

1. Copper
2. Steel

## Finishes

1. Copper—nickel-chromium 5. Cadmium, lacquered
2. Copper-nickel-oxide
3. Zinc, phosphated
4. Copper-nickel
5. Cadmium, phosphated
6. Zinc, lacquered

## Materials and finishes for tropical and marine use continued

Aluminum should always be anodized. Aluminum, steel, zinc, and cadmium should never be used bare.

Electrical contact surfaces should be given above finish B-1 or 3, and, in addition, they should be silver plated.

Variable capacitor plates should be silver plated.
All electrical circuit elements and uncoated metallic surfaces lexcept electrical contact surfaces) inside of cabinets should receive a coat of fungicidal moisture repellant varnish or lacquer.

## Wood parts should receive:

1. Dip coat of fungicidal water repellant sealer.
2. One coat of refinishing primer.
3. Suitable topcoat.

## Torque and horsepower

Torque varies directly with power and inversely with rotating speed of the shaft, or
$T=\frac{K P}{N}$
where $T=$ torque in inch-pounds, $P=h p, N=r p m, K$ (constant) $=63,000$. Example 1:-For a two-horsepower motor rotating at 1800 rpm ,
$T=\frac{63,000 \times 2}{1800}=70$ inch-pounds. .
If the shaft is 1 inch in diameter, the force at its periphery
$F=\frac{T}{\text { radius }}=\frac{70 \text { inch-pounds }}{0.5}=140$ pounds
Example 2: If 150 inch-pounds torque are required at 1200 rpm ,
$150=\frac{63,000 \mathrm{hp}}{1200} \quad \mathrm{hp}=\frac{150 \times 1200}{63,000}=2.86$

# Audio and radio design 

## Resistors and capacifors



* Letter used to indicate tolerance in type designations.
$\dagger$ Applies to copacitors only.


## Resistors, fixed composition

RMA Standard, American War Standard, and Joint Army-Navy Specifications for color coding of fixed composition resistors are identical in all respects.
The exterior body color of insulated axial-lead composition resistors is usually tan, but other colors, except black, are permitted. Non-insulated, axial-lead composition resistors have a black body color. Radial-lead composition resistors may have a body color representing the first significant figure of the resistance value.


Note: Low-power insulated wire-wound resistors hove axial leads and are color coded similar to the left-hond figure above except that band $A$ is double width.



## Capacitors, fixed mica dielectric

Fixed mica-dielectric capacitors of the American War Standards and Joint Army-Navy Specification are designated differently from the 1938 RMA Standard. AWS and JAN mica capacitors have a characteristic defined in Table I.

## Table I

| choracteristic | 0 | temperature coefficient parts/million $/{ }^{\circ} \mathrm{C}$ | maximum capacitonce drift | verificotion of chorocteristics by praduction test |
| :---: | :---: | :---: | :---: | :---: |
|  | * | Not specified | Not specified |  |
| B | $\dagger$ | Not specified | Not specified | Not required <br> Nol required |
| C | $\dagger$ | -200 to +200 | 0.5 percent | Not required |
| D |  | -100 to +100 | 0.2 percent | Not required |
| E | $\dagger$ | 0 to +100 | 0.05 percent | Nol required |
| F |  | 0 to +50 | 0.025 percent | Required |
| G | $\dagger$ | 0 to -50 | 0.025 percent | Required |

* $Q$ must be greater than $1 / 8$ of minimum allowable $Q$ for other characteristics (JAN].
$\dagger$ Minimum acceptable $Q$ at I MC is defined by a curve; value varies with copocitance.
Type designations of AWS or JAN fixed mica-dielectric capacitors are a comprehensive numbering system used to identify the component. The capacitor type designation is given in the following form:


Component designation: Fixed mica-dielectric capacitors are identified by the symbol CM.
Case designation: The case designation is a 2 -digit symbol which identifies a particular case size and shape.
Characteristic: The characteristic is indicated by a single letter in accordance with Table I.
Capacitance value: The nominal capacitance value in micromicrofarads is indicated by, a 3 -digit number. The first two digits are the first two digits of the capacitance value in micromicrofarads. The final digit specifies the number of zeros which follow the first two digits. If more than two significant figures are required, additional digits may be used, the last digit always indicating the number of zeros.
Capacitance tolerance: The symmetrical capacitance tolerance in percent is designated by a letter as shown on page 52 .

## 56

Capacitors, fixed mica dielectric continued

## AWS and JAN fixed capacitors (1946 RMA proposal)



## RMA fixed capacitors

The 1938 RMA Standard covers a simple 3-dot color code showing directly only the capacitance, and a more comprehensive 6 -dot color code showing 3 significant figures and tolerance of the capacitance value, and a voltage rating. Capacitance values are expressed in micromicrofarads up to 10,000 micromicrofarads.


RMA 6-dot

Examples

|  | top row |  |  | bottom row |  |  | description |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| type |  |  |  | left | :olerance center | multiplier right |  |
| RMA 13 dot) | red | green | brown | none | none | none | $250 \mu \mu \mathrm{f}=20 \%, 500$ volts |
| RMA | brown | black | black | blue | green | brown | $1000 \mu \mu f=5 \%, 600$ volts |
| RMA | brown | red | green | gold | red | brown | $1250 \mu \mu \mathrm{f}=2 \%, 1000$ volts |
| CM308681J | black | blue | gray | brown | gold | brown | $680 \mu \mu \mathrm{f} \pm 5 \%$, characteristic 8 |
| CM35E332G | black | orange | oronge | yellow |  | red | $3300 \mu \mu \mathrm{f} \pm 2 \%$, choracteristic E |

## Capacitors, fixed ceramic

Tubular ceramic dielectric capacitors are used for temperature compensation of tuned circuits and have many other applications as well. If the capacitance, tolerance, and temperature coefficient are not printed on the capacitor body, the following color code will be used. The change in capacitance per unit capacitance per degree centigrade is the temperature coefficient, usually stated in parts per million per centigrade (ppm/ ${ }^{\circ} \mathrm{C}$ ).

|  |  |  | capaelfan | tolerance | temperature |
| :---: | :---: | :---: | :---: | :---: | :---: |
| color | significant figure | multiplior | $\begin{gathered} \ln \% \\ \gg 10 \mu \mu \mathrm{l} \end{gathered}$ | $e \sum_{10 \mu \mu i}^{\text {in } \mu \mu \mathrm{f}}$ | $\begin{gathered} \text { coeffcient } \\ \text { ports } / \text { million } /{ }^{\circ} \mathrm{C} \end{gathered}$ |
| black | 0 | 1 | $\pm 20$ | 2.0 | 0 |
| brown | 1 | 10 | $\pm 1$ |  | -30 |
| red | 2 | 100 | $\pm 2$ |  | -80 |
| orange | 3 | 1,000 |  |  | -150 |
| yellow | 4 | - |  |  | -220 |
| green | 5 | - | $\pm 5$ | 0.5 | -330 |
| blue | 6 | - |  |  | -470 |
| violot | 7 | - |  |  | $-750$ |
| gray | 8 | 0.01 |  | 0.25 | +30 |
| white | 9 | 0.1 | $\pm 10$ | 1.0 | $-330 \pm 500$ |



## Examples

| wide | narrow bands or dofs |  |  |  | descriptlon |
| :---: | :---: | :---: | :---: | :---: | :---: |
| band | A | B | C | D |  |
| black blue violel | black red gray | red <br> red <br> red | black black brown | black green silver | $2.0 \mu \mu \mathrm{f} \pm 2 \mu \mu \mathrm{f}$, zero temp coeff $22 \mu \mu \mathrm{I} \pm 5 \%$. $-470 \mathrm{ppm} /{ }^{\circ} \mathrm{C}$ temp cooff $820 \mu \mu \mathrm{f} \pm 10 \%,-750 \mathrm{ppm} /{ }^{\circ} \mathrm{C}$ temp cosf |

## Inductance of single-layer solenoids

The approximate value of the low-frequency inductance of a single-layer solenoid is:
$L=F n^{2} d$ microhenries*
where $F=$ form factor, a function of the ratio $d / l$. The value of $F$ may be read from the accompanying chart, Fig. 1.
$n=$ number of turns, $d=$ diameter of coil linches), between centers of conductors, $1=$ length of coil (inches) $=n$ times the distance between centers of adjacent turns.

The formula is based on the assumption of a uniform current sheet, but the correction due to the use of spaced round wires is usually negligible for practical purposes. For higher frequencies skin effect alters the inductance slightly. This effect is not readily calculated, but is often negligibly small. However, it must be borne in mind that the formula gives approximately the true value of inductance. In contrast, the apparent value is affected by the shunting effect of the distributed capacitance of the coil.

Example: Required a coil of 100 microhenries inductance, wound on a form 2 inches diameter by 2 inches winding length. Then $d / 1=1.00$, and $F=0.0173$ on the chart.
$n=\sqrt{\frac{L}{f d}}=\sqrt{\frac{100}{0.0173 \times 2}}=54$ turns
Reference to Magnet Wire Data, page 60, will assist in choosing a desirable size of wire, allowing for a suitable spacing between turns according to the application of the coil. A slight correction may then be made for the increased diameter (diameter of form plus two times radius of wire), if this small correction seems justified.

In the use of various charts, tables, and calculators for designing inductors, the following relationships are useful in extending the range of the devices. They apply to coils of any type or design.

1. If all dimensions are held constant, inductance is proportional to $n^{2}$.
2. If the proportions of the coil remain unchanged, then for a given number of turns the inductance is proportional to the dimensions of the coil. A coil with all dimensions $m$ times those of a given coil thaving the same number of turns) has $m$ times the inductance of the given coil. That is, inductance has the dimensions of length.
[^6]
## Inductance of single-layer solenoids continued



Fig. 1-Inductance of a single-loyer solenoid, form foctor: F

## 60

## Magnef wire data

| size wire AWC | bore nom diam in inches | enam nom diam in Inches | scc ${ }^{*}$ <br> diam In Inches | DCC diam in Inches | SC. <br> diam in inches | ssc* <br> diom In inches | DSC ${ }^{*}$ diam in inches | 5s㡽 <br> diom in inches | bare |  | enamaled |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  |  |  | min <br> diam <br> inches | max <br> diam <br> Inches | min diam Inches |  |
| 10 | . 1019 | . 1039 | . 1079 | . 1129 | . 1104 |  |  |  | . 1009 | . 1029 | . 1024 | . 1044 |
| 11 | . 0907 | . 0927 | . 0957 | . 1002 | . 0982 |  |  |  | . 0398 | . 0917 | . 0913 | . 0932 |
| 12 | . 0808 | . 0827 | . 0858 | . 0903 | . 0882 |  |  |  | . 0800 | . 0816 | . 0814 | . 0832 |
| 13 | . 0720 | . 0738 | . 0770 | . 0815 | . 0793 |  |  |  | . 0712 | . 0727 | . 0726 | . 0743 |
| 14 | . 0641 | . 0659 | . 0691 | . 0736 | . 0714 |  |  |  | . 0634 | . 0647 | . 0648 | . 0664 |
| 15 | . 0571 | . 0588 | . 0621 | . 0666 | . 0643 | . 0591 | . 0611 | . 0613 | . 0565 | . 0576 | . 0578 | . 0593 |
| 16 | . 0508 | . 0524 | . 0558 | . 0603 | . 0579 | . 0528 | . 0548 | . 0549 | . 0503 | . 0513 | . 0515 | . 0529 |
| 17 | . 0453 | . 0469 | . 0503 | . 0548 | . 0523 | . 0473 | . 0493 | . 0493 | . 0448 | . 0457 | . 0460 | . 0473 |
| 18 | . 0403 | . 0418 | . 0453 | . 0498 | . 0472 | . 0423 | . 0443 | . 0442 | . 0399 | . 0407 | . 0410 | . 0422 |
| 19 | . 0359 | . 0374 | . 0409 | . 0454 | . 0428 | . 0379 | . 0399 | . 0398 | . 0355 | . 0363 | . 0366 | . 0378 |
| 20 | . 0320 | . 0334 | . 0370 | . 0415 | . 0388 | . 0340 | . 0360 | . 0358 | . 0316 | . 0323 | . 0326 | . 0338 |
| 21 | . 0285 | . 0299 | . 0335 | . 0380 | . 0353 | . 0305 | . 0325 | . 0323 | . 0282 | . 0287 | . 0292 | . 0303 |
| 22 | . 0253 | . 0266 | . 0303 | . 0343 | . 0320 | . 0273 | . 0293 | . 0290 | . 0251 | . 0256 | . 0261 | . 0270 |
| 23 | . 0226 | . 0238 | . 0276 | . 0316 | . 0292 | . 0246 | . 0266 | . 0262 | . 0223 | . 0228 | . 0232 | . 0242 |
| 24 | . 0201 | . 0213 | . 0251 | . 0291 | . 0266 | .0221 | . 0241 | . 0236 | . 0199 | . 0203 | . 0208 | . 0216 |
| 25 | . 0179 | . 0190 | . 0224 | . 0264 | . 0238 | . 0199 | . 0219 | . 0213 | . 0177 | . 0181 | . 0186 | . 0193 |
| 26 | . 0159 | . 0169 | . 0204 | . 0244 | . 0217 | . 0179 | . 0199 | . 0192 | . 0158 | . 0161 | . 0166 | . 0172 |
| 27 | . 0142 | . 0152 | . 0187 | . 0227 | . 0200 | . 0162 | . 0182 | . 0175 | . 0141 | . 0144 | . 0149 | . 0155 |
| 28 | . 0126 | . 0135 | . 0171 | . 0211 | . 0183 | . 0146 | . 0166 | . 0158 | . 0125 | . 0128 | . 0132 | . 0138 |
| 29 | . 0113 | . 0122 | . 0158 | . 0198 | . 0170 | . 0133 | . 0153 | . 0145 | . 0112 | . 0114 | . 0119 | . 0125 |
| 30 | . 0100 | . 0108 | . 0145 | . 0185 | . 0156 | . 0120 | . 0140 | . 0131 | . 0099 | . 0101 | . 0105 | . 0111 |
| 31 | . 0089 | . 0097 | . 0134 | . 0174 | . 0144 | . 0109 | . 0129 | . 0119 | . 0088 | . 0090 | . 0094 | . 0099 |
| 32 | . 0080 | . 0088 | . 0125 | . 0165 | . 0135 | . 0100 | . 0120 | . 0110 | . 0079 | . 0081 | .0085 | . 0090 |
| 33 | . 0071 | . 0078 | . 0116 | . 0156 | . 0125 | . 0091 | . 0111 | . 0100 | . 0070 | . 0072 | . 0075 | . 0080 |
| 34 | . 0063 | . 0069 | . 0108 | . 0148 | . 0116 | . 0083 | . 0103 | . 0091 | . 0062 | . 0064 | . 0067 | . 0071 |
| 35 | . 0056 | . 0061 | . 0101 | . 0141 | . 0103 | . 0076 | . 0096 | . 0083 | .005s | . 0057 | . 0059 | . 0063 |
| 36 | . 0050 | . 0055 | .0090 | . 0130 | .0097 | . 0070 | .0090 | . 0077 | . 0049 | . 0051 | .0053 | . 0057 |
| 37 | . 0045 | . 0049 | . 0085 | . 0125 | . 0091 | . 0065 | . 0085 | . 0071 | . 0044 | . 0046 | . 0047 | . 0051 |
| 38 | . 0040 | . 0044 | . 0080 | . 0120 | . 0386 | .0060 | . 0080 | . 0066 | . 0039 | . 0041 | . 0042 | . 0046 |
| 39 | . 0035 | . 0038 | . 0075 | . 0115 | . 0080 | . 0355 | . 0075 | . 0060 | . 0034 | . 0036 | . 0036 | . 0040 |
| 40 | . 0031 | . 0034 | . 0071 | . 0111 | . 0076 | . 0051 | . 0071 | .0056 | . 0030 | . 0032 | . 0032 | . 0036 |
| 41 | . 0028 | .0031 |  |  |  |  |  |  | . 0227 | . 0029 | . 0029 | . 0032 |
| 42 | . 0025 | . 0028 |  |  |  |  |  |  | . 0024 | . 0026 | . 0026 | . 0029 |
| 43 | . 0022 | . 0025 |  |  |  |  |  |  | . 0021 | . 0023 | . 0023 | . 0026 |
| 44 | . 0020 | . 0023 |  |  |  |  |  |  | . 0019 | .0021 | . 0021 | . 0024 |

[^7]
## Reactance charits



Figs 2, 3, and 4 give the relationships of capacitance, inductance, reactance, and frequency. Any one value may consideration in terms of two others by use of a straight edge laid across the correct chart for the frequency under

Fig. 2-I cycle to 1000 cycles.

Reactance charts canlinued


Example: Given a capacitance of $0.001 \mu \mathrm{f}$, find the reactance at 50 kilocycles and inductance required to resonate. Place a straight edge through these values and read the intersections on the other scales, giving 3,180 ohms and 10.1 millihenries.

Fig. 3-1 killocycle to 1000 kilocycles.


Fig. 4-1 megacycle to 1000 megocycles.

$$
\begin{array}{ll}
\text { Impedance } Z=R+j X \text { ohms } & \text { phase angle } \phi=\tan ^{-1} \frac{X}{R} \\
\text { magnitude }|Z|=\left[R^{2}+X^{2}\right]^{\frac{1}{2}} \text { ohms } & \text { admittance } Y=\frac{1}{Z} \text { mhos }
\end{array}
$$

phase angle of the admittance
Is $-\tan ^{-1} \frac{X}{R}$
(


| Impedance $\mathbf{Z}=\mathbf{R}$ <br> magnitude $\|\mathbf{Z}\|=[$ | $X$ ohms <br> $\left.+X^{2}\right]^{\frac{1}{2}}$ ohms | phase angle $\phi=\tan ^{-1} \frac{X}{R}$ <br> admittance $\mathbf{Y}=\frac{1}{\mathbf{Z}}$ mhos <br> phase angle of the admittance $\text { Is }-\tan ^{-1} \frac{X}{R}$ |
| :---: | :---: | :---: |
|  | impedance | $\frac{R+j \omega\left[L\left(1-\omega^{2} L C\right)-C R^{2}\right]}{\left(1-\omega^{2} L C\right)^{2}+\omega^{2} C^{2} R^{2}}$ |
|  | magnitude | $\left[\frac{R^{2}+\omega^{2} L^{2}}{\left(1-\omega^{2} L C\right)^{2}+\omega^{2} C^{2} R^{2}}\right]^{\frac{1}{2}}$ |
|  | phase angle | $\tan ^{-1} \frac{\omega\left[L \\|-\omega^{2} L C l-C R^{2}\right]}{R}$ |
|  | admittance | $\frac{R-j \omega\left[L\left(11-\omega^{2} L C\right)-C R^{2}\right]}{R^{2}+\omega^{2} L^{2}}$ |
|  | impedance | $X_{1} \frac{X_{1} R_{2}+j\left[R_{2}{ }^{2}+X_{2}\left(X_{1}+X_{2}\right)\right]}{R_{2}{ }^{2}+\left(X_{1}+X_{2}\right)^{2}}$ |
|  | magnitude | $x_{1}\left[\frac{R_{2}{ }^{2}+X_{2}{ }^{2}}{R_{2}{ }^{2}+\left(x_{1}+X_{2}\right)^{2}}\right]^{\frac{1}{2}}$ |
|  | phase angle | $\tan ^{-1} \frac{R_{2}^{2}+X_{2}\left(X_{1}+X_{2}\right)}{X_{1} R_{2}}$ |
|  | admittance | $\frac{R_{2} X_{1}-j\left(R_{2}{ }^{2}+x_{2}{ }^{2}+x_{1} x_{2}\right)}{X_{1}\left(R_{2}{ }^{2}+X_{2}{ }^{2}\right)}$ |



Parallel and series circuits and their equivalent relationships
Conductance $G=\frac{1}{R_{p}}$
$\omega=2 \pi f$


Admittance $Y=\frac{I}{E}=\frac{1}{Z}=G-j B$

$$
=\sqrt{G^{2}+B^{2}} \angle-\phi=|Y| \angle-\phi
$$

Impedance $Z=\frac{E}{I}=\frac{1}{Y}=\frac{R_{p} X_{p}}{R_{p}{ }^{2}+X_{p}{ }^{2}}\left(X_{p}+j R_{p}\right)$
$=\frac{R_{p} X_{p}}{\sqrt{R_{p}^{2}+X_{p}^{2}}} \angle \phi=|Z| \angle \phi$

parallel circuit

Phase anclo $-\phi=\tan ^{-1} \frac{-B}{O}=\cos ^{-1} \frac{G}{|Y|}=-\tan ^{-1} \frac{R_{p}}{X_{p}}$
Resistance $=R_{s}$
Reactance $X_{s}=\omega L_{s}-\frac{1}{\omega C_{s}}$
Impedance $Z=\frac{E}{I}=R_{s}+j X_{s}$
$=\sqrt{R_{s}{ }^{2}+X_{\delta}{ }^{2}} \angle \phi=|Z| \angle \phi$
Phase angle $\phi=\tan ^{-1} \frac{X_{s}}{R_{s}}=\cos ^{-1} \frac{R_{s}}{|Z|}$
Vectors $E$ and $I$, phasc angle $\phi$, and $Z, Y$ are identical for the parallcl circtit and its equivalent scrics circuit

$Q=|\tan \phi|=\frac{\left|X_{s}\right|}{R_{s}}=\frac{R_{p}}{\left|X_{p}\right|}=\frac{|B|}{G}$
$P F=\cos \phi=\frac{R_{s}}{|Z|}=\frac{|Z|}{R_{p}}=\frac{C}{|Y|}=\sqrt{\frac{R_{s}}{R_{p}}}=\frac{1}{\sqrt{Q^{2}+1}}=\frac{\mathrm{kw}}{\mathrm{kva}}$
$Z^{2}=R_{s}{ }^{2}+X_{s}{ }^{2}=\frac{R_{p}{ }^{2} X_{p}{ }^{2}}{R_{p}{ }^{2}+X_{p}{ }^{2}}=R_{s} R_{p}=X_{s} X_{p}$

$$
\begin{aligned}
& Y^{2}=G^{2}+B^{2}=\frac{1}{R_{p}^{2}}+\frac{1}{X_{p}^{2}}=\frac{G}{R_{s}} \\
& R_{s}=\frac{Z^{2}}{R_{p}}=\frac{G}{Y^{2}}=R_{p} \frac{X_{p}^{2}}{R_{p}^{2}+X_{p}^{2}}=R_{p} \frac{1}{Q^{2}+1} \\
& X_{s}=\frac{Z^{2}}{X_{p}}=\frac{B}{Y^{2}}=X_{p} \frac{R_{p}^{2}}{R_{p}^{2}+X_{p}^{2}}=X_{p} \frac{1}{1+\frac{1}{Q^{2}}} \\
& R_{p}=\frac{!}{G}=\frac{Z^{2}}{R_{s}}=\frac{R_{s}^{2}+X_{s}^{2}}{R_{s}}=R_{s}\left(Q^{2}+1\right) \\
& X_{p}=\frac{1}{B}=\frac{Z^{2}}{X_{s}}=\frac{R_{s}^{2}+X_{s}^{2}}{X_{s}}=X_{s}\left(1+\frac{1}{Q^{2}}\right)=\frac{R_{s} R_{p}}{X_{s}}= \pm R_{p} \sqrt{\frac{R_{s}}{R_{p}-R_{s}}}
\end{aligned}
$$

Approximate formulas
Reactor $R_{s}=\frac{X^{2}}{R_{p}}$ and $X=X_{s}=X_{p} \quad$ (See Note 1)
Resistor $R=R_{s}=R_{p}$ and $X_{s}=\frac{R^{2}}{X_{p}} \quad$ (See Note 2)

Simplified parallel and series circuifs

$$
X_{p}=\omega L_{p} \quad B=\frac{1}{\omega L_{p}} \quad X_{s}=\omega L_{s}
$$



$$
\tan \phi=\frac{\omega L_{s}}{R_{s}}=\frac{R_{p}}{\omega L_{p}} \quad Q=\frac{\omega L_{s}^{\prime}}{R_{s}}=\frac{R_{p}}{\omega L_{p}}
$$

$$
\begin{aligned}
& P F=\frac{R_{s}}{\sqrt{R_{s}^{2}+\omega^{2} L_{s}^{2}}}=\frac{\omega L_{p}}{\sqrt{R_{p}^{2}+\omega^{2} L_{p}^{2}}} \\
& P F=\frac{1}{Q} \text { approx (See Note 3) }
\end{aligned}
$$



$$
R_{s}=R_{p} \frac{1}{Q^{2}+1} \quad R_{p}=R_{s}\left(Q^{2}+1\right)
$$

$$
L_{s}=L_{p} \frac{1}{1+\frac{1}{Q^{2}}} \quad L_{p}=L_{s}\left(1+\frac{1}{\mathrm{Q}^{2}}\right)
$$

$$
\begin{aligned}
& X_{p}=\frac{-1}{\omega C_{p}} \quad \mathrm{~B}=-\omega \mathrm{C}_{p} \quad X_{s}=\frac{-1}{\omega C_{2}} \\
& \tan \phi=\frac{-1}{\omega C_{s} R_{s}}=-\omega C_{p} R_{p} \\
& Q=\frac{1}{\omega C_{s} R_{s}}=\omega C_{p} R_{p} \\
& P F=\frac{\omega C_{s} R_{s}}{\sqrt{1+\omega^{2} C_{s} R_{s}{ }^{2}}}=\frac{1}{\sqrt{1+\omega^{2} C_{p} R_{p}{ }^{2}}} \\
& P F=\frac{1}{Q} \text { approx } \quad(\operatorname{See} \operatorname{Note} 3) \\
& R_{s}=R_{p} \frac{1}{Q^{2}+1} \quad R_{p}=R_{s}\left(Q^{2}+1\right) \\
& C_{p}\left(1+\frac{1}{Q^{2}}\right) \quad C_{p}=C_{s} \frac{1}{1+\frac{1}{Q^{2}}}
\end{aligned}
$$

Approximate formulas

$$
\begin{aligned}
& \text { Inductor } R_{s}=\frac{\omega^{2} L^{2}}{R_{p}} \text { and } L=L_{p}=L_{s} \quad \text { (See Note 1) } \\
& \text { Resistor } R=R_{s}=R_{p} \text { and } L_{p}=\frac{R^{2}}{\omega^{2} L_{s}} \quad \text { (See Note 2) }
\end{aligned}
$$

Capacitor $R_{s}=\frac{1}{\omega^{2} C^{2} R_{p}}$ and $C=C_{p}=C_{s} \quad$ (See Note 1)
Resistor $R=R_{z}=R_{p}$ and $C_{s}=\frac{1}{\omega^{2} C_{p} R^{2}} \quad$ (See Note 2)
Note 1: (Small resistive component) Error in percent $=-\frac{100}{Q^{2}}$ (for $Q=10$,
error $=1$ percent low) error $=1$ percent lowl

Note 2: (Small reactive camponent) Error in percent $=-100 Q^{2}$ (for $Q=$ 0.1, error $=1$ percent low)

Note 3: Error in percent $=+\frac{50}{Q^{2}}$ approximately (for $Q=7$, error $=1,1$
percent high) percent high)

## Skin effect

A $=$ correction coefficient
$D=$ diameter of conductor in inches
$f=$ frequency in cycles per second
$R_{a c}=$ resistance at frequency $f$
$R_{d c}=$ direct-current resistance
$T=$ thickness of tubular conductor in inches
$T_{1}=$ depth of penetration of current
$\mu=$ permeability of conductor material $/ \mu=1$ for copper and other nonmagnetic materials)
$\rho=$ resistivity of conductor material at any temperature
$\rho_{c}=$ resistivity of copper at $20^{\circ} \mathrm{C} \quad 11.724$ microhm-centimeter)
Fig. 5 shows the relationship of $R_{a c} / R_{d c}$ versus $D \sqrt{f}$ for copper, or versus $D \sqrt{f} \sqrt{\mu_{\rho}} \bar{\rho}_{\rho}$ for any conductor material, for an isolated straight solid conductor of circular cross section. Negligible error in the formulas for $R_{a c}$ results when the conductor is spaced at least 100 from adjacent conductors. When the spacing between axes of parallel conductors carrying the same current is 4D, the resistance $R_{a c}$ is increased about 3 percent. The formulas are accurate for concentric lines due to their circular symmetry.

For values of $D \sqrt{f} \sqrt{\mu_{\rho}} \frac{\rho_{\rho}}{\rho}$ greater than 40,
$\frac{R_{a c}}{R_{d e}}=0.0960 D \sqrt{f} \sqrt{\mu} \frac{\rho_{\rho}}{\rho}+0.26$
The high-frequency resistance of an isolated straight conductor: either solid; or tubular for $T<\frac{D}{8}$ or $T_{1}<\frac{D}{8}$; is given in equation (2). If the current flow is along the inside surface of a tubular conductor, $D$ is the inside diameter.
$R_{a c}=A \frac{\sqrt{f}}{D} \sqrt{\mu} \frac{\rho}{\rho_{c}} \times 10^{-6}$ ohms per foot
The values of the correction coefficient A for solid conductors are shown in Table II and, for fubular conductors, in Table III.

The value of $T \sqrt{f} \sqrt{\mu} \frac{\rho_{\rho}}{\rho}$ that just makes $A=1$ indicates the penetration of

Skin effect continued


Fig. 5-Resistance ratio for isolated straight solid conductors of circular cross section.

## AUDIO AND RADIO DESIGN

Skin effect
the currents below the surface of the conductor. Thus, approximately,
$T_{1}=\frac{3.5}{\sqrt{f}} \sqrt{\frac{\rho}{\mu \rho_{c}}}$ inches.
When $T_{1}<\frac{D}{8}$ the value of $R_{a c}$ as given by equation (2) (but not the value of $\frac{R_{a c}}{R_{d o}}$ in Table III) is correct for any value $T \geqq T_{1}$.
Under the limitation that the radius of curvature of all parts of the cross section is appreciably greater than $T_{L}$, equations (2) and (3) hold for isolated straight conductors of any shape. In this case the term $D=$ (perimeter of cross section) $\div \pi$.

## Examples

1. At 100 megacycles, a copper conductor has a depth of penetration $T_{1}=0.00035$ inch.
2. A steel shield with 0.005 -inch copper plate, which is practically equivalent in $R_{a c}$ to an isolated copper conductor 0.005 -inch thick, has a value of $A=1.23$ at 200 kilocycles. This 23-percent increase in resistance over that of a thick copper sheet is satisfactorily low as regards its effect on the losses of the components within the shield. By comparison, a thick aluminum sheet has a resistance $\sqrt{\frac{\rho}{\rho_{c}}}=1.28$ times that of copper.

## Table II-Solid conductors

| $D \sqrt{f} \sqrt{\mu \frac{\rho_{e}}{\rho}}$ | $A$ |
| :---: | :---: |
| $>370$ | 1.000 |
| 220 | 1.005 |
| 160 | 1.010 |
| 98 | 1.02 |
| 48 | 1.05 |
| 26 | 1.10 |
| 13 | 1.20 |
| 9.6 | 1.30 |
| $<3.3$ | 2.00 |
| $<3.0$ | $R_{a c} \approx R_{d o}$ |
| $R_{d o}=\frac{10.37}{D^{2}} \frac{\rho}{\rho_{c}} \times 10^{-8}$ ohms per foot |  |

Table III-Tubular conductors

| $T \sqrt{f} \sqrt{\mu \frac{\rho_{c}}{\rho}}$ | A | $\mathbf{R a c}_{\text {ac }} / \mathbf{R}_{\text {dc }}$ |
| :---: | :---: | :---: |
| $\left.\begin{array}{l} =B \text { where } \\ B>3.5 \end{array}\right\}$ | 1.00 | 0.384 B |
| 3.5 | 1.00 | 1.35 |
| 3.15 | 1.01 | 1.23 |
| 2.85 | 1.05 | 1.15 |
| 2.60 | 1.10 | 1.10 |
| 2.29 | 1.20 | 1.06 |
| 2.08 | 1.30 | 1.04 |
| 1.77 | 1.50 | 1.02 |
| 1.31 | 2.00 | 1.00 |
| $\left.\begin{array}{rl} = & B \text { where } \\ B<1.3 \end{array}\right\}$ | $\frac{2.60}{B}$ | 1.00 |

## Network theorems

## Reciprocity theorem

If an emf of any character whatsoever located at one point in a linear network produces a current at any other point in the network, the same emf acting at the second point will produce the same current at the first point.

## Thévenin's theorem

If an impedance $Z$ is connected between two points of a linear network, the resulting steady-state current $I$ through this impedance is the ratio of the potential difference $V$ between the two points prior to the connection of $Z$, and the sum of the values of (1) the connected impedance $Z$, and (2) the impedance $Z_{1}$ of the network measured between the two points, when all generators in the network are replaced by their internal impedances

$$
I=\frac{V}{Z+Z_{1}}
$$

## Principle of superposition

The current which flows at any point in a network composed of constant resistances, inductances, and capacitances, or the potential difference which exists between any two points in such a network, due to the simultaneous action of a number of emf's distributed in any manner throughout the network, is the sum of the component currents at the first point, or the potential differences between the two points, which would be caused by the individual emf's acting alone. (Applicable to emf's of any character.)
In the application of this theorem, it is to be noted that: for any impedance element $Z$ through which flows a current $I$, there may be substituted a virtual source of voltage of value $-Z I$.

## Electrical circuit formulas

1. Self-inductance of circular ring of round wire af radio frequencies, for non-magnefic materials
$L=\frac{a}{100}\left[7.353 \log _{10} \frac{16 a}{d}-6.370\right]$
$L=$ inductance in microhenries
$a=$ mean radius of ring in inches
$d=$ diameter of wire in inches
$\frac{a}{d}>2.5$

## Electrical circuit formulas

continued

## 2. Capacitance of a parallel-plate capacitor

$C=0.0885 K \frac{(N-1) A}{1}$ micromicrofarads
$A=$ area of one side of one plate in square centimeters
$N=$ number of plates
$t=$ thickness of dielectric in centimeters
$K=$ dielectric constant
This formula neglects "fringing" at the edges of the plates.

## 3. Reactance of an inductor

$X=2 \pi f L$ ohms
$f=$ frequency in cycles per second
$L=$ inductance in henries
or $f$ in kilocycles and $L$ in millihenries; or $\dagger$ in megacycles and $L$ in microhenries

## 4. Reactance of a capacitor

$X=\frac{-1}{2 \pi f C}$ ohms
$f=$ frequency in cycles per second
$C=$ capacitance in farads
This may be written $\quad X=\frac{-159.2}{f C}$ ohms
$f=$ frequency in kilocycles per second
$C=$ capacitance in microfarads
or $f$ in megacycles and $C$ in milli-microfarads $10.001 \mu f)$.

## 5. Resonant frequency of a series-funed circuit

$f=\frac{1}{2 \pi \sqrt{L C}}$ cycles per second
$L=$ inductance in henries
C = capacitance in farads
This may be written $L C=\frac{25,330}{f^{2}}$
$f=$ frequency in kilocycles
$L=$ inductance in millihenries
$\mathrm{C}=$ capacitance in milli-microfarads $(0.001 \mu \mathrm{f})$
or $f$ in megacycles, $L$ in microhenries, and $C$ in micromicrofarads.

## Electrical circuit formulas continued

## 6. Dynamic resistance of a parallel-funed circuit at resonance

$r=\frac{X^{2}}{R}=\frac{L}{C R}$ ohms
$X=\omega L=\frac{1}{\omega C}$
$R=r_{1}+r_{2}$
$L=$ inductance in henries
C = capacitance in farads
$R=$ resistance in ohms
The formula is accurate for engineering purposes provided $\frac{X}{R}>10$.


## 7. Parallel impedances

If $Z_{1}$ and $Z_{2}$ are the two impedances which are connected in parallel, then the resultant impedance is

$$
\begin{aligned}
Z & =\frac{Z_{1} Z_{2}}{Z_{1}+Z_{2}}=\frac{\left(R_{1}+j X_{1}\right)\left(R_{2}+j X_{2}\right)}{\left(R_{1}+R_{2}\right)+j\left(X_{1}+X_{2}\right)}=\frac{\left(R_{1} R_{2}-X_{1} X_{2}\right)+j\left(R_{1} X_{2}+R_{2} X_{1}\right)}{\left(R_{1}+R_{2}\right)+j\left(X_{1}+X_{2}\right)} \\
Z & =\frac{|Z|\left|Z_{2}\right|}{\left|Z_{1}+Z_{2}\right|} \angle \phi \\
\phi & =\angle Z_{1}+\angle Z_{2}-\angle\left(Z_{1}+Z_{2}\right) \\
& =\tan ^{-1} \frac{X_{1}}{R_{1}}+\tan ^{-1} \frac{X_{2}}{R_{2}}-\tan ^{-1} \frac{X_{1}+X_{2}}{R_{1}+R_{2}}
\end{aligned}
$$

Given one impedance $Z_{1}$ and the desired resultant impedance $Z$, the other impedance is

$$
Z_{2}=\frac{z Z_{1}}{Z_{1}-Z}
$$

## 8. Impedence of a two-mesh network

$Z_{11}=R_{11}+j X_{11}$
is the impedance of the first circuit, measured at terminals $1-1$ with terminals $2-2$ open-circuited.
$Z_{22}=R_{22}+j \chi_{22}$

## AUdIO and radio design

Electrical circuit formulas
is the impedance of the second circuit, measured at terminals $2-2$ with terminals 1-1 open-circuited.
$Z_{12}=R_{12}+j X_{12}$
is the mutual! impedance between the two meshes, i.e., the open-circuit voltage apparing in either mesh when unit current flows in the other mesh.

Then the impedance looking into terminals
 1-1 with terminals $2-2$ short-circuited is
$Z_{1}^{\prime}=R_{1}^{\prime}+j X_{1}^{\prime}=Z_{11}-\frac{Z_{12}^{2}}{Z_{22}}=R_{11}+j X_{11}-\frac{R_{12}^{2}-X_{12}^{2}+2 j R_{12} X_{12}}{R_{22}+j X_{22}}$
When
$R_{12}=0$
$Z_{1}^{\prime}=R_{1}^{\prime}+j X_{1}^{\prime}=Z_{11}+\frac{X_{12}^{2}}{Z_{22}}=R_{11}+j X_{11}+\frac{X_{12}{ }^{2}}{R_{22}^{2}+X_{22}^{2}}\left(R_{22}-j X_{22}\right)$
Example 1: Two resistors in parallel.
$Z_{11}=R_{1} \quad Z_{22}=R_{1}+R_{2}$
$Z_{12}=R_{1}$
Hence $Z_{1}^{\prime}=R_{1}^{\prime}=R_{1}-\frac{R_{1}{ }^{2}}{R_{1}+R_{2}}=\frac{R_{1} R_{2}}{R_{1}+R_{2}}$


Example 2: A transformer with tuned secondary and negligible primary resistance.
$Z_{11}=j \omega L_{1}$
$Z_{22}=R_{2} \quad$ since $X_{22}=0$
$Z_{12}=j \omega M$
Then $Z_{1}^{\prime}=j \omega L_{1}+\frac{\omega^{2} \mathrm{M}^{2}}{R_{2}}$


## 9. Currents in a two-mesh network

$$
\begin{aligned}
i_{1} & =\frac{e_{1}}{Z_{1}^{\prime}} \\
& =e_{1} \frac{Z_{22}}{Z_{11} Z_{22}-Z_{12}{ }^{2}} \\
& =e_{1} \frac{R_{22}+j X_{22}}{\left(R_{11} R_{22}-X_{11} X_{22}-R_{12}{ }^{2}+X_{12}{ }^{2}\right)+j\left(R_{11} X_{22}+R_{22} X_{11}-2 R_{12} X_{12}\right)} \\
i_{2} & =e_{1} \frac{Z_{12}}{Z_{11} Z_{22}-Z_{12}{ }^{2}}
\end{aligned}
$$

## 10. Power transfer between two impedances connected direcily

Let $Z_{1}=R_{1}+j X_{1}$ be the impedance of the source, and $Z_{2}=R_{2}+j X_{2}$ be the impedance of the load.

The maximum power transfer occurs when

$$
R_{2}=R_{1} \text { and } X_{2}=-X_{1}
$$

The reflection loss due to connecting any two impedances directly is

$$
\frac{I_{2}}{I}=\frac{\left|Z_{1}+Z_{2}\right|}{2 \sqrt{R_{1} R_{2}}}
$$

In decibels

$$
\mathrm{db}=20 \log _{10} \frac{\left|Z_{1}+Z_{2}\right|}{2 \sqrt{R_{1} R_{2}}}
$$

$I_{2}=$ current which would flow in $Z_{2}$ were the two impedances connected through a perfect impedance matching network.
$I=$ current which flows when the impedances are connected directly.

## 11. Power transfer between two meshes coupled reactively

In the general case, $X_{11}$ and $X_{22}$ are not equal to zero and $X_{12}$ may be any reactive coupling. When only one of the quantities $X_{11}, X_{22}$, and $X_{12}$ can be varied, the best power transfer under the circumstances is given by


## Electrical circuif formulas

 continuedFor $X_{22}$ variable
$X_{22}=\frac{X_{12}{ }^{2} X_{11}}{R_{11}{ }^{2}+X_{11}{ }^{2}}$ (zero reactance looking into load circuit)
For $X_{11}$ variable
$X_{11}=\frac{X_{12}{ }^{2} X_{22}}{R_{22}{ }^{2}+X_{22^{2}}}$ (zero reactance looking into source circuit)
For $X_{12}$ variable
$X_{12}{ }^{2}=\sqrt{\left(\mathbb{R}_{11}{ }^{2}+X_{11}{ }^{2}\right)\left(R_{22}{ }^{2}+X_{22}{ }^{2}\right)}$
When two of the three quantities can be varied, a perfect impedance match is attained and maximum power is transferred when
$X_{12}{ }^{2}=\sqrt{\left(R_{11}{ }^{2}+X_{11}{ }^{2}\right)\left(R_{22}{ }^{2}+X_{22}{ }^{2}\right)}$
and
$\frac{X_{11}}{R_{11}}=\frac{X_{22}}{R_{22}}$ (both circuits of same $Q$ or phase angle)
For perfect impedance match the current is

$$
i_{2}=\frac{e_{1}}{2 \sqrt{R_{11} R_{22}}} \angle \tan ^{-1} \frac{R_{11}}{X_{11}}
$$

In the most common case, the circuits are funed to resonance $X_{11}=0$ and $X_{22}=0$. Then $X_{12}{ }^{2}=R_{11} R_{22}$ for perfect impedance match.

## 12. Optimum coupling between two circuits tuned to the same frequency

From the last result in the preceding section, maximum power transfer lor an impedance match) is obtained for $\omega^{2} M^{2}=R_{1} R_{2}$
where $M$ is the mutual inductance between the circuits, $R_{1}$ and $R_{2}$ are the resistances of the two circuits.

## 13. Coefficient of coupling

By definition, coefficient of coupling $k$ is

$$
k=\frac{M}{\sqrt{L_{1} L_{2}}} \quad \text { where } M=\text { mutual inductance }
$$

$L_{1}$ and $L_{2}$ are the inductances of the two coupled circuits.

Coefficient of coupling is a geometrical property, being a function of the proportions of the configuration of coils, including their relationship to any nearby, objects which affect the field of the system. As long as these proportions remain unchanged, the coefficient of coupling is independent of the physical size of the system, and of the number of turns of either coil.

## 14. Selective circuits

Formulas and curves are presented for the selectivity and phase shift
Of $n$ single tuned circuits
Of m pairs of coupled tuned circuits
The conditions assumed are

1. All circuits are tuned to the same frequency fo.
2. All circuits have the same $Q$, or each pair of circuits includes one circuit having $Q_{1}$, and the other having $Q_{2}$.
3. Otherwise the circuits need not be identical.
4. Each successive circuit or pair of circuits is isolated from the preceding and following ones by tubes, with no regeneration around the system.
Certain approximations have been made in order to simplify the formulas. In most actual applications of the types of circuits treated, the error involved is negligible from a practical standpoint. Over the narrow frequency band in question, it is assumed that
5. The reactance around each circuit is equal to $2 X_{0} \frac{\Delta f}{1_{0}}$.
6. The resistance of each circuit is constant and equal to $\frac{X_{0}}{Q}$.
7. The coupling between two circuits of a pair is reactive and constant. When an untuned limk is used to couple the two circuits, this condition frequently is far from satisfied, resulting in a lopsided selectivity curve.)
8. The equivalent input voltage, taken as being in series with the tuned circuit (or the first of a pair), is assumed to bear a constant proportionality to the grid voltage of the input tube or other driving source, at all frequencies in the band.
9. Likewise, the output voltage across the circuit lor the final circuit of a pairl is assumed to be proportional only to the current in the circuit.

## Electrical circuit formulas continued

The following symbols are used in the formulas.
$\frac{\Delta f}{f_{0}}=\frac{f-f_{0}}{f_{0}}=\frac{\text { deviation from resonance frequency }}{\text { resonance frequency }}$
$f=$ signal frequency
$f_{0}=$ frequency to which all circuits are independently tuned
$X_{0}=$ reactance at $f_{0}$ of inductor in tuned circuit
$Q=$ quality factor of tuned circuit. For a pair of coupled circuits; there is used $Q=\sqrt{Q_{1} Q_{2}}$
$Q_{1}$ and $Q_{2}$ are the values for the two circuits of a coupled paip
$Q^{\prime}=\frac{2 Q_{1} Q_{2}}{Q_{1}+Q_{2}}$
$E=$ amplitude of output voltage at frequency $f$ \} both for the same value
$E_{0}=$ amplitude of output voltage at frequency $f_{0} \int$ of input voltage
$n=$ number of single tuned circuits
$m=$ number of pairs of coupled circuits
$\phi=$ phase shift of signal at $f$ relative to shift at $f_{0}$, as signal passes through cascade of circuits
$k=$ coefficient of coupling between two coupled circuits
$p=k^{2} Q^{2}$ or $p=k^{2} Q_{1} Q_{2}$, a parameter determining the form of the selec: tivity curve of coupled circuits
$B=p-\frac{1}{2}\left(\frac{Q_{1}}{Q_{2}}+\frac{Q_{2}}{Q_{1}}\right)$

Selectivity and phase shift of single funed circuits'

$$
\frac{E}{E_{0}}=\left[\frac{1}{\sqrt{1+\left(2 Q \frac{\Delta f}{f_{0}}\right)^{2}}}\right]^{n}
$$

$\frac{\Delta f}{f_{0}}= \pm \frac{1}{2 Q} \sqrt{\left(\frac{E_{0}}{E}\right)^{\frac{2}{n}}-1}$


Decibel response $=20 \log _{10}\left(\frac{E}{E_{0}}\right)$
( db response of $n$ circuits) $=n$ times (db response of single circuit)
$\phi=n \tan ^{-1}\left(-2 Q \frac{\Delta f}{f_{0}}\right)$
These equations are plotted in Fig. 6 and Fig. 7, following:

$Q \frac{\Delta f}{f_{0}}=Q \frac{f-f_{0}}{f_{0}}$
db response of

- a single circuit $n=1$
- o pair of coupled circuits $m=1$

The selectivity curves ore symmetrical about the axis $Q \frac{\Delta f}{f_{0}}=0$ for practical purposes.

Extrapolation beyond lower limits of chart:

| $\Delta$ response for doubling $\Delta f$ | circuit | usoful Limit |  |
| :---: | :---: | :---: | :---: |
|  |  | at $\frac{\Delta f}{f_{0}}$ | error becomes |
| - 6 db | $\leftarrow$ single $\rightarrow$ | $\pm 0.3$ | 1 to 2 db |
| $-12 \mathrm{db}$ | $\leftarrow$ pair | $\pm 0.2$ | 3 to 4 db |

Fig. 6-Selectivity curves.

As an example of the use of the curves, suppose there are three single-tuned circuits $\ln =3$ ). Each circuit has a $Q=200$ and is tuned to 1000 kilocycles. The results of this example are shown in the following table:

| abscissa <br> $\mathbf{Q} \frac{\Delta f}{\mathbf{f}_{0}}$ | $\Delta f$ <br> $\mathbf{k c}$ | ordinate <br> db response <br> for $\mathbf{n}=1$ | db <br> response <br> for $\mathbf{n}=\mathbf{3}$ | $\boldsymbol{\phi}^{*}$ <br> for $\mathbf{n}=\mathbf{1}$ | $\phi^{*}$ <br> for $\mathbf{n}=\mathbf{3}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0.5 | $\pm 2.5$ | -3.0 | -9 | $\mp 45^{\circ}$ | $\mp 135^{\circ}$ |
| 1.5 | $\pm 7.5$ | -10.0 | -30 | $\mp 71 / 2^{\circ}$ | $\mp 215^{\circ}$ |
| 5.0 | $\pm 25.0$ | -20.2 | -61 | $\mp 84^{\circ}$ | $\mp 252^{\circ}$ |

[^8]
$$
Q \frac{\Delta f}{f_{0}}=Q \frac{f-f_{0}}{f_{0}}
$$
-relative phase angle $\phi$ in degrees
-a single circuit $n=1$

- a pair af coupled circuits $m=1$

The curves are symmetrical about the arigin. Far negative values af $Q \frac{\Delta f}{f_{0}}, \phi$ is positive and same numerical value as far corresponding negative value of $Q \frac{\Delta f}{f_{0}}$.

Fig. 7-Phase-shiff curves.

Selectivity and phase shift of pairs of coupled tuned circuits
Case 1: When $Q_{1}=Q_{2}=Q$
These formulas can be used with reasonable accuracy when $Q_{1}$ and $Q_{2}$ differ by ratios up to 1.5 or even 2 to 1 . In such cases use the value $Q=\sqrt{Q_{1} Q_{2}}$.


For very small values of $\frac{E}{E_{0}}$ the formulas reduce to
one of several types of coupling
$\frac{E}{E_{0}}=\left[\frac{\rho+1}{\left(2 Q \frac{\Delta f}{f_{0}}\right)^{2}}\right]^{m}$
Decibel response $=20 \log _{10}\left(\frac{E}{E_{0}}\right)$
( db response of $m$ pairs of circuits) $=m$ times ( db response of one pair)
$\phi=m \tan ^{-1}\left[\frac{-4 Q \frac{\Delta f}{f_{0}}}{(\rho+1)-\left(2 Q \frac{\Delta f}{f_{0}}\right)^{2}}\right]$
As $p$ approaches zero, the selectivity and phase shift approach the values for $n$ single circuits, where $n=2 m$ (gain also approaches zero).

The above equations are plotted in Figs. 6 and 7.
For overcoupled circuits ( $p>1$ )
Location of peaks: $\left(\frac{\Delta f}{f_{0}}\right)_{\text {peak }}= \pm \frac{1}{2 Q} \sqrt{p-1}$
Amplitude of peaks: $\left(\frac{E}{E_{0}}\right)_{\text {peak }}=\left(\frac{\rho+1}{2 \sqrt{\rho}}\right)^{m}$
Phase shift at peaks: $\quad \phi_{\text {peak }}=m \tan ^{-1}(\mp \sqrt{p-1})$

## Electrical circuit formulas

## continued

Approximate pass band (where $\frac{E}{E_{0}}=1$ ):
$\left(\frac{\Delta f}{f_{0}}\right)_{\text {coneer }}=0$ and $\left(\frac{\Delta f}{f_{0}}\right)_{\text {unity }}=\sqrt{2}\left(\frac{\Delta f}{f_{0}}\right)_{\text {peak }}= \pm \frac{1}{Q} \sqrt{\frac{p-1}{2}}$
Case 2: General formula for any $Q_{1}$ and $Q_{2}$

$$
\begin{aligned}
& \frac{E}{E_{0}}=\left[\frac{\rho+1}{\sqrt{\left[\left(2 Q \frac{\Delta f}{f_{0}}\right)^{2}-B\right]^{2}+(p+1)^{2}-B^{2}}}\right]^{m} \\
& \frac{\Delta f}{f_{0}}= \pm \frac{1}{2 Q \sqrt{B \pm\left[(p+1)^{2}\left(\frac{E_{0}}{E}\right)^{\frac{2}{m}}-(p+1)^{2}+B^{2}\right]^{\frac{2}{2}}}} \\
& \phi=m \tan ^{-2}\left[-\frac{2 Q \frac{\Delta f}{f_{0}}\left(\sqrt{\frac{Q_{1}}{Q_{2}}}+\sqrt{\frac{Q_{2}}{Q_{1}}}\right)}{(p+1)-\left(2 Q \frac{\Delta f}{f_{0}}\right)^{2}}\right]
\end{aligned}
$$

For overcoupled circuits
Location of peaks: $\left(\frac{\Delta f}{f_{0}}\right)_{\text {peak }}= \pm \frac{\sqrt{B}}{2 Q}= \pm \frac{1}{2} \sqrt{k^{2}-\frac{1}{2}\left(\frac{1}{Q_{1}{ }^{2}}+\frac{1}{Q_{2}{ }^{2}}\right)}$
Amplitude of peaks: $\left(\frac{E}{E_{0}}\right)_{\text {peak }}=\left[\frac{p+1}{\sqrt{(p+1)^{2}-B^{2}}}\right]^{m}$
Case 3: Peaks just converged to a single peak
Here $B=0 \quad$ or $\quad k^{2}=\frac{1}{2}\left(\frac{1}{Q_{2}{ }^{2}}+\frac{1}{Q_{2}{ }^{2}}\right)$
$\frac{E}{E_{0}}=\left[\frac{2}{\sqrt{\left(2 Q^{\prime} \frac{\Delta f}{f_{0}}\right)^{4}+4}}\right]^{m} ; \quad \frac{\Delta f}{f_{0}}= \pm \frac{\sqrt{2}}{4}\left(\frac{1}{Q_{1}}+\frac{1}{Q_{2}}\right) \sqrt[4]{\left(\frac{E_{0}}{E}\right)^{\frac{2}{m}}-1}$
$\phi=m \tan ^{-1}\left[-\frac{4 Q^{\prime} \frac{\Delta f}{f_{0}}}{2-\left(2 Q^{\prime} \frac{\Delta f}{f_{0}}\right)^{2}}\right]$
The curves of Figs. 6 and 7 may be applied to this case, using the value $p=1$, and substituting $Q^{\prime}$ for $Q$.

## Electrical circuit formulas continued

## 15. $T-\pi$ or $Y-\Delta$ transformation

The two networks are equivalent, as far as conditions at the terminals are concerned, provided the following equations are satisfied. Either the impedance equations or the admittance equations may be used.


Impedance equations

$$
\begin{aligned}
& Z_{12}=\frac{Z_{1} Z_{2}+Z_{1} Z_{3}+Z_{2} Z_{3}}{Z_{3}} \\
& Z_{13}=\frac{Z_{1} Z_{2}+Z_{1} Z_{3}+Z_{2} Z_{8}}{Z_{2}} \\
& Z_{28}=\frac{Z_{1} Z_{2}+Z_{1} Z_{3}+Z_{2} Z_{2}}{Z_{1}} \\
& Z_{1}=\frac{Z_{12} Z_{13}}{Z_{12}+Z_{13}+Z_{23}} \\
& Z_{2}=\frac{Z_{12} Z_{23}}{Z_{12}+Z_{18}+Z_{23}} \\
& Z_{3}=\frac{Z_{13} Z_{23}}{Z_{12}+Z_{13}+Z_{28}}
\end{aligned}
$$

Admittance equations
$Y_{12}=\frac{Y_{1} Y_{2}}{Y_{1}+Y_{2}+Y_{3}}$
$Y_{13}=\frac{Y_{1} Y_{3}}{Y_{1}+Y_{2}+Y_{3}}$
$\gamma_{23}=\frac{Y_{2} Y_{3}}{Y_{1}+Y_{2}+Y_{3}}$
$\gamma_{1}=\frac{\gamma_{12} Y_{18}+Y_{12} Y_{23}+Y_{13} Y_{23}}{Y_{23}}$
$\gamma_{2}=\frac{\gamma_{12} Y_{18}+Y_{12} Y_{23}+Y_{18} Y_{23}}{Y_{13}}$
$\gamma_{3}=\frac{\gamma_{12} \gamma_{13}+Y_{12} Y_{23}+\gamma_{18} Y_{23}}{\gamma_{12}}$

## 16. Amplitude modulation

In design work, usually the entire modulation is assumed to be in $\mathcal{M}_{1}$. Then $\mathcal{M}_{2}$, $M_{3}$ etc, would be neglected in the formulas below.
When the expression $\left(1+M_{1}+M_{2}+\ldots.\right)$ is used, it is assumed that $\omega_{1}$, $\omega_{2}$ etc, are incommensurate.
$i=I\left[1+M_{1} \cos \left(\omega_{1} \dagger+\phi_{1}\right)+M_{2} \cos \left(\omega_{2} t+\phi_{2}\right)+\ldots.\right] \sin \left(\omega_{0} t+\phi_{0}\right)$

Electrical circuit formulas continued


To determine the modulation percentage from an oscillogram of type lllustrated apply measurements $A$ and $B$ to scales $A$ and $B$ and read percentage from center scale. Example: $A=3$ inches, $B=0.7$ inches-Modulation $62 \%$. Any units of measurement may be used.

Fig. 8-Modulation percentoge from oseillograms.

## Electrical circuil formulas continued

$$
\begin{aligned}
& =l\left\{\sin \left(\omega_{0} t+\phi_{0}\right)+\frac{M_{1}}{2}\left[\sin \left(\overline{\omega_{0}+\omega_{1}} t+\phi_{0}+\phi_{1}\right)+\right.\right. \\
& \left.\left.\sin \overline{\left(\omega_{0}-\omega_{1}\right.} t+\phi_{0}-\phi_{1}\right)\right]+\frac{M_{2}}{2}\left[\sin \left(\overline{\omega_{0}+\omega_{2}} t+\phi_{0}+\phi_{2}\right)+\right. \\
& \left.\left.\left.\sin \overline{\left[\omega_{0}-\omega_{2}\right.} \mid+\phi_{0}-\phi_{2}\right)\right]+\ldots\right\} \\
& \text { Percent modulation }=\left(M_{1}+M_{2}+\ldots .\right) \times 100 \\
& =\frac{\text { crest ampl }- \text { trough ampl }}{\text { crest ampl }+ \text { trough ampl }} \times 100 \text {. }
\end{aligned}
$$

Percent modulation may be measured by means of an oscilloscope, the modulated carrier wave being applied to the vertical plates and the modulating voltage wave to the horizontal plates. The resulting trapezoidal pattern and a nomograph for computing percent modulation are shown in Fig. 8. The dimensions $A$
 and $B$ in that figure are proportional to the crest amplitude and trough amplitude, respectively.
Peak voltage at crest: $V_{\text {creas }}=V_{\text {carrier, rms }} 11+M_{1}+M_{2}+\ldots .1 \sqrt{2}$
Kilovolt-amperes at crest: $\mathrm{kva}_{\text {cras }}=\mathrm{kva}_{\text {earrier }}\left(1+M_{1}+M_{2}+\ldots\right)^{2}$
Average kilovolt-amperes over a number of cycles of lowest modulation frequency:
$k$ va $_{\text {averase }}=k v 0_{\text {carrier }}\left(1+\frac{M_{1}{ }^{2}}{2}+\frac{M_{2}{ }^{2}}{2}+\ldots\right)$
Effective current of the modulated wave:
$I_{a f f}=I_{\text {carrisr, } r_{m u}} \sqrt{1+\frac{M_{1}{ }^{2}}{2}+\frac{M_{2}{ }^{2}}{2}+\ldots}$

## 17. Elementary R-C, R-L, and L-C filters

Simple attentuating sections of broad frequency discriminating characteristics, as used in power supplies, grid-bias feed, etc. The output load impedance is assumed to be high compared to the impedance of the shunt element of the filter.

Electrical circuit formulas
continued

$R$ in ohms
$L$ in henries
$C$ in farads
$11 \mu f=10^{-6}$ farad)
$T=$ time constant (seconds) $f_{0}=$ resonant frequency (cps) $\quad \omega=2 \pi f$
$2 \pi=6.28 \quad \frac{1}{2 \pi}=0.1592 \quad 4 \pi^{2}=39.5 \quad \frac{1}{4 \pi^{2}}=0.0253$

## Electrical circuit formulas continued

The relationships for low-pass filters are plotted in Figs. 9 and 10.

## Examples

1. Low-pass R-C filters
a. $\quad R=100,000$ ohms, $C=0.1 \times 10^{-6} 10.1 \mu \mathrm{f}$

Then $T=R C=0.01$ second

$$
\begin{aligned}
& \text { At } t=100 \mathrm{cps}, \frac{E_{\text {out }}}{E_{\text {in }}}=0.16- \\
& \text { At } t=30,000 \mathrm{cps}, \frac{E_{\text {out }}}{E_{\text {in }}}=0.00053
\end{aligned}
$$



N is any convenient factor, usually taken as an integral power of 10.

Fig. 9-Low-pass R-C and R-L filters.

## Electrical circuit formulas continued

b. $\quad R=1,000$ ohms, $C=0.001 \times 10^{-6}$

$$
T=1 \times 10^{-6} \text { second }=0.1 \div \mathrm{N}, \text { where } \mathrm{N}=10^{5}
$$

$$
\text { At } f=10 \text { megacycles }=100 \times N, \frac{E_{\text {out }}}{E_{\text {in }}}=0.016-
$$

2. Low-pass $L-C$ filter

$$
\text { At } f=120 \mathrm{cps}, \text { required } \frac{E_{\text {ous }}}{E_{i \pi}}=0.03
$$

Then from curves: $L C=6 \times 10^{-5}$ approximately. Whence, for $C=4 \mu \mathrm{f}$, we require $L=15$ henries.


Fig. 10-Low-pass L-C filtors.

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## Electrical circuit formulas continued

## 18. Transients

The complete transient in a linear network is, by the principle of superposition, the sum of the individual transients due to the store of energy in each inductor and capacitor and to each external source of energy connected to the network. To this is added the steady state condition due to each external source of energy. The transient may be computed as starting from any arbitrary time $t=0$ when the initial conditions of the energy of the network are known.
Convention of signs: In the following formulas, one direction of current is assumed to be positive, and any emf on a capacitor or in an external source, tending to produce a current in the positive direction, is designated as positive. In the case of the charge of a capacitor, this results in the capacitor voltage being the negative of the value sometimes conventionally used, wherein the junction of the source and the capacitor is assumed to be grounded and potentials are computed with respect to ground.
Time constant (designated T ): of the discharge of a capacitor through a resistor is the time $f_{2}-f_{1}$ required for the voltage or current to decay to $\frac{1}{6}$ of its value at time $t_{1}$. For the charge of a capacitor the same definition $\epsilon$
applies, the voltage "decaying" toward its steady state value. The time constant of discharge or charge of the current in an inductor through a resistor follows an analogous definition.
Energy stored in a capacitor $=\frac{1}{2}$ CE $^{2}$ joules (watt-seconds).
Energy stored in an inductor $=\frac{1}{2} L I^{2}$ joules (watt-seconds).
$\epsilon=2.718 \quad \frac{1}{\epsilon}=0.3679 \quad \log _{10} \epsilon=0.4343 \quad T$ and $f$ in seconds
$R$ in ohms $L$ in henries $\quad C$ in farads $\quad E$ in volts $\quad I$ in amperes

## Capacitor charge and discharge

Closing of switch occurs at time $:=0$
Initial conditions lat $t=0$ ): Battery $=E_{b} ; \mathrm{e}_{\mathrm{c}}=\mathrm{E}_{a}$.
Steady state (at $t=\infty): i=0 ; \mathrm{e}_{c}=-E_{b}$.
Transient:

$$
\begin{aligned}
& i=\frac{E_{b}+E_{0}}{R} \epsilon^{-\frac{i}{R C}}=I_{0} \epsilon^{-\frac{i}{R C}} \\
& \log _{10}\left(\frac{i}{I_{0}}\right)=-\frac{0.4343}{R C}
\end{aligned}
$$



## Electrical circuif formulas

 continued$e_{c}=E_{0}-\frac{1}{C} \int_{0}^{t} i d t=E_{0} \epsilon^{-\frac{t}{R C}}-E_{b}\left(1-\epsilon^{-\frac{t}{R C}}\right)$
Time constant: $T=R C$
Fig. 11 shows current $\quad \frac{i}{I_{0}}=\epsilon^{-\frac{i}{T}}$
Fig. 11 shows discharge (for $E_{b}=0$ ) $\frac{e_{c}}{E_{0}}=\epsilon^{-\frac{i}{T}}$
Fig. 12 shows charge (for $\left.E_{0}=0\right) \quad-\frac{e_{c}}{E_{b}}=\left(1-\epsilon^{-\frac{?}{T}}\right)$


Fig. 11.


Fig. 12.

These curves are plotted on a larger scale in Fig. 13.

## Two capacitors

Closing of switch occurs at time $t=0$
Initial conditions (at $1=0$ ):
$e_{1}=E_{1 ;} e_{2}=E_{2}$.
Steady state lat $t=\infty$ ):
$e_{1}=E_{f ;} e_{2}=-E_{f ; i}=0$.
$E_{f}=\frac{E_{1} C_{1}-E_{2} C_{2}}{C_{1}+C_{2}} \quad C^{\prime}=\frac{C_{1} C_{2}}{C_{1}+C_{2}}$
Transient:

$$
i=\frac{E_{1}+E_{2}}{R} \epsilon^{-\frac{t}{R C^{\prime}}}
$$



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## Electrical circuit formulas

 continued$e_{1}=E_{f}+\left(E_{1}-E_{f}\right) \epsilon^{-\frac{1}{R C^{\prime}}}=E_{1}-\left(E_{1}+E_{2}\right) \frac{C^{\prime}}{C_{1}}\left(1-\epsilon^{-\frac{8}{R C^{\prime}}}\right)$
$\left.e_{2}=-E_{f}+\left|E_{2}+E_{f} \epsilon^{-\frac{b}{R C^{\prime}}}=E_{2}-\right| E_{1}+E_{2}\right\rangle \frac{C^{\prime}}{C_{2}}\left(1-\epsilon^{-\frac{f}{R C^{\prime}}}\right)$
Original energy $=\frac{1}{2}\left(C_{1} E_{1}{ }^{2}+C_{2} E_{2}{ }^{2}\right)$ joules
final energy $=\frac{1}{2}\left(C_{1}+C_{2}\right) E_{f}{ }^{2}$ ioules
Loss of energy $=\int_{0}^{\infty} i^{2} R d t=\frac{1}{3} C^{\prime}\left(E_{1}+E_{2}\right)^{2}$ ioules
thoss is independent of the value of R.I


Use exponential $\epsilon^{-\frac{1}{T}}$ for charge or discharge of capacitor or discharge of inductor:

$$
\frac{\text { current at time } t}{\text { initial current }}
$$

discharge of capacitor:

$$
\frac{\text { voltage at time } t}{\text { initial voltage }}
$$

Use exponential $1-\epsilon^{-\frac{8}{T}}$ for charge of capacitor:
voltage at time $t$ battery or final voltage charge of inductor: $\frac{\text { current at time } f}{\text { final current }}$

Fig. 13-Exponential functions $\epsilon^{-\frac{t}{T}}$ and $1-\epsilon^{-\frac{t}{T}}$ applied to transionts in R-C and L- $\mathbf{R}$ circulfs.

## Electrical circuit formulas continued

Inductor charge and discharge
Initial conditions lat $t=0$ ):
Battery $=E_{b} ; \mathbf{i}=I_{0}$
Steady state (at $t=\infty): i=I_{f}=\frac{E_{b}}{R}$
Transient, plus steady state:


$$
\begin{aligned}
i & =I_{f}\left(1-\epsilon^{-\frac{R t}{L}}\right)+\dot{I}_{0} \epsilon^{-\frac{R t}{L}} \\
e_{L} & =-L \frac{d i}{d t}=-\left(E_{b}-R I_{0}\right) \epsilon^{-\frac{R t}{L}}
\end{aligned}
$$

Time constant: $T=\frac{L}{R}$
Fig. 11 shows discharge (for $E_{b}=0$ ) $\frac{i}{I_{0}}=\epsilon^{-\frac{i}{T}}$
Fig. 12 shows charge (for $\left.I_{0}=0\right) \quad \frac{i}{I_{s}}=\left(1-\epsilon^{-\frac{1}{T}}\right)$
These curves are plotted on a larger scale in Fig. 13.
Series circuit of $R, L$, and $C$ charge and discharge
Initial conditions lat $t=0$ ):
Battery $=E_{b} ; \mathrm{e}_{c}=E_{0} ; \mathbf{i}=I_{0}$
Steady state lat $t=\infty): i=0 ; e_{c}=-E_{b}$
Differential equation:
$E_{b}+E_{0}-\frac{1}{C} \int_{0}^{t} i d t-R i-L \frac{d i}{d t}=0$

whence $L \frac{d^{2} i}{d t^{2}}+R \frac{d i}{d t}+\frac{i}{C}=0$
Solution of equation:

$$
i=\epsilon^{-\frac{R t}{2 L}}\left[\frac{2\left(E_{b}+E_{0}\right)-R I_{0}}{R \sqrt{D}} \sinh \frac{R t}{2 L} \sqrt{D}+I_{0} \cosh \frac{R t}{2 L} \sqrt{D}\right]
$$

where

$$
D=1-\frac{4 L}{R^{2} C}
$$

Case 1: When $\frac{L}{R^{2} C}$ is small

$$
\begin{aligned}
i & =\frac{1}{\left(1-2 A-2 A^{2}\right)}\left\{\left[\frac{E_{b}+E_{0}}{R}-I_{0}\left(A+A^{2}\right)\right] \epsilon^{-\frac{1}{R C}(1+A+2 A)}\right. \\
& \left.+\left[I_{0}\left(1-A-A^{2}\right)-\frac{E_{b}+E_{0}}{R}\right] \epsilon^{-\frac{R t}{L}\left(1-A-A^{R}\right)}\right\}
\end{aligned}
$$

where $A=\frac{L}{R^{2} C}$
For practical purposes, the terms $A^{2}$ can be neglected when $A<0.1$. The terms $A$ may be neglected when $A<0.01$.
Case 2: When $\frac{4 L}{R^{2} C}<1$ for which $\sqrt{D}$ is real

$$
\begin{aligned}
i & =\frac{\epsilon^{-\frac{R t}{2 L}}}{\sqrt{D}}\left\{\left[\frac{E_{b}+E_{0}}{R}-\frac{I_{0}}{2}(1-\sqrt{D})\right] \epsilon^{\frac{R t}{2 L} \sqrt{D}}\right. \\
& \left.+\left[\frac{I_{0}}{2}(1+\sqrt{D})-\frac{E_{b}+E_{0}}{R}\right] \epsilon^{-\frac{R t}{2 L} \sqrt{D}}\right\}
\end{aligned}
$$

Case 3: When $D$ is a small positive or negative quantity

$$
\begin{aligned}
i & =\epsilon^{-\frac{R t}{2 L}\left\{\frac{2\left(E_{b}+E_{0}\right)}{R}\left[\frac{R t}{2 L}+\frac{1}{6}\left(\frac{R t}{2 L}\right)^{8} D\right]\right.} \\
& \left.+I_{0}\left[1-\frac{R t}{2 L}+\frac{1}{2}\left(\frac{R t}{2 L}\right)^{2} D-\frac{1}{6}\left(\frac{R t}{2 L}\right)^{8} D\right]\right\}
\end{aligned}
$$

This formula may be used for values of $D$ up to $\pm 0.25$, at which values the error in the computed current $i$ is approximately 1 percent of $I_{0}$ or of $\frac{E_{b}+E_{0}}{R}$.
Case 3a: When $\frac{4 L}{R^{2} C}=1$ for which $D=0$, the formula reduces to
$i=\epsilon^{-\frac{R t}{2 L}\left[\frac{E_{b}+E_{0}}{R} \frac{R t}{L}+I_{0}\left(1-\frac{R t}{2 L}\right)\right]}$
or $i=i_{1}+i_{2}$, plotted in Fig. 14. For practical purposes, this formula may be used when $\frac{4 L}{R^{2} C}=1 \pm 0.05$ with errors of 1 percent or less.

## Electrical circuit formulas cantinued

Case 4: When $\frac{4 L}{R^{2} C}>1$ for which $\sqrt{D}$ is imaginary

$$
\begin{aligned}
i & =\epsilon^{-\frac{R t}{2 L}}\left\{\left[\frac{E_{b}+E_{0}}{\omega_{0} L}-\frac{R I_{0}}{2 \omega_{0} L}\right] \sin \omega_{0} t+I_{0} \cos \omega_{0} t\right\} \\
& =I_{m} \epsilon^{-\frac{R t}{2 L} \sin \left(\omega_{0} t+\psi\right)}
\end{aligned}
$$

where $\omega_{0}=\sqrt{\frac{1}{L C}-\frac{R^{2}}{4 L^{2}}}$
$I_{m}=\frac{1}{\omega_{0} L} \sqrt{\left(E_{b}+E_{0}-\frac{R I_{0}}{2}\right)^{2}+\omega_{0}{ }^{2} L^{2} I_{0}{ }^{2}}$
$\psi=\tan ^{-1} \frac{\omega_{0} L I_{0}}{E_{b}+E_{0}-\frac{R I_{0}}{2}}$


Fig. 14-Transients for $\frac{4 L}{R^{2} C}=1$.

The envelope of the voltage wave across the inductor is:
$\pm \epsilon^{-\frac{R t}{2 L}} \frac{1}{\omega_{0} \sqrt{L C}} \sqrt{\left(E_{b}+E_{0}-\frac{R I_{0}}{2}\right)^{2}+\omega_{0}{ }^{2} L^{2} I_{0}{ }^{2}}$
Example: Relay with transient suppressing capacitor.
Switch closed till time $t=0$, then opened.
Let $L=0.10$ henries, $R_{1}=100$ ohms,

$$
E=10 \text { volts }
$$

Suppose we choose $C=10^{-6}$ farads, $R_{2}$ $=100$ ohms.

Then $R=200$ ohms, $I_{0}=0.10$ amperes,

$$
E_{0}=10 \text { volts, } \omega_{0}=3 \times 10^{3}, f_{0}=480 \mathrm{cps}
$$



Maximum peak voltage across $L$ (envelope at $t=0$ ) is approximately 30 volts. Time constant of decay of envelope is 0.001 second.

If it had been desired to make the circuit just non-oscillating, (Case 3a):

$$
\frac{4 L}{R^{2} C}=1 \text { or } R=630 \text { ohms for } C=10^{-6} \text { farads. }
$$

$$
R_{2}=530 \text { ohms. }
$$

Initial voltage at $t=0$, across $L$ is $-E_{0}+R I_{0}=53$ volts.

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Electrical circuit formulas continued

## Series circuit of R, L, and C with sinusoidal applied voltage

By the principle of superposition, the transient and steady state conditions are the same for the actual circuit and the equivalent circuit shown in the accompanying illustrations, the closing of the switch occurring at time $t=0$. In the equivalent circuit, the steady state is due to the source e acting continuously from time $t=-\infty$, while the transient is due to short circuiting the source $-e$ at time $t=0$.

actual circuit

Source: $e=E \sin (\omega t+\alpha)$
Steady state: $i=\frac{e}{Z} \angle-\phi=\frac{E}{Z} \sin (\omega t+\alpha-\phi \mid$
where
$Z=\sqrt{R^{2}+\left(\omega L-\frac{1}{\omega C}\right)^{2}} ; \tan \phi=\frac{\omega^{2} L C-1}{\omega C R}$

equivalent circuit

The transient is found by determining current $i=I_{0}$ and capacitor voltage $e_{e}=E_{0}$ at time $t=0$, due to the source $-e$. These values of $I_{0}$ and $E_{0}$ are then substituted in the equations of Case $1,2,3$, or 4 , above, according to the values of $R, L$, and $C$.

At time $t=0$, due to the source -e:
$i=I_{0}=-\frac{E}{Z} \sin |\alpha-\phi|$
$e_{c}=E_{0}=\frac{-E}{\omega C Z} \cos (\alpha-\phi)$
This form of analysis may be used for any periodic applied voltage e. The steady-state current and the capacitor voltage for an applied voltage -e are determined, the periodic voltage being resolved into its harmonic components for this purpose, if necessary. Then the instantaneous values $i=I_{0}$ and $e_{c}=E_{0}$ at the time of closing the switch are easily found, from which the transient is determined. It is evident, from this method of analysis, that the wave form of the transient need bear no relationship to that of the applied voltage, depending only on the constants of the circuit and the hypothetical initial conditions $I_{0}$ and $E_{0}$.

Electrical circuit formulas continued

## 19. Effective and average values of alternating current

(Similar equations apply to a-c voltages)
$i=I \sin \omega t$
Average value $I_{a_{0}}=\frac{2}{\pi} I$
which is the direct current which would be obtained were the original current fully rectified, or approximately proportional to the reading of a rectifiertype meter.

Effective or root-mean-square (rms) value $I_{\text {afs }}=\frac{i}{\sqrt{2}}$
which represents the heating or power effectiveness of the current, and is proportional to the reading of a dynamometer or thermal-type meter.
When

$$
\begin{aligned}
i & =I_{0}+I_{1} \sin \omega_{1} t+I_{2} \sin \omega_{2} t+\ldots \\
I_{e f f} & =\sqrt{I_{0}^{2}+\frac{1}{2}\left(I_{1}^{2}+I_{2}^{2}+\ldots\right)}
\end{aligned}
$$

Note: The average value of a complex current is not equal to the sum of the average values of the components.

## 20. Constants of Iong transmission lines

$\alpha=\sqrt{\frac{1}{2}\left\{\sqrt{\left(R^{2}+\omega^{2} L^{2}\right)\left(G^{2}+\omega^{2} C^{2}\right)}+G R-\omega^{2} L C\right\}}$
$\beta=\sqrt{\frac{1}{2}\left\{\sqrt{\left(R^{2}+\omega^{2} L^{2}\right)\left(G^{2}+\omega^{2} C^{2}\right)}-G R+\omega^{2} L C\right\}}$
where
$\alpha=$ attenuation constant in nepers
$\beta=$ phase constant in radians
$R=$ resistance constant in ohms
$G=$ conductance constant in mhos per unit length of line.
$L=$ inductance constant in henries
C $=$ capacitance constant in farads
$\omega=2 \pi X$ frequency in cycles per second
Using values per mile for $R, G, L$, and $C$, the db loss per mile will be $8.686 \alpha$ and the wavelength in miles will be $\frac{2 \pi}{\beta}$.

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## Electrical circuif formulas continued

If vector formulas are preferred, $\alpha$ and $\beta$ may be determined from the following:
$\alpha+j \beta=\sqrt{Z Y}=\sqrt{(R+j \omega L)(G+j \omega C)}$
where all constants have the same meaning as above.
Characteristic impedance
$Z_{0}=\sqrt{\frac{\bar{Z}}{Y}}=\sqrt{\frac{R+j \omega L}{G+j \omega C}}$
Note: For radio frequency applications, see formulas under R-F Transmission Line Data.

## Attenuators

An attenuator is a network designed to introduce a known loss when working between resistive impedances $Z_{1}$ and $Z_{2}$ to which the input and output impedances of the attenuator are matched. Either $Z_{1}$ or $Z_{2}$ may be the source and the other the load. The attenuation of such networks expressed as a power ratio is the same regardless of the direction of working.
Three forms of resistance network which may be conveniently used to realize these conditions are shown on page 106. These are the $T$ section, the $\pi$ section, and the Bridged-T section. Equivalent balanced sections also are shown. Methods are given for the computation of attenuator networks, the hyperbolic expressions giving rapid solutions with the aid of tables of hyperbolic functions on pages 313 to 315 . Tables of the various types of attenuators are given on pages 108 to 114 .

## In the formulas

$Z_{1}$ and $Z_{2}$ are the terminal impedances (resistivel to which the attenuator is matched.
$N$ is the ratio of the power absorbed by the attenuator from the source to the power delivered to the load.
$K$ is the ratio of the attenuator input current to the output current into the load. When $Z_{1}=Z_{2}, K=\sqrt{N}$.

Attenuation in decibels $=10 \log _{10} N$
Attenuation in nepers $=\theta=\frac{1}{2} \log _{e} N$
For a table of decibels versus power and voltage or current ratio, see page 34. Factors for converting decibels to nepers, and nepers to decibels, are given at the foot of that table.

## Attenuators continued

## General remarks

The formulas and figures for errors, given in Tables IV to VIII, are based on the assumption that the attenuator is terminated approximately by its proper terminal impedances $Z_{1}$ and $Z_{2}$. They hold for deviations of the attenuator arms and load impedances up to $\pm 20$ percent or somewhat more. The error due to each element is proportional to the deviation of the element, and the total error of the attenuator is the sum of the errors due to each of the several elements.

When any element or arm $R$ has a reactive component $\Delta X$ in addition to a resistive error $\Delta R$, the errors in input impedance and output current are
$\Delta Z=A(\Delta R+j \Delta X)$
$\frac{\Delta i}{i}=B\left(\frac{\Delta R+j \Delta X}{R}\right)$
where $A$ and $B$ are constants of proportionality for the elements in question. These constants can be determined in each case from the figures given for errors due to a resistive deviation $\Delta R$.

The reactive component $\Delta X$ produces a quadrature component in the output current, resulting in a phase shift. However, for small values of $\Delta X$, the error in insertion loss is negligibly small.
For the errors produced by mismatched terminal load impedance, refer to Case 1, page 105.

## Ladder attenuafor



Fig. 15-Ladder aftenuator.

Ladder attenuator, Fig. 15 , input switch points $P_{0}, P_{1}, P_{2}, P_{3}$ at shunt arms. Also intermediate point $P_{m}$ tapped on series arm. May be either unbalanced, as shown, or balanced.

## Aftenuators continued

Ladder, for design purposes, Fig. 16, is resolved into a cascade of $\pi$ sections by imagining each shunt arm split into two resistors. Last section matches $\mathbf{Z}_{\mathbf{2}}$ to $2 Z_{1}$. All other sections are symmetrical, matching impedances $2 Z_{1}$, with a


Fig. 16-Ladder attenuator resolved Into a coscade of $\pi$ sections.
terminating resistor $2 Z_{1}$ on the first section. Each section is designed for the loss required between the switch points at the ends of that section.

Input to $P_{0}$ : Loss, $d b=10 \log _{10} \frac{\left(2 Z_{1}+Z_{2}\right)^{2}}{4 Z_{1} Z_{2}}$
Input impedance $Z_{1}^{\prime}=\frac{Z_{2}}{2}$
Output impedance $=\frac{Z_{1} Z_{2}}{Z_{1}+Z_{2}}$
Input to $P_{1}, P_{2}$, or $P_{3}$ : Loss, $\mathrm{db}=3 \mathrm{db}+$ sum of losses of $\pi$ sections between input and output. Input impedance $Z_{1}{ }^{\prime}=Z_{1}$

Input to $P_{m}$ (on a symmetrical $\pi$ section):
$\frac{e_{0}}{e_{m}}=\frac{1}{2} \frac{m(1-m)(K-1)^{2}+2 K}{K-m(K-1)}$
where
$\mathbf{e}_{0}=$ output voltage when $m=0$ (Switch on $P_{1}$ ).
$\mathbf{e}_{\boldsymbol{m}}=$ output voltage with switch on $P_{m}$.
and
$K=$ current ratio of the section (from $P_{1}$ to $P_{2}$ ). $K>1$.
Input impedance $Z_{1}^{\prime}=Z_{1}\left[m(1-m) \frac{(K-1)^{2}}{K}+1\right]$
$\operatorname{Max} Z_{1}^{\prime}=Z_{1}\left[\frac{(K-1)^{2}}{4 K}+1\right]$ for $m=0.5$.

## Attenuators continued

The unsymmetrical last section may be treated as a system of voltage dividing resistors. Solve for the resistance $R$ from $P_{0}$ to the tap, for each value of
output voltage with input on $P_{0}$
output voltage with input on tap
A useful case: $Z_{1}=Z_{2}=500$ ohms.
Then loss on $P_{0}$ is 3.52 db .
Let the last section be designed for loss of 12.51 db .
Then
$R_{13}=2444$ ohms (shunted by 1000 ohms)
$R_{23}=654$ ohms (shunted by 500 ohms)
$R_{12}=1409$ ohms.
The table shows the location of the tap and the input and output impedances for several values of loss, relative to the loss on $P_{0}$.

| rulative <br> loss <br> db | tap <br> $R$ <br> ohms | inpuf <br> impedance <br> ohms | oufput <br> impedance <br> ohms |
| :---: | :---: | :---: | :---: |
| 0 | 0 |  |  |
| 2 | 170 | 250 | 250 |
| 4 | 375 | 368 | 304 |
| 6 | 615 | 378 | 353 |
| 8 | 882 | 562 | 394 |
| 10 | 1157 | 600 | 428 |
| 12 | 1409 | 577 | 454 |
|  |  |  | 473 |



Fig. 17-A variation of the ladder attenvator, useful when $Z_{1}=Z_{2}=\mathbf{Z}$. Simpler in design, with improved impedance characteristics, but having minimum insertion loss 2.5 db higher then aftenvator of Fig. 16. All $\pi$ sections are symmetrical.

## Aftenuators continued

Input to $P_{0}$ : Output impedance $=0.6 Z \quad$ (See Fig. 17.)
Input to $\mathrm{P}_{0}, \mathrm{P}_{1}, \mathrm{P}_{2}$, or $\mathrm{P}_{3}$ : Loss $=6 \mathrm{db}+$ sum of losses of $\pi$ sections between input and output. Input impedance $=Z$

Input to $P_{m}: \quad \frac{e_{0}}{e_{m}}=\frac{1}{4} \frac{m(1-m)(K-1)^{2}+4 K}{K-m(K-1)}$
Input impedance $Z^{\prime}=Z\left[\frac{m(1-m)(K-1)^{2}}{2 K}+1\right]$
$\operatorname{Max} Z^{\prime}=Z\left[\frac{(k-1)^{2}}{8 K}+1\right]$ for $m=0.5$.

## Effect of incorrect load impedance on operation of an aftenuator

In the applications of attenuators the question frequently arises as to the effect upon the input impedance and the attenuation by the use of a load impedance which is different from that for which the network was designed. The following results apply to all resistive networks which, when operated between resistive impedances $Z_{1}$ and $Z_{2}$, present matching terminal impedances $Z_{1}$ and $Z_{2}$, respectively. The results may be derived in the general case by the application of the network theorems, and may be readily confirmed mathematically for simple specific cases such as the T section.

For the designed use of the network, let
$Z_{1}=$ input impedance of properly terminated network
$Z_{2}=$ load impedance which properly terminates the network
$N=$ power ratio from input to output
$K=$ current ratio from input to output
$K=\frac{i_{1}}{i_{2}}=\sqrt{\frac{N Z_{2}}{Z_{1}}}$ idifferent in the two directions of operation except when $Z_{2}=Z_{1}$.

For the actual conditions of operation, let
$\left(Z_{2}+\Delta Z_{2}\right)=Z_{2}\left(1+\frac{\Delta Z_{2}}{Z_{2}}\right)=$ actual load impedance
$\left(Z_{1}+\Delta Z_{1}\right)=Z_{1}\left(1+\frac{\Delta Z_{1}}{Z_{1}}\right)=$ resulting input impedance
$(K+\Delta K)=K\left(1+\frac{\Delta K}{K}\right)=$ resulting current ratio.

Altenuafors continued

While $Z_{1}, Z_{2}$, and $K$ are restricted to real quantities by the assumed nature of the network, $\Delta Z_{2}$ is not so restricted, e.g.,
$\Delta Z_{2}=\Delta R_{2}+j \Delta X_{2}$
As a consequence $\Delta Z_{1}$ and $\Delta K$ can become imaginary or complex. Further-: more $\Delta Z_{2}$ is not restricted to small values.

The results for the actual conditions are
$\frac{\Delta Z_{1}}{Z_{1}}=\frac{2 \frac{\Delta Z_{2}}{Z_{2}}}{2 N+\left(N-11 \frac{\Delta Z_{2}}{Z_{2}}\right.}$ and $\frac{\Delta K}{K}=\left(\frac{N-1}{2 N}\right) \frac{\Delta Z_{2}}{Z_{2}}$

## Certain special cases may be cited

Case 1: For small $\frac{\Delta Z_{2}}{Z_{2}}$
$\frac{\Delta Z_{1}}{Z_{1}}=\frac{1}{N} \frac{\Delta Z_{2}}{Z_{2}} \quad$ or $\quad \Delta Z_{1}=\frac{1}{K^{2}} \Delta Z_{2} \quad \frac{\Delta i_{2}}{i_{2}}=-\frac{1}{2} \frac{\Delta Z_{2}}{Z_{2}}$
but the error in insertion power loss of the attenuator is neglibly small.
Case 2: Short-circuited output $\frac{\Delta Z_{1}}{Z_{1}}=\frac{-2}{N+1}$
or input impedance $=\left(\frac{N-1}{N+1}\right) Z_{1}=Z_{1} \tanh \theta$
where $\theta$ is the designed attenuation in nepers.
Case 3: Open-circuited output $\quad \frac{\Delta Z_{1}}{Z_{1}}=\frac{2}{N-1}$
or input impedance $=\left(\frac{N+1}{N-1}\right) Z_{1}=Z_{1} \operatorname{coth} \theta$
Case 4: For $N=1$ (possible only when $Z_{1}=Z_{2}$ and directly connected)
$\frac{\Delta Z_{1}}{Z_{1}}=\frac{\Delta Z_{2}}{Z_{2}}$ and $\frac{\Delta K}{K}=0$
Case 5: For large $N \quad \frac{\Delta K}{K}=\frac{1}{2} \frac{\Delta Z_{2}}{Z_{2}}$


## nelwork design

| design formulas |  | cheeking formulas |
| :---: | :---: | :---: |
| hyperbolic | arithmetical |  |
| $R_{\mathrm{a}}=\frac{\sqrt{\bar{Z}_{1} \bar{Z}_{2}}}{\sinh \theta}$ | $R_{8}=\frac{2 \sqrt{N Z_{1} Z_{2}}}{N-1}$ |  |
| $R_{1}=\frac{Z_{1}}{\tanh \theta}-R_{8}$ | $R_{1}=Z_{1}\left(\frac{N+1}{N-1}\right)-R_{2}$ |  |
| $R_{2}=\frac{Z_{2}}{\tanh \theta}-R_{3}$ | $R_{2}=Z_{2}\left(\frac{N+1}{N-1}\right)-R_{2}$ |  |
| $\begin{aligned} & R_{3}=\frac{Z}{\sinh \theta} \\ & R_{1}=Z \tanh \frac{\theta}{2} \end{aligned}$ | $\begin{aligned} & R_{3}=\frac{2 Z \sqrt{N}}{N-1}=\frac{2 Z K}{K^{2}-1} \\ & R_{1}=Z \frac{\sqrt{N}-1}{\sqrt{N}+1}=z \frac{K-1}{K+1} \end{aligned}$ | $\begin{aligned} R_{1} R_{8} & =\frac{Z^{2}}{1+\cosh \theta}=Z^{2} \frac{2 K}{(K+1)^{2}} \\ \frac{R_{1}}{R_{3}} & =\cosh \theta-1=2 \sinh ^{2} \frac{\theta}{2} \\ & =\frac{(K-1)^{2}}{2 K} \\ Z & =R_{1} \sqrt{1+2 \frac{R_{3}}{R_{1}}} \end{aligned}$ |
| $\begin{aligned} \cosh \theta & =\sqrt{\frac{Z_{1}}{Z_{2}}} \\ \cosh 2 \theta & =2 \frac{Z_{1}}{Z_{2}}-1 \end{aligned}$ | $\begin{aligned} & R_{1}=Z_{1} \sqrt{1-\frac{Z_{2}}{Z_{1}}} \\ & R_{2}=\frac{Z_{2}}{\sqrt{1-\frac{Z_{3}}{Z_{1}}}} \end{aligned}$ | $\begin{aligned} R_{1} R_{2} & =Z_{1} Z_{2} \\ \frac{R_{1}}{R_{3}} & =\frac{Z_{1}}{Z_{2}}-1 \\ N & =\left(\sqrt{\frac{Z_{1}}{Z_{2}}}+\sqrt{\frac{Z_{1}}{Z_{2}}-1}\right)^{2} \end{aligned}$ |
| $\begin{aligned} & R_{3}=\sqrt{Z_{1} Z_{2}} \sinh \theta \\ & \frac{1}{R_{1}}=\frac{1}{Z_{1} \tanh \theta}-\frac{1}{R_{3}} \\ & \frac{1}{R_{2}}=\frac{1}{Z_{2} \tanh \theta}-\frac{1}{R_{8}} \end{aligned}$ | $\begin{aligned} & R_{i}=\frac{N-1}{2} \sqrt{\frac{Z_{1} Z_{2}}{N}} \\ & \frac{1}{R_{1}}=\frac{1}{Z_{1}}\left(\frac{N+1}{N-1}\right)-\frac{1}{R_{2}} \\ & \frac{1}{R_{2}}=\frac{1}{Z_{2}}\left(\frac{N+1}{N-1}\right)-\frac{1}{R_{3}} \end{aligned}$ |  |
| $\begin{aligned} & R_{8}=Z \sinh \theta \\ & R_{1}=\frac{Z}{\tanh \frac{\theta}{2}} \end{aligned}$ | $\begin{aligned} & R_{3}=z \frac{N-1}{2 \sqrt{N}}=2 \frac{K^{2}-1}{2 K} \\ & R_{1}=Z \frac{\sqrt{N}+1}{\sqrt{N}-1}=z \frac{K+1}{K-1} \end{aligned}$ | $\begin{aligned} R_{1} R_{3} & =Z^{2}(1+\cosh \theta)=Z^{2} \frac{(K+1)^{2}}{2 K} \\ \frac{R_{8}}{R_{1}} & =\cosh \theta-1=\frac{(K-1)^{2}}{2 K} \\ Z & =\frac{R_{1}}{\sqrt{1+2 \frac{R_{1}}{R_{3}}}} \end{aligned}$ |
| , | $\begin{aligned} & R_{1}=R_{2}=Z \\ & R_{4}=Z(K-1) \\ & R_{3}=\frac{Z}{K-1} \end{aligned}$ | $\begin{aligned} & R_{2} R_{4}=Z^{2} \\ & \frac{R_{1}}{R_{2}}=\{K-1\}^{2} \end{aligned}$ |

four-terminal networks: The hyperbolic formulas above are valid for passive linear fourterminal networks in general, working between input and output impedances moiching the respective imege impedances. In this cases $Z_{1}$ and $Z_{2}$ are the image impedances; $R_{1} R_{2}$ and $R_{2}$ become complex impedances, and $\theta$ is the image transfer constant, $\theta=\alpha+j B$, where $\alpha$ is the imoge attenuation constont and $\beta$ is the image phase constant.

Attenuafors continued

Table IV-Symmetrical T or H attenuator
$Z=500$ ohms resistive (diagram page 106)

| aftenuation db | series arm $\mathbf{R}_{\mathbf{l}}$ ohms | shunt arm $\mathbf{R}_{3}$ ohms | $\frac{1000}{R_{3}}$ | $\log _{10} \mathrm{R}_{\mathbf{1}}$ |
| :---: | :---: | :---: | :---: | :---: |
| 0.0 | 0.0 | inf | 0.0000 |  |
| 0.2 | 5.8 | 21,700 | 0.0461 |  |
| 0.4 | 11.5 | 10,850 | 0.0921 |  |
| 0.6 | 17.3 | 7,230 | 0.1383 |  |
| 0.8 | 23.0 | 5,420 | 0.1845 |  |
| 1.0 | 28.8 | 4,330 | 0.2308 |  |
| 2.0 | 57.3 | 2,152 | 0.465 |  |
| 3.0 | 85.5 | 1,419 | 0.705 |  |
| 4.0 | 113.1 | 1,048 | 0.954 |  |
| 5.0 | 140.1 | 822 | 1.216 |  |
| 6.0 | 166.1 | 669 | 1.494 | 2.826 |
| 7.0 | 191.2 | 558 |  | 2.747 |
| 8.0 | 215.3 | 473.1 |  | 2.675 |
| 9.0 | 238.1 | 405.9 |  | 2.608 |
| 10.0 | 259.7 | 351.4 |  | 2.546 |
| 12.0 | 299.2 | 268.1 |  | 2.428 |
| 14.0 | 333.7 | 207.8 |  | 2.318 |
| 16.0 | 363.2 | 162.6 |  | 2.211 |
| 18.0 | 388.2 | 127.9 |  | 2.107 |
| 20.0 | 409.1 | 101.0 |  | 2.004 |
| 22.0 | 426.4 | 79.94 |  | 1.903 |
| 24.0 | 440.7 | 63.35 |  | 1.802 |
| 26.0 | 452.3 | 50.24 |  | 1.701 |
| 28.0 | 461.8 | 39.87 |  | 1.601 |
| 30.0 | 469.3 | 31.65 |  | 1.500 |
| 35.0 | 482.5 | 17.79 |  | 1.250 |
| 40.0 | 490.1 | 10.00 |  | 1.000 |
| 50.0 | 496.8 | 3.162 |  | 0.500 |
| 60.0 | 499.0 | 1.000 |  | 0.000 |
| 80.0 | 499.9 | 0.1000 |  | -1.000 |
| 100.0 | 500.0 | 0.01000 |  | -2.000 |

Attenuafors continued

## Interpolation of symmetrical Tor H attenuators

Column $R_{1}$ may be interpolated linearly. Do not interpolate $R_{3}$ column. For 0 to 6 db , interpolate the $\frac{1000}{R_{3}}$ column. Above 6 db , interpolate the column $\log _{10} R_{3}$ and determine $R_{3}$ from the result.

Errors in symmetrical Tor H attenuators
Series arms $\boldsymbol{R}_{\mathbf{1}}$ and $\boldsymbol{R}_{\mathbf{2}}$ in error Error in input impedances:
$\Delta Z_{1}=\Delta R_{1}+\frac{1}{K^{2}} \Delta R_{2}$
and
$\Delta Z_{2}=\Delta R_{2}+\frac{1}{K^{2}} \Delta R_{1}$

nominally $\mathbf{R}_{\mathbf{1}}=\mathbf{R}_{\mathbf{2}}$ and $\mathbf{Z}_{\mathbf{1}}=\mathbf{Z}_{\mathbf{2}}$

Error in insertion loss, $\mathrm{db}=4\left(\frac{\Delta R_{1}}{Z_{1}}+\frac{\Delta R_{2}}{Z_{2}}\right)$, approximately.

Shunt arm $\mathrm{R}_{3}$ in error ( 10 percent high)

| designed loss, <br> $\mathbf{d b}$ | error in insertion <br> loss, $\mathbf{d b}$ | error in input <br> impedance <br> $\mathbf{1 0 0} \frac{\Delta \mathbf{Z}}{\mathbf{Z}}$ percent |
| :---: | :---: | :---: |
| 0.2 | -0.01 | 0.2 |
| $\mathbf{1}$ | -0.05 | 1.0 |
| 6 | -0.3 | 3.3 |
| 12 | -0.5 | 3.0 |
| 20 | -0.7 | 1.6 |
| 40 | -0.8 | 0.2 |
| 100 | -0.8 | 0.0 |

Error in input impedance: $\frac{\Delta Z}{Z}=2 \frac{K-1}{K(K+1)} \frac{\Delta R_{3}}{R_{3}}$
Error in output current: $\frac{\Delta i}{i}=\frac{K-1}{K+1} \frac{\Delta R_{3}}{R_{3}}$
See General Remarks on page 101.

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## Attenuators continued

## Table V-Symmefrical $\pi$ and $\mathbf{O}$ attenuators

The values of the series and shunt arms of these attenuators may be determined from Table IV of symmetrical $T$ attenuators by means of the following formulas.

Shunt ${ }_{\text {arms: }} R_{13}=R_{23}=R_{1}+2 R_{3}{ }^{\circ}=\frac{Z^{2}}{R_{1}}$
$\begin{aligned} & \text { Series } \\ & \text { arm: } \\ & R_{12}\end{aligned}=R_{1}\left(\frac{R_{1}}{R_{3}}+2\right)=\frac{Z^{2}}{R_{3}}$
Error in loss, $\mathrm{db}=-8 \frac{\Delta i_{2}}{i_{2}}$ (approximately)

$=4 \frac{K-1}{K+1}\left(-\frac{\Delta R_{13}}{R_{13}}-\frac{\Delta R_{23}}{R_{23}}+2 \frac{\Delta R_{12}}{R_{12}}\right)$
Error in input impedance:
$\frac{\Delta Z^{\prime}}{Z^{\prime}}=\frac{K-1}{K+1}\left(\frac{\Delta R_{13}}{R_{13}}+\frac{1}{K^{2}} \frac{\Delta R_{23}}{R_{28}}+\frac{2}{K} \frac{\Delta R_{12}}{R_{12}}\right)$
$\mathbf{R}_{18}=\mathbf{R}_{23}$ and $Z^{\prime}=\mathbf{Z}$


Table VI-Bridged T or H affenuafor


## Attenuators cominued

## Interpolation of bridged T or H attenuators

Bridge arm $R_{4}$ : Use the formula $\log _{10}\left(R_{4}+500\right)=2.699+\frac{\mathrm{db}}{20}$ for $Z=500$ ohms. However, if preferred, the tabular values of $R_{4}$ may be interpolated linearly, between 0 and 10 db only.

Shunt arm $R_{3}$ : Do not interpolate $R_{3}$ column. Compute $R_{3}$ by the formula $R_{3}=\frac{10^{6}}{4 R_{3}} \quad$ for $Z=500$ ohms.

Note: For attenuators of 60 db and over, the bridge arm $R_{4}$ may be omitted, provided a shunt arm is used having twice the resistance tabulated in the $R_{3}$ column. (This makes the input impedance 0.1 of 1 percent high at 60 db .)

Errors in bridged T or H attenuators
For resistance of any ene arm 10 percent higher than the correct value

| designed loss <br> $\mathbf{d b}$ | col 1* <br> $\mathbf{d b}$ | col 2* <br> percent | col 3* <br> percent |
| :---: | :---: | :---: | :---: |
|  |  |  |  |
| 0.2 | 0.01 | 0.005 | 0.2 |
| 1 | 0.05 | 0.1 | 1.0 |
| 6 | 0.2 | 2.5 | 2.5 |
| 12 | 0.3 | 5.6 | 1.9 |
| 20 | 0.4 | 8.1 | 0.9 |
| 40 | 0.4 | 10 | 0.1 |
| 100 | 0.4 | 10 | 0.0 |

* Refer to following rabulation.

| element in error (10 percent high) | error in loss | error in terminal impedance | remarks |
| :---: | :---: | :---: | :---: |
| Series orm $R_{1}$ lanologous for arm R2) | Zero | Col 2, for adiacent terminals | Error in impedance al opposite lerminols is zero |
| Shunt arm $R_{8}$ | - Coll | Col 3 | Loss is lower than designed loss |
| Bridge orm $\mathbf{R}_{\mathbf{4}}$ | +Col 1 | Col 3 | Loss is higher than designed loss |

Error in input impedance: $\frac{\Delta Z_{1}}{Z_{1}}=\left(\frac{K-1}{K}\right)^{2} \frac{\Delta R_{1}}{R_{1}}+\frac{K-1}{K^{2}}\left(\frac{\Delta R_{3}}{R_{8}}+\frac{\Delta R_{4}}{R_{4}}\right)$ For $\frac{\Delta Z_{2}}{Z_{2}}$ use subscript 2 in formula in place of subscript 1.
Error in output current: $\frac{\Delta i}{i}=\frac{K-1}{2 K}\left(\frac{\Delta R_{3}}{R_{3}}-\frac{\Delta R_{3}}{R_{4}}\right)$
See General Remarks on page 101.

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Attenuators
continued

## Table VII-Minimum loss pads

Matching $\mathbf{Z}_{1}$ and $\mathbf{Z}_{2}$ - both resistive (diagram page 106)

| $\begin{gathered} \mathbf{Z}_{1} \\ \text { ohms } \end{gathered}$ | $\begin{gathered} \mathbf{Z}_{2} \\ \text { ohros } \end{gathered}$ | $\frac{\mathbf{z}_{1}}{\mathbf{z}_{2}}$ | $\begin{aligned} & \text { loss } \\ & \mathrm{db} \end{aligned}$ | series arm $\mathrm{R}_{1}$ ohms | shunt arm $\mathbf{R}_{\mathbf{3}}$ ohms |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 10,000 | 500 | 20.00 | 18.92 | 9,747 | 513.0 |
| 8,000 | 500 | 16.00 | 17.92 | 7.746 | 516.4 |
| 6,000 | 500 | 12.00 | 16.63 | 5,745 | 522.2 |
| 5,000 | 500 | 10.00 | 15.79 | 4,743 | 527.0 |
| 4,000 | 500 | 8.00 | 14.77 | 3,742 | 534.5 |
| 3,000 | 500 | 6.00 | 13.42 | 2,739 | 547.7 |
| 2,500 | 500 | 5.00 | 12.54 | 2,236 | 559.0 |
| 2,000 | 500 | 4.00 | 11.44 | 1,732 | 577.4 |
| 1,500 | 500 | 3.00 | 9.96 | 1,224.7 | 612.4 |
| 1,200 | 500 | 2.40 | 8.73 | 916.5 | 654.7 |
| 1,000 | 500 | 2.00 | 7.66 | 707.1 | 707.1 |
| 800 | 500 | 1.60 | 6.19 | 489.9 | 816.5 |
| 600 | 500 | 1.20 | 3.77 | 244.9 | 1,224.7 |
| 500 | 400 | 1.25 | 4.18 | 223.6 | 894.4 |
| 500 | 300 | 1.667 | 6.48 | 316.2 | 474.3 |
| 500 | 250 | 2.00 | 7.66 | 353.6 | 353.6 |
| 500 | 200 | 2.50 | 8.96 | 387.3 | 258.2 |
| 500 | 160 | 3.125 | 10.17 | 412.3 | 194.0 |
| 500 | 125 | 4.00 | 11.44 | 433.0 | 144.3 |
| 500 | 100 | 5.00 | 12.54 | 447.2 | 111.80 |
| 500 | 80 | 6.25 | 13.61 | 458.3 | 87.29 |
| 500 | 65 | 7.692 | 14.58 | 466.4 | 69.69 |
| 500 | 50 | 10.00 | 15.79 | 474.3 | 52.70 |
| 500 | 40 | 12.50 | 16.81 | 479.6 | 41.70 |
| 500 | 30 | 16.67 | 18.11 | 484.8 | 30.94 |
| 500 | 25 | 20.00 | 18.92 | 487.3 | 25.65 |

Interpolation of minimum loss pads
This table may be interpolated linearly with respect to $Z_{1}, Z_{2}$, or $\frac{Z_{1}}{Z_{2}}$ except when $\frac{Z_{1}}{Z_{2}}$ is between 1.0 and 1.2. The accuracy of the interpolated value becomes poorer as $\frac{Z_{1}}{Z_{2}}$ passes below 2.0 toward 1.2 , especially for $R_{3}$.

Atfenuafors continued

## For other terminations

If the terminating resistances are to be $Z_{A}$ and $Z_{B}$ instead of $Z_{1}$ and $Z_{2}$, respectively, the procedure is as follows. Enter the table at $\frac{Z_{1}}{Z_{2}}=\frac{Z_{A}}{Z_{B}}$ and read the loss and the tabular values of $R_{1}$ and $R_{3}$. Then the series and shunt arms are, respectively, $M R_{1}$ and $M R_{3}$, where $M=\frac{Z_{A}}{Z_{1}}=\frac{Z_{B}}{Z_{2}}$.

## Errors in minimum loss pads

| impedance ratio <br> $\mathbf{Z}_{1}$ | col 1* <br> db | col 2* <br> percenf | col 3** <br> percent |
| :---: | :---: | :---: | :---: |
| 1.2 | 0.2 | +4.1 | +1.7 |
| 2.0 | 0.3 | 7.1 | 1.2 |
| 4.0 | 0.35 | 8.6 | 0.6 |
| 10.0 | 0.4 | 9.5 | 0.25 |
| 20.0 | 0.4 | 9.7 | 0.12 |

## * Notes

Series arm $R_{1} 10$ percent high: Loss is increased by col 1. Input impedance $Z_{1}$ is increased by col 2. Input impedance $Z_{2}$ is increased by col 3.
Shunt arm $R_{3} 10$ percent high: Loss is decreased by col 1 . Input impedance $Z_{2}$ is increased by col 2. Input impedance $Z_{1}$ is increased by col 3.

Errors in input impedance
$\frac{\Delta Z_{1}}{Z_{1}}=\sqrt{1-\frac{Z_{2}}{Z_{1}}}\left(\frac{\Delta R_{1}}{R_{1}}+\frac{1}{N} \frac{\Delta R_{3}}{R_{3}}\right)$
$\frac{\Delta Z_{2}}{Z_{2}}=\sqrt{1-\frac{Z_{2}}{Z_{1}}}\left(\frac{\Delta R_{3}}{R_{3}}+\frac{1}{N} \frac{\Delta R_{1}}{R_{1}}\right)$

Error in output current, working either direction
$\frac{\Delta i}{i}=\frac{1}{2} \sqrt{1-\frac{Z_{2}}{Z_{1}}}\left(\frac{\Delta R_{3}}{R_{3}}-\frac{\Delta R_{1}}{R_{1}}\right)$
See General Remarks on page 101.

Attenuators continued

## Table VIII-Miscellaneous $\mathbf{T}$ and $\mathbf{H}$ pads

(diagram page 106)

| resistive terminotions |  | $\begin{gathered} \text { loss } \\ \mathrm{db} \end{gathered}$ | aftenuator arms |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: |
| $\underset{\text { ohms }}{Z_{1}}$ | $\begin{gathered} \mathbf{Z}_{2} \\ \text { ohms } \end{gathered}$ |  | series $\mathbf{R}_{1}$ ohm: | serles $R_{2}$ ohms | shunt Rs -hms |
| 5,000 | 2,000 | 10 | 3,889 | 222 | 2,222 |
| 5,000 | 2,000 | 15 | 4,165 | 969 | 1,161 |
| 5,000 | 2,000 | 20 | 4,462 | 1,402 | 639 |
| 5,000 | 500 | 20 | 4,782 | 190.7 | 319.4 |
| 2,000 | 500 | . 15 | 1,763 | 165.4 | 367.3 |
| 2,000 | 500 | 20 | 1,838 | 308.1 | 202.0 |
| 2,000 | 200 | 20 | 1,913 | 76.3 | 127.8 |
| 500 | 200 | 10 | 388.9 | 22.2 | 222.2 |
| 500 | 200 | 15 | 416.5 | 96.9 | 116.1 |
| 500 | 200 | 20 | 446.2 | 140.2 | 63.9 |
| 500 | 50 | 20 | 478.2 | 19.07 | 31.94 |
| 200 | 50 | 15 | 176.3 | 16.54 | 36.73 |
| 200 | 50 | 20 | 183.8 | 30.81 | 20.20 |

Errors in T and H pads
Series arms $R_{1}$ and $R_{2}$ in error. Error in input impedances:
$\Delta Z_{1}=\Delta R_{1}+\frac{1}{N} \frac{Z_{1}}{Z_{2}} \Delta R_{2} \quad$ and $\quad \Delta Z_{2}=\Delta R_{2}+\frac{1}{N} \frac{Z_{2}}{Z_{1}} \Delta R_{1}$
Error in insertion loss, $\mathrm{db}=4\left(\frac{\Delta R_{1}}{Z_{1}}+\frac{\Delta R_{2}}{Z_{2}}\right)$, approximately.
Shunt afm $R_{3}$ In error ( 10 percent high)

| $\frac{\mathbf{Z}_{1}}{\mathbf{Z}_{2}}$ | designed loss <br> db | arror in loss <br> db | $\mathbf{1 0 0 \frac { \Delta \mathbf { Z } _ { 1 } } { \mathbf { Z } _ { 1 } }}$ | $\mathbf{1 0 0 \frac { \Delta \mathbf { Z } _ { 2 } } { \mathbf { Z } _ { \mathbf { 2 } } }}$ |
| :---: | :---: | :---: | :---: | :---: |
| 2.5 | 10 | -0.4 | $1.1 \%$ | $7.1 \%$ |
| 2.5 | 15 | -0.6 | 1.2 | 4.6 |
| 2.5 | 20 | -0.7 | 0.9 | 2.8 |
| 4.0 | 15 | -0.5 | 0.8 | 6.0 |
| 4.0 | 20 | -0.65 | 0.6 | 3.6 |
| 10 | 20 | -0.6 | 0.3 | 6.1 |

$\frac{\Delta Z_{1}}{Z_{1}}=\frac{2}{N-1}\left(\sqrt{\frac{N Z_{2}}{Z_{1}}}+\sqrt{\frac{Z_{1}}{N Z_{2}}}-2\right) \frac{\Delta R_{3}}{R_{3}}\left\{\right.$ for $\frac{\Delta Z_{2}}{Z_{2}}$ interchange subscripts
$\frac{\Delta i}{i}=\frac{N+1-\sqrt{N}\left(\sqrt{\frac{Z_{1}}{Z_{2}}}+\sqrt{\frac{Z_{2}}{Z_{1}}}\right)}{N-1} \frac{\Delta R_{3}}{R_{3}}\{$ where $i$ is the output current.

## Filter networks

Explanation: Table 1 X shows, in the first column, the fundamental series impedance, $Z_{1}$, and the fundamental shunt impedance, $Z_{2}$, from which the various types of filter sections shown in subsequent columns are composed. For example, a $T$ section (third column) is composed of two half-series arms, $\frac{Z_{1}}{2}$ in series, with a full shunt arm $Z_{2}$ connected to their junction point. The subsequent tables (Tables X, XI, XII, and XIIII give formulas for computing the full series arm and the full shunt arm. These must then be modified according to the type of section used.

Example: Design a series $M$ derived high-pass, $T$-section filter to terminate in 500 ohms, with cutoff frequency equal to 1000 cycles, and peak attenuation frequency equal to 800 cycles.

Using Table XIII:
$f_{c}=1000$
$f_{\infty}=800$
$R=500$
$m=\sqrt{1-\left(\frac{800}{1000}\right)^{2}}=0.6$

$C=\frac{1}{4 \pi f_{d} R}=\frac{1}{4 \pi \times 1000 \times 500}=0.159\left(10^{-6}\right)$ farad $=0.159 \mathrm{microfarad}$
$L=\frac{R}{4 \pi f_{c}}=\frac{500}{4 \pi \times 1000}=0.0398$ henry $=39.8$ millihenry
$C_{1}=\frac{C}{m}=\frac{0.159}{0.6}=0.265$ microfarad
$L_{2}=\frac{L}{m}=\frac{39.8}{0.6}=66.3$ millihenry
$C_{2}=\frac{4 m}{1-m^{2}} C=\frac{4 \times 0.6 \times 0.159}{0.64}=0.597$ microfarad
For a $T$-section, each series arm must be $\frac{Z_{1}}{2}$ while the full shunt arm is used.
Thus for the series arm use $2 \mathrm{C}_{1}$, or 0.53 microfarad. The accompanying figure shows the final result.

Filfer nełworks continued

## Table IX-Combination of filfer elements

$\frac{\text { configurafion }}{\text { half-section }}$

Table X-Band-pass filters

| ype | conflguration | serles arm | shunt arm | notations |
| :---: | :---: | :---: | :---: | :---: |
| Constant K |  | $\begin{aligned} L_{1} & =\frac{R}{\pi\left(f_{2}-f_{1}\right)} \\ C_{1} & =\frac{f_{2}-f_{1}}{4 \pi f_{2} f_{1} R} \end{aligned}$ | $\begin{aligned} & L_{2}=\frac{f_{2}-f_{1}}{4 \pi f_{1} f_{2}} R \\ & C_{2}=\frac{1}{\pi\left(f_{2}-f_{1}\right) R} \end{aligned}$ | $f_{2}=\begin{gathered} \text { upper cutoff } \\ \text { froquency } \end{gathered}$ |
| Three element series type | $\mathrm{c}_{0}^{\mathrm{L}_{1}} \mathrm{c}_{1}$ | $\begin{aligned} & L_{1}=\frac{R}{\pi\left(f_{2}-f_{1}\right)} \\ & C_{1}=\frac{f_{2}-f_{1}}{4 \pi f_{1}^{2} R} \end{aligned}$ | $C_{8}=\frac{1}{\pi\left(f_{1}+f_{2}\right) R}$ | $R=\begin{gathered} \text { nominal } \\ \\ \text { terminating } \\ \text { resistance } \end{gathered}$ |
| Three element shunt type |  | $C_{1}=\frac{f_{1}+f_{2}}{4 \pi f_{1} f_{2} R}$ | $\begin{aligned} & L_{3}=\frac{f_{2}-f_{1}}{4 \pi f_{1} f_{2}} R \\ & C_{2}=\frac{f_{1}}{\pi f_{2}\left(f_{2}-f_{1}\right) R} \end{aligned}$ |  |

Table XI-Band-elimination filters

| type | configuration | serles arm | shunt arm | notaflons |
| :---: | :---: | :---: | :---: | :---: |
| Constant K |  | $\begin{aligned} & L_{1}=\frac{f_{2}-f_{1}}{\pi f_{1} f_{2}} R \\ & C_{1}=\frac{1}{4 \pi\left(f_{2}-f_{1}\right) R} \end{aligned}$ | $\begin{aligned} & L_{2}=\frac{R}{4 \pi\left(f_{2}-f_{1}\right)} \\ & C_{2}=\frac{f_{2}-f_{1}}{\pi f_{1} f_{2} R} \end{aligned}$ | $f_{2}=$ upper cutoff frequency <br> $f_{1}=$ lower cutoff frequency <br> $R=$ nominal terminating resistance |

Filfer nefworks continued
Table XII—Low-pass filiers

| type | configuration | serles arm | shunt arm | notations |
| :---: | :---: | :---: | :---: | :---: |
| Constant K |  | $L=\frac{R}{\pi f_{c}}$ | $C=\frac{1}{\pi f_{c} R}$ | $\boldsymbol{i}_{c}=\underset{\text { frequency }}{\text { cuta }}$ |
| Series M derived |  | $L_{1}=m L$ | $\begin{gathered} L_{2}=\frac{1-m^{2}}{4 m} L \\ C_{2}=m C \end{gathered}$ | $\begin{aligned} & f_{\infty}= \begin{array}{l} \text { frequency of } \\ \\ \\ \text { patt } \\ \text { attenuation } \end{array} \\ & m=\sqrt{1-\left(\frac{f_{c}}{f_{\infty}}\right)^{2}} \end{aligned}$ |
| Shunt M derived |  | $\begin{gathered} L_{1}=m L \\ C_{1}=\frac{1-m^{2}}{4 m} C \end{gathered}$ | $C_{2}=m C$ | $R=\underset{\substack{\text { terminating } \\ \text { resistance }}}{\text { nominal }}$ |

Table XIII-High-pass filters

| type | configuration | series arm | shunt arm | notations |
| :---: | :---: | :---: | :---: | :---: |
| Constant K |  | $C=\frac{1}{4 \pi f_{c} R}$ | $L=\frac{R}{4 \pi f_{0}}$ | $f_{c}=\underset{\text { cutoff }}{\text { frequency }}$ |
| Series M derived |  | $C_{1}=\frac{C}{m}$ | $\begin{aligned} L_{2} & =\frac{l}{m} \\ C_{2} & =\frac{4 m}{1-m^{2}} c \end{aligned}$ | $f_{\infty}=$ frequency of peak attenuation $m=\sqrt{1-\left(\frac{f_{\infty}}{f_{c}}\right)^{2}}$ |
| Shunt M derived |  | $\begin{gathered} C_{1}=\frac{C}{m} \\ L_{1}=\frac{4 m}{1-m^{2}} L \end{gathered}$ | $L_{2}=\frac{1}{m}$ | $R=$ nominol terminating resistonce |

## Rectifiers and filters



Unless otherwise stated, factors shown express the ratlo of the RMS value of the circuit quantities designated to the average DC output values of the recilfier.
factors are based on a sine wave voliage input, infinite impedance choke and no transformer or rectifier losses.

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## connections and circuif dafa

| 6-phase half-wave delfa-star | 6-phese half-wove delta-6-phase fork | (doubhose half-wave delta-double wye with bolence coll | 3-phose full-wave <br> delta-wre | 3-phase full-wave <br> delta-delia |
| :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |
| 3 | 3 | 3 6 | 3 | $\begin{aligned} & 3 \\ & 6 \end{aligned}$ |
| $\begin{aligned} & 0.042 \\ & 6 f \end{aligned}$ | 0.042 67 | $\begin{aligned} & 0.042 \\ & 6 i \end{aligned}$ | $\frac{0.042}{6 f}$ | $\begin{aligned} & 0.042 \\ & 6 f \end{aligned}$ |
| 0.740 0.816 0.955 | 0.428 1.41 0.955 | 0.855 0.707 0.955 | 0.428 1.41 0.955 | $\begin{aligned} & 0.740 \\ & 0.816 \\ & 0.955 \end{aligned}$ |
| 0.740 | 0.428 | 0.855 | 0.428 | 0.740 |
| $\begin{aligned} & 0.577 \\ & 1.28 \end{aligned}$ | 0.816 1.05 | $\begin{aligned} & 0.408 \\ & 1.05 \end{aligned}$ | $\begin{aligned} & 0.816 \\ & 1.05 \end{aligned}$ | $\begin{aligned} & 0.471 \\ & 1.05 \end{aligned}$ |
| 1.55 | 1.42 | 1.26 | 1.05 | 1.05 |
| 0.740 W | 0.428 CN | 0.855 ${ }^{\text {W }}$ | 0.428 | 0.740 |
| 0.408 | $\left\{\begin{array}{c} 0.577(B) \\ 0.408 \mathrm{ic} \end{array}\right\}$ | 0.289 | 0.816 | 0.471 |
| 1.81 | 1.79 | 1.48 | 1.05 | 1.05 |
| ${ }_{1}^{2.09}$ | ${ }_{1}^{2.09}$ | $\begin{aligned} & 2.42 \\ & 0.5 \end{aligned}$ | $1.05$ | $1.05$ |
| 0.167 | 0.167 | 0.167 | 0.333 | 0.333 |

* These circult factors are equally applicable to tube or dry plate rectifying elements.
t line PF $=D C$ output wotts/line yolt-amperes


## Rectifier filter design



## Ripple voltage vs LC for choke-input fliters

Minimum inductance for a choke-input filter is determined from

$$
L=\frac{K E}{I f}
$$

where
$L=$ minimum inductance in henries
$E=$ d-c output in volts
$I=$ output current in amperes
f = supply frequency in cps
$K=0.0527$ for full-wave, single-phase
$=0.0132$ for half-wove, three-phase
$=0.0053$ for full-wave, two-phase
$=0.0016$ for full-wave, three-phase

## Rectifier filter design

continued


## Ripple veltage vs RC for capaciter-input filters

The above chart applies to a capacitance filter with resistance load as shown at the right.
For each additional $R^{\prime} C^{\prime}$ section, oblain $R$ by adding all resistances and add $\mathrm{db}=104-20 \log f R^{\prime} \mathrm{C}^{\prime}$.
For each additional $L C^{\prime}$ section, add $d b=88.2-40 \log f$ $-20 \log L C^{\prime}$.
The above assumes that the impedance of $C^{\prime}$ is small with respect to that of $, R, R^{\prime}$, and $L$
$f=$ ripple frequency in cps
$R^{\prime}=$ series filter resistance in ohms
$C^{\prime}=$ shunt filter capacitance in microfarads
L = series filter inductance in henries.


## - Iron-core transformers and reactors

## Major ircinsformer types

1. Audio transformers: Carry audio communication frequencies or some single control frequency.
a. Input transformers: Couple a signal source, e.g., microphone or line, to the grid (s) of an amplifier.
b. Interstage transformers (usually step-up voltage): Couple the plate (s) of a vacuum tube (except a driver stagel to the grid (s) of a succeeding stage of amplification.
c. Output transformers: Couple the plate(s) of an amplifier to an output load.
d. Driver transformers lusuallystep-down voltage): Couple the plate (s) of a driver stage (pre-amplifier) to the grid (s) of an amplifier stage in which grid current is drawn.
e. Modulation transformers: Couple the plate(s) of an audio output stage to the grid or plate of a modulated amplifier.
2. Power supply transformers: Supply appropriate plate and/or filament voltage to vacuum fubes in a unit of equipment.
a. Plate transformers: Supply potential to the plate(s) of high-vacuum or gasfilled tube (s) in a rectifier circuit.
b. Filament transformers: Supply current to heat the filaments of vacuum or gas-filled tubes.
c. Plate-filament transformers: Combinations of $2 a$ and $2 b$.
d. Isolation transformers: Insulate or isolate two circuits, such as a grounded circuit from an ungrounded circuit.
e. Scott-transformers: Scott-connection utilizes two transformers to transmit power from two-phase to three-phase systems, or vice versa.
f. Auto-transformers: Provide increased or decreased voltange by means of a single winding suitably tapped for the primary and secondary circuits, part of the winding being common to both circuits.

## Major reacłor types

1. Reactors: Single-winding units that smooth current flow, provide d-c feed, or act as frequency-selective units lin suitable arrangement with capacitors).
a. Audio reactors: Single-winding units that supply plate current to a vacuum tube in parallel with the output circuit.

## Major reactor Iypes <br> continued

b. Wave-filter reactors: Function as filter unit components which aid in the acceptance or rejection of certain frequencies.
c. Filter reactors: Smooth the d-c output current in rectifier circuits.
d. Saturable reactors: Regulate voltage, current, or phase in conjunction with glow-discharge tubes of the thyratron type. They are also used as voltage-regulating devices with dry-type rectifiers.

## Temperafure, humidity, and pressure effects

A maximum ambient temperature of $40^{\circ} \mathrm{C}$ is usually assumed. Final operating temperatures with organic insulation (Class A), such as silk, cotton, or paper, are restricted to values less than $95^{\circ} \mathrm{C}$. When weight and space requirements dictate undersized iron cores and wire, with resultant higher temperature rise, inorganic insulation and cooling expedients may be used. Cooling expedients include: open-frame; semi-enclosed (coil-covered, core-exposed) design; and fully-enclosed design having compound or liquid-filled insulant and cooling by convection, or forced cooling by air blast.
Relative humidities from zero to 97 percent should be assumed so that coils and leads should be impregnated with moisture-resistant insulating coatings or, alternafively, cases should be sealed vacuum tight. Pressure variation, in addition to moisture and temperature changes, due to altitude from sealevel up to 7,000 feet lgreater for aircraftl may be encountered.

## General limitations

## Core maferial

a. For audio transformers and reactors: Core material should be such that core distortion is not greater than 0.75 percent at the lowest frequency. b. For power supply transformers: Core loss should be less than 0.82 watts per pound at 60 cps , for a fux density of 10,000 gauss. Filter reactors may have a core loss of 1.2 watts per pound at 60 cps , for 10,000 gauss.

## Terminal facilifies

a. All leads or winding ends: Must remain inside the case for hermetically sealed units.
b. Leads may terminate: In studs in a Bakelite board or bushing when voltage is less than 1000 volts peak. For higher voltages, Isolantite or wet process porcelain may be used.

## Profective gaps

Protective gaps are frequently used on.filter reactors or plate transformers in rectifier circuits delivering more than 1000 volis dc.

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## Design of power-supply transformers

The following may be used as a guide in the design of power supply transformers for receivers and small transmitters.

## Nomenclature

$A_{c}=a b=$ cross section area of core in square inches
$a=$ stack width in inches
$b=$ stack height in inches
$B_{\text {max }}=$ maximum core flux density in gauss. Usually assumed to be 10,000 gauss ( 64.5 kilolines per square inch) at 60 cps , or 12,000 gauss at 25 cps
$E_{p}=$ primary terminal yoltage
$E_{s}=$ secondary terminal voltage
$f=$ frequency in cycles per second
$h=$ minimum height of a coil section above core in inches
$h^{\prime}=$ maximum height of a coil section above core in inches
$K=$ stacking factor lusually $K=0.9$ )
MLT $=$ mean length of turn of a coil section in feet
$T_{p}=$ number of primary turns
$T_{s}=$ number of secondary turns
$V D_{p}=$ voltage drop due to primary resistance
$V D_{z}=$ voltage drop due to secondary resistance

## Design procedure

1. Determine secondary output volt-ampere requirements.
2. Calculate primary current based on a wattage 10 percent greater than the volt-amperes determined in (1). Use the given primary voltage $E_{p}$.
3. The core area is determined roughly by the formula

Core area $=\frac{\sqrt{\text { wattage }}}{5.58} \sqrt{\frac{60}{f}}$
Select a lamination (from a transformer manufacturer's lamination data bookh that will fit the transformer space requirements and provide the proper core area when stacked to a sufficient height.
4. Compute the number of primary turns $T_{p}=\frac{E_{p} \times 10^{8}}{28.6 f B_{\max } A_{c} K}$
5. Compute the number of secondary turns $T_{s}=\frac{E_{s}}{E_{p}} T_{p}$
6. Determine the wire sizes needed for primary and secondary on the basis of on optimum current density of 1000 amperes per square inch, using Table I and the currents carried by the primary and secondary. Greater or smaller densities may be used as required. For very small transformers, densities up to 2500 amperes per square inch are sometimes used.

## IRON-CORE TRANSFORMERS AND REACTORS

Design of power-supply fransformers cantinued
7. Calculate the number of turns per layer that can be placed in the lamination window space, deducting margin space from the window length.
8. From this value, calculate the total number of primary and secondary layers needed.
9. Calculate the total wire height, using the wire diameter and the number of layers.
10. Determine the total insulation thickness required between wire layers (from Table 1), and under and over coil sections.
11. Add the results of 19 ) and (10) and multiply the figure obtained by $10 / 9$ to allow for bulge in winding wire and wrapping insulation. Revise the design, as necessary, to make this over-all thickness figure (coil build) slightly less than the lamination window width.
12. Calculate the mean length of turns for the primary and for each secondary coil section

MLT $=\frac{2 a+2 b+2 \pi \frac{\left(h^{\prime}+h l\right.}{2}}{12}$
13. Calculate the total wire length in feet of each primary and secondary coil by multiplying the MLT value of the coil by the corresponding total number of turns in that coil.
14. The resistance of each coil is obtained by multiplying the total wire length obtained above by the resistance per foot.
15. Calculate the voltage drop in each primary and secondary from the calculated resistance and the current flow.
16. Compensate for the voltage drop in the primary and in each secondary by determining the corrected number of turns
(corrected $T_{p}$ ) $=\frac{E_{p}-V D_{p}}{E_{p}} \times$ (original $T_{p}$ )
$\left(\right.$ corrected $\left.T_{s}\right)=\frac{E_{s}+V D_{s}}{E_{s}} \times\left(\right.$ original $\left.T_{s}\right)$
17. Revise the number of layers of each winding according to the corrected number of turns.
18. Calculate the copper loss in both primary and secondary windings from the resistance of each coil times the square of the current flowing in it.

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## Design of power-supply fransformers continued

19. Calculate the core loss from the weight (in pounds) of the core used and the core loss per pound obtained from the core loss curve given by the manufacturer for the iron used.
20. The efficiency of the transformer is

Percent efficiency $=\frac{\text { wattage output } \times 100}{\text { wattage output }+ \text { core loss }+ \text { copper loss }}$

Table 1-Round enameled copper wire

| $\begin{aligned} & \text { AWG } \\ & \text { (B\&S) } \end{aligned}$ | diametor inches |  | current capacity amperes* | ohms per 1000 ft at $50^{\circ} \mathrm{C}$ | coll margin inches | interlayer insulation $\dagger$ inches |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 10 | 0.1039 | 9 | 8.2 | 1.12 | 0.25 | 0.010 |
| 11 | 0.0927 | 10 | 6.5 | 1.41 | 0.25 | 0.010 |
| 12 | 0.0827 | 11 | 5.1 | 1.78 | 0.25 | 0.010 |
| 13 | 0.0738 | 12 | 4.1 | 2.24 | 0.25 | 0.010 |
| 14 | 0.0659 | 13 | 3.2 | 2.82 | . 0.25 | 0.010 |
| 15 | 0.0588 | 14 | 2.6 | 3.56 | 0.188 | 0.010 |
| 16 | 0.0524 | 16 | 2.0 | 4.49 | 0.188 | 0.010 |
| 17 | 0.0469 | 19 | 1.61 | 5.66 | 0.188 | 0.010 |
| 18 | 0.0418 | 21 | 1.28 | 7.14 | 0.125 | 0.005 |
| 19 | 0.0374 | 24 | 1.01 | 9.0 | 0.125 | 0.005 |
| 20 | 0.0334 | 26 | 0.80 | 11.4 | 0.125 | 0.005 |
| 21 | 0.0299 | 30 | 0.64 | 14.3 | 0.125 | 0.005 |
| 22 | 0.0266 | 34 | 0.50 | 18.1 | 0.125 | 0.003 |
| 23 | 0.0238 | 39 | 0.40 | 22.8 | 0.125 | 0.003 |
| 24 | 0.0213 | 43 | 0.32 | 28.7 | 0.125 | 0.003 |
| 25 | 0.0190 | 48 | 0.25 | 36.2 | 0.125 | 0.002 |
| 26 | 0.0169 | 54 | 0.20 | 45.6 | 0.125 | 0.002 |
| 27 | 0.0152 | 59 | 0.158 | 57.5 | 0.125 | 0.002 |
| 28 | 0.0135 | 68 | 0.126 | 72.6 | 0.125 | 0.002 |
| 29 | 0.0122 | 74 | 0.100 | 91 | 0.125 | 0.002 |
| 30 | 0.0108 | 84 | 0.079 | 115 | 0.125 | 0.0015 |
| 31 | 0.0097 | 94 | 0.063 | 146 | 0.125 | 0.0015 |
| 32 | 0.0088 | 104 | 0.050 | 183 | 0.094 | 0.0015 |
| 33 | 0.0078 | 117 | 0.039 | 231 | 0.094 | 0.0015 |
| 34 | 0.0069 | 131 | 0.031 | 292 | 0.094 | 0.001 |
| 35 | 0.0061 | 146 | 0.025 | 368 | 0.094 | 0.001 |
| 36 | 0.0055 | 162 | 0.0196 | 464 | 0.094 | 0.001 |
| 37 | 0.0049 | 183 | 0.0156 | 585 | 0.094 | 0.001 |
| 38 | 0.0044 | 204 | 0.0124 | 737 | 0.063 | 0.001 |
| 39 | 0.0038 | 227 | 0.0098 | 930 | 0.063 | 0.00075 |
| 40 | 0.0034 | 261 | 0.0078 | 1173 | 0.063 | 0.00075 |

[^9]
## V Vacuum fubes

## Nomenclafure:

$e_{c}=$ instantaneous total grid voltage
$e_{b}=$ instantaneous total plate voltage
$i_{c}=$ instantaneous total grid current
$i_{b}=$ instantaneous total plate current
$E_{c}=$ average value of grid voltage
$E_{b}=$ average or quiescent value of plate voltage
$I_{c}=$ average or quiescent value of grid current
$I_{b}=$ average or quiescent value of plate current
$e_{g}=$ instantaneous value of varying component of grid voltage
$e_{p}=$ instantaneous value of varying component of plate voltage
$i_{g}=$ instantaneous value of varying component of grid current
$i_{p}=$ instantaneous value of varying component of plate current
$E_{g}=$ effective or maximum value of varying component of grid voltage
$E_{p}=$ effective or maximum value of varying component of plate voltage
$I_{g}=$ effective or maximum value of varying component of grid current
$I_{p}=$ effective or maximum value of varying component of plate current
$I_{f}=$ filament or heater current
$I_{s}=$ total electron emission (from cathode)
$r_{2}=$ external plate load resistance
$\mathrm{C}_{a p}=$ grid-plate direct capacitance
$C_{0 k}=$ grid-cathode direct capacitance
$C_{p k}=$ plate-cathode direct capacitance
$\theta_{p}=$ plate current conduction angle
$r_{p}=$ variational (a-c) plate resistance
$R_{p b}=$ total (d-c) plate resistance
Note: In the following text, the superscript $M$ indicates the use of the maximum or peak value of the varying component, i.e., ${ }^{M} E_{p}=$ maximum or peak value of the alternating component of the plate voltage.

* From IRE standard symbols IElectronics Slandards, 1938


## Coefficients

Amplification factor $\mu$ : Ratio of incremental plate voltage to controlelectrode voltage change at a fixed plate current with constant voltage on other electrodes.

$$
\left.\begin{array}{c}
\mu=\left[\frac{\delta e_{b}}{\delta e_{c 1}}\right]_{I_{b}} \\
E_{d 2} \ldots \ldots . \ldots . . . E_{c n}
\end{array}\right\} \text { constant }
$$

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Coefftcients continued
Transconductance $\boldsymbol{s}_{m}$ : Ratio of incremental plate current to control-electrode voltage change at constant voltage on other electrodes.

$$
\begin{gathered}
s_{m}=\left[\frac{\delta i_{b}}{\delta \mathrm{e}_{c 1}}\right] E_{b}, E_{c 2} \ldots \ldots . . . E_{c n} \text { constant } \\
r_{l}=0
\end{gathered}
$$

When electrodes are plate and control grid, the ratio is the mutual conductance $g_{m}$ of the tube.
$g_{m}=\frac{\mu}{r_{p}}$
Variational (a-c) plate resistance $r_{p}$ : Ratio of incremental plate voltage to current change at constant voltage on other electrodes.

$$
\begin{gathered}
\boldsymbol{r}_{p}=\left[\frac{\delta \mathrm{e}_{b}}{\delta i_{b}}\right]_{E_{c 1-\ldots \ldots} \ldots \ldots-\ldots} E_{c n} \text { constant } \\
\boldsymbol{r}_{l}=0
\end{gathered}
$$

Total (d-c) plate resistance $R_{p}$ : Ratio of total plate voltage to current for constant voltage on other electrodes.

$$
\begin{gathered}
R_{p}=\left[\frac{e_{b}}{i_{b}}\right]_{E_{c l} \ldots \ldots \ldots . . . . . . E_{c n} \text { constant }} \\
r_{t}=0
\end{gathered}
$$

## Terminology

Control grid: Electrode to which plate-current-controlling signal voltage is applied.
Space-charge grid: Electrode, usually biased to constant positive voltage, placed adjacent to cathode to reduce current-limiting effect of space charge. Suppressor grid: Grid placed between two electrodes to suppress the effect of secondary electrons.
Screen grid: Grid placed between anode and control grid to reduce the capacitive coupling between them.
Primary emission: Thermionic emission of electrons from a surface.
Secondary emission: Usually of electrons, from a surface by direct impact not thermal action, of electronic or ionic bombardment.
Total emission $I_{s}$ : Maximum (saturated, temperature-limited) value of electron current which may be drawn from a cathode. Available total emission is that peak value of current which may safely be drawn.

## Terminology continued

Transfer characteristic: Relation, usually graphical, between voltage on one electrode and current to another, voltages on all other electrodes remaining constant.

Electrode characteristic: Relation, usually graphical, between the voltage on, and current to, a tube electrode, all other electrode voltages remaining constant.

Composite-diode lines: Relation, usually two curves, of the currents flowing to the control grid and the anode of a triode as a function of the equal voltage applied to them Igrid-plate tied).
Critical grid voltage: Instantaneous value of grid voltage (with respect to cathodel at which anode current conduction is initiated through a gas tube.
Constant current characteristics: Relation, usually graphical, between the voltages on two electrodes, for constant specified current to one of them and constant voltages on all other electrodes.

## Formulas

For unipoiential cathode and negligible seturation of cethode emission

| function | paroilel plane cothode and plate | cylindricol cothode ond plate |
| :---: | :---: | :---: |
| Diode plate current (amperes) | $\mathrm{G}_{1} \mathrm{e}^{\frac{3}{2}}$ | $\mathrm{G}_{1}{ }^{\text {e }}{ }^{\frac{3}{2}}$ |
| Triode plate current (amperes) | $\mathrm{G}_{2}\left(\frac{e_{b}+\mu e_{c}}{1+\mu}\right)^{\frac{3}{2}}$ | $\mathrm{G}_{2}\left(\frac{e_{b}+\mu \mathrm{e}_{0}}{1+\mu}\right)^{\frac{8}{2}}$ |
| Diode perveance $\mathrm{G}_{1}$ | $2.3 \times 10^{-6} \frac{A_{b}}{d_{b}^{2}}$ | $2.3 \times 10^{-6} \frac{A_{b}}{\beta^{2} r b^{2}}$ |
| Triode perveance $\mathrm{G}_{2}$ | $2.3 \times 10^{-6} \frac{A_{b}}{d_{b} d_{a}}$ | $2.3 \times 10^{-6} \frac{A_{b}}{\beta^{2} r_{b} r_{c}}$ |
| Amplification factor $\mu$ | $\frac{2.7 d_{c}\left(\frac{d_{b}}{d_{c}}-1\right)}{\rho \log \frac{\rho}{2 \pi r_{b}}}$ | $\frac{2 \pi d_{o}}{\rho} \frac{\log \frac{d_{b}}{d_{c}}}{\log \frac{\rho}{2 \pi r_{\theta}}}$ |
| Mutual conductance $\mathrm{g}_{\mathrm{m}}$ | $\begin{aligned} & 1.5 \mathrm{G}_{2} \frac{\mu}{\mu+1} \sqrt{\mathrm{e}_{\theta}^{\prime}} \\ & \mathrm{e}_{\theta}^{\prime}=\frac{E_{b}+\mu E_{c}}{1+\mu} \end{aligned}$ | $\begin{aligned} & 1.5 \mathrm{G}_{2} \frac{\mu}{\mu+1} \sqrt{\mathrm{e}_{\theta}^{\prime}} \\ & \mathrm{e}_{\theta}^{\prime}=\frac{E_{b}+\mu E_{c}}{1+\mu} \end{aligned}$ |

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Formulas continued
where
$A_{b}=$ effective anode area in square centimeters
$J_{b}=$ anode-cathode distance in centimeters
$d_{c}=$ grid-cathode distance in centimeters
$\beta=$ geometrical constant, a function of ratio of anode to cathode radius;

$$
\beta^{2} \cong 1 \text { for } \frac{r_{b}}{r_{k}}>10 \text { (see curve Fig. } 1 \text { ) }
$$

$\rho=$ pitch of grid wires in centimeters
$r_{g}=$ grid wire radius in centimeters
$r_{b}=$ anode radius in centimeters
$r_{k}=$ cathode radius in centimeters
$r_{c}=$ grid radius in centimeters
Note: These formulas ore based on theoretical considarations and do nat provide accurate results; for practical structures, however, they give a fair ideo of the relationship between the tube geometry and the constonts of the tube.


Fig. 1-Values of $\beta^{2}$ for values of $\frac{r_{b}}{r_{k}}<10$.

## Performance limitations

Tube performance limitation factors include electrode dissipation, filament emission, and the transit time of electrons in the active part of the tube. For a given tube, the ultimate limitation may be any one or a combination of these factors.

## Electrode dissipafion dafa

Tube performance is limited by electrode dissipation. In turn, tube dissipation is limited by the maximum safe operating temperatures of the glass-to-metal seals lapproximately $200^{\circ} \mathrm{Cl}$, glass envelope, and tube electrodes. Thus excessive dissipation may result in breakage, loss of vacuum, and destruction of the tube.

Typical operating data for common types of caoling are roughly

| type | average <br> cooling surface <br> femperature ${ }^{\circ} \mathbf{c}$ | specifte <br> dissipation <br> watts/em <br> cooling surface | cooling medium |
| :--- | :---: | :---: | :---: |
| supply |  |  |  |

The operating temperature of radiation-cooled anodes for a given dissipation is determined by the relative total emissivity of the anode material. Thus, graphite electrodes which approach black-body radiation conditions operate at the lower temperature range indicated, while untreated tantalum and molybdenum work at relatively high temperatures. In computing coolingmedium flow, a minimum velocity sufficient to insure turbulent flow at the dissipating surface must be maintained. In the case of water and forced-air cooled tubes, the figures above apply to clean cooling surfaces, and may be reduced to a small fraction of these values by heat-insulating coatings such as mineral scale or dust. Cooling surfaces should, thus, be closely observed and cleaned periodically.

Dissipation and temperature rise of cooling water
$K W=0.264 Q\left(T_{2}-T_{1}\right)$
where $K W=$ power in kilowatts, $\mathrm{Q}=$ flow in gallons per minute, $T_{2}$ and $T_{1}=$ outlet and inlet temperatures in degrees centrigrade. An alternate formula is
$K W=\frac{\text { liters per minute }\left(T_{2}-T_{1}\right)}{14.3}$
or $K W=$ liters per minute when the temperature rise is a reasonable figure, namely $14.3^{\circ} \mathrm{C}$.

Air flow and temperature rise
$Q=5.92\left(T_{1}+273\right) \frac{P}{T_{2}-T_{1}}$
where $Q=$ air flow in cubic feet per minute.

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## Filament characteristics

The sum of the instantaneous peak currents drawn by all of the electrodes must be within the available total emission of the filament. This emission is determined by the filament material, area, and temperature.

Typical data on the three types of floment most used ore

| type | officiency mo/wat | speciffe emission $I_{s}$ $\mathrm{amp} / \mathrm{cm}^{2}$ | wath/cm ${ }^{2}$ | operafing femperafure Kelvin | ratlo hat-fo-cold resistance |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Pure tungsten (W) | 5-10 | 0.25-0.7 | 70-84 | 2500-2600 | 14:1 |
| Thoriated tungsten (ThW) | 40-100 | 0.5-3 | 26-28 | 1950-2000 | 10:1 |
| Oxide coated ( BaCoSr ) | 50-150 | 0.5-2.5 | 5-10 | 1100-1250 | 2.5 to 5.5:1 |

In the cases of thoriated-tungsten and oxide-coated filament tubes, the emission data vary widely between tubes around the approximate range indicated in the table. The figures for specific emission refer to the peak or saturated value which is usually two or more times the total available value for these filaments. Instantaneous peak current values drawn during operation should never exceed the published available emission figure for the given tube.
Thoriated-tungsten and oxide-coated type filaments should be operated close to the specified published voltage. Deviation from these values will result in rapid destruction of the cathode surface.
In the case of pure tungsten, the filament may be operated over a considerable temperature range. It should be borne in mind, however, that the total filament-emission current available varies closely as the seventh power of the filament voltage. Likewise, the expected filament life is critically dependent on the operating temperature. The relationship between filament voltage and life is shown by Fig. 2. It will be seen that an increase of 5 percent above rated filament voltage reduces the life expectancy by 50 percent. Where the dull normal emission is not required, a corresponding increase in life may be secured by operating a pure tungsten filament below rated filament voltage.
From the above tabulated values of hot-to-cold resistance, it may be seen that a very high heating current may be drawn by a cold filament, particularly one of the tungsten type. In order to avoid destruction by mechanical stresses which are proportional to $I^{2}$, it is imperative to limit the current to a safe value, say, 150 percent of normal hot value for large tubes and 250 percent for medium types. This may be accomplished by resistance and time-delay relays, high-reactance transformers, or regulators.


Fig. 2-Effect of change in flament voltage on the life and emission of bright fungsten filament (based on $2575^{\circ} \mathrm{K}$ normal temperature).

## Filament characteristics

 continuedIn the case where a severe overload has temporarily impaired the emission of a thoriated-tungsten filament, the activity can sometimes be restored by operating the tube with filament voltage only in accordance with one of the following schedules:

1. At normal filament voltage for several hours or overnight. Or, if the emission fails to respond.
2. At 30 percent above normal for 10 minutes, then at normal for 20 to 30 minutes. Or, in extreme cases when 1 and 2 have failed to give results and at the risk of burning out the filament.
3. At 75 percent above normal for 30 seconds followed by schedule 2 .

## Ulira-high-frequency fubes

Tubes for u-h-f application differ widely in design among themselves and from those for lower frequency. The theory of their operation and the principles of their design have not been fully expounded, and great progress in this field still lies ahead.
Ultra-high-frequency tubes may be classified according to principle of operation as follows:

1. Negative-grid tubes
2. Positive-grid tubes
3. Velocity-modulated tubes
4. Magnetrons
1., Negative-grid tubes: Effectiveness of negative-grid tubes at ultra-highfrequencies is limited by two factors
a. difficulty of designing the circuit associated with the tube
b. effect of electron inertia.
a. Design of u-h-f circuit associated with negative-grid tubes: The circuit must be tunable at the operating frequency. This leads to the use of transmission lines as associated circuits of the parallel or coaxial type. The tubes themselves are constructed so as to be part of the associated transmission line.

Lines in some cases are tuned on harmonic modes, thus making possible the use of larger circuit elements.
Circuit impedance must match the optimum loading impedance of the tube, a requirement difficult to satisfy inasmuch as the capacitive reactances are very small and u-h-f losses are important in both conductors and insulators. Difficulty in obtaining the proper $Q$ of the circuit is increased with frequency.

## Ulira-high-frequency tubes

b. Effect of electron inertia: The theory of electron inertia effect in receiving tubes has been formulated by Llewelyn, but no comparable, complete theory is now available for transmitting tubes. In both cases the time of light of an electron from cathode to anode must be a small fraction of the oscillating period. When this period is so short as to be of the same order of magnitude as the transit time, receiving tubes cease to amplify and transmitting tubes cease to oscillate.
Small tubes with close spacing between electrodes have been built that can be operated up to about 3000 megacycles.
To compare results obtained with different tubes and circuits pertaining to a family ruled by the law of similitude, it is useful to know that dimensionless magnitudes, such as efficiency, or signal-noise ratio, are the same when the dimensionless parameter
$\phi=\frac{f \times d}{\sqrt{V}}$ remains constant
where
$f=$ frequency in megacycles
$d=$ cathode-to-anode distance in centimeters
$V=$ anode voltage in volts.
Transit-time effect appears when $\phi$ becomes greater than 1. Spacing between electrodes of u-h-f tubes then must be small, and operation at high voltage is necessary. In addition cathodes must be designed for high current density operation.
2. Positive-grid tubes: Utilize an oscillating space charge produced by acceleration of electrons through the positive grid toward a negative reflecting anode. This principle has been used for generating waves down to lengths of one centimeter. Low power output and low efficiency have hitherto limited their wide application.
3. Velocity-modulated fubes: Utilize the acceleration and retarding action of an alternating electron voltage on an electron beam to vary the velocity in the beam. After passage of the beam through a field-free drift space, the beam arrives with variations of space-charge density. In passing through the opening of a resonant cavity at this point, the variation of the beam density induces a current in the external circuit. Several types of amplifiers and oscillators employ this principle of operation; some, such as the reflex Klystron, have a single cavity. While a theoretical efficiency of about 50 percent may thereby be achieved, the actual efficiency in the frequency range around 10 centimeters is only a few percent.
4. Magnetrons: May be considered as another form of velocity-modulated tube in which the electron stream instead of being accelerated linearly is

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## Ulitra-high-frequency fubes continued

given a circular trajectory by means of a transverse magnetic field. Energy from this beam is not lost directly to an acceleration electrode at d-c potential as in the linear case and accordingly a higher operating efficiency may be obtained. Usually acceleration and retardation of the rotary beam is accomplished by one or more pairs of electrodes associated with one or more resonant circuits.

Wavelengths down to a centimeter are produced by the so-called first order ( $n=1$ ) oscillations generated in a magnetron having a single pair of plates. Relatively low efficiency and power output are obtained in this mode of operation. Design formulas relating dimensions, d-c anode voltage, magnetic field strength, and output frequency for this case are obtained from the basic relation for electron angular velocity

$$
\begin{aligned}
\omega_{m} & =H \frac{e}{m} \\
\lambda & =\frac{10,700}{H} \\
E_{b} & =0.022 r_{b}^{2}\left[1-\left(\frac{r_{k}}{r_{b}}\right)^{2}\right]^{2} H^{2}
\end{aligned}
$$

where

$$
H=\text { field intensity in gauss }
$$

$E_{b}=\mathrm{d}-\mathrm{c}$ accelerating voltage in volts
$\lambda=$ generated wavelength in centimeters
$r_{b}=$ anode radius in centimeters
$r_{k}=$ cathode radius in centimeters
Higher order oscillations of the magnetron may be obtained at high outputs and efficiencies exceeding that of the linear velocity-modulated tubes.

## Cathode-ray fubes

## Electrodes*

Control electrode (modulating electrode, grid, or grid No. 1): Is operated at a negative potential with respect to the cathode in conventional cathoderay tubes. The negative potential controls the beam current and, therefore, the trace brightness.

[^10]
## Cathode-ray fubes

continued
Screen grid (grid No. 2): Is not utilized in all cathode-ray tube designs. Its introduction makes the control characteristic independent of the accelerating potential when operated at fixed positive potential. In electrostaticfocus, it makes the screen current (beam current to fluorescent screen) substantially independent of the focusing electrode voltage over the focus region. In some tube designs, it is used to change the control characteristic dynamically by application of varying potential.


Fig. 3-fiectrede arrangement of iypical electrastatic foeus and deflection cathode-ray tube. A heater. B cathode. C centrol electrode. D screen grid or preaceelerator. E focusing electrode. F accelerating electrode. G defection plote peir. H defection plate poir. J conductive coating connected'to aceelerating electrode. K Intensifer electrode terminal. I Intenstifer electrode (conductive coating on glass). M fuorescent seraen.

Focusing electrode (anode No. 1): is used in electrostatic-focus cathode-ray tubes and operates at a positive potential," adjustable to focus the spot.
Accelerating electrode (anode No. 2 or anode): In usual usage, the second anode is the last electrode, prior to deflection, which produces acceleration. The second anode potential is the potential of the electron beam in the deflection region.
Intensifier electrode (post-accelerating electrode, anode No. 3): Provides acceleration after deflection.
Preaccelerating electrode: In common usage, is an electrode like a screen grid or second grid, but connected to the accelerating electrode internally. It makes the screen current (beam current to fluorescent screen) substantially independent of the focusing electrode voltage over the focus region.
Deflection plates (deflection electrodes): Conventional cathode-ray tubes have two pairs of deflection plates at right angles to each other. The electric field between the plates of a pair causes deflection of the beam and, therefore, displacement of spot, in a direction perpendicular to plates of a pair.

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## Cathode-ray fubes continued

## Characteristics

Cutoff voltage ( $E_{c o}$ ): Negative grid potential at which screen current becomes zero (as indicated by visual extinction of a focused undeflected spot), or some specified low value. It varies directly with the accelerating electrode potential except in tubes with independently connected screen grids where it varies approximately as the screen-grid potential, the accelerating electrode potential having a second order effect $\left(E_{\infty}\right.$ increases slightly with accelerating electrode potentiall. $E_{c o}$ is independent of intensifier electrode potential.

Control characteristic (modulation characteristic): is a curve of beam current versus grid potential. It is often expressed in terms of grid drive lgrid potential above cutoff) rather than actual grid potential. This method of expressing it has the advantage that the characteristic then varies less with accelerating potential and with individual tubes of a given design.

Focusing voltage: In electrostatic focus tubes, the focusing electrode voltage at which the spot comes to a focus varies directly with accelerating electrode voltage in most tube designs and is substantially independent of the intensifier electrode potential.

Focusing current or focusing ampere turns: Applies to magnetic-focus cathode-ray tubes and is usually expressed in terms of a definite focus coil in a definite location on the tube. While more than one value of current will focus, the best focus is obtained with the minimum value, i.e., the one ordinarily specified. The focusing current lor ampere turnsl increases with accelerating potential.

Deflection factor (for electrostatic-deflection tubes): Is defined as the voltage required between a pair of deflection plates to produce unit deflection of the spot, and is usually expressed in d-c volts per inch of displacement. It varies directly with the accelerating potential in intensifier-type tubes so long as the ratio of the intensifier potential to accelerating-electrode potential lall potentials with respect to cathodel is constant. The application of twice the accelerating electrode potential to the intensifier electrode increases the deflection factor 15 percent to 30 percent above the value with the accelerating electrode and intensifier electrode at the same potential, depending on the tube design.

Deflection factor (for magnetic deflection tubes): Usually expressed in terms of a definite deflection yoke in a definite location on the tube, in amperes or milliamperes per inch of spot deflection, it varies as the square root of the accelerating electrode potential.

Cathode-ray fubes continued

Deflection sensitivity: Is the reciprocal of the deflection factor. Usually, however, it is expressed in millimeters per volt for electrostatic deflection tubes.

Spot size: Must be expressed in terms of a defined method of measurement since spot edges are not usually sharp. When the accelerating potential is varied and the screen current maintained constant, the spot size usually decreases with increasing accelerating potential. If the brightness is held constant while varying the accelerating potential, the spot size decreases even more with increasing accelerating potential.
Brightness: Increases with beam current and with accelerating potential. At constant screen current, it usually increases with accelerating potential at a rate between the first and second power of the accelerating potential, approaching a maximum depending upon the screen material.

## Application notes

Grid voltage: To permit variation of brightness over the entire range, the grid voltage, should be variable from the maximum specified cutoff bias of a cathode-ray tube to zero. Allowance should be made for a-c grid voltages if they are applied, and for potential drops which may occur in d-c gridreturn circuits due to allowable grid leakage.

Focusing electrode voltage source (electrostatic-focus tubes): Bleeder design should be such as to cover the range of focus voltage over which tubes are permitted to vary by specifications, both at the value of focusing-electrode current that may be encountered in operation, and at cutoff (zero focusing-electrode currentl.

Deflection-plate potentials (electrostatic-deflection tubes): To avoid defocusing of the spot, the instantaneous average potential of the plates of each deflection-plate pair should always be the same as that of the accelerating electrode.

Magnetic shielding: Magnetic shielding is necessary if it is desired to eliminate magnetic effects on the beam. The earth's and other magnetic fields may shift the beam considerably.

## Approximate formulas

Electrostatic deflection: Is proportional to deflection voltage, inversely proportional to accelerating voltage, and at right angles to the plane of the plates and toward the more positive plate. For deflection electrode structures using straight parallel deflection plates

## Cathode-ray tubes continued

$D=\frac{E_{d} L l}{2 E_{a} A}$
$D=$ deflection
$E_{d}=$ deflection voltage
$E_{a}=$ accelerating voltage
$A=$ separation of plates
$I=$ length of plates
$L=$ length from center of plates to screen
$D, A, l, L$ are all in the same units
Electromagnetic deflection: Is proportional to flux or current in coil, inversely proportional to the square root of the accelerating voltage, and at right angles to the direction of the field

$$
\begin{aligned}
D= & \frac{0.3 L I H}{\sqrt{E_{a}}} \\
D= & \text { deflection in centimeters } \\
L= & \text { length in centimeters between screen } \\
& \text { and point where beam enters deflect- } \\
& \text { ing field } \\
I= & \text { length of deflection field in centimeters } \\
H= & \text { flux density in gauss } \\
E_{a}= & \text { accelerating voltage } \\
N I= & \text { deflecting coil ampere turns }
\end{aligned}
$$



Deflection sensitivity: Is linear up to frequency where phase of deflecting voltage begins to reverse before electron has reached end of deflecting field. Beyond this frequency, sensitivity drops off reaching zero and then passing through a series of maxima and minima as $n=1,2,3 \ldots$ Each succeeding maximum is of smaller magnitude

$$
\begin{aligned}
D_{z e r o} & =n \lambda\binom{v}{c} . \\
D_{m a x} & =12 n-11\left(\frac{\lambda}{2}\right)\left(\frac{v}{c}\right) \\
D & =\text { deflection } \\
v & =\text { electron velocity } \\
c & \left.=\text { speed of light } 13 \times 10^{10} \mathrm{~cm} / \mathrm{sec}\right)
\end{aligned}
$$

Electron velocity: For accelerating voltages up to 10,000 $v\left(\mathrm{~km}\right.$ per sec) $=593 \sqrt{E_{a}}$

## Cathode-ray fubes continued

Beyond 10,000 volts, apply Einstein's correction for the increase in mass of the electron.

## Earth's magnetic field:

Maximum 0.4 gauss horizontal (Philippine Islands)
0.6 gauss vertical (Canada)

City of New York 0.17 gauss horizontal; 0.59 gauss vertical
Magnetic focusing: There is more than one value of current that will focus. Best focus is at minimum value.
For an everage coil
$I N=220 \sqrt{\frac{V_{o d}}{f}}$
IN = ampere turns
$V_{0}=k v$ accelerating voltage
$d=$ mean diameter of coil
$f=$ focal length
$d$ and $f$ are in the same units
A well-designed, shielded coil will require fewer ampere turns.
Example of good shield design

$$
x=\frac{d_{1}}{20}
$$



I November 1945


## Transmilting

| trlades |  | tefrodes | twin tefrodes | penfodes | puise medulation | magnetrons |  | vocuum | $\begin{aligned} & \text { rectiflers } \\ & \text { gas } \end{aligned}$ |  | grid control | elipper | ges switching <br> ATR \| TR |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 2C26A | 811 | 807 | 815 | 2 E 22 | 3021A | 2130-34 | 4J31-35 | 122 | 3828 |  | 2021 | $3 \mathrm{B26}$ | 1835 | 1823 |
| 2C39 | 826 | 813 | 8298 | 2 E 25 | 3 C 45 | 2141 | 4J36-42 | 2×2A | 4826 |  | C5B | 4831 | 1837 | 1824 |
| 2 C 43 | 862A | 814 | 832A | 4 E 27 | 3 E 29 | 2 J 42 | 4143-44 | 3824 W | 4835 |  | 6D4 | 719A | 1844 | 1827 |
| 3 C 28 | 880 | $8278 \dagger$ |  | 803 | 4C35 | 2148 | 4150 | 5R4GY | 5821 |  | 393A |  | 1851 | 1832 |
| CV92 $\langle\mathrm{Br}\rangle \dagger$ | 889R-A | 1625 |  | 837 | 5 C 22 | 2149 | 4151 | 371 B | 6 C |  | 394A |  | 18.52 | 1850 |
| 100TH | 1626 |  |  |  | 6 C 21 | 2150 | 4152 | 836 | 83 |  | 884 |  | 1853 | 1855 |
| 250TH | 8025A |  |  |  | 715C $\dagger$ | 2151 | 5126 | 1616 | $857 B$ |  | 2050 |  | 1856 | IB58 |
| 304TH |  |  |  |  |  | 2153 | 5129 | 8016 | 866 A |  |  |  | 1857 |  |
| 450 TH |  |  |  |  |  | 2155-56 | 5130 | 8020 | 8698 |  |  |  |  |  |
| 527 |  |  |  |  |  | 2158 | 5131 |  | 872A |  |  |  | pre-TR | medulator: |
|  |  |  |  |  |  | 2160 | 5132 |  | 1c06 |  |  |  | 1838 | 1822 |
|  |  |  |  |  |  | 2161 A-62A |  |  |  |  |  |  | 1854 | 1841 |
|  |  |  |  |  |  |  |  |  |  |  |  |  |  | 1842 |

Specification hos been issued a counterport of the prototype indicated by suffix lefter(i) GT, GT/G, Y, W, A, B, etc. moy be used.

[^12]
## - Vacuum fube amplifiers

## Classification

It is common practice to differentiate between types of vacuum tube circuits, particularly amplifiers, on the basis of the operating regime of the tube.

Class A: Grid bias and alternating grid voltages such that plate current flows continuously throughout electrical cycle $1 \theta_{p}=360$ degrees).

Class $A B$ : Grid bias and alternating grid voltages such that plate current flows appreciably more than half but less than entire electrical cycle $\left(360^{\circ}>\theta_{p}>180^{\circ}\right.$ ).

Class B: Grid bias close to cut-off such that plate current flows only during approximately half of electrical cycle $\left(\theta_{p} \cong 180^{\circ}\right)$.

Class C: Grid bias appreciably greater than cut-off so that plate current flows for appreciably less than half of electrical cycle $\left(\theta_{p}<180^{\circ}\right)$.

A further classification between circuits in which positive grid current is conducted during some portion of the cycle, and those in which it is not, is denoted by subscripts 2 and 1 , respectively. Thus a class $A B_{2}$ amplifier operates with a positive swing of the alternating grid voltage such that positive electronic current is conducted, and accordingly in-phqse power is required to drive the tube.

## General design

For quickly estimating the performance of a tube from catalog data, or for predicting the characteristics needed for a given application, the ratios given in Table I may be used.

## Table 1-Typical amplifer operating data

Maximum signal conditions-per fube

| function | class A | $\begin{gathered} \text { class B } \\ a-f(p-p) \end{gathered}$ | $\begin{gathered} \text { class B } \\ \text { rof } \end{gathered}$ | $\underset{r-f}{c}$ |
| :---: | :---: | :---: | :---: | :---: |
| Plote efficiency $\eta$ \% | 20-30 | 35-65 | 60-70 | 65-85 |
| Peak instantaneous to d-c plate current ratio $\mathrm{Mib}_{\text {ib }} / \mathrm{I}_{b}$ | 1.5-2 | 3.1 | 3.1 | 3.1-4.5 |
| RMS alternating to d-c plate current ratio $I_{p} / I_{b}$ | 0.5-0.7 | 1.1 | 1.1 | 1.1-1.2 |
| RMS olternating to d-c plate voltoge ratio $E_{p} / E_{b}$ | 0.3-0.5 | 0.5-0.6 | 0.5-0.6 | 0.5-0.6 |
| D-C to peak instantaneous grid current $I_{c} / M_{i_{e}}$ |  | 0.25-0.1 | 0.25-0.1 | 0.15-0.1 |

## General design continued

Table I gives correlating data for typical operation of tubes in the various amplifier classifications. From this table, knowing the maximum ratings of a tube, the maximum power output, currents, voltages, and corresponding load impedance may be estimated. Thus, taking for example, a type F-124-A water-cooled transmitting tube as a class $C$ radio-frequency power amplifier and oscillator-the constant-current characteristics of which are shown in Fig. 1-published maximum ratings are as follows:
D.C plate voltage $E_{b}=20,000$ volts
D.C grid voltage $E_{c}=3,000$ volts

D-C plate current $I_{b}=7$ amperes
R-F grid current $\quad I_{o}=50$ amperes
Plate input $\quad P_{i}=135,000$ watts
Plate dissipation $P_{p}=40,000$ watts
Maximum conditions may be estimated as follows:
For $\eta=75 \% \quad P_{i}=135,000$ watts $\quad E_{b}=20,000$ volts
Power output $P_{0}=\eta P_{i}=100,000$ watts
Average d-c plate current $I_{b}=P_{i} / E_{b}=6.7$ amperes
From tabulated typical ratio ${ }^{\mathrm{M}} \mathrm{i}_{\mathrm{b}} / I_{b}=4$, instantaneous peak plate current ${ }^{M_{i b}}=4 l_{b}=27$ amperes
The rms alternating plate current component, taking ratio $I_{p} / I_{b}=1.2, I_{p}=$ $1.2 I_{b}=8$ amperes

The rms value of the alternating plate voltage component from the ratio $E_{p} / E_{b}=0.6$ is $E_{p}=0.6 E_{b}=12,000$ volts.

The approximate operating load resistance $r_{l}$ is now found from
$r_{l}=\frac{E_{p}}{I_{p}}=1500$ ohms.
An estimate of the grid drive power required may be obtained by reference to the constant current characteristics of the tube and determination of the peak instantaneous positive grid current ${ }^{M_{i}}$ and the corresponding instantaneous total grid voltage ${ }^{\mathrm{M}} \mathrm{e}_{c}$. Taking the value of grid bias $\mathrm{E}_{c}$ for the given operating condition, the peak a-c grid drive voltage is

$$
\left.{ }^{M} E_{0}=l^{M} e_{c}-E_{d}\right)
$$

from which the peak instantaneous grid drive power
${ }^{M} P_{c}={ }^{M} E_{g}{ }^{M} i_{c}$.

## General design continued

An approximation to the average grid drive power $P_{g}$, necessarily rough due to neglect of negative grid current, is obtained from the typical ratio
$\frac{I_{c}}{\mathrm{M}_{i_{c}}}=0.2$
of d-c to peak value of grid current, giving
$P_{g}=I_{c} E_{g}=0.2{ }^{\mathrm{M}_{c} E_{g}}$ watts.
Plate dissipation $P_{p}$ may be checked with published values since

$$
P_{p}=P_{i}-P_{0}
$$

grid amperes $I_{c}$


Fig. I-Constant-eurrent characteristics with typical load lines $A B$-elass $C, C D-$ class B, EFG-class A, and HJK-class AB.

## General design continued

It should be borne in mind that combinations of published maximum ratings as well as each individual maximum rating must be observed. Thus, for example in this case, the maximum d-c plate operating voltage of 20,000 volts does not permit operation at the maximum d-c plate current of 7 amperes since this exceeds the maximum plate input rating of 135,000 watts.
Plate load resistance $r_{l}$ may be connected directly in the tube plate circuit, as in the resistance-coupled amplifier, through impedance-matching elements as in audio-frequency transformer coupling, or effectively represented by a loaded parallel resonant circuit as in most radio-frequency amplifiers. In any case, calculated values apply only to effectively resistive loads, such as are normally closely approximated in radio-frequency amplifiers. With appreciably reactive loads, operating currents and voltages will in general be quite different and their precise calculation is quite difficult.
The physical load resistance present in any given set-up may be measured by audio-frequency or radio-frequency bridge methods. In many cases, the proper value of $r_{l}$ is ascertained experimentally as in radio-frequency amplifiers which are tuned to the proper minimum d-c plate current. Conversely, if the circuit is to be matched to the tube, $r_{l}$ is determined directly as in a resistance-coupled amplifier or as
$r_{l}=N^{2} r_{s}$
in the case of a transformer-coupled stage, where $N$ is the primary-to-secondary voltage transformation ratio. In a parallel-resonant circuit in which the output resistance $r_{s}$ is connected directly in one of the resistance legs,
$r_{l}=\frac{X^{2}}{r_{i}}=\frac{L}{C r_{g}}=Q X$,
where $X$ is the leg reactance at resonance (ohms).
$L$ and $C$ are leg inductance (henries) and capacitance (farads), respectively,
$Q=\frac{X}{r_{s}}$.

## Graphical design methods

When accurate operating data are required, more precise methods must be used. Because of the non-linear nature of tube characteristics, graphical methods usually are most convenient and rapid. Examples of such methods are given below.
A comparison of the operating regimes of class $A, A B, B$, and $C$ amplifiers is given in the constant-current current characteristics graph of Fig. 1. The

## Graphical design methods continued

lines corresponding to the different classes of operation are each the locus of instantaneous grid $e_{c}$ and plate $e_{b}$ voltages, corresponding to their respective load impedances.

For radio-frequency amplifiers and oscillators having tuned circuits giving an effective resistive load, plate and grid tube and load alternating voltages are sinusoidal and in phase (disregarding transit time), and the loci become straight lines.

For amplifiers having non-resonant resistive loads, the loci are in general non-linear except in the distortionless case of linear tube characteristios (constant $r_{p}$ ) for which they are again straight lines.

Thus, for determination of radio-frequency performance, the constantcurrent chart is convenient. For solution of audio-frequency problems, however, it is more convenient to use the ( $i_{b}-e_{c}$ ) transfer characteristics of Fig. 2 on which a dynamic load line may be constructed.
Methods for calculation of the most important cases are given below.

## Class C r-f ampliffer or oscillator

Draw straight line from $A$ to $B$ (Fig. II corresponding to chosen d-c operating plate and grid voltages, and to desired peak alternating plate and grid voltage excursions. The projection of $A B$ on the horizontal axis thus corresponds to ${ }^{M} E_{p}$. Using Chaffee's 11 -point method of harmonic analysis, lay out on $A B$ points:

$$
e_{p}^{\prime}={ }^{M} E_{p} \quad e_{p}^{\prime \prime}=0.866^{M} E_{p} \quad e^{\prime \prime \prime}{ }_{p}=0.5^{M} E_{p}
$$

to each of which correspond instantaneous plate currents $i^{\prime}{ }_{b}, i^{\prime \prime}{ }_{b}$ and $i^{\prime \prime \prime}{ }_{b}$ and instantaneous grid currents $i_{c}{ }_{c} i^{\prime \prime}{ }_{c}$ and $i^{\prime \prime \prime}{ }_{c}$. The operating currents are obtained from the following expressions:

$$
\begin{aligned}
I_{b} & =\frac{1}{12}\left[i_{b}^{\prime}+2 i_{b}^{\prime \prime}+2 i^{\prime \prime \prime}{ }_{b}\right] & I_{c} & =\frac{1}{12}\left[i_{c}+2 i_{c}^{\prime \prime}+2 i^{\prime \prime \prime}\right] \\
{ }^{\mathrm{M}} I_{p} & =\frac{1}{6}\left[i^{\prime}{ }_{b}+1.73 i_{b}^{\prime \prime}+i^{\prime \prime \prime}{ }_{b}\right] & { }^{\mathrm{M}} I_{q} & =\frac{1}{6}\left[i_{c}^{\prime}+1.73 i_{c}^{\prime \prime}+i^{\prime \prime \prime}{ }_{c}\right]
\end{aligned}
$$

Substitution of the above in the following give the desired operating data.
Power output $P_{0}=\frac{{ }^{M} E_{p}{ }^{M} I_{p}}{2}$
Power input $P_{i}=E_{b} I_{b}$
Average grid excitation power $=\frac{{ }^{M} E_{g}{ }^{M} I_{g}}{2}$


## Graphical design methods continued

Peak grid excitation power $={ }^{\mathrm{M}} E_{g} i_{c}$
Plate load resistance $r_{l}=\frac{{ }^{\mathrm{M}} E_{p}}{{ }^{\mathrm{M}} I_{p}}$
Grid bias resistance $R_{c}=\frac{E_{c}}{I_{c}}$
Plate efficiency $\eta=\frac{P_{0}}{P_{i}}$
Plate dissipation $P_{p}=P_{i}-P_{0}$
The above procedure may also be applied to plate-modulated class $C$ amplifiers. Taking the above data as applying to carrier conditions, the analysis is repeated for ${ }^{\text {crest }} E_{b}=2 E_{b}$ and ${ }^{\text {crest }} P_{0}=4 P_{0}$ keeping $r_{l}$ constant. After a cut-and-try method has given a peak solution, it will often be found that combination fixed and self grid biasing as well as grid modulation is indicated to obtain linear operation.

To illustrate the preceding exposition, a typical amplifier calculation is given below:

Operating requirements (carrier condition)
$E_{b}=12,000$ volts
$P_{0}=25,000$ watts
$\eta=75 \%$

Preliminary calculation (refer to Table 11 )

## Table II-Class C r-f amplifier data $\mathbf{1 0 0} \%$ plate modulation

| symbol | preliminary carrier | detailed |  |
| :---: | :---: | :---: | :---: |
|  |  | carrier | crest |
| $E_{b}$ lvolts) | 12,000 | 12,000 | 24,000 |
| $\mathrm{M} E_{p}$ (volts) | 10,000 | 10,000 | 20,000 |
| $E_{c}$ (valis) |  | -1,000 | -700 |
| ${ }^{M} E_{0}$ (volts) |  | 1.740 | 1,740 |
| 14 lampl | 2.9 | 2.8 | 6.4 |
| $\mathrm{M}_{\mathrm{p}}$ (amp) | 4.9 | 5.1 | 10.2 |
| $I_{c}$ (ampl |  | 0.125 | 0.083 |
| ${ }^{\mathrm{M}} \mathrm{I}_{\mathrm{p}}$ (amp) |  | 0.255 | 0.183 |
| $P_{i}$ (watts) | 35,000 | 33,600 | 154,000 |
| $P_{0}$ (watts) | 25,000 | 25,500 | 102,000 |
| $P_{0}$ (wotts) |  | 220 | 160 |
| $\eta$ (percent) | 75 | 76 | 66 |
| $r_{l}$ (ohms) | 2,065 | 1,960 | 1,960 |
| $R_{c}$ (0hms) |  | 7,100 | 7,100 |
| $E_{\text {ce }}$ (volts) |  | - 110 | -110 |

$$
\begin{aligned}
\frac{E_{p}}{E_{b}} & =0.6 \\
E_{p} & =0.6 \times 12,000=7200 \text { volts } \\
{ }^{{ }^{M}} E_{p} & =1.41 \times 7200=10,000 \text { volts } \\
I_{p} & =\frac{P_{0}}{E_{p}} \\
I_{p} & =\frac{25,000}{7200}=3.48 \text { amperes } \\
{ }^{{ }_{M} I_{p}} & =4.9 \text { amperes } \\
\frac{I_{p}}{I_{b}} & =1.2 \\
I_{b} & =\frac{3.48}{1.2}=2.9 \text { amperes } \\
P_{i} & =12,000 \times 2.9=35,000 \text { watts } \\
\frac{\mathrm{M}_{i_{b}}}{I_{b}} & =4.5 \\
{ }_{\mathrm{M}_{i_{b}}} & =4.5 \times 2.9=13.0 \text { amperes } \\
\mathrm{r}_{l} & =\frac{E_{p}}{I_{p}}=\frac{7200}{3.48}=2060 \text { ohms }
\end{aligned}
$$

## Complete calculation

Layout carrier operating line, $A B$ on constant current graph, Fig. I, using values of $E_{b},{ }^{M} E_{p}$, and ${ }^{M} i_{b}$ from preliminary calculated data. Operating carrier bias voltage, $E_{c}$, is chosen somewhat greater than twice cutoff value, 1000 volts, to locate point $A$.

## The following data are taken along $A B$ :

$$
\begin{aligned}
& i_{b}{ }^{\prime}=13 \mathrm{amp} \\
& i_{c}^{\prime}=1.7 \mathrm{amp} \\
& i_{6}{ }^{\prime \prime}=10 \mathrm{amp} \\
& i_{b}{ }^{\prime \prime \prime}=0.3 \mathrm{amp} \\
& \begin{aligned}
i_{c} & =1.7 \mathrm{amp} \\
i_{c}{ }^{\prime \prime} & =-0.1 \mathrm{amp} \\
i_{c}{ }^{\prime \prime \prime} & =0 \mathrm{amp}
\end{aligned} \\
& E_{c}=-1000 \text { volts } \\
& e_{e}^{\prime}=740 \text { volts } \\
& { }^{\mathrm{M}} E_{p}=10,000 \text { volts }
\end{aligned}
$$

From the formulas, complete carrier data as follows are calculated:

$$
\begin{aligned}
{ }^{M} I_{p} & =\frac{1}{6}[13+1.73 \times 10+0.3]=5.1 \mathrm{amp} \\
P_{0} & =\frac{10,000 \times 5.1}{2}=25,500 \mathrm{watts} \\
I_{b} & =\frac{1}{12}[13+2 \times 10+2 \times 0.3]=2.8 \mathrm{amp} \\
P_{i} & =12,000 \times 2.8=33,600 \mathrm{watts}
\end{aligned}
$$

## Graphical design methods continued

$$
\begin{aligned}
\eta & =\frac{25,500}{33,600} \times 100=76 \text { percent } \\
r_{l} & =\frac{10,000}{5.1}=1960 \mathrm{ohms} \\
I_{c} & =\frac{1}{12}[1.7+2(-0.11]=0.125 \mathrm{amp} \\
\mathrm{M}_{I_{g}} & =\frac{1}{6}[1.7+1.7(-0.11]+0.255 \mathrm{amp} \\
P_{g} & =\frac{1740 \times 0.255}{2}=220 \mathrm{watts}
\end{aligned}
$$

Operating data at 100 percent positive modulation crests are now calculated knowing that here
$E_{b}=24,000$ volts $\quad r_{b}=1960$ ohms
and for undistorted operation

$$
P_{0}=4 \times 25,500=102,000 \text { watts } \quad{ }^{M} E_{p}=20,000 \text { volts }
$$

The crest operating line $A^{\prime} B^{\prime}$ is now located by trial so as to satisfy the above conditions, using the same formulas and method as for the carrier condition.
It is seen that in order to obtain full-crest power output, in addition to doubling the alternating plate voltage, the peak plate current must be increased. This is accomplished by reducing the crest bias voltage with resultant increase of current conduction period, but lower plate efficiency.
The effect of grid secondary emission to lower the crest grid current is taken advantage of to obtain the reduced grid-resistance voltage drop required. By use of combination fixed and grid resistance bias proper variation of the total bias is obtained. The value of grid resistance required is given by

$$
R_{c}=\frac{-\left[E_{c}-{ }^{c r s e t} E_{c}\right]}{I_{c}-\operatorname{crest}^{2} t_{c}}
$$

and the value of fixed bias by
$E_{c c}=E_{c}-\left(U_{c} \cdot R_{c}\right)$
Calculations at carrier and positive crest together with the condition of zero output at negative crest give sufficiently complete data for most purposes. If accurate calculation of audio-frequency harmonic distortion is necessary the above method may be applied to the additional points required.

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## Graphical design methods

## Class B r-f amplifiers

A rapid approximate method is to determine by inspection from the tube $i_{b}$ - $e_{b}$ l characteristics the instantaneous current, $i_{b}^{\prime}$ and voltage $e^{\prime}{ }_{b}$ corresponding to peak alternating voltage swing from operating voltage $E_{b}$.
A.C plate current ${ }^{M} I_{p}=\frac{i^{\prime}}{2}$
D.C plate current $I_{b}=\frac{i_{b}}{\pi}$
A.C plate voltage ${ }^{\mathrm{M}} E_{p}=E_{b}-\mathrm{e}^{\prime}{ }_{b}$

Power output $P_{0}=\frac{\left(E_{b}-e^{\prime} b\right) i_{b}^{\prime}}{4}$
Power input $P_{i}=\frac{E_{b} i_{b}}{\pi}$
Plate efficiency $\eta=\frac{\pi}{4}\left(1-\frac{e^{\prime} b}{E_{b}}\right)$
Thus $\eta \cong 0.6$ for the usual crest value of ${ }^{\mathrm{M}} E_{p} \cong 0.8 E_{b}$.
The same method of analysis used for the class $C$ amplifier may also be used in this case. The carrier and crest condition calculations, however, are now made from the same $E_{b}$, the carrier condition corresponding to an alter-nating-voltage amplitude of $\frac{{ }^{M} E_{p}}{2}$ such as to give the desired carrier power output.

For greater accuracy than the simple check of carrier and crest conditions, the radio-frequency plate currents ${ }^{\mathrm{M}} I^{\prime}{ }_{p 1}{ }^{\mathrm{M}} I^{\prime \prime}{ }_{p 0}{ }^{\mathrm{M}} I^{\prime \prime \prime}{ }_{p,}{ }^{\mathrm{M}} I^{\circ}{ }_{p},-{ }^{\mathrm{M}} I^{\prime \prime \prime}{ }^{\prime}{ }_{p,}$ $-{ }^{\mathrm{M}} I^{\prime \prime}{ }_{p}$, and $-{ }^{\mathrm{M}} I^{\prime}{ }_{p}$ may be calculated for seven corresponding selected points of the audio-frequency modulation envelope $+{ }^{\mathrm{M}} E_{0}+0.707{ }^{\mathrm{M}} E_{0}$ $+0.5^{{ }^{\mathrm{M}}} E_{g}, 0,-0.5^{\mathrm{M}} E_{g},-0.707^{\mathrm{M}} E_{g}$, and $-{ }^{\mathrm{M}} E_{g}$, where the negative signs denote values in the negative half of the modulation cycle. Designating
$S^{\prime}={ }^{\mathrm{M}} I^{\prime}{ }_{p}+\left(-{ }^{\mathrm{M}} I^{\prime}{ }_{p}\right)$
$D^{\prime}={ }^{M} I_{p}^{\prime}-\left(-{ }^{M} I_{p}^{\prime}{ }_{p}\right)$, etc.,
the fundamental and harmonic components of the output audio-frequency current are obtained as
${ }^{\mathrm{M}} I_{p 1}=\frac{S^{\prime}}{4}+\frac{S^{\prime \prime}}{2 \sqrt{2}}$ (fundamental) $\quad{ }^{\mathrm{M}} I_{p 2}=\frac{5 D^{\prime}}{24}+\frac{D^{\prime \prime}}{4}-\frac{D^{\prime \prime \prime}}{3}$
${ }^{M} I_{p 3}=\frac{S^{\prime}}{6}-\frac{S^{\prime \prime \prime}}{3}$
${ }^{M} I_{p 6}=\frac{S^{\prime}}{12}-\frac{S^{\prime \prime}}{2 \sqrt{2}}+\frac{S^{\prime \prime \prime}}{3}$
${ }^{\mathrm{M}} l_{p 60}=\frac{D^{\prime}}{8}-\frac{D^{\prime \prime}}{4}$
${ }^{M} I_{p 6}=\frac{D^{\prime}}{24}-\frac{D^{\prime \prime}}{4}+\frac{D^{\prime \prime \prime}}{3}$
This detailed method of calculation of audio-frequency harmonic distortion may, of course, also be applied to calculation of the class $C$ modulated amplifier, as well as to the class A modulated amplifier.

## Class $A$ and $A B$ a-f amplifers

Approximate formulas assuming linear tube characteristics:
Maximum undistorted power output ${ }^{M} P_{0}=\frac{{ }^{M} E_{p}{ }^{M} I_{p}}{2}$
when plate load resistance $r_{l}=r_{p}\left[\frac{E_{c}}{\frac{{ }^{M} E_{p}}{\mu}-E_{c}}-1\right]$
and
Negative grid bias $E_{c}=\frac{{ }^{M} E_{p}}{\mu}\left(\frac{r_{b}+r_{p}}{r_{b}+2 r_{p}}\right)$
giving
Maximum plate efficiency $\eta=\frac{{ }^{M} E_{p}{ }^{M} I_{p}}{8 E_{b} I_{b}}$
Maximum maximum undistorted power output ${ }^{\mathrm{MM}} \mathrm{P}_{0}=\frac{{ }^{\mathrm{M}} E^{2}{ }_{p}}{16 r_{p}}$
when
$r_{i}=2 r_{p} \quad E_{c}=\frac{3}{4} \frac{{ }^{M} E_{p}}{\mu}$
An exact analysis may be obtained by use of a dynamic load line laid out on the transfer characteristics of the tube. Suct: a line is CKF of Fig. 2 which is constructed about operating point $K$ for a given load resistance $r_{3}$ from the following relation:
$i_{b}^{\mathrm{A}}=\frac{e_{b}^{\mathrm{R}}-\mathrm{e}_{b}^{\mathrm{g}}}{n_{l}}+i_{b}^{\mathrm{R}}$
where
R, S, etc., are successive conveniently spaced construction points.

Using the seven-point method of harmonic analysis, plot instantaneous plate currents $i^{\prime}{ }_{b} i^{\prime \prime}{ }_{b,} i^{\prime \prime \prime}{ }_{b,} i_{b,}-i^{\prime \prime \prime}{ }_{b,}-i^{\prime \prime}{ }_{b,}$ and $-i^{\prime}{ }_{b}$ corresponding to $+{ }^{\mathrm{M}} E_{q}+0.707^{\mathrm{M}} E_{0},+0.5^{\mathrm{M}} E_{0}, 0,-0.5^{\mathrm{M}} E_{q},-0.707^{\mathrm{M}} E_{q}$, and $-{ }^{\mathrm{M}} E_{q}$, where 0 corresponds to the operating point $K$. In addition to the formulas given under class $B$ radio-frequency amplifiers:
$I_{b}$ average $=I_{b}+\frac{D^{\prime}}{8}+\frac{D^{\prime \prime}}{4}$
from which complete data may be calculated.

## Class AB and Ba-f amplifiers

Approximate formulas assuming linear tube characteristics give (referring to Fig. I, line CDI for a class B audio-frequency amplifier:

$$
\begin{aligned}
{ }^{M} I_{p} & =i^{\prime}{ }_{b} \\
P_{0} & =\frac{{ }^{M} E_{p}{ }^{M} I_{p}}{2} \\
P_{i} & =\frac{2}{\pi} E_{b}{ }^{M} I_{p} \\
\eta & =\frac{\pi}{4} \frac{{ }^{M} E_{p}}{E_{b}} \\
R_{p p} & =4 \frac{{ }^{M} E_{p}}{i^{\prime}{ }_{b}}=4 r_{i}
\end{aligned}
$$

Again an exact solution may be derived by use of the dynamic load line JKL on the $\left(i_{b}-e_{c}\right)$ characteristic of Fig. 2. This line is calculated about the operating point $K$ for the given $r_{l}$ (in the same way as for the class $A$ case) However, since two tubes operate in phase opposition in this case, an identical dynamic load line MNO represents the other half cycle, laid out about the operating bias abscissa point but in the opposite direction (see Fig. 2).
Algebraic addition of instantaneous current values of the two tubes at each value of $e_{c}$ gives the composite dynamic characteristic for the two tubes OPL. Inasmuch as this curve is symmetrical about point $P$ it may be analyzed for harmonics along a single half curve PL by the Mouromtseff 5 -point method. A straight line is drawn from $P$ to $L$ and ordinate plate current differences $a, b, c, d, f$ between this line and curve, corresponding to $e^{\prime \prime}{ }_{9,} e^{\prime \prime \prime}{ }_{9,}$ $\mathrm{e}^{\mathrm{IV}}{ }_{g}, \mathrm{e}^{\mathbf{V}}{ }_{0}$, and $\mathrm{e}^{\mathrm{vI}^{\mathrm{I}}}{ }_{9}$, are measured. Ordinate distances measured upward from curve PL are taken positive.

## Graphical design methods

Fundamental and harmonic current amplitudes and power are found from the following formulas:

$$
\begin{aligned}
& { }^{\mathrm{M}} I_{p 1}=i^{\prime}{ }_{b}-{ }^{\mathrm{M}} I_{p 3}+{ }^{\mathrm{M}} I_{p 5}-{ }^{\mathrm{M}} I_{p 7}+{ }^{\mathrm{M}} I_{p 9}-{ }^{\mathrm{M}} I_{p 11} \\
& { }^{\mathrm{M}} I_{p 3}=0.4475 \mathrm{lb}+n+\frac{\mathrm{d}}{3}-0.578 \mathrm{~d}-\frac{1}{2}{ }^{\mathrm{M}} I_{p 5} \\
& { }^{\mathrm{M}} I_{p 5}=0.4 \mathrm{la}-n \\
& { }^{\mathrm{M}} I_{I_{7}}=0.4475 \mathrm{lb}+n-{ }^{\mathrm{M}} I_{p 3}+0 .{ }^{\mathrm{M}} I_{p 5} \\
& { }^{\mathrm{M}} I_{p 9}={ }^{\mathrm{M}} I_{p 3}-\frac{2}{3} \mathrm{~d} \\
& { }^{\mathrm{M}} I_{p 11}=0.707 \mathrm{c}-{ }^{\mathrm{M}} I_{p 3}+{ }^{\mathrm{M}} I_{p b} .
\end{aligned}
$$

Even harmonics are not present due to dynamic characteristic symmetry. The direct current and power input values are found by the 7 -point analysis from curve PL and doubled for two tubes.

## Classification of amplifier circuits

The classification of amplifiers in classes $\mathrm{A}, \mathrm{B}$, and C is based on the operating conditions of the tube.
Another classification can be used, based on the type of circuits associated with the tube.
A tube can be considered as a four-terminal network with two input terminals and two output terminals. One of the input terminals and one of the output terminals are usually common; this common junction or point is usually called "ground".
When the common point is connected to the filament or cathode of the tube, we can speak of a grounded-cathode circuit. It is the most conventional type of vacuum tube circuit. When the common point is the grid, we can speak of a grounded-grid circuit, and when the common point is the plate or anode, we can speak of the grounded-anode circuit.
This last type of circuit is most commonly known by the name of cathode follower.
A fourth and most general class of circuit is obtained when the common point or ground is not directly connected to any of the three electrodes of the tube. This is the condition encountered at $u$-h-f where the series impedances of the internal tube leads make it impossible to ground any of them. It is also encountered in such special types of circuits as the phase-splitter, in which the impedance from plate to ground and the impedance from cathode to ground are made equal in order to obtain an output between plate and cathode balanced with respect to ground.

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Table III-Classification of triode amplifier circuits

| circuit classiffation | groundedcathode | groundedgrid | grounded-plate or cathode follower |
| :---: | :---: | :---: | :---: |
| Circuit schematic |  |  |  |
| Equivalent circuit, o-c component, class $A$ operation |  |  |  |
| Voltage gain, $\gamma$ for output load impedance $=Z_{2}$ $\gamma=\frac{E_{2}}{E_{1}}$ | neglecting $\mathrm{C}_{g p}$ $\begin{aligned} \gamma & =\frac{-\mu Z_{2}}{r_{p}+z_{2}} \\ & =-g_{m} \frac{r_{p} z_{2}}{r_{p}+z_{2}} \end{aligned}$ <br> ( $Z_{2}$ includes $C_{p k}$ ) | neglecting $C_{p k}$ $\gamma=(1+\mu) \frac{z_{2}}{r_{p}+z_{2}}$ $\left(Z_{2} \text { includes } C_{o p}\right)$ | neglecting $C_{u k}$ $\gamma=\frac{\mu Z_{2}}{r_{p}+(1+\mu) Z_{2}}$ <br> $\left(Z_{2}\right.$ includes $\left.C_{p k}\right)$ |
| Input admittonce $Y_{1}=\frac{I_{1}}{E_{1}}$ | $Y_{1}=j \omega\left[C_{0 k}+(11-\gamma) C_{g p}\right]$ | $\begin{aligned} & Y_{1}=j \omega\left[C_{p k}+\right. \\ & \left.\quad(1-\gamma) C_{p k}\right]+\frac{1+\mu}{r_{p}+Z_{2}} \end{aligned}$ | $Y_{1}=j \omega\left[C_{a p}+(1-\gamma) C_{g k}\right]$ |
|  | neglecting $\mathrm{C}_{o p}$ | neglecting $\mathrm{C}_{\text {pk }}$ | neglecting $\mathrm{C}_{0 k}$ |
| Equivalent generator seen by load at output terminals |  |  |  |

## Classification of amplifler circuits continued

Design information for the first three classifications is given in Table III, where
$Z_{2}=$ load impedance to which output terminals of amplifier are connected
$E_{1}=\mathrm{rms}$ driving voltage across input terminals of amplifier
$E_{2}=$ rms output voltage across load impedance $Z_{2}$
$I_{1}=$ rms current at input terminals of amplifier
$\boldsymbol{\gamma}=$ voltage gain of amplifier $=\frac{E_{2}}{E_{1}}$
$Y_{1}=$ input admittance to input terminals of amplifier $=\frac{I_{1}}{E_{1}}$
$\omega=2 \pi \times$ frequency of excitation voltage $E_{1}$
$j=\sqrt{-1}$
and the remaining notation is in accordance with the nomenclature of pages 127 and 128.

## Cathode follower data

## General characterisfics

1. High impedance input, low impedance output.
2. Input and output have one side grounded.
3. Good wide-band frequency and phase response.
4. Output is in phase with input.
5. Voltage gain or transfer is always less than one.
6. A power gain can be obtained.
7. Input capacitance is reduced.

## General case

Transfer $=\frac{g_{m} R_{L}}{g_{m} R_{L}+1}$ or $g_{m} Z_{r}$
$Z_{r}=$ resultant cathode to ground impedance $=R_{\text {out }}$ in parallel with $R_{e}$
$R_{\text {out }}=$ output resistance
$=\frac{R_{p}}{\mu+1}$ or approximately $\frac{1}{g_{m}}$
$R_{L}=$ total load resistance
Input capacitance $=C_{o p}+\frac{C_{p k}}{1+g_{m} R_{L}}$
$g_{m}=$ transconductance in mhos 11000
micromhos $=0.001$ mhosl


## Cathode follower data continued

## Specific cases

1. To match the characteristic impedance of the transmission line, $R_{\text {oue }}$ must equal $Z_{0}$. The transfer is approximately 0.5 .
2. If $R_{\text {out }}$ is less than $Z_{0}$, add resistor $R_{c}{ }^{\prime}$ in series so that $R_{c}{ }^{\prime}=Z_{0}-R_{\text {outo }}$ The transfer is approximately 0.5.

3. If $R_{o u}$ is greater than $Z_{0}$ add resistor $R_{c}$ in parallel so that
$R_{c}=\frac{Z_{0} R_{\text {out }}}{R_{\text {out }}-Z_{0}}$
Transfer $=\frac{g_{m} Z_{0}}{2}$
Note: Normal operating blas must be provided.


For coupling a high impedance into a low impedance transmission line, for maximum transfer choose a tube with a high 9 m .

## Resistance-coupled audio amplifier design

Stage gain at
Medium frequencies $=A_{m}=\frac{\mu R}{R+R_{p}}$
High frequencies $=A_{h}=\frac{A_{m}}{\sqrt{1+\omega^{2} C_{1}^{2} r^{2}}}$
Low frequencies* $=A_{1}=\frac{A_{m}}{\sqrt{1+\frac{1}{\omega^{2} C_{2}^{2} \rho^{2}}}}$

* The low-frequency stage gain also is offected by the volues of the cathode by-poss capacitor and the screen ty-pass copocitor.


## Resisfance coupled audio amplifler design

cantinued
where

$$
R=\frac{r_{l} R_{2}}{r_{l}+R_{2}}
$$

$$
r=\frac{R r_{p}}{R+r_{p}}
$$

$\rho=R_{2}+\frac{r_{l} r_{p}}{r_{l}+r_{p}}$


A-plate.
B-grid.
C-ground or cathode.
. $\mu=$ amplification factor of tube
$\omega=2 \pi \times$ frequency
$r_{l}=$ plate load resistance in ohms
$\mathbf{R}_{\mathbf{2}}=$ grid leak resistance in ohms
$r_{p}=a-c$ plate resistance in ohms
$\mathrm{C}_{1}=$ total shunt capacitance in farads
$C_{2}=$ coupling capacitance in farads
Given $C_{1}, C_{2}, R_{2}$ and $X=$ fractional response required
At highest frequency

$$
r=\frac{\sqrt{1-X^{2}}}{\omega C_{1} X} \quad R=\frac{r r_{p}}{r_{p}-r} \quad r_{l}=\frac{R R_{2}}{R_{2}-R}
$$

At lowest frequency*
$C_{2}=\frac{X}{\omega \rho \sqrt{1-X^{2}}}$

* The low-frequency stage gain also is affected by the values of the cathode by-pass capocitor and the screen by-pass capacitor.


## Negafive feedback

The following quantities are functions of frequency with respect to magnitude and phase:
$E, N$, and $D=$ signal, noise, and distortion output voltage with feedback $e, n$, and $d=$ signal, noise, and distortion output voltage without feedback
$A=$ voltage amplification of amplifier at a given frequency
$\beta=$ fraction of output voltage fed back; for usual negative feedback, $\boldsymbol{\beta}$ is negative
$\phi=$ phase shift of amplifier and feedback circuit at a given frequency

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## Reduction in gain caused by feedback



Fig. 3-In negative-feedback amplifer consideralions $\beta$, expressed as a percentage, hos a negative value. A line across the $\beta$ and $A$ scales intersects the center scale to indicate change in gain. It also indicates the amount, In decibels, the inpul must be increased to maintaln original output.


## Negative feedback cantinued

The total output voltage with feedback is
$E+N+D=e+\frac{n}{1-A \beta}+\frac{d}{1-A \beta}$
It is assumed that the input signal to the amplifier is increased when negative feedback is applied, keeping $E=\mathrm{e}$.
$(1-A \beta)$ is a measure of the amount of feedback. By definition, the amount of feedback expressed in decibels is
$20 \log _{10}|1-A \beta|$
Voltage gain with feedback $=\frac{A}{1-A \beta}$
and change of gain $=\frac{1}{1-A \beta}$
If the amount of feedback is large, i.e., $-A \beta>1$, the voltage gain becomes $-\frac{1}{\beta}$ and so is independent of $A$.

In the general case when $\phi$ is not restricted to 0 or $\pi$
the voltage gain $=\frac{A}{\sqrt{1+|A \beta|^{2}-2|A \beta| \cos \phi}}$
and change of gain $=\frac{1}{\sqrt{1+|A \beta|^{2}-2|A \beta| \cos \phi}}$
Hence if $|A \beta| \gg 1$, the expression is substantially independent of $\phi$.
On the polar diagram relating $(\mathbb{A} \beta)$ and $\phi$ (Nyquist diagram), the system is unstable if the point (1,0) is enclosed by the curve.

## Feedback amplifier with single beam power fube

The use of the foregoing negative feedback formulas is illustrated by the amplifier circuit shown in Fig. 4.

The amplifier consists of an output stage using a 6V6-G beam power tetrode with feedback driven by a resistance-coupled stage using a 6J7-G in a pentode connection. Except for resistors $R_{1}$ and $R_{2}$ which supply the feedback voltage, the circuit constants and tube characteristics are taken from published data.

## Negative feedback

The fraction of the output voltage to be fed back is determined by specifying that the total harmonic distortion is not to exceed 4 percent. The plate supply voltage is taken as 250 volts. At this voltage, the $6 \mathrm{~V} 6-\mathrm{G}$ has 8 percent


Fig. 4-Feedback amplifier with single beam power tube.
total harmonic distortion. From equation (1), it is seen that the distortion output voltage with feedback is
$D=\frac{d}{1-A \beta}$
This may be written as
$1-A \beta=\frac{d}{D}$
where
$\frac{d}{D}=\frac{8}{4}=2 \quad 1-A \beta=2 \quad \beta=-\frac{1}{A}$
and where $A=$ the voltage amplification of the amplifier without feedback.
The peak a-f voltage output of the 6V6-G under the assumed conditions is
$E_{o}=\sqrt{4.5 \times 5000 \times 2}=212$ volts
This voltage is obtained with a peak $a_{-}$f grid voltage of 125 volts so that the voltage gain of this stage without feedback is
$A=\frac{212}{12.5}=17$

## Negative feedback

Hence $\quad \beta=-\frac{1}{A}=-\frac{1}{17}=-0.0589$ or $5.9 \%$ approximately
The voltage gain of the output stage with feedback is computed from equation (3) as follows
$A^{\prime}=\frac{A}{1-A \beta}=\frac{17}{2}=8.5$
and the change of gain due to feedback by equation (4) thus
$\frac{1}{1-A \beta}=0.5$
The required amount of feedback voltage is obtained by choosing suitable values for $R_{1}$ and $R_{2}$. The feedback voltage on the grid of the 6V6-G is reduced by the effect of $R_{g}, R_{\mathrm{L}}$ and the plate resistance of the $6 \mathrm{~J} 7 . \mathrm{G}$. The effective grid resistance is
$R_{g}{ }^{\prime}=\frac{R_{g} r_{p}}{R_{g}+r_{p}}$
where $\quad R_{g}=0.5$ megohm.
This is the maximum allowable resistance in the grid circuit of the 6V6-G with cathode bias.
$r_{p}=4$ megohms, the plate resistance of the 6J7-G tube
$R_{g}{ }^{\prime}=\frac{4 \times 0.5}{4+0.5}=0.445$ megohm
The fraction of the feedback voltage across $R_{2}$ which appears at the grid of the $6 \mathrm{~V} 6-\mathrm{G}$ is
$\frac{R_{g}^{\prime}}{R_{g}^{\prime}+R_{L}}=\frac{0.445}{0.445+0.25}=0.64$
where $\quad R_{L}=0.25$ megohm.
Thus the voltage across $R_{2}$ to give the required feedback must be .
$\frac{5.9}{0.64}=9.2 \%$ of the output voltage.
This voltage will be obtained if $R_{1}=50,000$ ohms and $R_{2}=5000$ ohms.
This resistance combination gives a feedback voltage ratio of
$\frac{5000 \times 100}{50,000+5000}=9.1 \%$ of the output voltage.

## Negative feedback

In a transformer-coupled output stage, the effect of phase shift on the gain with feedback does not become appreciable until a noticeable decrease in gain without feedback also occurs. In the high-frequency range, a phase shift of 25 degrees lagging is accompanied by a 10 percent decrease in gain. For this frequency, the gain with feedback is computed from equation (6).

$$
A^{\prime}=\frac{A}{\sqrt{1+|A \beta|^{2}-2|A \beta| \cos \phi}}
$$

where $A=15.3, \phi=180^{\circ}, \cos \phi=0.906, \beta=0.059$.

$$
A^{\prime}=\frac{15.3}{\sqrt{1+|0.9|^{2}+2|0.9| 0.906}}=\frac{15.3}{\sqrt{3.44}}=\frac{15.3}{1.85}=8.27
$$

The change of gain with feedback is computed from equation (7).
$\frac{1}{\sqrt{1+|A \beta|^{2}-2|A \beta| \cos \phi}}=\frac{1}{1.85}=0.541$
If this gain with feedback is compared with the value of 8.5 for the case of no phase shift, it is seen that the effect of frequency on the gain is only 2.7 percent with feedback compared to 10 percent without feedback.

The change of gain with feedback is 0.541 times the gain without feedback whereas in the frequency range, where there is no phase shift, the corresponding value is 0.5 . This quantity is 0.511 when there is phase shift but no decrease of gain without feedback.

## Distortion

A rapid indication of the harmonic content of an alternating source is given by the distortion factor which is expressed as a percentage.


If this factor is reasonably small, say less than 10 percent, the error involved in measuring it

```
sum of squares of amplitudes of harmonics
sum of squares of amplitudes of fundamental and harmonics}\times100
```

is also small. This latter is measured by the distortion factor meter.

## ■ Room acoustics*

## General considerations for good room acoustics

The following information is intended primarily to aid field engineers in appraising acoustical properties of existing structures and not as a complete treatise on the subject.

## Good acoustics-governing factors

a. Reverberation time or amount of reverberation: Varies with frequency and is measured by the time required for a sound, when suddenly interrupted, to die away or decay to a level 60 decibels (db) below the original sound.

The reverberation time and the shape of the reverberation-time/frequency curve can be controlled by selecting the proper amounts and varieties of sound-absorbent materials and by the methods of application. Room occupants must be considered inasmuch as each person present contributes a fairly definite amount of sound absorption.
b. Standing sound waves: Resonant conditions in sound studios cause standing waves by reflections from opposing parallel surfaces, such as ceilingfloor and parallel walls, resulting in serious peaks in the reverberation-time/ frequency curve. Standing sound waves in a room can be considered comparable to standing electrical waves in an improperly terminated transmission line where the transmitted power is not fully absorbed by the load.

## Room sizes and proportions for good acoustics

The frequency of standing waves is dependent on room sizes: frequency decreases with increase of distances between walls and between floor and ceiling. In rooms with two equal dimensions, the two sets of standing waves occur at the same frequency with resultant increase of reverberation time at resonant frequency. In a room with walls and ceilings of cubical contour this effect is tripled and elimination of standing waves is practically impossible.

The most advantageous ratio for height: width: length is in the proportion of $1: 2^{1 / 5}: 2^{3 / 3}$ or separated by $1 / 3$ or $2 / 3$ of an octave.

In properly proportioned rooms, resonant conditions can be effectively reduced and standing waves practically eliminated by introducing numerous surfaces disposed obliquely. Thus, large-order reflections can be avoided by breaking them up into numerous smaller reflections، The object is to pre-

[^13]
## Room sizes and proportions for good acoustics continued

vent sound reflection back to the point of origin until after several rereflections.

Most desirable ratios of dimensions for broadcast studios are given in Fig. 1.

 percent.

## Optimum reverberation time

Optimum, or most desirable reverberation time, varies with (1) room size, and (2) use, such as music, speech, etc. (see Figs. 2 and 3).

## Optimum reverberation time continued



Fig. 2-Oplimum reverberation time in seconds for various reom valumes at 512 cyeles per second.


Fig. 3-Desirable relative reverberation time versus frequency for various structures ond auditoriums.

[^14]A small radio studio for speech broadcasts represents a special case. The acoustic studio design should be such that the studio neither adds nor detracts from the speaker's voice, which on reproduction in the home should sound as though he were actually present.


FIg. 4.

For optimum characteristics of a speech studio, the reverberation time should be about one-half a second throughout the middle and lower audio-frequency range. At high frequencies, the reverberation time may be 20 percent to 25 percent greater than at 512 cycles. This rise at the higher frequencies enhances intelligibility and allows for the presence in the studio of one or two extra persons without materially affecting the reverberation-time/ frequency curve.

## Optimum reverberation time

 continuedSpeech sounds above about 1000 cycles promote intelligibility. Apparent intensity of speech sounds is provided by frequencies below this value.
Preponderance of low bass reverberation and standing waves tends to make the voice sound "boomy" and impairs speech intelligibility.


Fig. 5-Value of aftenuation constont $m$ different frequencies and relative humidthles.*

## Computation of reverberation time

Reverberation time at different audio frequencies may be computed from room dimensions and average absorption. Each portion of the surface of a room has a certain absorption coefficient a dependent on the material of the surface, its method of application, etc. This absorption coefficient is equal to the ratio of the energy absorbed by the surface to the total energy impinging thereon at various audio frequencies. Total absorption for a given surface area in square feet $S$ is expressed in terms of absorption units, the number of units being equal to $a_{a v} S$.
$a_{a v}=\frac{\text { total number of absorption units }}{\text { total surface in square feet }}$
One absorption unit provides the same amount of sound absorption as one square foot of open window. Absorption units are sometimes referred to as "open window" or "OW" units.
$T=\frac{0.05 \mathrm{~V}}{-S \log _{e}\left(1-a_{a v}\right)}$
where $T=$ reverberation time in seconds, $V=$ room volume in cubic feet, $S=$ total surface of room in square feet, $a_{a v}=$ average absorption coeffi. cient of room at frequency under consideration.

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Computation of reverberation time continued
For absorption coefficients a of some typical building materials, see Table I. As an aid in using the formula for reverberation time, Fig. 4 (page 1681 may be used for obtaining $\left[-\log _{\varepsilon}\left(1-a_{a v}\right)\right]$ from known values of $a_{a v}$.

Table Il shows absorption coefficients for some of the more commonly used materials for acoustical correction.

Table I-Acoustical coefficients of materials and persons*

| description | sound obsorption coefficients cycles per second |  |  |  |  |  | authority |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 128 | 256 | 512 | 1024 | 2048 | 4096 |  |
| Brick wall unpaintod | 0.024 | 0.025 | 0.031 | 0.042 | 0.049 | 0.07 | W. C. Sabine |
| Brick wall pointed | 0.012 | 0.013 | 0.017 | 0.02 | 0.023 | 0.025 | W. C. Sobine |
| Ploster + finish coot <br> Wood lath-wood studs | 0.020 | 0.022 | 0.032 | 0.039 | 0.039 | 0.028 |  |
| Plaster + finish coot on metal lath | 0.038 | 0.049 | 0.060 | 0.085 | 0.043 | 0.056 | V. E. Sabine |
| Poured concrete unpainted | 0.010 | 0.012 | 0.016 | 0.019 | 0.023 | 0.035 | V. O. Knudsen |
| Poured concrete painted and varnished | 0.009 | 0.011 | 0.014 | 0.016 | 0.017 | 0.018 | V. O. Knudsen |
| Carpet, pile on cancrete | 0.09 | 0.08 | 0.21 | 0.26 | 0.27 | 0.37 | Building Research Stotion |
| Carpet, plle on $36{ }^{\prime \prime}$ felt | 0.11 | 0.14 | 0.37 | 0.43 | 0.27 | 0.25 | Building Research Station |
| Draperies, volour, 18 oz per sq yd in contact with wall | 0.05 | 0.12 | 0.35 | 0.45 | 0.38 | 0.36 | P. E. Sabine |
| Ozite 36" | 0.051 | 0.12 | 0.17 | 0.33 | 0.45 | 0.47 | P. E. Sabine |
| Rug, oxminstar | 0.11 | 0.14 | 0.20 | 0.33 | 0.52 | 0.82 | Wento and Bodell |
| Audience, seated per sq it of area | 0.72 | 0.89 | 0.95 | 0.99 | 1.00 | 1.00 | W. C. Sabine |
| Eoch person, seated | 1.4 | 2.25 | 3.8 | 5.4 | 6.6 | - | Bureau of Standards, averages of 4 tests |
| Each parson, seated Glass surfoces | 0.05 | $0 . \overline{04}$ | 0.03 | 0.025 | 0.02 | 7.0 0.02 | Estimated |

* Reprinted by permission from Architectural Acoustics by V. O. Knudsen, published by John

Wiley and Sons, Inc.
Table II-Acoustical coefficients of materials used for acoustical correction

| material | cycles per second |  |  |  |  |  | nolsered coef ${ }^{*}$ | manufactured by |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 128 | 236 | 512 |  | 2048 | 1096 |  |  |
| Corkoustic-B4 | 0.08 | 0.13 | 0.51 | 0.75 | 0.47 | 0.46 | 0.45 | Armstrong Cork Co. |
| Corkoustic-86 | 0.15 | 0.28 | 0.82 | 0.60 | 0.58 | 0.38 | 0.55 | Armstrong Cork Co. |
| Cushiontone A-3 | 0.17 | 0.58 | 0.70 | 0.90 | 0.76 | 0.71 | 0.75 | Armstrang Cork Co. |
| Koustex | 0.10 | 0.24 | 0.64 | 0.92 | 0.77 | 0.75 | 0.65 | David E. Kannedy, Inc. |
| Sanocousitc tmetoll piles | 0.25 | 0.56 | 0.99 | 0.99 | 0.91 | 0.82 | 0.85 | Johns-Manville Sales Corp. |
| Pormacoustic tiles $4^{\prime \prime}$ | 0.19 | 0.34 | 0.74 | 0.76 | 0.75 | 0.74 | 0.65 | Johns-Manville Soles Corp. |
| Low-frequency element | 0.66 | 0.60 | 0.50 | 0.50 | 0.35 | 0.20 | 0.50 | Johns-Manville Solos Corpa |
| Triplo-funed element | 0.66 | 0.61 | 0.80 | 0.74 | 0.79 | 0.75 | 0.75 | Johns-Manvillo Salos Corp. |
| High -frequency element | 0.20 | 0.46 | 0.55 | 0.66 | - 0.79 | 0.75 | 0.60 | Johns-Manvilla Salas Corp. |
| Absorbotone A | 0.15 | 0.28 | 0.82 | 0.99 | 0.87 | 0.98 | 0.75 | Luse Stevenson Co. |
| Accusfex 60\% | 0.14 | 0.28 | 0.81 | 0.94 | 0.83 | 0.80 | 0.70 | Notional Gypsum Co. |
| Econacoustic 1" | 0.25 | 0.40 | 0.78 | 0.76 | 0.79 | 0.68 | 0.70 | National Gypsum Co. |
| Fiberglos acoustical tiletype TWPF 9D | 0.22 | 0.46 | 0.97 | 0.90 | 0.68 | 0.52 | 0.75 | Owens-Corning Fiberglas Corp. |
| Acousione D ${ }^{11} 10$ | 0.13 | 0.26 | 0.79 | 0.88 | 0.76 | 0.74 | 0.65 | U. S. Gypsum Company |
| Acoustona $\mathrm{F}^{18} \mathrm{IH}^{2}$ | 0.16 | 0.33 | 0.85 | 0.89 | 0.80 | 0.75 | 0.70 | U. S. Gypsum Company |
| Acousti-celotex type $\mathrm{C}-611 / 4{ }^{\text {a }}$ | 0.30 | 0.56 | 0.94 | 0.96 | 0.69 | 0.56 | 0.80 | The Colotox Corp. |
| Absorbex type A ${ }^{\text {Acousteal B matal }}$ | 0.41 | 0.71 | 0.96 | 0.88 | 0.85 | 0.96 | 0.85 | The Colotex Corp. |
| Acousteal B matal facing 1\%\% | 0.29 | 0.57 | 0.98 | 0.99 | 0.85 | 0.57 | 0.85 | The Colotex Corp. |

[^15]
## Computation of reverberation time continued

Considerable variation of sound-absorption in air at frequencies above 1000 cycles occurs at high relative humidities (see Fig. 5). Calculation of reverberation time, therefore, should be checked at average relative humidities applicable to the particular location involved. For such check calculations the following formula may be used:

$$
T=\frac{0.05 V}{-S \log _{e}\left(1-a_{a \nu}\right)+4 m V}
$$

where $m$ is the coefficient in feet ${ }^{-1}$ as indicated in Fig. 5, page 169.

## Elecirical power levels for public address requirements

a. Indoor: See Fig. 7, page 172.
b. Outdoor: See Fig. 8, page 173.

Note: Curves are for an exponential trumpet-type horn. Spoech levels above referenco-average 70 ob , peak 80 db . For a loudspeaker of 25 percent eificlency, 4 times the power output would be required or an equivalent of 6 decibels. For one of 10 percent efficiency, 10 times tho power output would be required or 10 decibels.


Fig. 6-Wire sizes for loudspeaker circuits assuming maximum loss of 0.5 decibel.

Electrical power levels for public address requirements continued


Fig. 7-Room volume and relative amplifer power capacity. To the indicated power level depending on loudspeaker efficiency, there must be added a correction factor which may vary from 4 decibels for the most efficient horn-type reproducers to 20 decibels for less efficient cone loudspeakers.

Electrical power levels for public address requirements continued

relative ampliffer power capacity-moximum single-frequency output rating in decibels above 0.001 watt

Courtesy Western Electric Compony

Fig. 8-Distance from loudspeaker and relative amplifier power capacity required for speech, average for $30^{\circ}$ angle of coveragè. For angles over $30^{\circ}$, more loudspeakers and proportional output power are required. Depending on loudspeaker efficiency, a correction factor must be added to the indicated power level, varying approximately from 4 to 7 decibels for the more-efficient type of horn loudspeakers.

## Acoustical music ranges and levels



Fig. 9-Frequency ranges of musical instruments. Intensity levels of music. Zero level equals $10^{-10}$ watt per square centimeter.

Acoustical speech levels and ranges of other sounds


Fig. 10-Frequency ranges of male and female speech ond other sounds. Intensity levels of conversational speech. Zere level equals $10^{-16} \mathrm{waft}$ per square centimeter.

Acoustical sound level and pressure

sound level in decibels obove $10^{-16}$ wott per squore centimeter
Courtesy Western Electric Company
Fig. 11-One dyne per square centimatar is equivalent to an acoustieal feval of piys. 74 decibels.

## Table III-Noise levels



Zero lavel $=10^{-16}$ woll per squore centimeter
Courtesy Western Electric Company

## General

a. Loudspeaker wire sizes: See Fig. 6, page 171.
b. Acoustical musical ranges and levels: See. Fig. 9, page 174.
c. Acoustical speech levels and ranges of other sounds: See Fig. 10, page 175.
d. Acoustical sound levels: See Fig. 11, page 176.
e. Noise levels: See Table III.
f. Equal loudness contours: Fig. 12 gives average hearing characteristics of the human ear at audible frequencies and at loudness levels of zero to 120 db versus intensity levels expressed in decibels above $10^{-16}$ watt per square centimeter. Ear sensitivity varies considerably over the audible range of sound frequencies at various levels. A loudness level of 120 db is heard fairly uniformly throughout the entire audio range but, as indicated in Fig. 12,


Fig. 12-Equal loudness contours.
a frequency of 1000 cycles at a 20 db level will be heard at very nearly the same intensity as a frequency of 60 cycles at a 60 db level. These curves explain why a loudspeaker operating at lower than normal level sounds as though the higher frequencies were accentuated and the lower tones seriously attenuated or entirely lacking; also, why music, speech, and other sounds, when reproduced, should have very nearly the same intensity as the original rendition. To avoid perceptible deficiency of lower tones, a symphony orchestra, for example, should be reproduced at an acoustical level during the loud passages of 90 to 100 db (see Fig. 91.

## Wire transmission

## Telephone fransmission line dała

## line constants of copper open-wire pairs

40 pairs DP (deuble petticoat) insulotors per mile 12-inch spacing temperature $68^{\circ} \mathrm{F}$

| frequency cycles per second | resistance ohms per loop mile |  |  | inductance <br> millihenries per loop mile |  |  | leakance <br> micromhos per loop mile: <br> 165, 128, or 104 mil |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 165 mil | 128 mil | 104 mil | 165 mil | 128 mil | 104 mil | diy | well |
| 0 | 4.02 | 6.68 | 10.12 | 3.37 | 3.53 | 3.66 | 0.01 |  |
| 500 | 4.04 | 6.70 | 10.13 | 3.37 | 3.53 | 3.68 3.66 | 0.015 | 2.5 3.0 |
| 1000 | 4.11 | 6.74 | 10.15 | 3.37 | 3.53 | 3.66 | 0.29 | 3.5 |
| 2000 | 4.35 | 6.89 | 10.26 | 3.36 | 3.53 | 3.66 | 0.57 | 4.5 |
| 3000 | 4.71 | 7.13 | 10.43 | 3.35 | 3.52 | 3.66 | 0.85 | 5.5 |
| 5000 | 5.56 | 7.83 | 10.94 | 3.34 | 3.52 | 3.66 | 1.4 | 7.5 |
| 10000 | 7.51 10.16 | 9.98 | 12.86 | 3.31 | 3.49 | 3.64 | 2.8 | 12.1 |
| 20000 30000 | 10.16 | 13.54 | 17.08 | 3.28 | 3.46 | 3.61 | 5.6 | 20.5 |
| 30000 40000 | 12.19 13.90 | 16.15 | 20.42 | 3.26 | 3.44 | 3.59 | 8.4 | 28.0 |
| 40000 50000 | 13.90 15.41 | 18.34 | 23.14 | 3.28 | 3.43 | 3.58 | 11.2 | 35.0 |
| 50000 infin | 15.41 | 20.29 | 25.51 | 3.25 | 3.43 3.37 | 3.57 | 14.0 | 41.1 |
| infin |  |  |  | 3.21 | 3.37 | 3.50 |  |  |

Capacifance on 40-wire Ilnes
microforod per loop mile

In spece
On 40-wire line, dry
On 40-wire line, wet lopprox)

| 165 mil | 128 mil | 104 mil |
| :--- | :--- | :--- |
| 0.00898 | 0.00855 | 0.00822 |
| 0.00915 | 0.00871 | 0.00837 |
| 0.00928 | 0.0088 s | 0.00850 |

## Line constants of copper open-wire pairs

53 pairs CS (speciol glass with steel pin) insulators per mile
8-inch spacing
temperature $68^{\circ} \mathrm{F}$

| frequency trilocycles per second | resistance ohms per loop mile |  |  | Inductance millihenries per loop mile |  |  | leakance micromhos per loop mile: 165, 128, or 104 mil |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 165 mil | 128 mil | 104 mil | 165 mil | 128 mil | 104 mil | dry | wof |
| 0.0 | 4.02 | 6.68 | 10.12 | 3.11 | 3.27 | 3.40 |  |  |
| 1.0 | 4.11 | 6.74 | 10.15 | 3.10 | 3.26 | 3.40 | 0.052 | 1.75 |
| 2.0 | 4.35 | 6.89 | 10.28 | 3.10 | 3.26 | 3.40 | 0.052 | 1.75 |
| 3.0 | 4.71 | 7.13 | 10.43 | 3.09 | 3.26 | 3.40 |  |  |
| 5.0 | 5.56 | 7.83 | 10.94 | 3.08 | 3.25 | 3.40 | 0.220 | 3.40 |
| 10.0 | 7.51 | 9.98 | 12.86 | 3.04 | 3.23 | 3.38 | 0.408 | 5.14 |
| 20.0 | 10.16 | 13.54 | 17.08 | 3.02 | 3.20 | 3.35 | 0.748 | 8.06 |
| 50,0 | 15.41 | 20.29 | 25.51 | 2.99 | 3.16 | 3.31 | 1.69 | 15.9 |
| 100.0 200.0 | 21.30 | 27.90 | 34.90 | 2.98 | 3.15 | 3.29 | 3.12 | 27.6 |
| 200.0 500.0 | 29.77 | 38.77 | 48.25 | 2.97 | 3.14 | 3.28 | 3.2 | 27.6 |
| 500.0 | 46.45 | 60.30 | 74.65 | 2.96 | 3.13 | 3.27 |  |  |
| 1000.0 | 65.30 | 84.50 | 104.5 | 2.96 | 3.12 | 3.26 |  |  |
| infin |  |  |  | 2.95 | 3.11 | 3.24 |  |  |

Capacifance on 40-wire lines
microforad per loop mile

| In spoce Ino insulotorsi | $165 \mathrm{ml\mid}$ | 126 mil | 104 mil |
| :--- | :--- | :--- | :--- |
| On 40-wire Ine, dry | 0.00978 | 0.00928 | 0.00888 |
|  | 0.01003 | 0.00951 | 0.00912 |

Characterisfics of standard types of aerial copper wire telephone circuifs af 1000 cycles per second

| type of circuil | gaugeofwiros(mils) | $\left\|\begin{array}{c\|} \text { spac- } \\ \text { Ing } \\ \text { of } \\ \text { wires } \\ \text { (inches) } \end{array}\right\|$ | primary constants |  |  |  | propagation constant |  |  |  | Ine Impedance |  |  |  | wavelength miles | velocity miles par second | aftenuation -db <br> por <br> mile |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  | pola |  | ctangular |  | polar |  | etangular |  |  |  |  |
|  |  |  | $\stackrel{R}{\mathrm{ohms}^{2}}$ | L henries | $\underset{\mu}{C}$ | $\underset{\mu \mathrm{mho}}{\mathbf{G}}$ | mag. nitude | angle <br> dag |  | $\beta$ | mag-mitude | ongle dog $\qquad$ | $\underset{\text { ohms }}{R}$ | $\begin{gathered} \text { X } \\ \text { ohms } \\ \hline \end{gathered}$ |  |  |  |
| Non-Pole Pair Phys | 165 | 8 | 4.11 | .00311 | . 00996 | . 14 | . 0353 | 83.99 | . 00370 | . 0351 | 565 | 5.88 | 562 | 58 | 179.0 | 179,000 | . 0321 |
| Non-Pole Pair Side | 165 | 12 | 4.11 | . 00337 | . 00915 | . 29 | . 0352 | 84.36 | . 00346 | . 0350 | 612 | 5.35 | 610 | 57 | 179.5 | 179,500 | . 0300 |
| Pole Pair Side | 165 | 18 | 4.11 | . 00364 | . 00863 | . 29 | . 0355 | 84.75 | . 00325 | . 0353 | 653 | 5.00 | 651 | 57 | 178.0 | 178,000 | . 0282 |
| Non-Pole Pair Phan | 165 | 12 | 2.06 | . 00208 | . 01514 | . 58 | . 0355 | 85.34 | . 00288 | . 0354 | 373 | 4.30 | 372 | 28 | 177.5 | 177,500 | . 0250 |
| Non-Pole Pair Phys | 128 | 8 | 6.74 | . 00327 | . 00944 | . 14 | . 0358 | 80.85 | . 00569 | . 0353 | 603 | 8.97 | 596 | 94 | 178.0 | 178,000 | . 0494 |
| Non.Polo Pair Sido | 128 | 12 | 6.74 | . 00353 | . 00871 | . 29 | . 0356 | 81.39 | . 00533 | . 0352 | 650 | 8.32 | 843 | 94 | 178.5 | 178,500 | . 0462 |
| Pole Pair Side | 128 | 18 | 6.74 | . 00380 | . 00825 | . 29 | . 0358 | 81.95 | . 00502 | . 0355 | 693 | 7.72 | 686 | 93 | 177.0 | 177,000 | . 0436 |
| Non-Polo Pair Phan | 128 | 12 | 3.37 | . 00216 | . 01454 | . 58 | . 0357 | 82.84 | . 00445 | . 0355 | 401 | 6.73 | 398 | 47 | 177.0 | 177,000 | . 0386 |
| Non-Pole Pair Phys | 104 | 8 | 10.15 | . 00340 | . 00905 | . 14 | . 0367 | 77.22 | . 00811 | . 0358 | 644 | 12.63 | 629 | 141 | 175.5 | 175,500 | . 0704 |
| Non-Pole Pair Sido | 104 | 12 | 10.15 | . 00366 | . 00837 | . 29 | . 0363 | 77.93 | . 00760 | . 0355 | 692 | 11.75 | 677 | 141 | 177.0 | 177,000 | .0660 |
| Pole Pair Side | 104 | 18 | 10.15 | . 00393 | . 00797 | . 29 | . 0365 | 78.66 | . 00718 | . 0358 | 730 | 10.97 | 717 | 139 | 175.5 | 175,500 | . 0624 |
| Non-Pole Pair Pha | 104 | 12 | 5.08 | . 00223 | . 01409 | . 58 | . 0363 | 79,84 | . 00640 | . 0357 | 421 | 9.70 | 415 | 71 | 176.0 | 178,000 | . 0556 |

Notes: 1. All values are for dry weather condifions.
2. All capacitonce values assume a line carrying 40 wires.
3. Resistance values are for temperature of $20^{\circ} \mathrm{C} 168^{\circ} \mathrm{f}$.
4. DP IDouble Patticoat) Insulators assumed for all 12-inch and 18 -inch spaced wiros-CS (Special Glass with Sieal Pin) Insulators assumed for all 8 -inch spaced wiros.

Telephone fransmission line data continued

## Aftenuation of 12-inch spaced open-wire pairs

Tall and DP (double petticoat) insulators

| size wire wealher | aftenuation in $\mathrm{db}^{\text {per mill }}$ |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 165 mil |  |  | 128 mil |  | 104 mll |  |
|  | dry |  | wet | dry | wet | dry | wel |
| frequency |  |  |  |  |  |  |  |
|  | . 0127 |  | . 0279 | . 0163 | . 0361 | . 0198 | . 0444 |
| 100 | . 0231 |  | . 0320 | . 0318 | . 0427 | . 0402 | . 0535 |
| 500 | . 0288 |  | . 0367 | . 0445 | . 0530 | . 0620 | . 0715 |
| 1000 | . 0300 |  | . 0387 | . 0464 | . 0557 | . 0661 | . 0760 |
| 2000 | . 0326 |  | . 0431 | . 0486 | . 0598 | . 0686 | . 0804 |
| 3000 |  |  | . 0485 | . 0511 | . 0642 | . 0707 | . 0845 |
| 5000 | . 0439 |  | . 0598 | . 0573 | . 0748 | . 0757 | . 0938 |
| 7000 | . 051 |  | . 070 | . 084 | . 085 | . 082 | . 103 |
| 10000 | . 081 |  | . 085 | . 076 | . 102 | . 093 | . 120 |
| 15000 | . 076 |  | . 108 | . 094 | . 127 | . 111 | . 147 |
| 20000 | . 088 |  | . 127 | . 108 | . 150 | . 129 | . 173 |
| 30000 | .110 |  | . 161 | . 135 | . 188 | . 159 | . 216 |
| 40000 | . 130 |  | . 192 | . 158 | . 223 | . 185 | . 254 |
| 50000 | . 148 |  | . 220 | . 179 | . 253 | .209 | 287 |

CS (special gloss with steel pln) insulotors

| 20 | .0126 | .0252 | .0162 | .0326 | .0197 | .0402 |
| ---: | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| 100 | .0230 | .0303 | .0317 | .0406 | .0401 | .0509 |
| 500 | .0286 | .0348 | .0441 | .0510 | .0618 | .069 |
| 1000 | .0296 | .0364 | .0458 | .0532 | .0655 | .0735 |
| 2000 | .0318 | .0399 | .0475 | .0561 | .0676 | .0767 |
| 3000 | .0346 | .0437 | .0495 | .0593 | .0694 | .0797 |
| 5000 | .0412 | .0531 | .0547 | .0668 | .0731 | .0856 |
| 7000 | .048 | .061 | .062 | .075 | .078 | .093 |
| 10000 | .057 | .072 | .071 | .087 | .088 | .104 |
| 15000 | .068 | .087 | .086 | .105 | .104 | .123 |
| 20000 | .078 | .099 | .099 | .121 | .119 | .141 |
| 30000 | .111 | .121 | .120 | .146 | .145 | .171 |
| 40000 | .125 | .153 | .138 | .166 | .166 | .195 |
| 50000 |  | .154 | .184 | .185 | .215 |  |

Attenuation of 8-inch spaced open-wire pairs
CS Insulators


Line and propagation constants of 16- and 19-AWG toll cable loop mile basis non-loaded temperature $55^{\circ} \mathrm{F}$

| $\begin{gathered} \text { frequeney } \\ \text { ke } \\ \text { per sec } \end{gathered}$ | resistance ohms per mille | Inductance millihenries per mile | conductance umhe per mile | $\qquad$ | $\begin{aligned} & \text { aftenuation } \\ & \text { db } \\ & \text { per mile } \end{aligned}$ | phase shift radians per mille | characteristic impedence ohms |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 16-gauge |  |  |  |  |  |  |  |
| 1 | 40.1 | 1.097 | 1 | 0.0588 | 0.69 | 0.09 | 251-/215 |
| 2 | 40.3 | 1.095 | 2 | 0.0588 | 0.94 | 0.14 | 190-j141 |
| 3 | 40.4 | 1.094 | 4 | 0.0587 | 1.05 | 0.19 | 170-1108 |
| 5 | 40.7 | 1.092 | 8 | 0.0588 | 1.15 | 0.28 | 154-171 |
| 10 | 42.5 | 1.085 | 19 | 0.0587 | 1.30 | 0.54 | 142-j42 |
| 20 | 47.5 | 1.086 | 49 | 0.0585 | 1.54 | 1.01 | 137- 123 |
| 30 | 53.5 | 1.046 | 83 | 0.0584 | 1.77 | 1.49 | $135-117$ |
| 50 | 66.5 | 1.013 | 164 | 0.0582 | 2.25 | 2.43 | 133-j13 |
| 100 | 91.6 | 0.963 | 410 | 0.0580 | 3.30 | 4.71 | 129- $/ 9$ |
| 150 | 111.0 | 0.934 | 690 | 0.0578 | 4.17 | 6.94 | $127-17$ |
| 19 -gaue |  |  |  |  |  |  |  |
| 1 | 83.6 | 1.108 | 1 | 0.0609 | 1.05 | 0.132 | 345-/319 |
| 2 | 83.7 | 1.108 | 3 | 0.0609 | 1.44 | 0.190 | 254-/215 |
| 3 | 83.8 | 1.107 | 4 | 0.0609 | 1.73 | 0.249 | 215- 1170 |
| 5 | 84.0 | 1.106 | 9 | 0.0609 | 2.02 | 0.347 | 181- 1121 |
| 10 | 85.0 | 1.103 | 22 | 0.0608 | 2.43 | 0.584 | 153-j72 |
| 20 | 88.5 | 1.094 | 56 | 0.0607 | 2.77 | 1.07 | $141-141$ |
| 30 | 93.5 | 1.083 | 98 | 0.0506 | 3.02 | 1.56 | 137-129 |
| 50 | 105.4 | 1.062 | 193 | 0.0304 | 3.53 | 2.55 | 134- $/ 20$ |
| 100 | 136.0 | 1.016 | 484 | 0.0601 | 4.79 | 4.94 | 131-j13 |
| 150 | 164.4 | 0.985 | 830 | 0.0599 | 6.01 | 7.27 | 129-j10 |

Approximate characteristics of standard types of paper-insulated

|  |  |  | ioad coll per loe | nstants ection | constants ostumed to be distributed per loop mile |  |  |  | propagation polor |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| eauge <br> AWG | $\begin{gathered} \text { type } \\ \text { of } \\ \text { locding } \end{gathered}$ | colls milles | ohms | henries | $\begin{gathered} \text { R } \\ \text { ohms } \end{gathered}$ | $\stackrel{\mathrm{L}}{\text { henries }}$ | $\underset{\mu f}{\mathbf{C}}$ | $G$ mbhe | magnitude | $\begin{aligned} & \text { angle } \\ & \text { deg }+ \end{aligned}$ |


| side circulf |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 19 | N.L.S. |  | - | - | 85.8 | . 001 | . 062 | 1.5 | . 183 | 47.0 |
| 19 | H-31.S | 1.135 | 2.7 | . 031 | 88.2 | . 028 | . 052 | 1.5 | . 277 | 76.6 |
| 19 | H-44-S | 1.135 | 4.1 | . 043 | 89.4 | . 039 | . 062 | 1.5 | . 319 | 79.9 |
| 19 | H-88-S | 1.135 | 7.3 | . 088 | 92.2 | . 078 | . 062 | 1.5 | . 441 | 84.6 |
| 19 | H.172-S | 1.135 | 13.0 | . 170 | 97.3 | . 151 | . 062 | 1.5 | . 810 | 87.0 |
| 19 | B.88-S | 0.568 | 7.3 | . 088 | 98.7 | . 156 | . 052 | 1.5 | . 620 | 87.0 |
| 16 | N.L.S. | - | - | - | 42.1 | . 001 | . 052 | 1.5 | . 129 | 49.1 |
| 16 | H.31.S | 1.135 | 2.7 | . 031 | 44.5 | . 028 | . 052 | 1.5 | . 266 | 82.8 |
| 16 | H-44.S | 1.135 | 4.1 | . 043 | 45.7 | . 039 | . 052 | 1.5 | . 315 | 84.6 |
| 16 | H-88-S | 1.135 | 7.3 | . 088 | 48.5 | . 078 | . 062 | 1.5 | . 438 | 87.6 |
| 16 | H.172-S | 1.135 | 13.0 | . 170 | 53.6 | . 151 | . 062 | 1.5 | . 608 | 88.3 |
| 16 | B-88.5 | 0.568 | 7.3 | . 088 | 54.9 | . 156 | . 062 | 1.5 | . 618 | 88.3 |
| 13 | N.L.S. |  |  | . | 21.9 | . 001 | . 062 | 1.5 | . 094 | 52.9 |
| phanfom circuil |  |  |  |  |  |  |  |  |  |  |
| 19 | N.L.P. | - | - | - | 42.9 | . 0007 | . 100 | 2.4 | . 165 | 47.8 |
| 19 | H-18.P | 1.135 | 1.4 | . 018 | 44.1 | . 017 | .100 | 2.4 | . 270 | 78.7 |
| 19 | H-25.P | 1.135 | 2.1 | . 025 | 44.7 | . 023 | .100 | 2.4 | . 308 | 81.3 |
| 19 | H. $50 . \mathrm{P}$ | 1.135 | 3.7 | . 050 | 46.2 | . 045 | .100 | 2.4 | . 424 | 85.3 |
| 19 | H.63.P | 1.135 | 6.1 | . 063 | 48.3 | . 056 | . 100 | 2.4 | . 472 | 86.0 |
| 19 | 8-50.P | 0.568 | 3.7 | . 050 | 49.4 | . 089 | .100 | 2.4 | . 594 | - 87.4 |
| 16 | N.L.P. | - | - | -18 | 21.0 | . 0007 | . 100 | 2.4 | . 116 | 50.0 |
| 16 | H.18.P | 1.135 | 1.4 | . 018 | 22.2 | . 017 | .100 | 2.4 | . 262 | 84.0 |
| 16 | H.25-P | 1.135 | 2.1 | . 025 | 22.8 | . 023 | .100 | 2.4 | . 303 | 85.4 |
| 16 | H.50-P | 1.135 | 3.7 | . 050 | 24.3 | . 045 | . 100 | 2.4 | . 422 | 87.4 87.7 |
| 16 | H-63-P | 1.135 | 6.1 | . 063 | 26.4 | . 056 | . 100 | 2.4 | . 471 | 87.7 88.5 |
| 16 | B-50-P | 0.568 | 3.7 | . 050 | 27.5 10.9 | . 089 | . 100 | 2.4 2.4 | . 593 | 88.5 55.1 |
| 13 | N.L.P. |  |  |  | 10.9 | . 0007 | . 100 | 2.4 | . 086 | 55.1 |
| physical circuif |  |  |  |  |  |  |  |  |  |  |
| 16 | B-22 | 0.568 | 1.25 | . 022 | 143.1 | . 040 | . 062 | 11.5 | . 315 | 85.0 |
| - The | ers H and | indicate | ding | spacing | of 6000 | d 3000 | 1, respe | ctivoly. |  |  |

## Telephone transmission line dafa continued

## Line constants of shielded 16 -gauge spiral-four toll-entrance cable

loop mile basis non-loaded temperature $70^{\circ} \mathrm{F}$

| frequency ke por sec | nesistance ohms per mile | inductonce mh per mile | sonductance umho per mile | copocitance $\mu^{\text {ti }}$ per mile | aftenwalion db per mile |
| :---: | :---: | :---: | :---: | :---: | :---: |
| tide clrcuit |  |  |  |  |  |
| 0.4 | 43.5 | 1.913 | 0.02 | 0.0247 | 0.92 |
| 0.6 | 43.5 | 1.907 | 0.04 | 0.0247 | 0.93 |
| 0.8 | 43.6 | 1.901 | 0.06 | 0.0247 | 0.93 |
| 1.0 | 43.9 | 1.891 | 0.08 | 0.0247 | 0.94 |
| 2 | 44.2 | 1.857 | 0.20 | 0.0247 | 0.95 |
| 3 | 45.2 | 1.821 | 0.32 | 0.0247 | 0.96 |
| 5 | 49.0 | 1.753 | 0.53 | 0.0247 | 0.97 |
| 10 | 55.1 | 1.626 | 1.11 | 0.0247 | 1.00 |
| 20 | 61.6 | 1.539 | 2.49 | 0.0247 | 1.06 |
| 30 | 66.1 | 1.507 | 3.77 | 0.0247 | 1.15 |
| 40 | 71.0 | 1.490 | 5.50 | 0.0247 | 1.26 |
| 60 | 81.5 | 1.467 | 8.80 | 0.6247 | 1.44 |
| 80 | 90.1 | 1.450 | 12.2 | 0.0247 | 1.60 |
| 100 | 97.8 | 1.438 | 15.81 | 0.0247 | 1.77 |
| 120 | 104.9 | 1.429 | 19.6 | 0.0247 | 1.90 |
| 140 | 111.0 | 1.421 | 23.3 | 0.0247 | 2.03 |
| 200 | 127.3 | 1.411 | 35.1 | 0.0246 | 2.35 |
| 250 | 137.0 | 1.408 | 46.0 | 0.0246 | - |
| 300 | 149.5 | 1.406 | 56.5 | 0.0246 | - |
| 350 | 159.9 | 1.405 | 67.8 | 0.0246 | - |

Chorocteristic lmpedance of this cable at 140 kiloeyclas approximately 240 ohms.
For a description and illustration of this type coble see Kendall and Afel, "A Twelve-Channal Caprier Telephone System for Open-Wire Lines," B.S.T」., Janvary 1939, pp. 129-131.

## foll telephone cable circuits at 1000 cycles per second



Approximate characteristics of standard types of paper-insulated exchange felephone cable circuits

| wire gauge AWG | code no | $\begin{gathered} \text { type } \\ \text { of } \\ \text { looding } \\ \hline \end{gathered}$ | loop mille constonts |  | propagotion constant |  |  |  | mid-section characteristic impedonce |  |  |  | wave length mifes | ```volocity miles per second``` | $\begin{aligned} & \text { cut- } \\ & \text { off } \\ & \text { frea } \end{aligned}$ | $\begin{aligned} & \text { difen } \\ & \text { db } \\ & \text { per } \\ & \text { mill } \\ & \hline \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | $\mathrm{C}_{\mu} \mathrm{F}$ | in | mag | angle <br> (deg) | $\boldsymbol{\alpha}$ | $\beta$ | mog | angle <br> (deg) | $Z_{01}$ | $\mathbf{Z}_{02}$ |  |  |  |  |
| 26 | BST | NL | .083 .069 | 1.6 1.6 | . 439 | 45.30 | . 307 | . 310 | $\begin{array}{r} 910 \\ 1007 \end{array}$ | 44.5 | 719 | 706 | 20.4 | 20,400 | - | $\begin{aligned} & 2.9 \\ & 2.67 \end{aligned}$ |
| 24 | DSM | NL | . 085 | 1.9 1.9 |  |  |  |  | 725 |  |  |  |  |  | - | 2.3 2.15 |
|  | ASM | NL | . 075 | 1.9 | . 355 | 45.53 | 247 .151 | . 251 | 778 987 | 44.2 23.7 | 558 904 | 543 396 | 25.0 14.9 | 25,000 14,900 | - | 2.15 |
|  |  | M88 | . 075 | 1.9 | . 448 | 70.25 | . 151 | . 421 | 987 | 23.7 | 904 | 396 | 14.9 | $14,900$ | 3100 | 1.31 |
|  |  | H88 | . 075 | 1.9 | . 512 | 75.28 | . 130 | . 495 | 1160 | 14.6 | 1122 | 292 | 12.7 | 12,700 | 3700 | 1.13 |
|  |  | B88 | . 075 | 1.9 | . 684 | 81.70 | . 099 | . 677 | 1532 | 8.1 | 1515 | 215 | 9.3 | 9,270 | 5300 | 0.86 |
| 22 | CSA | NL | . 083 | 2.1 | . 297 | 45.92 | . 207 | . 213 | 576 | 43.8 | 416 | 399 | 29.4 | 29,400 | - | 1.80 |
|  |  | M88 | . 083 | 2.1 | . 447 | 76.27 | . 106 | . 434 | 905 | 13.7 | 880 | 214 | 14.5 | 14,500 | 2900 | 0.92 |
|  |  | H88 | . 083 | 2.1 | . 526 | 80.11 | . 0904 | . 519 | 1051 | 9.7 | 1040 | 177 | 12.1 | 12,100 | 3500 | 0.79 |
|  |  | H135 | . 083 | 2.1 | . 644 | 83.50 | . 0729 | . 640 | 1306 | 6.3 | 1300 | 144 | 9.8 • | 9,800 | 2800 | 0.63 |
|  |  | B88 | . 083 | 2.1 | . 718 | 84.50 | . 0689 | . 718 | 1420 | 5.3 | 1410 | 130 | 8.75 | 8,750 | 5000 | 0.60 |
|  |  | B135 | . 083 | 2.1 | . 890 | 86.50 | . 0549 | . 890 | 1765 | 3.3 | 1770 | 102 | 7.05 | 7,050 | 4000 | 0.48 |
| 19 | CNB | NL | . 085 | 1.6 | - | - | - | - | 400 | - | - | - | - | - | - | 1.23 |
|  | DNB | NL | . 066 | 1.6 | . 188 | 47.00 | . 128 | . 138 | 453 | 42.8 | 333 | 308 | 45.7 | 45,700 | - | 1.12 |
|  |  | M88 | . 066 | 1.6 | . 383 | 82.42 | . 0505 | . 380 | 950 | 8.9 | 939 | 146 | 16.6 | 16,600 | 3200 | 0.44 |
|  |  | H88 | . 066 | 1.6 | . 459 | 84.60 | . 0432 | . 459 | 1137 | 5.2 | 1130 | 103 | 13.7 | 13,700 | 3900 | 0.38 |
|  |  | H135 | . 066 | 1.6 | . 569 | 86.53 | . 0345 | . 570 | 1413 | 4.0 | 1410 | 99 | 11.0 | 11,000 | 3200 | 0.30 |
|  |  | H175 | . 066 | 1.6 | . 651 | 87.23 | . 0315 | . 651 | 1643 | 3.3 | 1640 | 95 | 9.7 | 9,700 | 2800 | 0.27 |
|  |  | B88 | . 066 | 1.6 | . 641 | 86.94 | . 0342 | . 641 | 1565 | 2.8 | 1560 | 77 | 9.8 | 9,800 | 5500 | 0.30 |
| 16 | NH | NL | . 064 | 1.5 | . 133 | 49.10 | . 0868 | . 1004 | 320 | 40.6 | 243 | 208 | 62.6 | 62,600 | - | 0.76 |
|  |  | M88 | . 064 | 1.5 | . 377 | 85.88 | . 0271 | . 377 | 937 | 4.3 | 934 | 76 | 16.7 | 16,700 | 3200 | 0.24 |
|  |  | H88 | . 064 | 1.5 | . 458 | 87.14 | . 0238 | . 458 | 1130 | 2.8 | 1130 | 55 | 13.7 | 13,700 | 3900 | 0.21 |

In the third column of the above table the letters $M, H$, and $B$ indicate loading coll apacings of 9000 feet, 6000 feet, and 3000 feet, respectivaly, and the figures show the inductance of the loading colls used.

Frequency allocation chart for type J and $\mathbf{K}$ carrier systems
Type J


Frequency allocation chant for carrier systems
Carrier telephone



frequency in kilocycles per second

## Noise and noise measurement

## Definitions

The following definitions are based upon those given in the Proceedings of the tenth Plenary Meeting (1934) of the Comite Consultatif International Tèléphonique (C.C.I.F.).

Note: The unit in which noise is expressed in many of the European countries differs from the two American standards, the noise unit and the db above reference noise. The European unit is referred to as the psophometric electromotive force.

Noise: Is a sound which tends to interfere with a correct perception of vocal sounds, desired to be heard in the course of a telephone conversation.

It is customary to distinguish between:

1. Room noise: Present in that part of the room where the telephone apparatus is used.
2. Frying noise (transmitter noise): Produced by the microphone, manifest even when conversation is not taking place.
3. Line noise: All noise electrically transmitted by the circuit, other than room noise and frying noise.

## Psophomełric electromotive force

In the case of a complete telephone connection the interference with a telephone conversation produced by extraneous currents may be compared with the interference which would be caused by a parasitic sinusoidal current of 800 cycles per second. The strength of the latter current, when the interference is the same in both cases, can be determined.

If the receiver used has a resistance of 600 ohms and o negligible reactance lif necessary it should be connected through a suitable transformerl, the psophometric electromotive force at the end of a circuit is defined as twice the voltage at 800 cycles per second, measured at the terminals of the receiver under the conditions described.

The psophometric electromotive force is therefore the electromotive force of a source having an internal resistance of 600 ohms and zero internal reactance which, when connected directly to a standard receiver of 600 ohms resistance and zero reactance, produces the same sinusoidal current at 800 cycles per second as in the case with the arrangements indicated above.

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## Noise and noise measurement continued

An instrument known as the psophometer has been designed. When connected directly across the terminals of the 600 -ohm receiver, it gives a reading of half of the psophometric electromotive force for the particular case considered.

In a general way, the term psophometric voltage between any two points refers to the reading on the instrument when connected to these two points.
If, instead of a complete connection, only a section thereof is under consideration, the psophometric electromotive force with respect to the end of that section is defined as twice the psophometric voltage measured at the terminals of a pure resistance of 600 ohms, connected at the end of the section, if necessary through a suitable transformer.
The C. C. I. F. has published a Specification for a psophometer which is included in Volume II of the Proceedings of the Tenth Plenary Meeting in 1934. An important part of this psophometer is a filter network associated with the measuring circuit whose function is to weight each frequency in accordance with its interference value relative to a frequency of 800 cycles.

## Noise levels

The amount of noise found on different circuits, and even on the same circuit at different times, varies through quite wide limits. Further, there is no definite agreement as to what constitutes a quiet circuit, a noisy circuit, etc. The following values should therefore be regarded merely as a rough indication of the general levels which may be encountered under the different conditions:

| Open-wire circuit | db above <br> ref noise |
| :--- | :---: |
| Quiet | 20 |
| Average | 35 |
| Noisy | 50 |
| Cable ciraif |  |
| Quiet | 15 |
| Average | 25 |
| Noisy | 40 |

## Relationship of European and American noise units

The psophometric emf can be related to the American units: the noise unit and the decibel above reference noise.

The following chart shows this relationship together with correction factors for psophometric measurements on circuits of impedance other than 600 ohms.

## Relationship of European and American units



## Teıegraph facilifies

|  | spese of usual types <br> frequency <br> cycles |  |
| :--- | :---: | ---: |
| Grounded wire | 75 | bauds |
| Simplex (telephone) | 50 | 150 |
| Composite | 15 | 100 |
| Metallic telegraph | 85 | 30 |
| Carrier channel |  | 170 |
| Narrow band | 40 | 80 |
| Wide band | 75 | 150 |

## Telegraph prinfer systems

Speed depends on two factors: 1. Code used, and 2. frequency handling capacity of transmission facilities. One (1) word $=5$ letters and 1 space.

## Frequency of printing telegraph systems in cycles per second

Let
$S=$ number of units in code (plus allowance for synchronizing)
$N=$ number of channels
$W=$ revolutions per second
$=\frac{\text { words per minute } X \text { characters per transmitted word }}{60}$
(1 word is assumed to consist of 5 letters and 1 space, or 6 characters.)
$f=$ frequency in cycles per second $f=\frac{1}{3}$ SNW

## Examples

1. Three-channel multiplex operating at 60 words per minute, 5 -unit code.
$f=\frac{1}{2} \times 5 \times 3 \times \frac{60 \times 6}{60}=45$ cycles or 90 bauds
2. Single-printer circuit operating at 60 words per minute, 5 -unit code + $2 \frac{1}{2}$ units for synchronizing.
$f=\frac{1}{2} \times 7 \frac{1}{2} \times 1 \times \frac{60 \times 6}{60}=22 \frac{1}{2}$ cycles or 45 bauds
3. Two-channel Baudot operating at 50 words per minute, 5 -unit code + 2 units for synchronizing.
$f=\frac{1}{2}(5+2) \times 2 \times \frac{50 \times 6}{60}=35$ cycles or 70 bauds

## Comparison of telegraph codes


Hughes

| Murray Autamatic |  |  |
| :---: | :---: | :---: |
| Baudot |  | Add 2 units to each chonnel for 2 -channel and 1 unit ta each character far 4 -channel operation. These conditions allaw for synchronization and refardation. |



Cook $\stackrel{P}{\text { P }}$

Multiple
P A R $\quad \mathrm{B}$ upace



|  | P | A | R | 1 | S | spoee |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| RCA |  |  |  |  |  | cr |

## Radio frequency transmission lines

## Formulas for uniform transmission lines losses neglected

$$
\begin{aligned}
& Z_{0}=\sqrt{\frac{L}{C}} \\
& L=1016 \sqrt{\epsilon} Z_{0} \\
& C=1016 \frac{\sqrt{\epsilon}}{Z_{0}} \\
& \frac{V}{c}=\frac{1}{\sqrt{\epsilon}} \\
& Z_{s}=Z_{0} \frac{Z_{r}+j Z_{0} \tan l^{\circ}}{Z_{0}+j Z_{r} \tan l^{\circ}} \\
& Z_{s}=\frac{Z_{0}^{2}}{Z_{r} \quad \text { for } l^{\circ}=90^{\circ} \text { (quarter wave) }} \\
& Z_{s s}=+j Z_{0} \tan l^{\circ} \\
& Z_{s 0}=-\frac{j Z_{0}}{\tan l^{\circ}} \\
& l^{\circ}=360 \frac{l}{\lambda} \\
& \lambda=\lambda_{0}\left(\frac{V}{c}\right)
\end{aligned}
$$

where
$L=$ inductance of transmission line in micromicrohenries per foot
$C=$ capacitance of transmission line in micromicrofarads per foot
$V=$ velocity of propagation in transmission line
$c=$ velocity of propagation in free space $\}$ same units
$Z_{s}=$ sending end impedance of transmission line in ohms
$Z_{e}=$ surge impedance of transmission line in ohms
$Z_{r}=$ terminating impedance of transmission line in ohms
$I^{\circ}=$ length of line in electrical degrees
$1=$ length of line
$\lambda=$ wavelength in transmission line same units
$\lambda_{0}=$ wavelength in free space
$\boldsymbol{\epsilon}=$ dielectric constant of transmission line medium
$=1$ for air
$Z_{s t}=$ sending end impedance (ohms) of transmission line shorted at far end
$Z_{* 0}=$ sending end impedance (ohmsl of transmission line open at far end

Surge impedance of uniform lines- $\mathbf{0}$ to 210 ohms


Surge impedance of uniform lines- $\mathbf{0}$ to 700 ohms



## Transmission line data

| Pype of line | characteristic Impodance |
| :---: | :---: |
| A single coaxial line | $\begin{aligned} Z_{0} & =\frac{138}{\sqrt{\epsilon}} \log _{10} \frac{D}{d} \\ \epsilon & =\text { dielectric constant } \\ & =1 \text { in air } \end{aligned}$ |
| B balanced shielded line | $\begin{aligned} & \text { for } D \gg d, h \gg d \\ Z_{o} & \cong \frac{276}{\sqrt{\epsilon}} \log _{10}\left[2 v \frac{1-\sigma^{2}}{1+\sigma^{2}}\right] \\ \sigma & =\frac{h}{D} \\ v & =\frac{h}{d} \end{aligned}$ |
| C beads-dielectric $\epsilon_{1}$ | for cases ( $A$ ) and ( $B$ ) <br> if ceramic beads are used at frequent intervals-call new surge impedance $Z_{0}{ }^{\prime}$ $Z_{0}^{\prime}=\frac{Z_{0}}{\sqrt{1+\left(\frac{\epsilon_{1}}{\epsilon}-1\right) \frac{W}{S}}}$ |
| D open twa-wire line | $\begin{aligned} Z_{0} & =120 \cosh ^{-1} \frac{D}{d} \\ & \cong 276 \log _{10} \frac{2 D}{d} \end{aligned}$ |


| type of line | characteristic impedance |
| :---: | :---: |
|  | $Z_{0}=69 \log _{10}\left[\frac{4 h}{d} \sqrt{1+\left(\frac{2 h}{D}\right)^{2}}\right]$ |
|  | $Z_{0}=276 \log _{10}\left[\frac{4 h}{d \sqrt{1+\left(\frac{2 h}{D}\right)^{2}}}\right]$ |
|  | $Z_{0}=138 \log _{10} \frac{4 h}{d}$ |
|  | $Z_{0}=138 \log _{10} \frac{D}{d}\left[1.078-0.078\left(\frac{d}{D}\right)^{2}\right]$ |
|  | $Z_{0}=138 \log _{10} \frac{2 D_{2}}{d \sqrt{1+\left(\frac{D_{2}}{D_{1}}\right)^{2}}}$ |
|  | $l \ggg_{w}$ $\mathrm{Z}_{0} \cong 377 \frac{\mathrm{w}}{\mathrm{l}}$ |

Transmission line aftenuation due to load mismatch


Ao normol line offenuotion in decibels

- A $-A_{0}$ oftenuotion in decibels due to load mismotch


## Impedance matching with shorted słub



Impedance matching with open stub


Impedance matching with coupled section


Defuning from resonance for a particular type of section


$A=$ coupled section-two 0.75 -inch diameter copper tubes, coplanar with line
B = transmission line-two 0.162-inch diameter wires
C = alternative positions of shorting bar for impedance matching

D = position of shorting bar for maximum current in section conductors

| class of cables |  | ArmyNavy type number | inner conductor | dielec maferial (1) | nominal diam of dialectric (in) | shiolding braid | protective covering | nominal overall diam (in) | woight $16 / 4$ | nominal impedance | nominal caparitance $\mu \mu \mathrm{f} / \mathrm{h}$ | moximum operating voliage FMIT | remarks |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{aligned} & 50-55 \\ & \text { ohms } \end{aligned}$ | Single braid | RG-58/U | 20 AWG copper | $\wedge$ | 0.116 | Tinned Copper | Vinyl | 0.195 | 0.025 | 53.5 | 28.5 | 1,900 | General purpose small size floxible cable |
|  |  | RG-8/U | $\begin{aligned} & 7 / 21 \text { AWG } \\ & \text { copper } \end{aligned}$ | $\wedge$ | 0.285 | Copper | Vinyl | 0.405 | 0,106 | 52.0 | 29.5 | 4,200 | General purpose medium size flexible cable |
|  |  | RG-10/U | 7/21 AWG copper | $\wedge$ | 0.285 | Copper | Vinyl Inoncontaminatingl armor | $\begin{aligned} & \text { lmax } \\ & 0.475 \end{aligned}$ | 0.146 | 52.0 | 29.5 | 4,000 | Same as RG-8/U armored for noval equipment |
|  |  | RG-17/U | $\begin{aligned} & 0.188 \\ & \text { copper } \end{aligned}$ | A | 0.680 | Copper | Vinyl Inon-contaminating! | 0.870 | 0.460 | 52.0 | 29.5 | 11,000 | large high power low aftenuation transmission cable |
|  |  | RG-18/U | 0.188 copper | A | 0.680 | Copper | Vinyl inon. contaminating) armor | $\begin{aligned} & (\max ) \\ & 0.945 \end{aligned}$ | 0.585 | 52.0 | 29.5 | 11,000 | Same as RG-17/U armored for noval equip. ment |
|  |  | RG-19/U | $\begin{aligned} & 0.250 \\ & \text { copper } \end{aligned}$ | A | 0.910 | Copper | Vinyl Inon contaminotingl | 0.120 | 0.740 | 52.0 | 29.5 | 14,000 | Very lorge high power low aftenuation fransmission cab'e |
|  |  | RG-20/U | $\begin{aligned} & 0.250 \\ & \text { copper } \end{aligned}$ | A | 0.910 | Copper | Vinyl Inoncontaminatingl armor | $\begin{aligned} & \text { Imax) } \\ & 1.19 .5 \end{aligned}$ | 0.925 | 52.0 | 29.5 | 14,000 | Same ar RG-19/U armored for naval equipment |
|  | Double braid | RG-55/U | 20AWG copper | $\wedge$ | 0.116 | Tinned copper | Polyothylene | $\begin{aligned} & \text { Imax) } \\ & 0.206 \end{aligned}$ | 0.034 | 53.5 | 28.5 | 1,900 | Small size flexible cable |
|  |  | RG-5/U | 16 AWG copper | $\wedge$ | 0.185 | Copper | Vinyl | 0.332 | 0.087 | 53.5 | 28.5 | 2,000 | Smoll microwove cable |
|  |  | RG-9/U | 7/21 AWG silvered copper | $\wedge$ | 0.280 | Inner-silver coated copper. Outer-copper | Vinyl Inon-contami natingl | 0.420 | 0.150 | 51.0 | 30.0 | 4,000 | Medium size, low levol circult cable |

## Noses:

1. Diolectric materiols

A Stabilized polyothylene
C Synthetic rubber compound
D layer of synthetic rubber dielectric berween thin layers of conducting rubber

| class of cables |  | ArmyNavy type number | Inner conducter | dieles material (1) | nominal diam of dielectric (in) | $\begin{aligned} & \text { shielding } \\ & \text { bralid } \end{aligned}$ | protective covering | $\begin{aligned} & \text { nominal } \\ & \text { overoll } \\ & \text { diam } \\ & \text { (in) } \end{aligned}$ | weight <br> /b/f | $\begin{aligned} & \text { nominul } \\ & \text { imped- } \\ & \text { ance } \\ & \text { ohms } \end{aligned}$ | $\begin{aligned} & \text { nominal } \\ & \text { capaci- } \\ & \text { fance } \\ & \mu \mu / / h \\ & \hline \end{aligned}$ | $\left\|\begin{array}{c} \text { maximum } \\ \text { operafing } \\ \text { voltage } \\ \text { rms } \end{array}\right\|$ | remarks |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | RG-14/U | 10 AWG coppor | A | 0.370 | Coppar | Vinyl Inon-contaml. natingl | 0.545 | 0.216 | 52.0 | 29.5 | 5,500 | General purpose samiflexible power transmission cable |
|  |  | RG-74/U | 10 AWG copper | A | 0.370 | Copper | Vinyl Inoncontaminatingl armor | 0.615 | 0.310 | 52.0 | 29.5 | 5,500 | Same os RG-14/U ar. mored for noval equip. ment |
| $\begin{aligned} & 70-80 \\ & \text { ohms } \end{aligned}$ | Single braid | RG-59/U | 22 AWG copperweld | A | 0.146 | Copper | Vinyl | 0.242 | 0.032 | 73.0 | 21.0 | 2,300 | General purpose small size video coble |
|  |  | RG-11/U | $\begin{aligned} & 7 / 26 \text { AWG } \\ & \text { tinned } \\ & \text { copper } \end{aligned}$ | A | 0.285 | Copper | Vinyl | 0.405 | 0.096 | 75.0 | 20.5 | 4,000 | Medium sizo, fexible video and communication sable |
|  |  | RG-12/U | $\begin{aligned} & 7 / 26 \text { AWG } \\ & \text { tinned } \\ & \text { coppor } \end{aligned}$ | A | 0.285 | Copper | Vinyl inoncontaminatingl armor | 0.475 | 0.141 | 75.0 | 20.5 | 4,000 | Some as RG.11/U armored for noval equip. ment |
|  | Double braid | RG-6/U | 21 AWG copperweld | A | 0.185 | Inner-silvar coated copper. Outor-copper | Vinyl Inon-contaminatingl | 0.332 | 0.682 | 76.0 | 20.0 | 2,700 | Small size video and I.F cable |
|  |  | RG-13/U | 7/26 AWG finned copper | A | 0.280 | Copper | Vinyl | 0.420 | 0.126 | 74.0 | 20.5 | 4,000 | I.F coble |
| Cables of special charac. toristics | Twin conductor | RG-22/U | $\begin{aligned} & 2 \text { Cond. } \\ & 7 / 18 \text { AWG } \\ & \text { copper } \end{aligned}$ | A | 0.285 | $\begin{aligned} & \text { Single-finned } \\ & \text { copper } \end{aligned}$ | Vinyl | 0.405 | 0.107 | 95.0 | 16.0 | 1,000 | Smail size twin conductor cable |
|  |  | RG-57/U | 2 Cond. <br> 7/21 AWG <br> copper | A | 0.472 | Single-finned coppor | Vinyl | 0.625 | 0.225 | 95.0 | 16.0 | 3,000 | Large size twin conductor cable |
|  | High attonuation | RG-21/U | 16 AWG resistance wire | A | 0.185 | Inner-silvar coated copper. Outer-copper | Vinyl inon-contami. nating) | 0.332 | 0.087 | 53.0 | 29.0 | 2,700 | Special attenuating cable with small temperature cosfficient of aftonuation |
|  | High imped. ance | RG-65/U | No. 32 Formex $F$ helix diam 0.128 in . | A | 0.285 | $\begin{aligned} & \text { Single-cop. } \\ & \text { por } \end{aligned}$ | Vinyl | 0.405 | $0.0 \%$ | 950 | 44.0 | 1,000 | High Impedance video cable. High delay |


| class of cables |  | ArmyNovy type number | Inner conductor | dielec maferial (1) | nominal diam of dielectric (in) | shielding braid | protective covering | $\begin{aligned} & \text { nominal } \\ & \text { overall } \\ & \text { dlam } \\ & \text { (ln) } \end{aligned}$ | weight lb/ 1 | nominal imped. ance ohms | nominal capacitance $\mu \mu \boldsymbol{f} / \mathrm{h}$ | $\begin{gathered} \text { maximum } \\ \text { operating } \\ \text { veltoge } \\ \text { rms } \end{gathered}$ | remorlas |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Low copocitance | Single braid | RG-62/U | 22 AWG copperweld | A | 0.146 | Copper | Vixyl | 0.242 | 0.0382 | 93.0 | $\begin{gathered} 13.5 \\ \max 14.5 \end{gathered}$ | 750 | Smoll size low copocitance alr-spoced coble |
|  |  | RG-63/U | 22 AWG copperweld | A | 0.285 | Copper | Vinyl | 0.405 | 0.0832 | 125 | $\begin{gathered} 10.0 \\ \max 11.0 \end{gathered}$ | 1,000 | Medium size low copacltance air-spoced coble |
|  | Double brald | RG-71/U | 22 AWG copperweld | A | 0.146 | Inner-plain copper. Outer -linnedcopper | Polyethy ene | 0.250 | 0.0457 | 93.0 | $\begin{gathered} 13.5 \\ \max 14.5 \end{gathered}$ | 750 | Smoll size low capocitance air-spoced cable for I.f purposes |
| Pulse appli. cations | Single brald | RG-26/U | $19 / \mathrm{C} .0117$ <br> tinned copper | D | ${ }_{0.308}^{20}$ | Tinned copper | Synthetic rubber and armor | $\begin{aligned} & \text { (mox) } \\ & 0.525 \end{aligned}$ | 0.189 | 48.0 | 50.0 | $\begin{array}{r} 8,000 \\ \text { (peok) } \end{array}$ | Medium slze pulse cable armored for noval equip. ment |
|  |  | RG-27/U' | 19/0.0185 tinned copper | D | ${ }_{0.455}^{(2)}$ | Single-linned copper | Vinyl and armor | $\begin{aligned} & \text { (max }) \\ & 0.675 \end{aligned}$ | 0.304 | 48.0 | 50.0 | $\begin{aligned} & \text { 15,000 } \\ & \text { 〔peak] } \end{aligned}$ | Large slize pulse cable armored for noval equip. ment |
|  | Double brald | RG-64/U. | $19 / 0.0117$ <br> Hinned copper | D | $\stackrel{(2)}{0.308}$ | Tinned copper | Neoprene | 0.495 | 0.205 | 48.0 | 50.0 | $\begin{array}{r} 8,000 \\ \text { (peok) } \end{array}$ | Medium size puise cable |
|  |  | RG-25/U | $19 / 0.0117$ <br> tinned copper | D | $\begin{gathered} { }^{122} \\ 0.308 \end{gathered}$ | Tinned copper | Neoprene | 0.565 | 0.205 | 48.0 | 50.0 | $\begin{array}{r} 8,000 \\ \text { tpeakJ } \end{array}$ | Special twisting pulse cable for noval equipment |
|  |  | RG-28/U | 19/0.0185 finned copper | D | $\begin{gathered} 127 \\ 0.455 \end{gathered}$ | Inner-Hnned copper. Outer -golvanized steel | Synthetic rubber | 0.805 | 0.370 | 48.0 | 50.0 | $\begin{aligned} & 15,000 \\ & \text { [peok] } \end{aligned}$ | Large size pulso coble |
| Twisting applicafion | Single braid | RG-41/U | 16/30 AWG tinned copper | C | 0.250 | Tinned copper | Neoprene | 0.425 | 0.150 | 67.5 | 27.0 | 3,000 | Special twist cable |

Notes:

1. Dielectric materiols

A Siabilized polyethylene
C Synthetic rubber compound
C Synthetic rubber compound
2. This value is the diamater over the outer loyer of conducting rubber.

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## Attenuation of standard r-f cables vs frequency


frequency in megacycles

The above chart refers to cables listed in the Army-Navy standard list of radio-frequency cables on pages 201, 202, and 203. For an explanation of the letters accompanying the curves, see the table below. Each letter refers to one or more A-N standard cables. The number following the letter in the table is the numerical part of the RG- $U$ number as listed under "Army-Navy type number" in the third column of the preceding list.

RC-number

| A $55 / U$ | D | $5 / U$ | F $10 / U$ | $I 63 / U$ | M $17 / U$ | $O 26 / U$ |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| A $58 / U$ | D | $6 / U$ | G $11 / U$ | $J 65 / U$ | M $18 / U$ | $O 64 / U$ |
| B $59 / U$ | E $21 / U$ | G $12 / U$ | K $14 / U$ | N $19 / U$ | P $27 / U$ |  |
| C $62 / U$ | F $8 / U$ | G $13 / U$ | K $74 / U$ | N $20 / U$ | P $28 / U$ |  |
| C $71 / U$ | F $9 / U$ | H $22 / U$ | L $57 / U$ | O $25 / U$ | Q $4 / U$ |  |

## RADIO FREQUENCY TRANSMISSION LINES <br> 205

## Length of transmission line



This chart gives the actual length of line in centimeters and inches when given the length in electrical degrees and the frequency provided the velocity of propagation on the transmission line is equal to that in free space. The length is given on the L scale intersection by a line between $\boldsymbol{\lambda}$ and $I^{\circ}$ where $I^{\circ}=\frac{360 \mathrm{l} \text { in centimeters }}{\lambda \text { in centimeters }}$

Exomple: $f=600$ megacycles $I^{0}=30$ Length $\mathrm{l}=1.64$ inches or 4.2 centimetors

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## Aftenuation and resistance of fransmission

## lines af ultra-high frequencies

$A=4.35 \frac{R_{t}}{Z_{0}}+2.78 \sqrt{\epsilon \rho F}$
where
$A=$ attenuation in decibels per 100 feet
$R_{t}=$ total line resistance in ohms per 100 feet
$p=$ power factor of dielectric medium
$F=$ frequency in megacycles

$$
\begin{aligned}
R_{t} & =0.1\left(\frac{1}{d}+\frac{1}{D}\right) \sqrt{F} \quad \text { for coaxial copper line } \\
& =\frac{0.2}{d} \sqrt{F} \quad \text { for open two-wire copper line }
\end{aligned}
$$

where
$d=$ diameter of conductors (center conductor for the coaxial line) in inches
$D=$ diameter of inner surface of outer coaxial conductor in inches

## Wave guides and resonators

## Propagation of electromagnetic waves in hollow wave guides

For propagation of energy at ultra-high frequencies through a hollow metal tube under fixed conditions, a number of different types of waves are available, namely:

1. TE waves: Transverse electric waves, sometimes called H waves, characterized by the fact that the electric vector (E vector) is always perpendicular to the direction of propagation. This means that
$E_{x} \equiv 0$
where $x$ is the direction of propagation.
2. TM waves: Transverse magnetic waves, also called E waves, characterized by the fact that the magnetic vector (H vector) is always perpendicular to the direction of propagation.

This means that
$H_{x} \equiv 0$
where $x$ is the direction of propagation.
Note: TEM waves: Transverse electromagnetic waves. These waves are characterized by the fact that both the electric vector (E vector) and the magnetic vector ( H vector) are perpendicular to the direction of propagation. This means that
$E_{x}=H_{x}=0$
where x is the direction of propagation. This is the mode commonly excited in coaxial and open-wire lines. It cannot be propagated in a wave guide.

The solutions for the field configurations in wave guides are characterized by the presence of the integers $n$ and $m$ which can take on separate values from 0 or 1 to infinity. Only a limited number of these different $n, m$ modes can be propagated, depending on the dimensions of the guide and the frequency of excitation. For each mode there is a definite lower limit or cutoff frequency below which the wave is incapable of being propagated. Thus, a wave guide is seen to exhibit definite properties of a high-pass filter.

The propagation constant $\gamma_{n, m}$ determines the amplitude and phase of each component of the wave as it is propagated along the length of the guide. With $x$ the direction of propagation and $\omega$ equal to $2 \pi$ times the frequency, the factor for each component is
$e^{j \omega R-\gamma_{m, ~}^{m}}{ }^{x}$

## Propagation of electromagnefic waves in hollow wave guides continued

Thus, if $\gamma_{n, m}$ is real, the phase of each component is constant, but the amplitude decreases exponentially with $x$. When $\gamma_{n, m}$ is real, it is said that no propagation takes place. The frequency is considered below cutoff. Actually, propagation with high attenuation does take place for a small distance, and a short length of guide below cutoff is often used as a calibrated attenuator.

When $\boldsymbol{\gamma}_{n, m}$ is imaginary, the amplitude of each component remains constant, but the phase varies with $x$. Hence, propagation takes place. $\gamma_{n, m}$ is a pure imaginary only in a lossless guide. In the practical case, $\gamma_{n, m}$ usually comprises both a real part, which is the attenuation constant, and an imaginary part, which is the phase propagation constant.

## Rectangular wave guides

Fig. I shows a rectangular wave guide and a rectangular system of coordinates, disposed so that the origin falls on one of the corners of the wave guide; $x$ is the direction of propagation along the guide, and the crosssectional dimensions are $y_{0}$ and $z_{0}$.
For the case of perfect conductivity of the guide walls with a non-conducting interior dielectric (usually airl, the equations for the $\mathrm{TM}_{n, m}$ or $\mathrm{E}_{n, m}$ waves in the dielectric are:
$E_{x}=A \sin \left(\frac{n \pi}{y_{0}} y\right) \sin \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega t-\gamma_{n, m^{x}}}$
$E_{y}=-A \frac{\gamma_{n, m}}{\gamma^{2}{ }_{n, m}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{n \pi}{y_{0}}\right) \cos \left(\frac{n \pi}{y_{0}} y\right) \sin \left(\frac{m \pi}{z_{0}} z\right) e^{\rho \omega t-\gamma_{n, m} x}$
$E_{z}=-A \frac{\gamma_{n, m}}{\gamma^{2}{ }_{n, m}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{m \pi}{z_{o}}\right) \sin \left(\frac{n \pi}{\gamma_{0}} y\right) \cos \left(\frac{m \pi}{z_{o}} z\right) e^{j \omega t-\gamma_{n, m^{x}}}$
$H_{x}=0$
$H_{\nu}=A \frac{j \omega \epsilon_{k}}{\gamma^{2}{ }_{n, m}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{m \pi}{z_{0}}\right) \sin \left(\frac{n \pi}{y_{0}} y\right) \cos \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega t-\gamma_{n, m^{x}}}$
$H_{z}=-A \frac{j \omega \epsilon_{k}}{\gamma^{2}{ }_{n, m}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{n \pi}{y_{0}}\right) \cos \left(\frac{n \pi}{y_{0}} y\right) \sin \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega t-\gamma_{n, m^{2}}}$
where $\epsilon_{k}$ is the dielectric constant and $\mu_{k}$ the permeability of the dielectric material in MKS (rationalized) units.

## Rectangular wave guides continued

Constant A is determined solely by the exciting voltage. It has both amplitude and phase. Integers $n$ and $m$ may individually take on values from 1 to infinity. No TM waves of the 0,0 type or 0,1 type are possible in a rectangular guide so that neither $n$ nor $m$ may be 0 .
Equations for the $T E_{n, m}$ waves or $H_{n, m}$ waves in a dielectric are:

$$
\begin{aligned}
& H_{x}=B \cos \left(\frac{n \pi}{y_{0}} y\right) \cos \left(\frac{m \pi}{z_{0}} z\right) e^{f \omega t-\gamma_{n, m^{x}}} \\
& H_{y}=B \frac{\gamma_{n, m}}{\gamma_{n, m}^{2}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{n \pi}{y_{0}}\right) \sin \left(\frac{n \pi}{y_{0}} y\right) \cos \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega l-\gamma_{n, m} m^{x}} \\
& H_{z}=B \frac{\gamma_{n, m}}{\gamma_{n, m}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{m \pi}{z_{0}}\right) \cos \left(\frac{n \pi}{y_{0}} y\right) \sin \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega t-\gamma_{n, m^{m}}} \\
& E_{x} \equiv 0 \\
& E_{y}=B \frac{j \omega \mu_{k}}{\gamma_{n, m}^{2}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{m \pi}{z_{0}}\right) \cos \left(\frac{n \pi}{y_{0}} y\right) \sin \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega t-\gamma_{n, m}} \\
& E_{z}=-B \frac{j \omega \mu_{k}}{\gamma_{n, m}^{2}+\omega^{2} \mu_{k} \epsilon_{k}}\left(\frac{n \pi}{y_{0}}\right) \sin \left(\frac{n \pi}{y_{0}} y\right) \cos \left(\frac{m \pi}{z_{0}} z\right) e^{j \omega \omega-\gamma_{a, m}}
\end{aligned}
$$

where $\epsilon_{k}$ is the dielectric constant and $\mu_{k}$ the permeability of the dielectric material in MKS (rationalized) units.

Constant $B$ again depends only on the original exciting voltage and has both magnitude and phase; $n$ and $m$ individually may assume any integer value from 0 to infinity. The 0,0 type of wave where both $n$ and $m$ are 0 is not possible, but all other combinations are.

As stated previously, propagation only takes place when $\gamma_{n, m}$ the propagation constant is imaginary;
$\gamma_{n, m}=\sqrt{\left(\frac{n \pi}{y_{0}}\right)^{2}+\left(\frac{m \pi}{z_{0}}\right)^{2}-\omega^{2} \mu_{k \epsilon_{k}}}$
This means, for any $n, m$ mode, propagation takes place when
$\omega^{2} \mu_{k} \epsilon_{k}>\left(\frac{n \pi}{y_{0}}\right)^{2}+\left(\frac{m \pi}{z_{0}}\right)^{2}$
or, in terms of frequency $f$ and velocity of light $c$, when

$$
f>\frac{c}{2 \pi \sqrt{\mu_{1} \epsilon_{1}}} \sqrt{\left(\frac{n \pi}{\gamma_{0}}\right)^{2}+\left(\frac{m \pi}{z_{0}}\right)^{2}}
$$

where $\mu_{1}$ and $\epsilon_{1}$ are the relative permeability and relative dielectric constant, respectively, of the dielectric material with respect to free space.

## Rectangular wave guides continued



Fig. 2-Field conflguration for $\mathrm{TE}_{0,1}$ wave.


Fig. 3-Field conflguration for a TE 1,2 wave.


Fig. 4-Characteristic E lines for TE waves.

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## Rectangular wave guides continued

The wavelength in the wave guide is always greater than the wavelength in an unbounded medium. If $\lambda$ is the wavelength in free space, the wavelength in the guide with air as a dielectric for the $n, m$ mode is
$\lambda_{\theta(n, m)}=\frac{\lambda}{\sqrt{1-\left(\frac{n \lambda}{2 y_{0}}\right)^{2}-\left(\frac{m \lambda}{2 z_{0}}\right)^{2}}}$
The phase velocity within the guide is also always greater than in an unbounded medium. The phase velocity $v$ and group velocity $u$ are related by the following equation:
$u=\frac{c^{2}}{v}$
where the phase velocity is given by $v=c \frac{\lambda_{0}}{\lambda}$ and the group velocity is the velocity of propagation of the energy.

To couple energy into wave guides, it is necessary to understand the configuration of the characteristic electric and magnetic lines. Fig. 2 illustrates the field configuration for a $\mathrm{TE}_{0,1}$ wave. Fig. 3 shows the instantaneous field configuration for a higher mode, a $\mathrm{TE}_{1,2}$ wave.

In Fig. 4 are shown only the characteristic $E$ lines for the $T E_{0,1}, T E_{0,2}, T E_{1,1}$ and $\mathrm{TE}_{1,2}$ waves. The arrows on the lines indicate their instantaneous relative directions. In order to excite a TE wave, it is necessary to insert a probe to coincide with the direction of the $E$ lines. Thus, for a $T E_{0,1}$ wave, a single probe projecting from the side of the guide parallel to the $E$ lines would be sufficient to couple into it. Several means of coupling from a coaxial line to a rectangular wave guide to excite the $\mathrm{TE}_{0,1}$ mode are shown in Fig. 5. With structures such as these, it is possible to make the standing wave ratio due to the junction less than 1.15 over a 10 to 15 percent frequency band.

Fig. 6 shows the instantaneous configuration of a TM1.1 wave; Fig. 7, an instantaneous field configuration for a $\mathrm{TM}_{1,2}$ wave. Coupling to this type of wave is accomplished by inserting a probe, which is again parallel to the $E$ lines. Since the $E$ lines in this case extend along the length of the tube, it is necessary to position a probe along its length at the center of the $E$ configuration. Fig. 8 illustrates a method of coupling to an $E_{1,1}$ wave and an $E_{1,2}$ wave.

## Rectangular wave guides continued





Fig. 5—Methods of coupling to $\mathrm{TE}_{0,1}$ mode $(a \approx \lambda \mathrm{~g} / 4$ ).

electric intensity
magnetic intensity
Fig. 6-Instantaneous fieid configuration for a TM1.1 wave.


Fig. 7-Instantaneous field configuration for a $T M_{1,2}$ wave.

Rectangular wave guides continued


Fig. 8-Methods of coupling to rectangular wave guides for TM(E) modes.

## Circular wave guides

The usual co-ordinate system is $\rho, \theta, z$, where $\rho$ is in radial direction; $\theta$ is the angle; $z$ is in the longitudinal direction.
TM waves (E waves): $H_{z} \equiv 0$
$E_{z}=A J_{n}\left(k_{n, m} \rho\right) \cos n \theta \epsilon^{j \omega t-\gamma_{n, m^{z}}}$
By the boundary conditions, $E_{2}=0$ when $\rho=a$, the radius. Thus, the only permissible values of $k$ are those for which $J_{n}\left(k_{n, m} a\right)=0$ because $E_{z}$ must be zero at the boundary.
The numbers $n, m$ take on all integral values from zero to infinity. The waves are seen to be characterized by the numbers, $n$ and $m$, where $n$ gives.the order of the bessel functions, and $m$ gives the order of the root of $J_{n}$ $\left(k_{n, m} a\right)$. The bessel function has an infinite number of roots, so that there are an infinite number of $k$ 's which make $J_{n}\left(k_{n, m} a\right)=0$.
The other components of the electric vector $E_{\theta}$ and $E_{\rho}$ are related to $E_{z}$ as are $H_{\theta}$ and $H_{p}$.

TE waves ( $H$ waves): $E_{z} \equiv 0$
$H_{z}=B J_{n}\left(k_{n, m} \rho\right) \cos n \theta e^{j \omega t-\gamma_{n, m^{2}}}$
$H \rho, H_{\theta}, E_{\rho}, E_{\theta,}$ are all related to $H_{z}$.

## Circular wave guides

Again $n$ takes on integral values from zero to infinity. The boundary condition $E_{z}=0$ when $\rho=a$ still applies. To satisfy this condition $k$ must be such as to make $J_{n}^{\prime} \mathbb{k}_{n, m}$ al equal to zero where the superscript indicates the derivative of $J_{n}\left(k_{n, m} a\right)$. It is seen that $m$ takes on values from 1 to infinity since there are an infinite number of roots of $J^{\prime}{ }_{n}\left(k_{n, m} a\right)$.
For circular wave guides, the cut-off frequency for the $n, m$ mode is $f_{c_{n, m}}=\frac{c k_{n, m}}{2 \pi}$ where $c=$ velocity of light and $k_{n, m}$ is evaluated from the roots of the bessel functions
and
$k_{n, m}=\frac{U_{n, m}}{a}$ or $\frac{U_{n, m}^{\prime}}{a}$ where $a=$ radius of guide or pipe and $U_{n, m}$ is the root of the particular bessel function of interest lor its derivativel. The wavelength in the guide is

$$
\lambda_{\theta}=\frac{2 \pi}{\sqrt{\left(\frac{2 \pi}{\lambda_{0}}\right)^{2}-{k^{2}{ }_{n, m}}}}
$$

where $\lambda_{o}$ is the wavelength in an unbounded medium.

The following tables are useful in determining the values of $k$. For $H$ waves the roots $U^{\prime}{ }_{n, m}$ of $J^{\prime}{ }_{n}(U)=0$ are given in the following table, and the corresponding $k_{n, m}$ values are $\frac{U_{n, m}^{\prime}}{a}$

Values of $U^{\prime}{ }_{n, m}$

| $m{ }^{n}$ | 0 | 1 | 2 |
| :---: | :---: | :---: | :---: |
| 1 | 3.832 | 1.841 | 3.054 |
| 2 | 7.016 | 5.332 | 6.705 |
| 3 | 10.173 | 8.536 | 9.965 |

For $E$ waves the roots $U_{n, m}$ of $J_{n}|U|=0$ are given in the following table, and the corresponding $k_{n, m}$ values are $\frac{U_{n, m}}{a}$

Values of $U_{n, m}$

| $m \mathbf{n}^{\mathbf{n}}$ | 0 | 1 | 2 |
| :---: | :---: | :---: | ---: |
| 1 | 2.405 | 3.832 | 5.135 |
| 2 | 5.520 | 7.016 | 8.417 |
| 3 | 8.654 | 10.173 | 11.620 |

where $n$ is the order of the bessel function and $m$ is the order of the root.

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Circular wave guides continued


Fig. 9
Patterns of mognetic force of TM waves in circular wave guides.


Fig. 10
Method of coupling to circular wave guide for $T M_{0,1}$ wave.


Fig. 11
Patterns of electric force of TE waves in circular wave guides.


Fig. 12
Method of coupling to circuiar wave guide for $T E_{1,1}$ wave.

Table I-Cut-off wavelengths and attenuation factors

|  | $\begin{gathered} \text { eouxial } \\ \text { cable }(a, b) \end{gathered}$ | rectangular plpe $a, b$ TE $0_{0} m$ or $H_{0, m}$ | TM $\mathrm{M}_{0,1}$ or $\mathrm{E}_{0}$ | cireular pipe of radius a $T E_{1,1}$ or $H_{1}$ | TE $\mathrm{E}_{1}$ or $\mathrm{H}_{0}$ |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Cut-off wavelength | 0 | $\frac{2 b}{m}$ | 2.6130 | 3.412a | 1.640a |
| Attenuation constant $=\alpha$ | $\alpha_{0} \sqrt{\frac{c}{\lambda}} \frac{\left(\frac{1}{a}+\frac{1}{b}\right)}{\log \frac{b}{a}}$ | $\frac{4 \alpha_{0}}{b} A\left(\frac{b}{2 a}+\frac{\lambda^{2}}{\lambda_{c}^{2}}\right)$ | $\frac{2 \alpha_{0}}{a} A$ | $\frac{2 \alpha_{0}}{a} A\left(0.415+\frac{\lambda^{2}}{\lambda_{c}^{2}}\right)$ | $\frac{2 \alpha_{0}}{a} A\left(\frac{\lambda}{\lambda_{c}}\right)^{2}$ |

where
$\lambda_{c}=$ cut-off wavelength

$$
A=\frac{\sqrt{c / \lambda}}{\sqrt{1-\left(\frac{\lambda}{\lambda_{c}}\right)^{2}}}, \quad \alpha_{0}=\frac{1}{4} \sqrt{\frac{\mu_{2} \epsilon_{1}}{\sigma_{2} \mu_{1}}} \text { (emu) }
$$

## Circular wave guides cantinued

The pattern of magnetic force of TM waves in a circular wave guide is shown in Fig. 9. Only the maximum lines are indicated. In order to excite this type of pattern, it is necessary to insert a probe along the length of the wave guide concentric with the $H$ lines. For instance, in the $\mathrm{TM}_{0,1}$ type of wave, a probe extending down the length of the wave guide at the very center of the guide would provide the proper excitation. This method of excitation is shown in Fig. 10. Corresponding methods of excitation may be used for the other types of TM waves shown in Fig. 9.

Fig. 11 shows the patterns of electric force for TE waves. Again only the maximum lines are indicated. This type of wave may be excited by an antenna which is parallel to the electric lines of force. For instance, the $\mathrm{TE}_{0,1}$ wave would be excited by a small circular loop placed where the maximum $E$ line is indicated in the diagram. The $T E_{1,1}$ wave may be excited by means of an antenna extending across the wave guide. This is illustrated in Fig. 12.

## Attenuation constants

All the attenuation constants consain a common coefficient
$\alpha_{0}=\frac{1}{4} \sqrt{\frac{\mu_{2} \epsilon_{1}}{\sigma_{2} \mu_{1}}}$
$\epsilon_{1}, \mu_{1}$ dielectric constant and magnetic permeability for the insulator
$\sigma_{2}, \mu_{2}$ electric conductivity and magnetic permeability for the metal
For air and copper $\alpha_{0}=0.35 \times 10^{-9}$ nepers per centimeter or $0.3 \times 10^{-3} \mathrm{db}$ per kilometer

Table I summarizes some of the most important formulas. The dimensions $a, b$ are measured in centimefers.

## Electromagnetic horns

Radiation from the wave guide may be obtained by placing an electromagnetic horn of a particular size at the end of the wave guide.

Fig. 13 gives data for designing a horn to have a specified gain with the shortest length possible. The length $L_{1}$ is given by $L_{1}=L\left(1-\frac{a}{2 A}-\frac{b}{2 B}\right)$ where $a=$ wide dimension of wave guide in the $H$ plane, and $b=$ narrow dimension of wave guide in E plane.

## Electromagnetic horns continued



$\mathrm{L}=\mathrm{axial}$ length to apex
$\mathrm{A}=$ width of aperture in $\mathbf{H}$ plane
$B=$ width of aperture in $E$ plane
Fig. 13.

## Electromagnetic horns

continued


Fig. 14-10-decibel widths of horns.
If $L \geqq \frac{a^{2}}{\lambda}$ la $=$ longer dimension of aperfurel the gain is given by $G=$ $\frac{10 a b}{\lambda^{2}}$, the half power width in the $E$ plane is given by $51^{\circ} \frac{\lambda}{b^{\prime}}$ and the half power width in the $H$ plane is given by $70^{\circ} \frac{\lambda}{a^{\prime}}$, where E is the electric vector and $H$ is the magnetic vector,
Fig. 14 shows how the angle between 10-decibel points varies with aperture.

## Parabolas

If the intensity across the aperture of the parabola is of constant phase and tapers smoothly from the center to the edges so that the intensity at the edges is 10 decibels down from that at the center, the gain is given by $G=\frac{8 A}{\lambda^{2}} L A=$ area of aperture $)$. The half power width is given by $70^{\circ} \frac{\lambda}{D}$ ( $D=$ diameter of parabola).

## Resonant cavities

A cavity enclosed by metal walls will have an infinite number of natural frequencies at which resonance will occur. The lowest frequency or mode of oscillation is determined by the geometry of the cavity. One of the

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## Resonant cavities continued

more common types of cavity resonators is a length of transmission line (coaxial, or waveguidel short circuited at both ends.

## Resonance occurs when

$2 h=l \frac{\lambda g}{2}$ where $l$ is an integer
$2 h=$ length of the resonator
$\lambda_{\theta}=$ guide wavelength in resonator

$$
\lambda_{0}=\frac{\lambda}{\sqrt{1-\left(\frac{\lambda}{\lambda_{c}}\right)^{2}}}
$$

$\lambda=$ free space wavelength $\lambda_{c}=$ guide cut-off wavelength
For $T E_{n, m}$ or $T M_{n, m}$ waves in a rectangular cavity with cross section $a, b$.
$\lambda_{c}=\frac{2}{\sqrt{\left(\frac{n}{a}\right)^{2}+\left(\frac{m}{b}\right)^{2}}}$ where $n$ and $m$ are integers
For $\mathrm{TE}_{n, m}$ waves in a cylindrical cavity
$\lambda_{c}=\frac{2 \pi a}{U_{n, m}^{\prime}}$
where $a$ is the guide radius and $U_{n, m}^{\prime}$ is the $m$ th root of the equation $J_{n}(U)=0$
For $\mathrm{TM}_{n, m}$ waves in a cylindrical cavity
$\lambda_{c}=\frac{2 \pi \mathrm{a}}{U_{n, m}}$
where $a$ is the guide radius and $U_{n, m}$ is the $m$ th root of the equation $J_{n}(U)=0$.

For TM waves $l=0,1,2 \ldots$
For TE waves $1=1,2 \ldots$ but not 0

## Rectangular cavity of dimensions $a \mathbf{b} \mathbf{2 h}$

$$
\lambda=\frac{2}{\sqrt{\left(\frac{l}{2 h}\right)^{2}+\left(\frac{n}{a}\right)^{2}+\left(\frac{m}{b}\right)^{2}}} \text { where only one of } l, n, m \text { may be zero. }
$$

Resonant cavilies continued

## Cylindrical cavities of radius a and length $\mathbf{2 h}$

$$
\lambda=\frac{1}{\sqrt{\left(\frac{1}{4 h}\right)^{2}+\left(\frac{1}{\lambda_{c}}\right)^{2}}}
$$

where $\lambda_{c}$ is the guide cut-of wavelength.

## Spherical resonators of radius a

$\lambda=\frac{2 \pi a}{U_{n, m}}$ for a TE wave
$\lambda=\frac{2 \pi a}{U_{n, m}^{\prime}}$ for a TM wave.
Values of $U_{n, m}$ :
$U_{1,1}=4.5, U_{2,1}=5.8, U_{1,2}=7.64$
Values of $U^{\prime}{ }_{n, m}$ :
$U_{1,1}^{\prime}=2.75=$ lowest order root

## Additional cavity formulas

| type of cavity | mode | $\lambda_{0}$ resonant wavelength | 0 |
| :---: | :---: | :---: | :---: |
| Right circular cylind or |  | 4 | $\lambda_{0} \circ 1$ |
|  |  | $\sqrt{\left(\frac{1}{h}\right)^{2}+\frac{2.35}{a^{2}}}$ | $\bar{\delta} \overline{\lambda_{0}} \overline{1+\frac{a}{2 h}}$ |
|  | TE $\mathrm{E}_{0,1,2}\left(\mathrm{H}_{0}\right)$ | $\frac{4}{\sqrt{\left(\frac{1}{h}\right)^{2}+\frac{5.93}{a^{2}}}}$ | $\frac{\lambda_{0}}{\delta} \frac{o}{\lambda_{0}}\left[\frac{1+0.168\left(\frac{o}{h}\right)^{2}}{1+0.168 \cdot\left(\frac{o}{h}\right)^{3}}\right]$ |
|  | $\mathrm{TE}_{1,1,1}\left(\mathrm{H}_{1}\right)$ | $\frac{4}{\sqrt{\left(\frac{1}{h}\right)^{2}+\frac{1.37}{a^{2}}}}$ | $\frac{\lambda_{0}}{\delta} \frac{h}{\lambda_{0}}\left[\frac{2.39 h^{2}+1.73 a^{2}}{3.39 \frac{h^{3}}{a}+0.73 a h+1.73 \sigma^{2}}\right]$ |

Some characteristics of various types of resonators
$\delta$ is the skin depth

|  | typeresonator | wavalength, $\lambda$ | 0 |
| :---: | :---: | :---: | :---: |
| Square prism $T E_{0,1,1}$ |  | $2 \sqrt{2} a$ | $\frac{0.353 \lambda}{\delta} \frac{1}{1+\frac{0.177 \lambda}{h}}$ |
| Circular cylinder TM $M_{0.1,0}$ |  | 2.61a | $\frac{0.383 \lambda}{\delta} \frac{1}{1+\frac{0.192 \lambda}{h}}$ |
| Sphere |  | 2.28a | $0.318 \frac{\lambda}{\delta}$ |
| Sphere with cones |  | 40 | Optimum Q for $\theta=34^{\circ}$ $0.1095 \frac{\lambda}{\delta}$ |
| $\begin{aligned} & \text { Coaxlal } \\ & \text { TEM } \end{aligned}$ |  | 4h | Optimum Q $\text { for } \frac{b}{a}=3.6$ $\left(Z_{0}=77\right. \text { ohms) }$ $\frac{\lambda}{4 \delta+7.2 \frac{h \delta}{b}}$ |

$\delta=\sqrt{\frac{\rho}{2 \pi \omega \mu}}$ where $\rho=$ resistivity of wall in abohm-cm, $\mu=$ permeability of volume lunity for free space), $\delta=$ skin depth in centimeters.

## Recommended rectangular wave guides



## - Radio propagation and noise

## Propagation of medium and long waves*

For a theoretical short vertical antenna over perfect ground:
$E=186 \sqrt{P_{r}}$ millivolts per meter at 1 mile
or,
$E=300 \sqrt{P_{r}}$ millivalts per meter at 1 kilometer
where $P_{r}=$ radiated power in kilowatts.
Actual inverse-distance fields at one mile for a given transmitter output power depend on the height and efficiency of the antenna and the efficiency of coupling devices.

Typical values found in practice for well-designed stations are: Small L or $T$ antennas as on ships; $25 \sqrt{P_{t}}$ millivolts per meter at 1 mile Vertical radiators 0.15 to $0.25 \lambda$ high; $150 \sqrt{P_{t}}$ millivolts per meter at 1 mile Vertical radiators 0.25 to $0.40 \lambda$ high; $175 \sqrt{P_{t}}$ millivolts per meter at 1 mile Vertical radiators 0.40 to $0.60 \lambda$ high or top-loaded vertical radiators; $220 \sqrt{P_{t}}$ millivalts per meter at 1 mile, where $P_{\ell}=$ transmitter output power in kilowatts.
These values can be increased by directive arrangements.
The surface-wave field (commonly called ground wave) at greater distances can be found from Figs. 1, 2, and 3. These are based on a field strength of 186 millivolts per meter at one mile. The ordinates should be multiplied by the ratio of the actual field at 1 mile to 186 millivolts per meter.

## Table 1-Ground conductivities and dielectric constants

| ferroin | conductivity emu | dielectric constant esu |
| :---: | :---: | :---: |
| Sea water | $4 \times 10^{-11}$ | 80 |
| Fresh water | $5 \times 10^{-11}$ | 80 |
| Dry, sandy fiat coastal land | $2 \times 10^{-14}$ | 10 |
| Marshy, forested flap land | $8 \times 10^{-14}$ | 12 |
| Rich agricultural land, low hills | $1 \times 10^{-13}$ | 15 |
| Pastoral land, medium hills and forestation | $5 \times 10^{-14}$ | 13 |
| Rocky land, steep hills | $2 \times 10^{-14}$ | 10 |
| Mountainous thils up to 3000 feet) | $1 \times 10^{-14}$ | 5 |
| Cities, residential areas | $2 \times 10^{-14}$ | 5 |
| Cities, industrial areas | $1 \times 10^{-15}$ | 3 |

[^16]

## Propagation of medium and long waves continued

Figs. 1, 2, and 3 do not include the effect of sky waves reflected from the ionosphere. Sky waves cause fading at medium distances and produce higher field intensities than the surface wave at longer distances, particularly at night and on the lower frequencies during the day. Sky-wave field intensity, in addition to the usual diurnal, seasonal, and irregular variations due to changing properties of the ionosphere, depends on frequency and the vertical radiation pattern of the antenna. Fig. 4 shows the average of nighttime measurements on a number of broadcast stations for about l-kilowatt output.


Fig. 3-Sirength of surface waves as a function of distance with a vertical anfonna for sea woter ( $\sigma=4 \times 10^{-11} \mathrm{emu}$ and $\epsilon=80$ esu).

## Propagation of short waves

At frequencies be tween about 3 and 25 megacycles and distances greater than about 100 miles, transmission depends entirely on sky waves reflected from the ionosphere. The ionosphere la region high above the earth's surface where the rarefied air is sufficiently ionized to reflect or absorb radio waves) is usually considered as consisting of the following layers.

D layer: At heights from about 50 to 90 kilometers, it exists only during daylight hours and ionization density corresponds with the altitude of the sun.

Propagation of short waves continued

This layer reflects low- and medium-frequency waves and weakens highfrequency waves through partial absorption.

Elayer: At height of about 110 kilometers, this layer is of importance for shortwave daytime propagation at distances less than 1000 miles and for medium wave nighttime propagation at distances in excess of about 100 miles. lonization density corresponds closely with the altitude of the sun. Irregular cloud-like areas of unusually high ionization, called sporadic $E$ may occur up to more than 50 percent of the time on certain days or nights. Sporadic $E$ occasionally prevents frequencies that normally penetrate the E layer reaching higher layers and also causes occasional longdistance transmission at very high frequencies.


Fig. 4-Average sky-wave field intensity (corresponding to the second hour after sunset at the recording station).
$\mathbf{F}_{1}$ layer: At heights of about 175 to 250 kilometers, it exists only during daylight. This layer occasionally is the reflecting region for shortwave transmission, but usually oblique incidence waves that penetrate the Elayer also penetrate the $F_{1}$ layer to be reflected by the $F_{2}$ layer. The $F_{1}$ layer introduces additional absorption of such waves.

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## Propagation of short waves

conlinued
$F_{2}$ layer: At heights of about 250 to 400 kilometers, $F_{2}$ is the principal reflecting region for long-distance shortwave communication. Height and ionization density vary diurnally, seasonally, and over the sunspot cycle. Ionization does not correspond closely to the altitude of the sun. At night, the $F_{1}$ layer merges with the $F_{2}$ layer at a height of about 300 kilometers. The absence of the $F_{1}$ layer, and reduction in absorption of the E layer, causes nighttime field intensities and noise to be generally higher than during daylight hours.
As indicated to the right on Fig. 6, these layers are contained in a thick region throughout which ionization generally increases with height. The layers are said to exist where the ionization gradient is capable of refracting waves back to earth. Obliquely incident waves follow a curved path through the ionosphere due to gradual refraction or bending of the wave front.
Depending on the ionization density at each layer, there is a critical or highest frequency $f_{c}$ at which the layer reflects a vertically incident wave. Frequencies higher than $f_{c}$ pass through the layer af vertical incidence. At oblique incidence the layer reflects frequencies higher than $f_{c}$ as given by the approximate relation:
muf $=f_{c} \sec \phi$
where muf $=$ maximum usable frequency for the particular layer and distance, $\phi=$ angle of incidence at reflecting layer.
$f_{c}$ and height, and hence $\phi$ for a given distance, for each layer vary with local time of day, season, latitude, and throughout the eleven-year sunspot cycle. The various layers change in different ways with these parameters. In addition, ionization is subject to frequent abnormal variations.
The loss at reflection for each layer is a minimum at the maximum usable frequency and increases rapidly for frequencies lower than maximum usable frequency.
Short waves travel from the transmitter to the receiver by reflections from the ionosphere and earth in one or more hops as indicated in Figs. 5 and 6. Additional reflections may occur along the path between the bottom edge of a higher layer and the top edge of a lower layer, the wave finally returning to earth near the receiver.
Fig. 5 illustrates single-hop transmission, Washington to Chicago, via the E layer $\left(\phi_{1}\right)$. At higher frequencies over the same distance, single-hop transmission would be obtained via the $\mathrm{F}_{2}$ layer $\left(\phi_{2}\right)$. Fig. 5 also shows two-hop transmission, Washington to San Francisco, via the $F_{2}$ layer $\left(\phi_{3}\right)$. Fig. 6 indicates transmission on a common frequency, (1.) single-hop via $E$ layer, Denver to Chicago, and, (2.) single-hop via $F_{2}$, Denver to Washington, with, (3.) the wave failing to reflect at higher angles, thus producing a skip region of no signal between Denver and Chicago.

Actual transmission over long distances is more complex than indicated by Figs. 5 and 6, because the layer heights and critical frequencies differ with time (and hence longitudel and with latitude. Further, scattered reflections occur at the various surfaces.


Fig. 5.


Fig. 6.

Maximum usable frequencies (muf) for single-hop transmission at various distances throughout the day are given in Fig. 7. These approximate values apply to latitude $39^{\circ} \mathrm{N}$ for the approximate minimum years 11944 and 1955) and approximate maximum years (1949 and 1960) of the sunspot cycle. Since the maximum usable frequency and layer heights change from month to month, the latest predictions should be obtained whenever available. This information is published by the National Bureau of Standards in the U. S. A. and by similar organizations in other countries.

Operating frequencies should be selected from 50 to 85 percent of the maximum usable frequency, preferably nearer the higher limit in order to reduce absorption loss. The 85 percent limit provides some margin for day-to-day deviation of the ionospheric characteristics from the predicted monthly average value. Maximum usable frequency changes continuously throughout the day, whereas it is ordinarily impractical to change operating frequencies correspondingly. Each operating frequency, therefore, should be selected to fall within the above limits for a substantial portion of the daily operating period.
For single-hop transmission, frequencies should be selected on the basis of local time and other conditions existing at the mid-point of the path. In view of the layer heights and the fact that practical antennas do not operate effectively below angles of about three degrees, single-hop trans-

## Propagation of short waves continued

mission cannot be achieved for distances in excess of about 2200 miles $(3500$ kilometers) via $F$ layers or in excess of about 1050 miles ( 1700 kilometersl via the E layer. Multiple-hop transmission must occur for longer distances and, even at distances of less than 2200 miles, the major part of the received signal frequently arrives over a two- or more-hop path. In analyzing two-hop paths, each hop is treated separately and the lowest frequency required on either hop becomes the maximum usable frequency for the circuit. It is usually impossible to predict accurately the course of radio waves on circuits involving more than two hops because of the large number of possible paths and the scattering that occurs at each reflection. For such longdistance circuits, it is customary to consider the conditions existing at points 1250 miles along the path from each end as the points at which the maximum usable frequencies should be calculated.


Pig. 7.

## Propagation forecasts for short waves

In addition to forecasts for ionospheric disturbances, the Central Radio Propagation Laboratories of the National Bureau of Standards issues monthly Basic Radio Propagation Predictions 3 months in advance used to determine the optimum working frequencies for shortwave communication. Indication of the general nature of the CRPL data and a much abbreviated example of their use follows:

## Example

To determine working frequencies for use between San Francisco and Wellington, N. Z.

## Method

1. Place a transparent sheet over Fig. 8 and mark thereon the equator, a line across the equator showing the meridian of time desired (viz., GCT or PST), and locations of San Francisco and Wellington.
2. Transfer sheet to Fig. 9, keeping equator lines of chart and transparency aligned. Slide from left to right until terminal points marked fall along a Great Circle line. Sketch in this Great Circle between terminals and mark "control points" 2000 kilometers along this line from each end.
3. Transfer sheet to Fig. 10, showing muf for transmission via the $F_{2}$ layer. Align equator as before. Slide sheet from left to right placing meridian line on time desired and record frequency contours at control points. This illustration assumes that radio waves are propagated over this path via the $F_{2}$ layer. Eliminating all other considerations, 2 sets of frequencies, corresponding to the control points, are found as listed in Table II, the lower of which is the muf. The muf, decreased by 15 percent, gives the optimum working frequency.
Transmission may also take place via other layers. For the purpose of illustration only and without reference to the problem above, Figs. 11 and 12 have been reproduced to show characteristics of the $E$ and sporadic $E$ layers. The complete detailed step-by-step procedure, including special considerations in the use of this method, are contained in the complete CRPL forecasts.

Table II-Maximum usuable frequency

| at San Francisco |
| :---: | :---: | :---: | :---: |
| control point |
| (2000 km from |
| San Francisco) |$\quad$| Wellington, N. Z. |
| :---: |
| control point |
| (2000 km from |
| Wellington) |$\quad$| optimum working |
| :---: |
| frequency |
| (lower of |
| muf $\times \mathbf{0 . 8 5 )}$ |



Fig. 8-World map showing zones covered by predicied charts and auroral zones.

Flg. 9-Great circle chart centered on equator. Solid lines reprosent great clrcles. Dot-dosh lines indicate dislances in thousands of kllometers.


## 



Fig. 11-E layer 2000 .
kllometer maximum
usable frequency in megacycles predicted for July, 1946.


Fig. 12-Median $\mathbf{f E}$ in megacycles (sporadic E layer) predicted for July, 1946


## Propagation of very short waves

For propagation over distance within the radio path horizon, the field intensity is given approximately by

$$
\begin{equation*}
E=\frac{14.0 \sqrt{W}}{d} \sin \left(\frac{2 \pi h_{i} h_{r}}{\lambda d}\right) \text { volts per meter } \tag{1}
\end{equation*}
$$

## where

$W=$ watts radiated, $h_{t}=$ height of transmitting antenna in meters, $h_{r}=$ height of receiving antenna in meters, $\boldsymbol{\lambda}=$ wavelength in meters, $d=$ distance in meters.

The following approximate formula is useful for transmission below 100 megacycles within the radio path horizon.

$$
\begin{equation*}
E=\frac{0.33 \sqrt{P} H_{t} H_{r} f_{m c}}{D^{2}} \text { microvolts per meter } \tag{2}
\end{equation*}
$$

where
$P=$ kilowatts radiated, $H_{t}=$ height of transmitting antenna in feet, $H_{r}=$ height of receiving antenna in feet, $f_{m c}=$ frequency in megacycles, $D=$ distance in statute miles.

Equations (1) and (2) apply to both vertical and horizontal polarization. It is assumed that the antennas are small dipoles. The equations hold only when the transmission distance is large compared to antenna heights, i.e.,
for equation (1) $d>10 \mathrm{hr}_{r}$
for equation (2) $D>4 \mathrm{H}_{t} \mathrm{H}_{\mathrm{r}} f_{m c} \times 10^{-6}$
Multiplying the true radius of the earth by correction factor 1.33 to provide for average atmospheric refraction gives the radio path horizon as
$D_{l}=\sqrt{2 H_{t}}+\sqrt{2 H_{r}}$ statute miles
If the refractive effect of the atmosphere is ignored, line-of-sight horizon is reduced to the geometric range
$D_{b}=1.23\left(\sqrt{H_{t}}+\sqrt{H_{r}}\right)$
These distances may be obtained from the nomograph, Fig. 13.
When the transmission distance is not large compared with antenna height, the field strength oscillates with distance and height as indicated by the sine term of equation (1).
The number of oscillations for a given distance increases with frequency as illustrated in Fig. 14. This is due to interference between the space wave and the ground-reflected wave as these two components fall in or out of phase at various distances and heights.

## U-H-F path İength and optical line-of-sight

## disfance range of radio waves



The theoretical maximum poth of a radio wave, the sum of the "optical" horizon distances of each antenna, is found on "iine-of-sight" scale by a line connecting points representing the two antenna heights. Atmospheric diffraction increases this path an amount generally considered as $2 / \sqrt{3}$ times optical line of sight, given on the radio path scale.
Example shown: Height of receiving antenna 60 feet, height of transmitting antenna 500 feet, and maximum radio path length $\mathbf{4 1 . 5}$ miles.

Fig. 13.

Propagation of very short waves cantinued


Fig. 14-Effect of frequency on ground-wave fleld intensity.
To compute the field accurately under these conditions, it is necessary to calculate the two components separately and to add them in correct phase relationship as determined by the geometry of the path and the change in magnitude and phase at ground reflection. For horizontally-polarized waves, the reflection coefficient can be taken as approximately one, and the phase
shift at reflection as 180 degrees, for nearly all types of ground and angles of incidence. For vertically-polarized waves, the reflection coefficient and phase shift vary with the ground constants and angle of incidence.*
For methods of computing field intensities when equations (1) and (2) do not hold beyond the radio path horizon, or when the antenna height is not negligible compared to distance, see reference below. $\dagger$
At points beyond the radio path horizon, field intensity decreases more rapidly than the square of the distance; and, if the antennas are raised, the field intensity increases more rapidly than the product of antenna heights.
Measured field intensities usually show large deviations from point to point due to reflections from irregularities in the ground, buildings, trees, etc. In addition, fields at the longer distances are subject to fading and day-to-day variations due to changes in the refractive index of the atmosphere and tropospheric reflections.

[^17]
## Great circle calculations

Referring to Figs. 15,16 , and $17, A$ and $B$ are two places on the earth's surface the latitudes and longitudes of which are known. The angles $X$ and $Y$ at $A$ and $B$ of the great circle passing through the two places and the distance $Z$ between $A$ and $B$ along the great circle can be calculated as follows:
$B$ is the place of greater latitude, i.e., nearer the pole
$L_{A}$ is the latitude of $A$
$L_{B}$ is the latitude of $B$
$C$ is the difference of longitude between $A$ and $B$
Then, $\tan \frac{Y-X}{2}=\cot \frac{C}{2} \frac{\sin \frac{L_{B}-L_{A}}{2}}{\cos \frac{L_{B}+L_{A}}{2}}$
and, $\tan \frac{Y+X}{2}=\cot \frac{C \cdot \cos \frac{L_{B}-L_{A}}{2}}{\sin \frac{L_{B}+L_{A}}{2}}$
give the values of $\frac{Y-X}{2}$ and $\frac{Y+X}{2}$

## Great circle calculations

## continued

from which
$\frac{Y+X}{2}+\frac{Y-X}{2}=Y$
and
$\frac{Y+X}{2}-\frac{Y-X}{2}=X$
In the above formulas, north latitudes are taken as positive and south latitudes as negative. For example, if $B$ is latitude $60^{\circ} \mathrm{N}$ and $A$ is latitude $20^{\circ} \mathrm{S}$
$\frac{L_{B}+L_{A}}{2}=\frac{60+(-20)}{2}=\frac{60-20}{2}=\frac{40}{2}=20^{\circ}$
and
$\frac{L_{B}-L_{A}}{2}=\frac{60-(-20)}{2}=\frac{60+20}{2}=\frac{80}{2}=40^{\circ}$
If both places are in the southern hemisphere and $L_{B}+L_{A}$ is negative, it is simpler to call the place of greater south latitude $B$ and to use the above method for calculating bearings from true south and to convert the results afterwards to bearings east of north.

The distance $Z$ (in degrees) along the great circle between $A$ and $B$ is given by the following:
$\tan \frac{Z}{2}=\tan \frac{L_{B}-L_{A}}{2} \frac{\sin \frac{Y+X}{2}}{\sin \frac{Y-X}{2}}$
The angular distance $Z$ lin degrees) between $A$ and $B$ may be converted to linear distance as follows:
$Z$ lin degrees) $\times 111.195=$ kilometers
$Z$ lin degrees) $\times 69.093=$ statute miles
$Z$ lin degrees) $\times 60.000=$ nautical miles
In multiplying, the minutes and seconds of arc must be expressed in decimals of a degree. For example, $Z=37^{\circ} 45^{\prime} 36^{\prime \prime}$ becomes $37.755^{\circ}$.

Example:-Find the great circle bearings at Brentwood, Long Island, Longitude $73^{\circ} 15^{\prime} 10^{\prime \prime} \mathrm{W}$, Latitude $40^{\circ} 48^{\prime} 40^{\prime \prime} \mathrm{N}$, and at Rio de Janeiro, Brazil, Longitude $43^{\circ} 22^{\prime} 07^{\prime \prime} \mathrm{W}$, Latitude $22^{\circ} 57^{\prime} 09^{\prime \prime} \mathrm{S}$, and the great circle distance in statute miles between the two points.


Fig. 16
$L_{A}=$ latitude of $A$
$\mathrm{L}_{\mathrm{B}}=$ latitude of B
$\mathbf{C}=$ difference of longitude

Fig. 17
$L_{A}=$ latitude of $A$
$L_{B}=$ latitude of $B$
$\mathbf{C}=$ difference of longitude

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## Great circle calculations continued

|  | longitude | latifude |  |
| :---: | :---: | :---: | :---: |
| Brentwood Rio de Janeiro | $\begin{aligned} & 73^{\circ} 15^{\prime} 10^{\prime \prime} \mathrm{W} \\ & 43^{\circ} 22^{\prime} 07^{\prime \prime} \mathrm{W} \end{aligned}$ | $\begin{array}{r} 40^{\circ} 48^{\prime} 40^{\prime \prime} \mathrm{N} \\ \left(-122^{\circ} 57^{\prime} 09^{\prime \prime} \mathrm{S}\right. \end{array}$ | $\begin{aligned} & L_{B} \\ & L_{A} \end{aligned}$ |
| C | $29^{\circ} 53^{\prime} 03^{\prime \prime}$ | $17^{\circ} 51^{\prime} 31^{\prime \prime}$ $63^{\circ} 45^{\prime} 49^{\prime \prime}$ | $\begin{aligned} & L_{n}+L_{A} \\ & L_{n}-L_{A_{n}} \end{aligned}$ |
| $\frac{C}{2}=14^{\circ} 56^{\prime} 31^{\prime \prime}$ | $\frac{L_{B}+L_{A}}{2}=8^{\circ} 55^{\prime} 45^{\prime \prime}$ | $\frac{L_{8}-L_{A}}{2}=31^{\circ} 52^{\prime} 54^{\prime \prime}$ |  |

$\log \cot 14^{\circ} 56^{\prime} 31^{\prime \prime}=10.57371$
plus log $\cos 31^{\circ} 52^{\prime} 54^{\prime \prime}=\frac{9.92898}{0.50269}$
minus $\log \sin 8^{\circ} 55^{\prime} 45^{\prime \prime}=9.19093$
$\log \tan \frac{Y+X}{2}=\overline{1.31176}$

$$
\frac{Y+X}{2}=87^{\circ} 12^{\prime} 26^{\prime \prime}
$$

$\log \cot 14^{\circ} 56^{\prime} 31^{\prime \prime}=10.57371$
plus $\log \sin 31^{\circ} 52^{\prime} 54^{\prime \prime}=\frac{9.72277}{0.29648}$
minus log $\cos 8^{\circ} 55^{\prime} 45^{\prime \prime}=9.99471$
$\log \tan \frac{Y-X}{2}=0.30177$

$$
\frac{Y-X}{2}=63^{\circ} 28^{\prime} 26^{\prime \prime}
$$

$\frac{Y+X}{2}+\frac{Y-X}{2}=Y=150^{\circ} 40^{\prime} 52^{\prime \prime}$ East of North-bearing at Brentwood
$\frac{Y+X}{2}-\frac{Y-X}{2}=X=23^{\circ} 44^{\prime} 00^{\prime \prime}$ West of North-bearing at Rio de Janeiro

$$
\left.\begin{array}{rl}
\frac{L_{B}-L_{A}}{2}=31^{\circ} 52^{\prime} 54^{\prime \prime} \\
\frac{Y+X}{2}=87^{\circ} 12^{\prime} 26^{\prime \prime} \\
\frac{Y-X}{2}=63^{\circ} 28^{\prime} 26^{\prime \prime} & \log \tan 31^{\circ} 52^{\prime} 54^{\prime \prime}
\end{array}=9.79379, \begin{array}{r}
9.79327 \\
\text { plus } \log \sin 87^{\circ} 12^{\prime} 26^{\prime \prime}
\end{array}\right) \quad \begin{aligned}
& 9.99948 \\
& \text { minus } \log \sin 63^{\circ} 28^{\prime} 26^{\prime \prime}=9.95170 \\
& \log \tan \frac{Z}{2}=9.84157 \\
& \frac{Z}{2}=34^{\circ} 46^{\prime} 24^{\prime \prime} \\
& Z=69^{\circ} 32^{\prime} 48^{\prime \prime}
\end{aligned}
$$

$69^{\circ} 32^{\prime} 48^{\prime \prime}=69.547^{\circ}$
linear distance $=69.547 \times 69.093=4805.21$ statute miles

## Time inferval between transmission and reception of reflected signal

Fig. 18 gives the time interval between transmission and reception of a reflected signal based on a velocity of propagation in free space of 985 feet per microsecond or 300 meters per microsecond. A statute mile of 5280 feetor 1760 yards or 1.609 kilometers is used.


Note: Ordinates show distance to point of reflection
Fig. 18.

## Radio noise and noise measurement*

Radio noise may be divided into four classifications, depending on origin:

1. Atmospheric noise (static)
2. Cosmic noise
3. Man-made noise
4. Receiver and antenna noise
[^18]
## RADIO PROPAGATION AND NOISE 24

## Radio noise and noise measurement continued

Radio noise, as in Fig. 19, is usually expressed in terms of peak values. Atmospheric noise is shown in the figure as the average peaks would be read on the indicating instrument of an ordinary field intensity meter. This is lower than the true peaks of atmospheric noise. Man-made noise is shown as the peak values that would be read on the EEI-NEMA-RMA standard noise meter. Receiver and antenna noise is shown with the peak values 13 decibels higher than the values obtained with an energy averaging device such as a thermoammeter.

1. Atmospheric noise: is produced mostly by lightning discharges in thunderstorms. The noise level is thus dependent on frequency, time of day, weather, season of the year, and geographical location.
Subject to variations due to local stormy areas, noise generally decreases with increasing latitude on the surface of the globe. Noise is particularly severe during the rainy seasons in certain areas such as Caribbean, East Indies, equatorial Africa, northern India, etc. Fig. 19 shows median values of atmospheric noise for the U.S. A. and these values may be assumed to apply approximately to other regions lying between 30 and 50 degrees latitude north or south.

Rough approximations for atmospheric noise in other regions may be obtained by multiplying the values of Fig. 19 by the factors in Table III.

Table III-Multiplying factors for atmospheric noise in regions not shown on Fig. 19

| latitude | nightime |  | daytime |  |
| :---: | :---: | :---: | :---: | :---: |
|  | 100 kc | $\mathbf{1 0} \mathbf{~ m e}$ | $\mathbf{1 0 0} \mathbf{k c}$ | $\mathbf{1 0} \mathbf{m e}$ |
| $90^{\circ}-50^{\circ}$ | 0.1 | 0.3 | 0.05 | 0.1 |
| $50^{\circ}-30^{\circ}$ | 1 | 1 | 1 | 1 |
| $35^{\circ}-10^{\circ}$ | 2 | 2 | 3 | 2 |
| $10^{\circ}-0^{\circ}$ | 5 | 4 | 6 | 3 |

Atmospheric noise is the principal limitation of radio service on the lower frequencies. At frequencies above about 30 megacycles, the noise falls to levels generally lower than receiver noise.
The peak amplitude of atmospheric noise usually may be assumed to be proportional to the square root of receiver bandwidth.
2. Cosmic noise: originates outside the earth's atmosphere and appears as a random noise like thermal agitation. Cosmic noise has been observed and measured at frequencies from 10 to 20 megacycles and at frequencies of about 160 megacycles. It is reasonable to assume that it exists at all frequencies between 10 and 1000 megacycles and higher.

The intensity of cosmic noise is generally lower than interference produced by other sources. In the absence of atmospheric and man-made noise, it may be the principal limiting factor in reception between 10 and 30 megacycles.


## Notes:

1. All noise curves ossume a bondwith of 10 kilocycles.
2. Receiver noise is bosed on the use of o holf-wove dipole ontenno ond is worse thon on ideal receiver by 10 decibels of 50 megocycles ond 15 decibels of 1000 megocycles.
3. Refer to Fig. 20 for converting mon-mode noise curves to bondwiths greoter thon 10 kilocycles.
4. For oll other curves, noise vories os the squore root of bondwith.

Fig. 19.

## Radio noise and noise measurement continued

3. Man-made noise: includes interference produced by sources such as motorcar ignition, electric motors, electric switching gear, high-tension line leakage, diathermy, industrial heating generators. The field intensity from these sources is greatest in densely populated and industrial areas.

The nature of man-made noise is so variable that it is difficult to formulate a simple rule for converting 10 kilocycle bandwidth receiver measurements to other bandwidth values. For instance, the amplitude of the field strength radiated by a diathermy device will be the same in a 100 - as in a 10 -kilocycle bandwidth receiver. Conversely, peak noise field strength due to automobile ignition will be considerably greater with a 100 - than with a 10 -kilocycle bandwidth. According to the best available information, the peak field strengths of man-made noise lexcept diathermy and other narrow-band noise) increases as the receiver bandwidth is increased, substantially as shown in Fig. 20.

receiver bandwidth in kilocycles
Fig. 20-Bandwidth factor. Multiply value of man-made noise from Fig. 19 by the factor above for receiver bandwidths higher than 10 kilocycles.

The man-made noise curves in Fig. 19 show typical median values for the U.S.A. In accordance with statistical practice, median values are interpreted to mean that 50 percent of all sites will have lower noise levels than the values of Fig. 19; 70 percent of all sites will have noise levels less than 1.9 times these values; and 90 percent of all sites, less than seven times these values.
4. Receiver and antenna noise: is caused by thermal agitation in resistance components of the antenna and receiver circuits and by electronic current flow in the tubes.

The basic equation for thermal agitation noise is
$E^{2}=4 k T R \Delta f$
where
$E=$ rms volts
$k=$ Boltzmann's constant $=1.374 \times 10^{-23}$
$T=$ absolute temperature in degrees Kelvin
$R=$ resistance in ohms
$\Delta f=$ bandwidth in cycles per second
For application of this formula to receiver input circuits see Herold, E. W., An Analysis of the Signal-to-Noise Ratio of Ulira-High-Frequency Receivers; and North, D. O., The Absolute Sensitivity of Radio Receivers. RCA Review, vol. 6 (January, 1942).
The ideal receiver is one in which the only noise is that generated by thermal agitation in the radiation resistance of the antenna and in the input coupling resistance. The calculated values shown in Fig. 19 are based on the assumption that an actual receiver has a noise level greater than the ideal receiver by a factor varying from 10 decibels at 50 megacycles to 15 decibels at 1000 megacycles.

The peak value of this type of noise is approximately 13 decibels greater than its rms value. The amplitude is proportional to the square root of receiver bandwidth. Fig. 19 shows the field intensities required to equal the peak receiver noise values calculated on the above basis. These equivalent field intensities assume the use of a half-wave dipole receiving antenna. Transmission-line loss is omitted in the calculations. For antennas delivering more power to the receiver than a half-wave dipole, equivalent noise field intensities are less than indicated in Fig. 19 in proportion to the net gain of the antenna plus transmission line.
5. Signal-to-noise ratio: for satisfactory reception varies over, wide limits dependent on the type of communication, bandwidth, type of modulation, directivity of receiving antenna, character of noise, etc. A rough general relationship applicable to many services is that the average value of field intensity should be at least 10 decibels higher than the peak noise intensity, both measured on nondirective antennas with the noise peaks as observed on the usual type of measuring devices. Due to the relationship between peak and average values for noise, this means that the average field intensity should exceed the average noise intensity by at least 20 to 25 decibels.

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## Radio noise and noise measurement continued

Considerably higher ratios of signal-to-noise fields are required for many uses such as AM program transmission, television, loop direction finding, etc.
6. Measurement of radio noise: External noise fields, such as atmospheric, cosmic, and man-made, are measured in the same way as radio wave field strengths* with the exception that peak rather than average values of noise are usually of interest and that the overall bandpass action. of the measuring apparatus must be accurately known in measuring noise. When measuring noise varying over wide limits with time, such as atmospheric noise, it is generally best to employ automatic recorders.
Internal receiver and antenna noise may be measured by a standard signal generator connected to the receiver through a resistance equal to the calculated antenna radiation resistance. The amplitude of a single-frequency signal at the center of the pass band, when receiver output is $\sqrt{2}$ times the noise output with no signal, may be taken as equal to the noise amplitude.

* For methods of measuring field strengths and, hence, noise. see I.R.E. Siandards on Radio Wova Prooogation. Meas* uring Methods (1942). For information on sultable circuits to obtoin peak values, partieularly with resoect to man emade noise, see Agger, C. V., Foster, D. E., ond Young, C. S. Instruments and Methods of Measuring Radio Naise. Trans. Ad.E.E. Elec. Eng., Morch, 1940, vol. 59.


## Antennas

## Field intensity from an elementary dipole:

The elementary dipole forms the basis for many antenna computations. Since dipole theory assumes an antenna with current of constant magnitude and phase throughout its length, approximations to the elementary dipole are realized in practice only for antennas shorter than one-tenth wavelength. The theory can be applied directly to a loop whose circumference is less than one-tenth wavelength, thus forming a magnetic dipole. For larger antennas, the theory is applied by assuming the antenna to consist of a large number of infinitesimal dipoles with differences between individual dipoles of space position, polarization, current magnitude, and phase corresponding to the distribution of these parameters in the actual antenna. Field intensity equations for large antennas are then developed by integrating or otherwise summing the field vectors of the many elementary dipoles.

The outline below concerns electric dipoles. It also can be applied to magnetic dipoles by installing the loop perpendicular to the PO line at the center of the sphere in Fig. I. In this case, vector $h$ becomes $\epsilon$, the electric field; $\epsilon_{t}$ becomes the magnetic tangential field; and $\epsilon_{r}$ the radial magnetic field.

Fig. 1
Electric and magnetic components in spherical coordinatos for electric dipoles.


In the case of a magnetic dipole, Table I, showing variations of the field in the vicinity of the dipole, can also be used. $A_{r}$ is then the coefficient for the radial magnetic field; $A_{t}$ is the coefficient for the tangential magnetic field; $A_{h}$ is the coefficient for the electric field; $\phi_{r} ; \phi_{t ;}$ and $\phi_{h}$ being the phase angles corresponding to the coefficients.

[^19]Field infensily from an elementary dipole continued
For electric dipoles, Fig. 1 indicates the electric and magnetic field components in spherical coordinates with positive values shown by the arrows.
$r=$ distance $O M$

$$
\begin{aligned}
& \omega=2 \pi f \\
& \alpha=\frac{2 \pi}{\lambda}
\end{aligned}
$$

$\theta=$ angle POM measured
from $P$ toward $M$
$I=$ current in dipole
$c=$ velocity of light (see page 28)
$\lambda=$ wavelength
$v=\omega t-\alpha r$
$f=$ frequency
$1=$ length of dipole

The following equations expressed in electromagnetic units* (in vacuum) result:
$\epsilon_{r}=-\frac{c / \lambda I}{\pi} \frac{\cos \theta}{r^{3}}(\cos v-\alpha r \sin v)$ $\epsilon_{\ell}=+\frac{c \Lambda I}{2 \pi} \frac{\sin \theta}{r^{3}}\left(\cos v-\alpha r \sin v-\alpha^{2} r^{2} \cos v\right)$
$h=-I I \frac{\sin \theta}{r^{2}}(\sin v-\alpha r \cos v)$

- See pages 16 and 17.

Table I-Variations of the fleld in the vicinity of a dipole

| $\mathbf{r} / \boldsymbol{\lambda}$ | $\mathbf{I} / \boldsymbol{\alpha} \mathbf{r}$ | $\mathbf{A}_{\mathbf{r}}$ | $\boldsymbol{\phi}$ | $\mathbf{A}_{\mathbf{1}}$ | $\phi_{\mathbf{l}}$ | $\mathbf{A}_{\mathbf{h}}$ | $\boldsymbol{\phi}_{\mathbf{h}}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  |  |
| 0.01 | 15.9 | 4,028 | $3^{\circ} .6$ | 4,012 | $3^{\circ} .6$ | 253 | $93^{\circ} .6$ |
| 0.02 | 7.96 | 508 | $7^{\circ} .2$ | 500 | $7^{\circ} .3$ | 64.2 | $97^{\circ} .2$ |
| 0.04 | 3.98 | 65 | $14^{\circ} .1$ | 61 | $15^{\circ} .0$ | 16.4 | $104^{\circ} .1$ |
| 0.06 | 2.65 | 19.9 | $20^{\circ} .7$ | 17.5 | $23^{\circ} .8$ | 7.67 | $110^{\circ} .7$ |
| 0.08 | 1.99 | 8.86 | $26^{\circ} .7$ | 7.12 | $33^{\circ} .9$ | 4.45 | $116^{\circ} .7$ |
| 0.10 | 1.59 | 4.76 | $32^{\circ} .1$ | 3.52 | $45^{\circ} .1$ | 2.99 | $122^{\circ} .1$ |
| 0.15 | 1.06 | 1.66 | $42^{\circ} .3$ | 1.14 | $83^{\circ} .1$ | 1.56 | $132^{\circ} .3$ |
| 0.20 | 0.80 | 0.81 | $51^{\circ} .5$ | 0.70 | $114^{\circ} .0$ | 1.02 | $141^{\circ} .5$ |
| 0.25 | 0.64 | 0.47 | $57^{\circ} .5$ | 0.55 | $133^{\circ} .1$ | 0.75 | $147^{\circ} .5$ |
| 0.30 | 0.56 | 0.32 | $62^{\circ} .0$ | 0.48 | $143^{\circ} .0$ | 0.60 | $152^{\circ} .0$ |
| 0.35 | 0.45 | 0.23 | $65^{\circ} .3$ | 0.42 | $150^{\circ} .1$ | 0.50 | $155^{\circ} .3$ |
| 0.40 | 0.40 | 0.17 | $68^{\circ} .3$ | 0.37 | $154^{\circ} .7$ | 0.43 | $158^{\circ} .3$ |
| 0.45 | 0.35 | 0.134 | $70^{\circ} .5$ | 0.34 | $158^{\circ} .0$ | 0.38 | $160^{\circ} .5$ |
| 0.50 | 0.33 | 0.106 | $72^{\circ} .3$ | 0.30 | $160^{\circ} .4$ | 0.334 | $162^{\circ} .3$ |
| 0.60 | 0.265 | 0.073 | $75^{\circ} .1$ | 0.26 | $164^{\circ} .1$ | 0.275 | $165^{\circ} .1$ |
| 0.70 | 0.228 | 0.053 | $77^{\circ} .1$ | 0.22 | $166^{\circ} .5$ | 0.234 | $167^{\circ} .1$ |
| 0.80 | 0.199 | 0.041 | $78^{\circ} .7$ | 0.196 | $168^{\circ} .3$ | 0.203 | $168^{\circ} .7$ |
| 0.90 | 0.177 | 0.032 | $80^{\circ} .0$ | 0.175 | $169^{\circ} .7$ | 0.180 | $170^{\circ} .0$ |
| 1.00 | 0.159 | 0.026 | $80^{\circ} .9$ | 0.157 | $170^{\circ} .7$ | 0.161 | $170^{\circ} .9$ |
| 1.20 | 0.133 | 0.018 | $82^{\circ} .4$ | 0.132 | $172^{\circ} .3$ | 0.134 | $172^{\circ} .4$ |
| 1.40 | 0.114 | 0.013 | $83^{\circ} .5$ | 0.114 | $173^{\circ} .5$ | 0.114 | $173^{\circ} .5$ |
| 1.60 | 0.100 | 0.010 | $84^{\circ} .3$ | 0.100 | $174^{\circ} .3$ | 0.100 | $174^{\circ} .3$ |
| 1.80 | 0.088 | 0.008 | $84^{\circ} .9$ | 0.088 | $174^{\circ} .9$ | 0.088 | $174^{\circ} .9$ |
| 2.00 | 0.080 | 0.006 | $85^{\circ} .4$ | 0.080 | $175^{\circ} .4$ | 0.080 | $175^{\circ} .4$ |
| 2.50 | 0.064 | 0.004 | $86^{\circ} .4$ | 0.064 | $176^{\circ} .4$ | 0.064 | $176^{\circ} .4$ |
| 5.00 | 0.032 | 0.001 | $88^{\circ} .2$ | 0.032 | $178^{\circ} .2$ | 0.032 | $178^{\circ} .2$ |

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## Field intensity from an elementary dipole cantinued

These formulas are valid for the elementary dipole at distances which are large compared with the dimensions of the dipole. Length of the dipole must be small with respect to the wavelength, say $\frac{l}{\lambda}<0.1$. The formulas are for a dipole in free space. If the dipole is placed vertically on a plane of infinite conductivity, its image should be taken into account, thus doubling the above values.

## Field of an elementary dipole at great distance

When distance $r$ exceeds five wavelengths, as is generally the case in radio applications, the product $\alpha r=2 \pi \frac{r}{\lambda}$ is large and lower powers in $\alpha r$ can be neglected. The radial electric field $\epsilon_{r}$ then becomes negligible with respect to the tangential field and

$$
\left.\begin{array}{l}
\epsilon_{r}=0  \tag{2}\\
\epsilon_{\ell}=-\frac{2 \pi c l I}{\lambda_{r}} \sin \theta \cos (\omega f-\alpha r) \\
h=-\frac{\epsilon_{\ell}}{c}
\end{array}\right\}
$$

## Field of an elementary dipole at short disfance

In the vicinity of the dipole $\left(\frac{r}{\lambda}<0.01\right)$, $\alpha r$ is very small and only the first terms between parantheses in equations (1) remain. The ratio of the radial and tangential field is then
$\frac{\epsilon_{r}}{\epsilon_{i}}=-2 \cot \theta$
Hence, the radial field at short distance has a magnitude of the same order as the tangential field. These two fields are in opposition. Further, the ratio of the magnetic and electric tangential field is
$\frac{h}{\epsilon_{\ell}}=-\frac{\alpha r}{c} \frac{\sin v}{\cos v}$
The magnitude of the magnetic field at short distances is, therefore, extremely small with respect to that of the tangential electric field, relative to their relationship at great distances. The two fields are in quadrature. Thus, at short distances, the effect of the dipole on an open circuit is much greater than on a closed circuit as compared with the effect at remote points.

## Field of an elementary dipole at intermediate distance

At intermediate distance, say between 0.01 and 5.0 wavelengths, one should take into account all the terms of the equations (1). This case occurs, for instance, when studying reactions between adjacent antennas. To calculate the fields, it is convenient to transform the equations as follows:
$\left.\begin{array}{l}\epsilon_{r}=-2 \alpha^{2} c l I \cos \theta A_{r} \cos \left(v+\phi_{r}\right) \\ \epsilon_{\ell}=\alpha^{2} c l I \sin \theta A_{l} \cos \left(v+\phi_{l}\right) \\ h=\alpha^{2} I I \sin \theta A_{h} \cos \left(v+\phi_{h}\right)\end{array}\right\}$
where
$\left.\begin{array}{ll}A_{r}=\frac{\sqrt{1+(\alpha r)^{2}}}{(\alpha r)^{3}} & \tan \phi_{r}=\alpha r \\ A_{t}=\frac{\sqrt{1-(\alpha r)^{2}+(\alpha r)^{4}}}{(\alpha r)^{3}} \cot \phi_{r}=\frac{1}{\alpha r}-\alpha r \\ A_{h}=\frac{\sqrt{1+(\alpha r)^{2}}}{(\alpha r)^{2}} \quad \cot \phi_{h}=-\alpha r\end{array}\right\}$
Values of A's and $\phi$ 's are given in Table I as a function of the ratio between the distance $r$ and the wavelength $\lambda$. The second column contains values of $\frac{1}{\alpha r}$ which would apply if the fields $\epsilon_{\boldsymbol{t}}$ and $h$ behaved as at great distances. $\alpha r$

## Field intensity from a vertically polarized

## antenna with base close to ground

The following formula is obtained from elementary dipole theory. and is applicable to low frequency antennas. It assumes that the earth is a parfect reflector, the antenna dimensions are small compared with $\lambda$, and the actual height does not exceed $\frac{\lambda}{4}$.
The vertical component of electric field radiated in the ground plane, at distances so short that ground attenuation may be neglected lusually when $D<10 \lambda 1$, is given by
$E=\frac{377 I H^{\prime}}{\lambda D}$
where
$E=$ field intensity in millivolts per meter
$I=$ current at base of antenna in amperes
$H_{s}=$ effective height of antenna
$\lambda=$ wavelength in same units as $H$
$D=$ distance in kilometers

## Field infensity from a vertically polarized

anfenna with base close to ground
continued
The effective height of a grounded vertical antenna is equivalent to the height of a vertical wire producing the same field along the horizontal as the actual antenna, provided the vertical wire carries a current that is constant along its entire length and of the same value as at the base of the actual antenna. Effective height depends upon the geometry of the antenna and varies slowly with $\boldsymbol{\lambda}$. For types of antennas normally used at low and medium frequencies, it is roughly one-half to two-thirds the actual height of the antenna.

For certain antenna configurations effective height can be calculated by the following formulas

1. Straight vertical antenna $\left(h ₹ \frac{\lambda}{4}\right)$
$H_{e}=\frac{\lambda}{\pi \sin \frac{2 \pi h}{\lambda}} \sin ^{2}\left(\frac{\pi h}{\lambda}\right)$
where $h=$ actual height
2. Loop antenna $\left(A<0.001 \lambda^{2}\right)$
$H_{e}=\frac{2 \pi n A}{\lambda}$
where $A=$ mean area per turn of loop
$n=$ number of turns
3. Adcock antenna
$H_{a}=\frac{2 \pi a b}{\lambda}$
where
$a_{0}=$ height of antenna
$b=$ spacing between antennas
In the above formulas, if $H_{c}$ is desired in meters or feet, all dimensions $h, A$, $a, b$, and $\lambda$ must be in meters or feet respectively.

## Vertical radiafors

The field intensity from a single vertical tower insulated from ground and either of self-supporting or guyed construction, such as is commonly used for medium-frequency broadcasting, may be calculated by the following

## Vertical radiafors continued

formula. This is more accurate than the formula given on page 253. Near ground level the formula is valid within the range $2 \lambda<D<10 \lambda$.
$E=\frac{60 I}{D \sin 2 \pi \frac{h}{\lambda}}\left[\frac{\cos \left(2 \pi \frac{h}{\lambda} \cos \theta\right)-\cos 2 \pi \frac{h}{\lambda}}{\sin \theta}\right]$
where
$E=$ field intensity in millivolts per meter
$I=$ current at base of antenna in amperes
$h=$ height of antenna
$\lambda=$ wavelengths in same units as $h$
$D=$ distance in kilometers
$\theta=$ angle from the vertical
Radiation patterns in the vertical plane for antennas of various heights are shown in Fig. 2. Field intensity along the horizontal as a function of antenna height for one kilowatt radiated is shown in Fig. 3.


Fig. 2-Field strength as a function of ongle of elevation for vertical rodiafors of different heights.

## Vertical radiators continued

Both Figs. 2 and 3 assume sinusoidal distribution of current along the antenna and perfect ground conductivity. Current magnitudes for one-kilowatt power used in calculating Fig. 3 are also based on the assumption that the only resistance is the theoretical radiation resistance of a vertical wire with sinusoidal current.

Since inductance and capacitance are not uniformly distributed along the tower and since current is attenuated in traversing the tower, it is impossible to obtain sinusoidal current distribution in practice. Consequently actual radiation patterns and field intensities differ from Figs. 2 and 3.* The closest approximation to sinusoidal current is found on constant cross-section towers.


Fig. 3-Fieid strength alang the horizontol os a function of antenna height for a vertical grounded rodiatior with one kilowatt rodiated power.

In addition, antenna efficiencies vary from about 70 percent for 0.15 wavelength physical height to over 95 percent for 0.6 wavelength height. The input power must be multiplied by the efficiency to obtain the power radiated.

Average results of measurements of impedance at the base of several actual

[^20]
## Vertical radiafors continued

vertical radiators, as given by Chamberlain and Lodge, are shown in Fig. 4. For design purposes when actual resistance and current of the projected radiator are unknown, resistance values may be selected from Fig. 4 and


Fig. 4-Resistance and reactonce components of impedance between tower base and ground of vertical radiators as given by Chamberialn and Lodge. Solid lines show average results for 5 guyed towers; dotted lines show average results for 3 selfsupporting towers.

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## Vertical radiafors continued

the resulting effective current obtained from the following equation

$$
\begin{equation*}
I_{\theta}=\sqrt{\frac{W \eta}{R}} \tag{6}
\end{equation*}
$$

where

$$
I_{\theta}=\text { current effective in producing radiation in amperes }
$$

W = watts input

$$
\begin{aligned}
& \eta=\text { antenna efficiency, varying from } 0.70 \text { at } \frac{h}{\lambda}=0.15 \\
& \text { to } 0.95 \text { at } \frac{h}{\lambda}=0.6
\end{aligned}
$$

$R=$ resistance at base of antenna in ohms
If $I_{e}$ from (6) is substituted in (5), reasonable approximations to the field intensity at unit distances, such as one kilometer or one mile, will be obtained.

The practical equivalent of a higher tower may be secured by adding a capacitance "hat" with or without tuning inductance at the top of a lower tower.*

A good ground system is important with vertical-radiator antennas. It should consist of at least 120 radial wires, each one-half wavelength or longer, buried 6 to 12 inches below the surface of the soil. A ground screen of highconductivity metal mesh, bonded to the ground system, should be used on or abcve the surface of the ground adjacent to the tower.

For additional infomation see Brown, G. H., Proc. I.R.E., vol. 24, p. 48 Uanuary, 1936) and Brown, G. H. and leitch J. G., vol. 25, p. 533 imay, 1937.

## Field infensity and radiafed power from

## a half-wave dipole in free space

Fig. 5 on page 259 shows the field intensity and radiated power from a half-wave dipole in free space. The following formulas apply:
input power $W=I^{2} R=I^{2}$ (73.12) watts
Radiated power $P=\frac{30 I^{2}}{\pi d^{2}}=\frac{0.1306 \mathrm{~W}}{d^{2}}$ watts per square meter
Electric field intensity $E=\frac{60 I}{d}=\frac{7.02 \sqrt{ } \bar{W}}{d}$ volts per meter
$I=$ maximum current on dipole in rms amperes
$R=$ radiation resistance $=73.12$ ohms
$d=$ distance from antenna in meters


[^21]
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Table II-Radiation from an end-fed conductor of any length in space

| configuration (length of radiator) | $\begin{gathered} \text { expression for intensity } \\ F(\theta) \\ \hline \end{gathered}$ |
| :---: | :---: |
| Half wave resonant | $\frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta}$ |
| Any odd number of half waves resonant | $\frac{\cos \left(\frac{l^{\circ}}{2} \sin \theta\right)}{\cos \theta}$ |
| Any even number of half waves resonant | $\frac{\sin \left(\frac{l^{\circ}}{2} \sin \theta\right)}{\cos \theta}$ |
| Any length resonant | $\begin{aligned} \frac{1}{\cos \theta}[ & 1+\cos ^{2} l^{\circ}+\sin ^{2} \theta \sin ^{2} l^{\circ} \\ & -2 \cos \left(l^{\circ} \sin \theta\right) \cos l^{\circ} \\ & \left.-2 \sin \theta \sin \left(l^{\circ} \sin \theta\right) \sin l^{\circ}\right]^{\frac{1}{2}} \end{aligned}$ |
| Any length non-resonant | $\tan \frac{\theta}{2} \sin \frac{l^{\circ}}{2}(1-\sin \theta)$ |

$1^{\circ}=$ Length of radiator in electrical degrees, energy to flow from left-hand end of radiator.
$\theta=$ angle from the vertical
$\lambda=$ wavelength


## Maxima and minima of radiation from a single-wire radiator



Fig. 6.

## Rhombic antennas

Linear radiators may be combined in various ways to form antennas such as the horizontal vee, inverted vee, etc. The type most commonly used at high frequencies is the horizontal terminated rhombic shown in Fig. 7.


Fig. 7.
In designing rhombic antennas* for high-frequency radio circuits, the desired vertical angle $\Delta$ of radiation above the horizon must be known or assumed. When the antenna is to operate over a wide range of radiation angles or is to operate on several frequencies, compromise values of $H, L$, and $\phi$ must

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Rhombic antennas continued
be selected. Gain of the antenna increases as the length of $L$ of each side is increased; however, to avoid too-sharp directivity in the vertical plane, it is usual to limit $L$ to less than six wavelengths.


Fig. 8-Rhombic antenna design chart.
Knowing the side length and radiation angle desired, the height H above ground and the tilt angle $\phi$ can be obtained from Fig. 8 as in the following example:
Problem: Find $H$ and $\phi$ if $\Delta=20^{\circ}$ and $L=4 \lambda$.
Solution: On Fig. 8 draw a vertical line from $\Delta=20^{\circ}$ to meet $\frac{L}{\lambda}=4$ curve and $\frac{H}{\lambda}$ curves. From intersection at $\frac{L}{\lambda}=4$, read on the right-hand

## Rhombic antennas

cantinued
scale $\phi=71.5^{\circ}$. From intersection on $\frac{H}{\lambda}$ curves, there are two possible values on the left-hand scale

1. $\frac{H}{\lambda}=0.74$ or $H=0.74 \lambda$
2. $\frac{H}{\lambda}=219$ or $H=219 \lambda$

Similarly, with an antenna $4 \lambda$ on the side and a tilt angle $\phi=71.5^{\circ}$, working backwards, it is found that the angle of maximum radiation $\Delta$ is $20^{\circ}$, if the antenna is $0.74 \lambda$ or $2.19 \lambda$ above ground.

## Antenna arrays

The basis for all directivity control in antenna arrays is wave interference. By providing a large number of sources of radiation, it is possible with a fixed' amount of power greatly to reinforce radiation in a desired direction by suppressing the radiation in undesired directions. The individual sources may be any type of antenna.
Expressions for the radiation pattern of several common types of individual elements are shown in Table III but the array expressions are not limited to them. The expressions hold for linear radiators, rhombics, vees, horn radiators, or other complex antennas when combined into arrays, provided a suitable expression is used for $A$, the radiation pattern of the individual antenna. The array expressions are multiplying factors. Starting with an individual antenna having a radiation pattern given by $A$, the result of combining it with similar antennas is obtained by multiplying $A$ by a suitable array factor, thus obtaining an $A^{\prime}$ for the group. The group may then be treated as a single source of radiation. The result of combining the group with similar groups or, for instance, of placing the group above ground, is obtained by multiplying $A^{\prime}$ by another of the array factors given.
The expressions given here assume negligible mutual coupling between individual antennas. When coupling is not negligible, the expressions apply only if the feeding is adjusted to overcome the coupling and thus produce resultant currents which are equal or binomial in amplitude and of the relative phases indicated.

One of the most important arrays is the linear multi-element array where a large number of equally spaced antenna elements are fed equal currents in phase to obtain maximum directivity in the forward direction. Table IV gives expressions for the radiation pattern of several particular cases and the general case of any number of broadside elements.

In this type of array, a great deal of directivity may be obtained. A large number of minor lobes, however, are apt to be present and they may be undesirable under some conditions, in which case a type of array, called the Binomial array, may be used. Here again all the radiators are fed in phase

## Table III-Radiation patterns of several common types of antennas

| type of radiat or | $\begin{aligned} & \text { current } \\ & \text { distribution } \end{aligned}$ | directivity <br> horizontal $F(\theta)$ | vertical |
| :---: | :---: | :---: | :---: |
| Half-wave dipole |  | $\begin{aligned} F(9) & = \\ K & \frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta} \\ & \cong K \cos \theta \end{aligned}$ | $F(\beta)=K(1)$ |
| Shortened dipole |  | $F(\theta) \cong K \cos \theta$ | $F(\beta)=K(1)$ |
| Lengthened dipole |  | $\begin{aligned} & F(\theta)= \\ & K\left[\frac{\cos \left(\frac{\pi l}{\lambda} \sin \theta\right)-\cos \frac{\pi l}{\lambda}}{\cos \theta}\right] \end{aligned}$ | $F(\beta)=K(1)$ |
| Horizontal loop |  | $F(\theta) \cong K(1)$ | $F(\beta)=K \cos \beta$ |
| Horizontal turnstile | $i_{1}$ and $i_{2}$ phased $90^{\circ}$ | $F(\theta) \cong K^{\prime}(1)$ | $F(\beta) \cong K^{\prime}(1)$ |

$\theta=$ horizontal angle measured from perpendicular bisecting plane
$\beta=$ vertical angle measured from horizon
$K$ and $K^{\prime}$ are constants and $K^{\prime} \cong 0.7 K$

## Antenna arrays

 continuedbut the current is not distributed equally among the array elements, the center radiators in the array being fed more current than the outer ones. Table $V$ shows the configuration and general expression for such an array. In this case the configuration is made for a vertical stack of loop attennas

Table IV—Linear multi-element array broadside directivity
expression for intensity $F(\theta)$
$A=1$ for horizontal loop, vertical dipole
$A=\frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta}$ for horizontal dipole
$s^{\circ}=$ spacing of successive elements in degrees

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## Antenna arrays continued

in order to obtain single-lobe directivity in the vertical plane. If such an array were desired in the horizontal plane, say $n$ dipoles end to end, with the specified current distribution the expression would be
$F(\theta)=2^{n-1}\left[\frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta}\right] \cos ^{n-1}\left(\frac{1}{2} S^{\circ} \sin \theta\right)$
The term binomial results from the fact that the current intensity in the successive array elements is in accordance with the binomial expansion $11+1)^{n-1}$, where $n$ is the number of elements.

## Examples af use af Tables III, IV, V, and VI

Problem 1: Find horizontal radiation pattern of four colinear horizontal dipoles, spaced successively $\frac{\lambda}{2}\left(180^{\circ}\right)$.
Solution: From Table IV radiation from four radiators spaced $180^{\circ}$ is given by $F(\theta)=4 A \cos \left(180^{\circ} \sin \theta\right) \cos \left(90^{\circ} \sin \theta\right)$.

From Table III the horizontal radiation of a half-wave dipole is given by
$A=K \frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta} ;$
therefore, the total radiation
$F(\theta)=K\left[\frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta}\right] \cos \left(180^{\circ} \sin \theta\right) \cos \left(90^{\circ} \sin \theta\right)$
Problem 2: Find vertical radiation pattern of four horizontal dipoles, stacked one above the other, spaced $180^{\circ}$ successively.

Solution: From Table IV we obtain the general equation of four radiators, but since the spacing is vertical, the expression should be in terms of vertical angle $\beta$.
$F(\beta)=4 A \cos \left(180^{\circ} \sin \beta\right) \cos \left(90^{\circ} \sin \beta\right)$.
From Table Ill we find that the vertical radiation from a horizontal dipole lin the perpendicular bisecting planel is non-directional. Therefore the vertical pattern is
$F(\beta)=K(1) \cos \left(180^{\circ} \sin \beta\right) \cos \left(90^{\circ} \sin \beta\right)$

## Table V-Development of binomial array

configuration of array
and in general:

$$
2^{n-1} \cos \beta\left[\cos ^{n-1}\left(\frac{5^{\circ}}{2} \sin \beta\right)\right]
$$

where $n$ is the number of loops in the array

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## Anfenna arrays continued

Problem 3: Find horizontal radiation pattern of group of dipoles in problem 2.

Solution: From Table III.
$F(\theta)=K \frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta} \cong K \cos \theta$
Problem 4: Find the vertical radiation pattern of stack of five loops spaced $2 / 3 \lambda\left(240^{\circ}\right)$ one above the other, all currents equal in phase and amplitude.

Solution: From Table IV, using vertical angle because of vertical stacking,
$F(\beta)=A \frac{\left.\sin \left(51120^{\circ}\right) \sin \beta\right]}{\sin \left(120^{\circ} \sin \beta\right)}$
From Table III, we find A for a horizontal loop in the vertical plane
$A=F(\beta)=K \cos \beta$
Total radiation pattern
$F(\beta)=K \cos \beta \frac{\sin \left[5\left(120^{\circ}\right) \sin \beta\right]}{\sin \left(120^{\circ} \sin \beta\right)}$
Problem 5: Find radiation pattern (vertical directivity) of the five loops in problem 4, if they are used in binomial array. Find also current intensities in the various loops.

Solution: From Table V
$F(\beta)=K \cos \beta\left[\cos ^{4}\left(120^{\circ} \sin \beta\right)\right]$
(all terms not functions of vertical angle $\beta$ combined in constant $K$ )
Current distribution $(1+1)^{4}=1+4+6+4+1$, which represent the current intensities of successive loops in the array.

Problem 6: Find horizontal radiation pattern from two vertical dipoles spaced one-quarter wavelength apart when their currents differ in phase by $90^{\circ}$.

Solution: From Table VI
$s^{\circ}=\frac{\lambda}{4}=90^{\circ}=$ spacing
$\phi=90^{\circ}=$ phase difference
$F(\theta)=2 A \cos \left(45 \sin \theta+45^{\circ}\right)$

Anfenna arrays continued

## Table VI-Supplementary problems

A—two radiators any phase $\phi$ expression for intenstity

B-radiator above ground thorizontal polarizationl


C-radiator parallel to screen


$$
\begin{aligned}
& F(\beta)=2 A \sin \left(d^{\circ} \cos \beta\right) \\
& \text { or } \\
& F(\theta)=2 A \sin \left(d^{\circ} \cos \theta\right)
\end{aligned}
$$

$s^{0}=$ spacing in electrical degrees
$h_{1}{ }^{\circ}=$ height of radiator in electrical degrees
$d^{\circ}=$ spacing of radiator from.screen in electrical degrees

Anfenna arrays cantinued
Problem 7: Find the vertical radiation pattern and the number of nulls in the vertical pattern $(0 \leq \beta \leq 90$ ) from a horizontal loop olaced three wavelengths above ground.

## Solution:

$h_{1}{ }^{\circ}=3(360)=1080^{\circ}$
From Table VI
$F(\beta)=2 A \sin (1080 \sin \beta)$
From Table III for loop antennas
$A=K \cos \beta$
Total vertical radiation pattern
$F(\beta)=K \cos \beta \sin (1080 \sin \beta)$
A null occurs wherever $F(\beta)=0$.
The first term, $\cos \beta$, becomes 0 when $\beta-90^{\circ}$.
The second term, $\sin (1080 \sin \beta)$, becomes 0 whenever the value inside the parenthesis becomes a multiple of $180^{\circ}$. Therefore, number of nulls equal
$1+\frac{h_{1}{ }^{\circ}}{180}=1+\frac{1080}{180}=7$.
Problem 8: Find the vertical and horizontal patterns from a horizontal half-wave dipole spaced $\frac{\lambda}{8}$ in front of a vertical screen.

## Solution:

$d^{\circ}=\frac{\lambda}{8}=45^{\circ}$
From Table VI
$F(\beta)=2 A \sin \left(45^{\circ} \cos \beta\right)$
$F(\theta)=2 A \sin \left(45^{\circ} \cos \theta\right)$
From Table III for horizontal half-wave dipole
Vertical pattern $A=K(l)$
Horizontal pattern $A=K \frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta}$
Total radiation patterns are
Vertical: $\mathrm{F}(\beta)=K \sin \left(45^{\circ} \cos \beta\right)$
Horizontal: $F(\theta)=K \frac{\cos \left(\frac{\pi}{2} \sin \theta\right)}{\cos \theta} \sin \left(45^{\circ} \cos \theta\right)$.

## Anfenna arrays continued


spacing $s^{\circ}$ (electrical degrees)
$F(\beta)=\frac{\sin \left(\frac{n s^{\circ}}{2} \sin \beta\right)}{\sin \left(\frac{s^{\circ}}{2} \sin \beta\right)} \cos \beta$
$\mathrm{n}=$ number of loops


Gain $(d b)=10 \log _{10}\left[\frac{1}{\frac{1}{n}+\frac{3}{n^{2}} \sum_{k=1}^{n-1}(n-k)\left[-\frac{2 \cos k s^{\circ}}{\left(k s^{\circ}\right)^{2}}+\frac{2 \sin k s^{\circ}}{\left(k s^{\circ}\right)^{3}}\right]}\right]$
Fig. 9-Gain of linear array of loops vertically stacked.

## $\square$ Non-sinusoidal and modulated wave forms

## Relaxation oscillators

## Gas tube oscillator


$A=$ pulse output
$B=$ sawtonth output
Typical circuit
$V_{1}=884$
$C_{1}=0.05 \mu \mathrm{f}$
$C_{2}=0.05 \mu \mathrm{f}$
$R_{1}=100,000$ ohms
$R_{2}=500$ ohms
$R_{3}=100,000$ ohms
Frequency controlling elements $C_{2}, R_{3}$

Feedback relaxation oscillator


Typical circuit
$V_{1}=6 F 6$
$T_{1}=3: 1$ audio transformer
0.3 henry primary
$R_{1}=100,000$ ohms
$R_{2}=5000$ ohms
$C_{1}=1 \mu \mathrm{f}$
$C_{2}=0.1 \mu \mathrm{f}$
Frequency controlling elements $C_{2}, R_{2}$

Blocking oscillator


Typical circuit
$V_{1}=6 \mathrm{~J} 5$
$C_{1}=0.01 \mu \mathrm{f}$
$C_{2}=0.25 \mu \mathrm{f}$
$R_{1}=1$ megohm
$R_{2}=1$ megohm
$R_{3}=1000$ ohms
Frequency controlling elements
$R_{1}, C_{2}, R_{2}$

## Relaxation oscillators continued

## Squegging oscillator



Typical circuit
$V_{1}=615$
$\left.\begin{array}{l}L_{1} \\ L_{2}\end{array}\right\}$ tightly coupled
$R_{1}=500,000$ ohms
$\mathrm{C}_{1}=0.01 \mu \mathrm{f}$
Frequency controlling elements $R_{L} C_{1}$

Multivibrator


Typical circuit
$V_{1}=658$
$R_{1}=100,000$ ohms
$R_{2}=1000$ ohms
$R_{3}=25,000$ ohms
$R_{4}=250,000$ ohms
$R_{5}=25,000$ ohms
$\mathrm{C}_{1}=0.01 \mu \mathrm{f}$
$\mathrm{C}_{2}=250 \mu \mu \mathrm{f}$
Frequency controlling elements
$R_{1} R_{2}, R_{1}, C_{2}$
van der Pol oscillator


Typical circuit
$V_{1}=6 \mathrm{SJ7}$
$R_{1}=100,000$ ohms
$R_{2}=5000 \mathrm{hms}$
$R_{3}=1000 \mathrm{hms}$
$R_{4}=3,000$ ohms
$R_{5}=10,000$ ohms
$R_{6}=25,000$ ohms
$R_{7}=25,000$ ohms
Frequency controlling elements
$R_{1}, R_{G}, C_{1}$ lalso $B+1$

## Electronic infegration methods



## Electronic infegration methods



## Mathods I and II

a. Voltage $V$ must be obtained from a low-impedance source.
b. $\frac{L}{R} \gg T$ or $\frac{M}{R} \gg T$
c. The output $E$ should not react back on the input voltage $V$.
d. The impedance into which the integrator circuit works should be large compared with $R$. If this impedance is resistive, it should be included as part of $R$ (this also applies to the input source impedance).

## Method III

a. Voltage $V$ must be obtained from a low-impedance source.
b. $R C \gg T$
c. The output $E$ should not react back on the input voltage $V$.
d. The impedance into which the integrator circuit works should be as large as possible. If this impedance is resistive $r$ then

$$
r C \gg R C
$$

The source impedance should be included in $R$.

## Method IV

a. Current $I$ should be a replica of the input voltage wave-form $V$.
b. The discharge device allows for integration between limits. If discharge device is not used, the circuit will integrate until $E$ equals the $B+$ voltage.
c. The impedance into which the integrator circuit works should be as large as possible. If this impedance is resistive $r$ then $r C \gg T$.

## Electronic differentiation methods



I or $V$ is the change of current or voltage in time $T$


## NON-SINUSOIDAL AND MODULATED WAVE FORMS

## Electronic differentiation methods

 continued
## Methods I and II

a. Current I should be a replica of the input voltage wave-form V.
b. The voltage $V$ must be substantially independent of the back emf developed by the inductance $L$.
c. The output shunt impedance placed across $E$ should be high compared to the network impedance.
d. The resonant period associated with the inductance caused by shunting circuit capacitances should be at least one-third the build-up time $T$.

## Method III

a. Voltage $V$ must be obtained from a low-impedance source.
b. The $R C$ product should be one-fiftieth of the build-up time $T$ or smaller.
c. The output voltage $E$ should not react back on the input voltage $V$.
d. The impedance into which the differentiator circuit works should be large compared with $R$. If this impedance is resistive, it should be included as part of R. (This also applies to the input source impedance.)

## Fourier analysis of recurrent wave forms

## General formulas



$$
\begin{align*}
F(\theta)= & \frac{B_{0}}{2}+A_{1} \sin \theta+A_{2} \sin 2 \theta+\ldots+A_{n} \sin n \theta \\
& +B_{1} \cos \theta+B_{2} \cos 2 \theta+\ldots B_{n} \cos n \theta \tag{I}
\end{align*}
$$

Formula (1) may be written

$$
\begin{align*}
F(\theta)= & \frac{B_{0}}{2}+C_{1} \cos \left(\theta-\phi_{1}\right)+C_{2} \cos \left(2 \theta_{2}^{*}-\phi_{2}\right)+\ldots \\
& +C_{n} \cos \left(n \theta-\phi_{n}\right) \tag{2}
\end{align*}
$$

where

$$
\begin{align*}
C_{n} & =\sqrt{A_{n}^{2}+B_{n}^{2}}  \tag{3}\\
\phi_{n} & =\arctan \frac{A_{n}}{B_{n}} \tag{4}
\end{align*}
$$

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## Fourier analysis of recurrent wave forms continued

The coefficients $A_{n}$ and $B_{n}$ are determined by the following formulas:

$$
\begin{align*}
& A_{n}=\frac{1}{\pi} \int_{-\pi}^{\pi} F(\theta) \sin n \theta d \theta  \tag{5}\\
& B_{n}=\frac{1}{\pi} \int_{-\pi}^{\pi} F(\theta) \cos n \theta d \theta \tag{6}
\end{align*}
$$

By a change of limits equations (5) and (6) may also be written

$$
\begin{align*}
& A_{n}=\frac{1}{\pi} \int_{0}^{2 \pi} F(\theta) \sin n \theta d \theta  \tag{7}\\
& B_{n}=\frac{1}{\pi} \int_{0}^{2 \pi} F(\theta) \cos n \theta d \theta \tag{8}
\end{align*}
$$

If the function $F(\theta)$ is an odd function, that is
$F(\theta)=-F(-\theta)$
the coefficients of all the cosine terms $\left(B_{n}\right)$ of equation (6) become equal to zero.

Similarly if the function $F(\theta)$ is an even function, that is
$F(\theta)=F(-\theta)$
the coefficients of all the sine terms $\left(A_{n}\right)$ of equation (5) become equal to zero.

If the function to be analyzed is thus a symmetrical function defined by either equation (9) or (10) the function should be disposed about the zero axis and an analysis obtained by means of equations (5) or (6) for the simplest solution.

Fourier analysis of recurrent wave forms continued

## Graphical solution

If the function to be analyzed is not known analytically, a solution of the Fourier integral may be approximated by graphical means.

The period of the function is divided into a number of ordinates as indicated by the graph.


The values of these ordinates are recorded and the following computations made:

|  | $Y_{0}$ | $Y_{1}$ | $Y_{2}$ | $Y_{3}$ | $Y_{4}$ | $Y_{5}$ | $Y_{6}$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- |
|  |  | $Y_{11}$ | $Y_{10}$ | $Y_{9}$ | $Y_{8}$ | $Y_{7}$ |  |
|  | $S_{0}$ | $S_{1}$ | $S_{2}$ | $S_{3}$ | $S_{4}$ | $S_{5}$ | $S_{6}$ |
| Sum ifference | $d_{1}$ | $d_{2}$ | $d_{3}$ | $d_{4}$ | $d_{5}$ |  |  |

The sum terms are arranged as follows:

Sum

| $S_{0}$ | $S_{1}$ | $S_{2}$ | $S_{3}$ | $(12)$ | $\overline{S_{0}}$ | $\overline{S_{1}}$ |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- |
| $S_{6}$ | $S_{5}$ | $S_{4}$ |  |  | $\overline{S_{2}}$ | $\overline{S_{3}}$ |
| $\overline{S_{0}}$ | $\overline{S_{1}}$ | $\overline{S_{2}}$ | $\overline{S_{3}}$ |  | $\overline{S_{7}}$ | $\overline{S_{3}}$ |
| $\overline{D_{0}}$ | $\overline{D_{1}}$ | $D_{2}$ |  |  |  |  |

Difference $\begin{array}{lll}\overline{D_{0}} & \overline{D_{1}} & D_{2}\end{array}$

The difference terms are as follows:


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Fourier analysis of recurrent wave forms continued

The coefficients of the Fourier series are now obtained as follows, where $A_{0}$ equals the average value, the $B_{1} \ldots n$ expressions represent the coefficients of the cosine terms, and the $A_{1} \ldots n_{n}$ expressions represent the coefficients of the sine terms:

$$
\begin{align*}
& B_{0}=\frac{\overline{S_{7}}+\overline{S_{8}}}{12}  \tag{16}\\
& B_{1}=\frac{\overline{D_{0}}+0.866 \overline{D_{1}}+0.5 \overline{D_{2}}}{6}  \tag{17}\\
& B_{2}=\frac{\overline{S_{0}}+0.5 \overline{S_{1}}-0.5 \overline{S_{2}}-\overline{S_{3}}}{6}  \tag{18}\\
& B_{3}=\frac{\overline{D_{6}}}{6} \tag{19}
\end{align*}
$$

$B_{4}=\frac{\overline{S_{0}}-0.5 \overline{S_{1}}-0.5 \overline{S_{2}}+\overline{S_{8}}}{6}$
$B_{5}=\frac{\overrightarrow{D_{0}}-0.866 \overline{D_{1}}+0.5 \overline{D_{2}}}{6}$
$B_{6}=\frac{\overline{S_{7}}-\overline{S_{8}}}{12}$
also
$A_{1}=\frac{0.5 \overline{S_{4}}+0.866 \overline{S_{5}}+\overrightarrow{S_{6}}}{6}$
$A_{2}=\frac{0.866\left(\overline{D_{3}}+\overline{\left.D_{4}\right)}\right.}{6}$
$A_{8}=\frac{\overrightarrow{D_{5}}}{6}$
$A_{1}=\frac{0.866\left(D_{3}-D_{4}\right)}{6}$
$A_{5}=\frac{0.5 \overline{S_{4}}-0.866 \overrightarrow{S_{5}}+\overline{S_{6}}}{6}$

## Analyses of commonly encounfered wave forms

The following analyses include the coefficients of the Fourier series for all harmonics $\mathrm{In}^{\text {th }}$ orderl. By the use of the graph for the $\left(\frac{\sin x}{x}\right)$ function, where $f(x)$ is even, the amplitude coefficients may be evaluated in a simple manner.


The symbols used are defined as follows:

| $A=$ pulse amplitude | $r=$ pulse decay time |
| :--- | :--- |
| $T=$ periodicity | $n=$ order of harmonic |
| $d=$ pulse width | $C_{n}=$ amplitude of $n^{t h}$ harmonic |
| $f=$ pulse build-up time | $\theta_{n}=$ phase angle of $n^{\text {n }}$ harmonic |

$A_{a v}=$ average value of function $=\frac{1}{T} \int_{0}^{T} F(t) d t$
$A_{r m s}=$ root-mean square value of function $=\sqrt{\frac{1}{T}} \int_{0}^{T}[F(t)]^{2} d t$

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## Analyses of commonly encounfered wave forms continued

## 1. Rectangular wave



$$
A_{a v}=\frac{A d}{T}
$$

$$
A_{\mathrm{omo}}=A \sqrt{\frac{d}{T}}
$$

$$
C_{n}=2 A_{a v}\left[\frac{\sin \frac{n \pi d}{T}}{\frac{n \pi d}{T}}\right]
$$

2. Symmetrical trapezoid wave


$$
\begin{array}{ll}
A_{a v}=A \frac{(f+d)}{T} & A_{r m s}=A \sqrt{\frac{2 f+3 d}{3 T}} \\
C_{n}=2 A_{a v}\left[\frac{\sin \frac{n \pi f}{T}}{\frac{n \pi f}{T}}\right]\left[\frac{\sin \frac{n \pi(f+d)}{T}}{\frac{n \pi(f+d)}{T}}\right] &
\end{array}
$$

## Analyses of commonly encountered wave forms continued

## 3. Unsymmetrical trapezoid wave



$$
A_{a v}=\frac{A}{T}\left[\frac{f}{2}+\frac{r}{2}+d\right] \quad A_{r m s}=A \sqrt{\frac{f+r+3 d}{3 T}}
$$

If $f \cong r$

$$
C_{n}=2 A_{a v}\left[\frac{\sin \frac{n \pi f}{T}}{\frac{n \pi f}{T}}\right]\left[\frac{\sin \frac{n \pi(f+d)}{T}}{\frac{n \pi(f+d)}{T}}\right]\left[\frac{\sin \frac{n \pi(r-f}{T}}{\frac{n \pi(r-f)}{T}}\right]
$$

4. Isosceles triangle wave


$$
A_{a v}=\frac{A f}{T} \quad A_{r m s}=A \sqrt{\frac{2 f}{3 T}}
$$

$$
C_{n}=2 A_{a v}\left[\frac{\sin \frac{n \pi f}{T}}{\frac{n \pi f}{T}}\right]^{2}
$$

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Analyses of commonly encountered wave forms continued

## 5. Clipped sawtooth wave



$$
\begin{aligned}
& A_{a v}=\frac{A d}{2 T} \quad A_{r m s}=A \sqrt{\frac{d}{3 T}} \\
& C_{n}=\frac{A T}{2 \pi^{2} n^{2 d}}\left[2\left(1-\cos \frac{2 \pi n d}{T}\right)+\frac{4 \pi n d}{T}\left(\frac{\pi n d}{T}-\sin \frac{2 \pi n d}{T}\right)\right]^{\frac{1}{2}}
\end{aligned}
$$

If $d$ is small

$$
C_{n}=\frac{2 A_{a \pi}}{\frac{\pi n d}{T}}\left[\frac{\sin \frac{\pi n d}{T}}{\frac{\pi n d}{T}}-1\right]
$$

6. Sawtooth wave

$A_{a v}=\frac{A}{2}$
$A_{r m 0}=\frac{A}{\sqrt{3}}$
$C_{n}=-\frac{2 A_{a v}}{n \pi} \cos (n \pi)$

## 7. Sawtooth wave



$$
A_{a v}=\frac{A}{2} \quad A_{r m s}=\frac{A}{\sqrt{3}}
$$

$$
C_{n}=\frac{2 A_{a v} T}{\pi^{2} n^{2 f}\left(1-\frac{f}{T}\right)} \sin \frac{\pi f}{T}
$$

## 8. Fractional sine-wave



$$
A_{a v}=\frac{A\left(\sin \frac{\pi d}{T}-\frac{\pi d}{T} \cos \frac{\pi d}{T}\right)}{\pi\left(1-\cos \frac{\pi d}{T}\right)}
$$

$$
A_{r m s}=
$$

$$
\frac{A}{\left(1-\cos \frac{\pi d}{T}\right)}\left[\frac{1}{2 \pi}\left(\frac{\pi d}{T}+\frac{1}{2} \sin \frac{2 \pi d}{T}-4 \cos \frac{\pi d}{T} \sin \frac{\pi d}{T}+\frac{2 \pi d}{T} \cos ^{2} \frac{\pi d}{T}\right)\right]^{\frac{2}{2}}
$$

$$
C_{n}=\frac{A_{a v} \frac{\pi d}{T}}{n\left(\sin \frac{\pi d}{T}-\frac{\pi d}{T} \cos \frac{\pi d}{T}\right)}\left[\frac{\sin \ln -11 \frac{\pi d}{T}}{\ln -11 \frac{\pi d}{T}}-\frac{\sin \ln +11 \frac{\pi d}{T}}{\ln +11 \frac{\pi d}{T}}\right]
$$

## Analyses of commonly encounfered wave forms contioued

## 9. Half sine-wave


10. Full sine-wave

$C_{n}=A_{a v}\left[2 \frac{\sin \left(n \pi \frac{d}{T}\right)}{n \pi \frac{d}{T}}+\frac{\sin \pi\left(1-n \frac{d}{T}\right)}{\pi\left(1-n \frac{d}{T}\right)}+\frac{\sin \pi\left(1+n \frac{d}{T}\right)}{\pi\left(1+n \frac{d}{T}\right)}\right]$

## Analyses of commonly encountered wave forms

continued

## 11. Critically damped exponential wave


$f(t)=\frac{A_{\epsilon}}{f} t_{\epsilon}^{-\frac{1}{f}}$ where $\epsilon=2.718$ for $T>10 f \quad A_{\text {rme }}=\frac{A_{\epsilon}}{2} \sqrt{\frac{f}{T}}$ $A_{a v}=\frac{A \epsilon f}{T}$
$C_{n}=2 A_{a,}\left[\frac{1}{1+\left(\frac{2 \pi n f}{T}\right)^{2}}\right]=2 A_{a v} \cos ^{2} \frac{\theta_{n}}{2}$
$\frac{\theta_{n}}{2}=\tan ^{-1}\left(\frac{2 \pi n f}{T}\right)$
12. Full-wave rectifled sine-wave

$A_{a,}=\frac{2 A}{\pi}$
$A_{r m e}=\frac{A}{\sqrt{2}}$
$C_{n}=\frac{\pi}{2} A_{a n}\left[\frac{\sin \frac{\pi}{2}(1-n)}{\frac{\pi}{2}(1-n)}+\frac{\sin \frac{\pi}{2}(1+n)}{\frac{\pi}{2}(1+n)}\right]$

## Modulated wave forms

Starting from a carrier $i=A \sin \theta$ modulated waveforms are obtained when either or both $A$ and $\theta$ are functions of time.

## 1. Amplitude modulation

$\theta=\omega t+\phi \quad$ where $\omega$ and $\phi$ are constants
$A=A_{0}\left[1+m_{a} f(t)\right]$
$i=A_{0}\left[1+m_{a} f(t)\right] \sin (\omega t+\phi)$
where $f(t)$ is a continuous function of time representing the signal and $|f(t)| \leq 1$. Then $m_{a}$ is the degree of amplitude modulation; $0 \leq m_{a} \leq 1$. Generally the frequency spectrum of $f(t)$ will be limited up to a value $\alpha$ $\ll \omega$ and the total frequency spectrum will comprise:
the carrier $\omega$
the lower side band from $\omega$ to $\omega-\alpha$
the upper side band from $\omega$ to $\omega+\alpha$
For correct transmission of intelligence it is sufficient to transmit one of the side bands only.

For a sinusoidal signal $f(t)=\cos p t$ where $p=$ angular frequency of the signal; $i=A_{0}\left\{\sin \omega t+\frac{m_{a}}{2}[\sin (\omega+\rho) t+\sin (\omega-p) t]\right\}$

## 2. Frequency modulation

wherein $A$ is constant
$\omega_{t}=\frac{d \theta}{d t}=\omega[1+m f(t)]$
$\omega=2 \pi \times$ mean carrier frequency (a constant), $\omega_{i}=2 \pi \times$ instantaneous frequency, $m=$ degree of frequency modulation, $\Delta \omega=m \omega=2 \pi \times$ frequency wing, $f(t)$ is the signal to be transmitted; $|f(t)| \leq 1$.
Even when the frequency spectrum of $f(t)$ extends only up to $\alpha \ll \omega$ the resulting frequency spectrum of the modulated wave is complex, depending on the relative values of $\alpha$ and $m$. Generally $\Delta \omega \geq \alpha$ and the spectrum is composed of groups of upper and lower side bands even when $f(f)$ is a sinusoidal function of time.

For a sinusoidal signal $f(t)=\cos p t$
$\omega_{t}=\omega[1+m \cos p t]$
$\theta=\omega t+\frac{\Delta \omega}{\rho} \sin p t$
$m_{f}=\frac{\Delta \omega}{\rho}=$ frequency modulation index (radians)

## Modulated wave forms

continued

In this case the carrier and side bands include a number of components at frequencies $(\omega \pm n p) / 2 \pi$ where $n=0$ or a positive integer.

$$
\begin{aligned}
\frac{i}{A_{0}}= & \sin (\omega t+m f \sin p t) \\
= & J_{0}\left(m_{f}\right) \sin \omega t \\
& +J_{1}\left(m_{f}\right)[\sin (\omega+p) t-\sin (\omega-p) t] \\
& +J_{2}\left(m_{f}\right)[\sin (\omega+2 p) t+\sin (\omega-2 p) t] \\
& +\ldots \\
& +J_{m}\left(m_{f}\right)\left[\sin (\omega+n p) t+(-1)^{n} \sin (\omega-n p) t\right] \\
= & J_{0}\left(m_{f}\right) \sin \omega t+2 J_{1}\left(m_{f}\right) \sin p t \cos \omega t \\
& +2 J_{2}(m f) \cos 2 p t \sin \omega t+\ldots \\
& +(-1)^{n} 2 J_{n}\left(m_{f}\right) \cos \left(n p t+n^{\frac{\pi}{2}}\right) \sin \left(\omega t+n \frac{\pi}{2}\right)
\end{aligned}
$$

Where $J_{n}\left(m_{f}\right)$ is the Bessel function of the first kind and $n^{2 m}$ order. An expansion of $J_{n}\left(m_{f}\right)$ in a series is given on page 299 and tables of Bessel functions on pages 319 to 322.


Amplitude of carrier and side bands far $\mathrm{mp}_{\mathrm{f}}=10$. The carrier amplifude is $0.246 \mathrm{~A}_{0}$ and is represented by the heavy line in the center. The separation between each twa adjacent campanents $=$ signal frequency f.
a. For small values of $m_{f}$ up to about 0.2

$$
\begin{aligned}
i & =A_{0}\left\{\sin \omega t+\frac{m_{f}}{2}[\sin (\omega+\rho) t-\sin (\omega-p) t]\right\} \\
& =A_{0}\left(\sin \omega t+m_{f} \sin \rho t \cos \omega t\right)
\end{aligned}
$$

Compare with amplitude modulation above.
b. The carrier amplitude varies with $m_{f}$ as does also that of each pair of side bands.
$\begin{array}{llllll}\text { Carrier vanishes for } m_{f}=2.40 & 5.52 & 8.65 & 11.79 & 14.93 \text { etc. }\end{array}$
$\begin{array}{lllll}\text { First side band vanishes for } m_{f}=3.83 & 7.02 & 10.17 & 13.32 \text { etc. }\end{array}$
This property of vanishing components is used frequently in the measurement of $m$.
c. The approximate number of important side bands and the corresponding band width necessary for transmission are as follows (where $f=p / 2 \pi$ and $\Delta F=\Delta \omega / 2 \pi l:$

| mp | 5 | 10 | 20 |
| :---: | :---: | :---: | :---: |
| signal frequency f | $0.2 \Delta F$ | $0.1 \Delta F$ | $0.05 \Delta F$ |
| number of pairs of side bands | 7 | 13 | 23 |
| band width | $\begin{gathered} 14 f \\ 2.8 \Delta F \end{gathered}$ | $\begin{gathered} 26 f \\ 2.6 \Delta f \end{gathered}$ | $\begin{gathered} 46 f \\ 2.3 \Delta F \end{gathered}$ |

This table is based on neglecting side bands in the outer regions where all amplitudes are less than $0.02 \mathrm{~A}_{0}$. The amplitude below which the side bands are neglected, and the resultant band width, will depend on the particular application and the quality of transmission desired.

## 3. Pulse modulation

Pulse modulation is obtained when $A$ or $\frac{d \theta}{d t}$ are keyed periodically. Then $f(t)$ is generally a pulsing waveform of the type previously described. See 4, page 283 (with $f \ll T$ ).

In pulse modulation generally $f(t)$ has no simple relation to the signal to be transmitted. Various forms of pulse modulation have been described:
a. Pulse-time modulation: The timing of the pulse $f(t)$ relative to a reference pulse is varied around a fixed mean value and conforms to the amplitude of the signal to be transmitted.
b. Pulse-width modulation: The duration of the pulse $f(t)$ is varied around a fixed mean value and conforms to the amplitude of the signal to be transmitted.
c. Pulse-frequency modulation: The repetition rate of the pulse $f(t)$ is varied around a fixed mean value and conforms to the amplitude of the signal to be transmitted.

- Mathematical formulas


## Mensuration formulas

## Areas of plane figures

Ararallelogram $=$ bh

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Mensuration formulas continued

Areas of plane figures
Circle formula

## Area of irregular plane surface



Trapezoidal rule:
Area $=\Delta\left(\frac{y_{1}}{2}+y_{2}+y_{3}+\ldots+y_{n-2}+y_{n-1}+\frac{y_{n}}{2}\right)$
Simpson's rule:
$n$ must be odd
Area $=\frac{\Delta}{3}\left(y_{1}+4 y_{2}+2 y_{3}+4 y_{4}+2 y_{5}+\ldots+2 y_{n-2}+4 y_{n-1}+y_{n}\right)$ $y_{1}, y_{2}, y_{3} \ldots y_{n}$ are measured lengths of a series of equidistant parallel chords

## Volumes and surface areas

$$
\text { Sphere: } \begin{aligned}
& \text { Surface }=4 \pi r^{2} \\
& \text { Volume }=\frac{4 \pi r^{3}}{3} \\
& r=\text { radius of sphere }
\end{aligned}
$$

Cylinder: Cylindrical portion of surface $=2 \pi$ rh

$$
\text { Volume }=\pi r^{2} h
$$

$r=$ radius of cylinder
$h=$ height of cylinder
Pyramid or cone: Volume $=$ Area of base $\times \frac{1}{3}$ of height

## Formulas for complex quantities

$(A+j B)(C+j D)=(A C-B D)+j(B C+A D)$

$$
\frac{A+j B}{C+j D}=\frac{A C+B D}{C^{2}+D^{2}}+j \frac{B C-A D}{C^{2}+D^{2}}
$$

$\frac{1}{A+j B}=\frac{A}{A^{2}+B^{2}}-j \frac{B}{A^{2}+B^{2}}$
$\mathrm{A}+j \mathrm{~B}=\rho(\cos \theta+j \sin \theta)$
$\sqrt{A+j B}= \pm \sqrt{p}\left(\cos \frac{\theta}{2}+j \sin \frac{\theta}{2}\right)$
where $\rho=\sqrt{A^{2}+B^{2}} ; \cos \theta=\frac{A}{\rho}$

$$
\begin{aligned}
\sin \theta & =\frac{B}{\rho} \\
\mathrm{e}^{j \theta} & =\cos \theta+j \sin \theta \\
\mathrm{e}^{-j \theta} & =\cos \theta-j \sin \theta
\end{aligned}
$$

## Algebraic and frigonometric formulas

$1=\sin ^{2} A+\cos ^{2} A=\sin A \operatorname{cosec} A=\tan A \cot A=\cos A \sec A$
$\sin A=\frac{\cos A}{\cot A}=\frac{1}{\operatorname{cosec} A}=\cos A \tan A=\sqrt{1-\cos ^{2} A}$
$\cos A=\frac{\sin A}{\tan A}=\frac{1}{\sec A}=\sin A \cot A=\sqrt{1-\sin ^{2} A}$
$\tan A=\frac{\sin A}{\cos A}=\frac{1}{\cot A}=\sin A \sec A$
$\cot A=\frac{1}{\tan A} \quad \sec A=\frac{1}{\cos A}$
$\operatorname{cosec} A=\frac{1}{\sin A}$
$\sin (A \pm B)=\sin A \cos B \pm \cos A \sin B$
$\tan (A \pm B)=\frac{\tan A \pm \tan B}{1 \mp \tan A \tan B}$

Algebraic and frigonometric formulas continued

$$
\begin{aligned}
& \cos |A \neq B|=\cos A \cos B \mp \sin A \sin B \\
& \cot (A \pm B)=\frac{\cot A \cot B \mp 1}{\cot B \pm \cot A} \\
& \sin A+\sin B=2 \sin \frac{1}{2}(A+B) \cos \frac{1}{2}(A-B) \\
& \sin ^{2} A-\sin ^{2} B=\sin (A+B) \sin (A-B) \\
& \tan A \pm \tan B=\frac{\sin (A \pm B)}{\cos A \cos B} \\
& \sin A-\sin B=2 \cos \frac{1}{2}(A+B) \sin \frac{1}{2}(A-B) \\
& \cos A+\cos B=2 \cos \frac{1}{2}(A+B) \cos \frac{1}{2}(A-B) \\
& \cot A \pm \cot B=\frac{\sin (B \pm A)}{\sin A \sin B} \\
& \cos B-\cos A=2 \sin \frac{1}{2}(A+B) \sin \frac{1}{2}(A-B) \\
& \sin 2 A=2 \sin A \cos A \quad \cos 2 A=\cos ^{2} A-\sin ^{2} A \\
& \cos ^{2} A-\sin ^{2} B=\cos (A+B) \cos (A-B) \\
& \tan 2 A=\frac{2 \tan A}{1-\tan ^{2} A} \\
& \sin \frac{1}{2} A= \pm \sqrt{\frac{1-\cos A}{2}} \\
& \cos \frac{1}{2} A= \pm \sqrt{\frac{1+\cos A}{2}} \\
& \tan \frac{1}{2} A=\frac{\sin A}{1+\cos A} \quad \sin ^{2} A=\frac{1-\cos 2 A}{2} \\
& \cos ^{2} A=\frac{1+\cos 2 A}{2} \quad \tan ^{2} A=\frac{1-\cos 2 A}{1+\cos 2 A} \\
& \frac{\sin A \pm \sin B}{\cos A+\cos B}=\tan \frac{1}{2}(A \pm B) \\
& \frac{\sin A \pm \sin B}{\cos B-\cos A}=\cot \frac{1}{2}(A \mp B) \\
& \sin A \cos B=\frac{1}{2}[\sin (A+B)+\sin (A-B)] \\
& \cos A \cos B=\frac{1}{2}[\cos (A+B)+\cos (A-B)] \\
& \sin A \sin B=\frac{1}{2}[\cos (A-B)-\cos (A+B)]
\end{aligned}
$$

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## Algebraic and trigonometric formulas continued

$\sin x+\sin 2 x+\sin 3 x+\ldots+\sin m x=\frac{\sin \frac{1}{2} m x \sin \frac{1}{2}(m+1) x}{\sin \frac{1}{2} x}$
$\cos x+\cos 2 x+\cos 3 x+\ldots+\cos m x=\frac{\sin \frac{1}{2} m x \cos \frac{1}{2}(m+1) x}{\sin \frac{1}{2} x}$
$\sin x+\sin 3 x+\sin 5 x+\ldots+\sin (2 m-1) x=\frac{\sin ^{2} m x}{\sin x}$
$\cos x+\cos 3 x+\cos 5 x+\ldots+\cos (2 m-1) x=\frac{\sin 2 m x}{2 \sin x}$
$\frac{1}{2}+\cos x+\cos 2 x+\ldots+\cos m x=\frac{\sin \left(m+\frac{1}{2}\right) x}{2 \sin \frac{1}{2} x}$

| angle | 0 | 0 | $30^{\circ}$ | $45^{\circ}$ | $60^{\circ}$ | $90^{\circ}$ | $180^{\circ}$ | $270^{\circ}$ |
| :--- | :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $360^{\circ}$ |  |  |  |  |  |  |  |
| $\sin$ | 0 | $1 / 2$ | $1 / 2 \sqrt{2}$ | $1 / 2 \sqrt{3}$ | 1 | 0 | -1 | 0 |
| $\cos$ | 1 | $1 / 2 \sqrt{3}$ | $1 / 2 \sqrt{2}$ | $1 / 2$ | 0 | -1 | 0 | 1 |
| $\tan$ | 0 | $1 / 3 \sqrt{3}$ | 1 | $\sqrt{3}$ | $\pm \infty$ | 0 | $\pm \infty$ | 0 |

versine $\theta=1-\cos \theta$
$\sin 14 \frac{1}{2}^{\circ}=\frac{1}{4}$ approximately
$\sin 20^{\circ}=11 / 32$ approximately

## Approximations for small angles

| $\sin \theta$ | $=\left(\theta-\theta^{3} / 6 \ldots \ldots\right)$ | $\theta$ in radians |
| ---: | :--- | ---: | :--- |
| $\tan \theta$ | $=\left(\theta+\theta^{3} / 3 \ldots \ldots\right)$ | $\theta$ in radians |
| $\cos \theta$ | $=\left(1-\theta^{2} / 2 \ldots \ldots\right)$ | $\theta$ in radians |

## Quadratic equation

If $a x^{2}+b x+c=0$, then $x=\frac{-b \pm \sqrt{b^{2}-4 a c}}{2 a}$

## Arithmetical progression

$S=n l a+n / 2=n[2 a+l n-1) d] / 2$
where $S=$ sum, $a=$ first term, $l=$ last term, $n=$ number of terms, $d=$ common difference $=$ the value of any term minus the value of the preceding term.

## Geometrical progression

$S=\frac{a\left(r^{n}-1\right)}{r-1}=\frac{a\left(1-r^{n}\right)}{1-r}$
where $S=$ sum, $a=$ first term, $n=$ number of terms, $r=$ common ratio $=$ the value of any term divided by the preceding term.

## Combinations and permutations

The number of combinations of $n$ things, all different, taken $r$ at a time is
${ }_{n} C_{r}=\frac{n!}{r!(n-r)!}$
The number of permutations of $n$ things $r$ at a time $={ }_{n} P_{r}$
${ }_{n} P_{r}=n(n-1)(n-2) \ldots \ldots(n-r+1)=\frac{n!}{(n-r)!}$
${ }_{n} P_{n}=n!$

## Binomial theorem

$(a \pm b)^{n}=a^{n} \pm n a^{n-1} b+\frac{n(n-1)}{2!} a^{n-2} b^{2} \pm \frac{n(n-1)(n-2)}{3!} a^{n-8} b^{8}+\ldots$.
If $n$ is a positive integer, the series is finite and contains $n+1$ terms; otherwise it is infinite, converging for $\left|\frac{b}{a}\right|<1$ and diverging ior $\left|\frac{b}{a}\right|>1$.

## Maclaurin's theorem

$f(x)=f(0)+x f^{\prime}(0)+\frac{x^{2}}{1 \cdot 2} f^{n}(0)+\ldots+\frac{x^{n}}{n!} f^{n}(0)+\ldots$.

## Taylor's theorem

$$
\begin{aligned}
f(x) & =f\left(x_{0}\right)+f^{\prime}\left(x_{0}\right)\left(x-x_{0}\right)+\frac{f^{\prime \prime}\left(x_{0}\right)}{2!}\left(x-x_{0}\right)^{2}+\ldots \\
f(x+h) & =f(x)+f^{\prime}(x) \cdot h+\frac{f^{\prime \prime}(x)}{2!} h^{2}+\ldots+\frac{f^{n}(x)}{n!} h^{n}+\ldots .
\end{aligned}
$$

## Trigonometric solution of triangles

## Right-angled friangles (right angle at C)

$$
\begin{aligned}
\sin A & =\cos B=\frac{a}{c} \\
\tan A & =\frac{a}{b} \quad B=90^{\circ}-A \\
\text { vers } A & =1-\cos A=\frac{c-b}{c} \\
c & =\sqrt{a^{2}+b^{2}} \\
b & =\sqrt{c^{2}-a^{2}}=\sqrt{(c+a)(c-a)} \\
\text { Area } & =\frac{a b}{2}=\frac{a}{2} \sqrt{c^{2}-a^{2}}=\frac{a^{2} \cot A}{2}=\frac{b^{2} \operatorname{ton} A}{2}=\frac{c^{2} \sin A \cos A}{2}
\end{aligned}
$$

## Oblique-angled friangles

$$
\begin{aligned}
\sin \frac{1}{2} A & =\sqrt{\frac{(s-b)(s-c)}{b c}} \\
\cos \frac{2}{2} A & =\sqrt{\frac{s(s-a)}{b c}} \\
\text { where } s & =\frac{a+b+c}{2}
\end{aligned}
$$


$A+B+C=180^{\circ}$
$\tan \frac{\frac{1}{2}}{} A=\sqrt{\frac{(s-b)(s-c)}{s(s-a)}}$, similar values for angles $B$ and $C$

$$
\begin{aligned}
\text { Area } & =\sqrt{s(s-a)(s-b)(s-c)}=\frac{1}{2} a b \sin C=\frac{a^{2} \sin B \sin C}{2 \sin A} \\
c & =\frac{a \sin C}{\sin A}=\frac{a \sin (A+B)}{\sin A}=\sqrt{a^{2}+b^{2}-2 a b \cos C}
\end{aligned}
$$

$$
\tan A=\frac{a \sin C}{b-a \cos C}, \tan \frac{1}{2}(A-B)=\frac{a-b}{a+b} \cot \frac{1}{2} C
$$

$a^{2}=b^{2}+c^{2}-2 b c \cos A$, similar expressions for other sides.

## mathematical formulas

## Complex hyperbolic and other functions

Properties of "e"

$$
\begin{aligned}
e= & 1+1+\frac{1}{2!}+\frac{1}{3!}+\ldots .=2.71828 \\
& \frac{1}{e}=0.3679 \\
e^{x}= & 1+x+\frac{x^{2}}{2!}+\frac{x^{3}}{3!}+\ldots .
\end{aligned}
$$

$\log _{10} \mathrm{e}=0.43429 ; \log _{0} 10=2.30259$
$\log _{6} N=\log _{6} 10 \times \log _{10} N ; \log _{10} N=\log _{10} e \times \log _{0} N$.
$\sin x=x-\frac{x^{3}}{3!}+\frac{x^{5}}{5!}-\frac{x^{7}}{7!}+\ldots$
$\cos x=1-\frac{x^{2}}{2!}+\frac{x^{4}}{4!}-\frac{x^{6}}{6!}+\ldots$
$x$ is in radians. The series are con-
$\left.\sinh x=x+\frac{x^{3}}{3!}+\frac{x^{5}}{5!}+\frac{x^{7}}{7!}+\ldots.\right\}$ vergent for all finite values of $x$.
$\cosh x=1+\frac{x^{2}}{2!}+\frac{x^{4}}{4!}+\frac{x^{6}}{6!}+\ldots$.
For $n=0$ or a positive integer, the expansion of the Bessel function of the first kind, $n^{\text {th }}$ order, is given by the convergent series

$$
\begin{aligned}
& J_{n}(x)=\frac{x^{n}}{2^{n} n!}\left[1-\frac{x^{2}}{2(2 n+2)}+\frac{x^{4}}{2 \cdot 4(2 n+2)(2 n+4)}\right. \\
& \left.\left.-\frac{x^{6}}{2 \cdot 4 \cdot 6(2 n+2)\{2 n+4)\{2 n+6)}+\ldots\right]\right] \\
& \text { and } J_{-n}(x)=(-1)^{n} J_{n}(x) \quad \text { Note: } 0!=1 \\
& \sin x=\frac{e^{f x}-e^{-f x}}{2 j} \\
& \begin{aligned}
e^{j x} & =\cos x+j \sin x \\
e^{-j x} & =\cos x-j \sin x
\end{aligned} \\
& j=\sqrt{-1} \\
& \cos x=\frac{\mathrm{e}^{j x}+\mathrm{e}^{-j x}}{2} \\
& \sinh x=\frac{e^{x}-e^{-x}}{2} \\
& \sinh (-x)=-\sinh x_{i} \cosh (-x)=\cosh x \\
& \sinh j x=j \sin x ; \cosh j x=\cos x \\
& \cosh ^{2} x-\sinh ^{2} x=1 \\
& \sinh 2 x=2 \sinh x \cosh x \\
& \cosh 2 x=\cosh ^{2} x+\sinh ^{2} x \\
& \sinh (x \pm j y)=\sinh x \cos y \pm j \cosh x \sin y \\
& \cosh (x \pm j y)=\cosh x \cos y \pm i \sinh x \sin y
\end{aligned}
$$

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## Table of integrals

## Indefinite integrals

In the following formulas, $a, b$, and $m$ are constants. The constant of integration is not shown, but is added to each result.
$\int d x=x$
$\int a f(x) d x=a \int f(x) d x$
$\int(u+v-s) d x=\int u d x+\int v d x-\int s d x$
$\int x^{m} d x=\frac{x^{m+1}}{m+1} \quad m \neq-1$
$\int \frac{d x}{x}=\log _{e} x$
$\int(a x+b)^{m} d x=\frac{(a x+b)^{m+1}}{a(m+1)} \quad m \neq-1$
$\int \frac{d x}{a x+b}=\frac{1}{a} \log _{e}(a x+b)$
$\int \frac{x d x}{a x+b}=\frac{1}{a^{2}}\left[a x+b-b \log _{e}(a x+b)\right]$
$\int \frac{x d x}{(a x+b)^{2}}=\frac{1}{a^{2}}\left[\frac{b}{a x+b}+\log _{e}(a x+b)\right]$
$\int \frac{x^{2} d x}{a x+b}=\frac{1}{a^{3}}\left[\frac{(a x+b)^{2}}{2}-2 b(a x+b)+b^{2} \log _{e}(a x+b)\right]$
$\int \frac{d x}{x^{2}+\sigma^{2}}=\frac{1}{a} \tan ^{-1} \frac{x}{a}$
$\int \frac{d x}{\sqrt{a^{2}-x^{2}}}=\sin ^{-1} \frac{x}{a}$
$\int \log _{a} x d x=x \log _{a} \frac{x}{e}$ where $e=2.718$
$\int a^{x} d x=\frac{a^{x}}{\log _{e} a}$

## Table of infegrals

## continued

$$
\begin{aligned}
& \int x e^{x} d x=e^{x}(x-1) \\
& \int x^{m} e^{x} d x=x^{m} e^{x}-m \int x^{m-1} e^{x} d x \\
& \int \sin x d x=-\cos x \\
& \int \sin ^{2} x d x=\frac{1}{2}(x-\sin x \cos x) \\
& \int \cos x d x=\sin x \\
& \int \cos ^{2} x d x=\frac{1}{2}(x+\sin x \cos x) \\
& \int \tan ^{x} x d x=-\log _{e} \cos x \\
& \int \cot ^{x} x d x=\log _{e} \sin x \\
& \int \sec ^{x} x d x=\log _{e} \cdot(\sec x+\tan x) \\
& \int \sec ^{2} x d x=\tan x \\
& \int \operatorname{cosec}^{2} x d x=-\cot x \\
& \int \operatorname{cosec}^{x} x d x=\log _{e}(\operatorname{cosec} x-\cot x) \\
& \int \sin ^{-1} x d x=x \sin ^{-1} x+\sqrt{1-x^{2}} \\
& \int \cos { }^{-1} x d x=x \cos { }^{-1} x-\sqrt{1-x^{2}} \\
& \int \tan ^{-1} x d x=x \tan ^{-1} x-\log e \sqrt{1+x^{2}}
\end{aligned}
$$

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Table of infegrals continued

## Definite integrals

$\int_{0}^{\infty} x^{n-1} e^{-x} d x=\Gamma(n)^{*}$
$\int_{0}^{1} x^{m-1}(1-x)^{m-1} d x=\frac{\Gamma(m) \Gamma(n) *}{\Gamma(m+n)}$
$\int_{0}^{\frac{\pi}{2}} \sin ^{n} x d x=\int_{0}^{\frac{\pi}{2}} \cos ^{n} x d x=\frac{1}{2} \sqrt{\pi} \frac{\Gamma\left(\frac{n+1}{2}\right)^{*}}{\Gamma\left(\frac{n}{2}+1\right)^{n}}, n>-1$
$\int_{0}^{\infty} \frac{\sin m x d x}{x}=\frac{\pi}{2}$ if $m>0 ; 0$ if $m=0 ;-\frac{\pi}{2}$ if $m<0$
$\int_{0}^{\infty} \frac{\cos m x d x}{1+x^{2}}=\frac{\pi}{2} e^{-|m|}$
$\int_{0}^{\infty} \frac{\cos x d x}{\sqrt{x}}=\int_{0}^{\infty} \frac{\sin x d x}{\sqrt{x}}=\sqrt{\frac{\pi}{2}}$
$\int_{0}^{\infty} e^{-\sigma^{2} x^{2}} d x=\frac{1}{2 a} \sqrt{\pi}$
$\int_{\frac{-\pi}{2}}^{\frac{\pi}{2}} \frac{\cos ^{2}\left(\frac{\pi}{2} \sin x\right) d x}{\cos x}=1.22$

- Volves of $\Gamma$ inf are tobulated in Jahnko \& Emdo, Tables of functioash

| $n$ | 10* dif | $n$ | -n diff | $n$ | * | n | -* diff | n | - ${ }^{-3}$ | n | ${ }^{-7}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.00 | 1.000 | 0.50 | 1.649 | 1.0 | $2.718^{\text {\# }}$ | 0.00 | 1.000 | 0.50 | . 607 | 1.0 | 348** |
| . 01 | 1.01010 | . 51 | 1.66517 | . 1 | 3.004 | . 01 | 0.990 | . 51 | . 600 | . 1 | 333 |
| 02 | 1.02010 | . 52 | 1.682 | 2 | 3.320 | . 02 | $.980=10$ <br> 10 | . 52 | . 595 | 2 | 301 |
| . 03 | 1.03011 | . 53 | 1.69917 | 3 | 3.669 | . 03 | . $970=9$ | . 53 | . 589 | 3 | 273 |
| 04 | 1.04110 | . 54 | 1.71617 | 4 | 4.055 | . 04 | . $961-10$ | . 54 | . 583 | 4 | . 247 |
| 0.05 | 1.05111 | 0.55 | 1.73318 | 1.3 | 4.482 | 0.03 | . 951 | 0.35 | . 577 | 1.5 | . 223 |
| 06 | 1.06211 | . 56 | 1.75117 | 6 | 4.953 | . 06 | .942-9 | . 56 | . 571 | 6 | . 202 |
| . 07 | 1.07310 | . 58 | 1.76818 | 7 | 5.474 | . 07 | . $932=9$ | . 57 | ${ }^{5} 568$ | 7 | . 183 |
| . 08 | 1.08311 | . 59 | 1.786 1.804 | . 8 | 6.050 6.686 | . 08 | .923-9 | . 58 | . 560 .544 | 8 | . 165 |
|  |  |  |  |  |  |  |  | . 5 | . 554 | . 9 | . 150 |
| 0.10 | 1.105 | 0.60 | 1.822 | 2.0 | 7.389 | 0.10 | . 905 | 0.60 | . 54 | 2.0 | . 135 |
| . 11 | 1.116 | . 61 | 1.840 | . 1 | 8.166 | . 11 | .896 | . 61 | . 543 | . 1 | . 122 |
| . 12 | 1.12712 | . 62 | 1.85919 | 2 | 9.025 | . 12 | .887 | . 62 | . 536 | 2 | . 111 |
| . 13 | 1.13911 | . 63 | 1.878 | 3 | 9.974 | . 13 | .878 | . 63 | . 533 | 3 | . 100 |
| . 14 | 1.150 | . 64 | 1.89620 | . 4 | 11.02 | . 14 | ${ }^{869}$ - 8 | S4 | . 527 | 4 | 0907 |
| 0.15 | $1.162 \quad 12$ | 0.65 | 1.916 | 2.5 | 12.18 | 0.15 | . 861 | 0.65 | . 522 | 2.5 | . 0221 |
| . 16 | 1.17411 | . 66 | 1.93519 | 8 | 13.46 | . 16 | .852-8 | . 66 | . 517 | 6 | . 0743 |
| .17 | 1.18512 | . 67 | 1.95420 | 7 | 14.88 | . 17 | .840 ${ }^{8}$ | . 6 | . 512 | 7 | . 0672 |
| .18 |  | 68 69 | 1.97420 | 8 | 16.44 | . 18 | ${ }_{827} 88$ | . 68 | . 507 | 8 | . 0508 |
| . 19 | 1.20912 | . 69 | 1.99420 | 9 | 18.17 | . 19 | . $827=8$ | . 69 | . 502 | . 9 | . 0550 |
| 0.20 | 1.201 | 0.70 | 2014 | 3.0 | 20.09 | 0.20 | 819 | 0.70 | . 497 | 3.0 | . 0498 |
| 21 | 1.23413 | 71 | 2.03420 | . 1 | 22.20 | . 21 | $811=$ | . 71 | . 492 | 1 | . 0450 |
| 22 | 1.24612 | 72 | 2.05420 | 2 | 24.53 | . 22 | ${ }^{803}=$ | . 72 | . 487 | 2 | . 0408 |
| 23 | 1.259 | . 73 | 2075 | . 3 | 27.11 | . 23 | .795-8 | . 73 | . 482 | 3 | . 0369 |
| 24 | 1.27113 | 74 | 200621 | 4 | 27.96 | 24 | .787-8 | . 74 | . 477 | 4 | 1034 |
| 0.23 | 1.28413 | 0.75 | 211721 | 3.5 | 33.12 | 0.25 | .779-8 | 0.75 | . 472 | 3.3 | . 0302 |
| 26 | 1.29713 |  |  |  |  | 26 |  | . 77 | . 468 | 6 | . 0273 |
| - 27 | 1.31013 1.323 | . 77 | 2160 2181 21 | 3 | 40.45 44.70 | .27 | .763 <br> 756 <br> 58 | . 77 | . 463 | 7 | 0247 |
| -29 | 1.33613 | 78 | 220322 | 8 | 49.40 49.40 | . 28 | .746-8 | . 79 | . 4.54 | . | . 0224 |
| 0.30 | 1.350 | 0.00 | 2226 | 4.0 | 54.60 | 0.30 | .741- | 0.80 | . 449 | 4.0 | . 0183 |
| . 31 | 1.36314 | . 81 | 224822 | . 1 | 60.34 | . 31 | 733-8 | . 81 | . 445 | . 1 | . 0166 |
| . 32 | 1.37714 | . 82 | 2.27022 | 2 | 68.69 | . 32 | .725-7 | . 82 | . 440 | 2 | . 0150 |
| . 33 | 1.39114 | . 83 | $2.293{ }_{23}^{23}$ | 3 | 73.70 | . 33 | .719-7 | . 83 | . 436 | 3 | . 0136 |
| 34 | 1.40514 | 84 | $2.316{ }_{24}^{23}$ | . 4 | 81.45 | . 34 | $.712=7$ | . 84 | . 432 | 4 | . 0123 |
| 0.35 | 1.41914 | 0.85 | 2.34023 | 4.3 | 90.02 | 0.35 | .705-9 | 0.05 | 427 | 4.3 | . 0111 |
| . 36 |  |  | 23,363 23 |  |  | . 36 | .698-7 | . 86 | 423 |  |  |
| .37 .38 | 1.44814 1.462 | 87 | 238724 2411 24 |  |  |  | . 691 - 7 | . 87 |  | 5.0 | . 00674 |
| . 38 | 1.462 <br> 1.477 | . 88 | 2.41124 2.435 | 6.0 7.0 | 1093.4 | .38 .39 | . $6894^{-7}$ | . 88 | . 415 | 6.0 | . 00248 |
| $\cdots$ | 1.4715 | 89 | 2.43525 | 7.0 | 1097. | . 39 | ${ }^{.677}$ - 7 | . 29 | . 111 | 7.0 | . 000912 |
| 0.40 | $1.40215$ | 0.90 | ${ }_{2} 2.460$ | 8.0 | 2981. | 0.40 | . 670 - 6 | 0.90 | . 407 | 2.0 | . 000335 |
| . 41 | 1.50715 |  | $2.484{ }^{24}$ |  | 8103. |  | . 664 - 7 | . 91 | . 403 | 9.0 | . 000123 |
| . 42 | 1.52215 | .92 | 2.50926 | 10.0 | 22086. | . 42 | . $657^{-7}$ | . 92 | 399 | 10.0 | . 000045 |
| . 44 | 1.537 1.553 | 9 | ${ }_{2}^{2.5350} 25$ |  |  | . 43 | . 651 - 7 | . 93 | 385 |  |  |
|  | 1.0515 | . 9 | 2.56026 | $\pi / 2$ | 4.810 | 4 | . 644 - 6 | . 94 | 391 | T/2 | . 208 |
| 0.45 |  | 0.95 |  | $2 \pi / 2$ | 23.14 |  |  |  |  | 2x/2 | . 0432 |
| . 46 | 1.58416 | . 96 | $2.612{ }^{26}$ | $3 \pi / 2$ | ${ }_{5355}$ | 0.45 | .638-7 | 0.93 | 388 | 3x/2 | . 00878 |
| . 47 | 1.600 | . 97 | 2.638 | 5m/2 | 2576. | . 47 | . $625-6$ | . 97 | . 379 | 4x/2 | . 00187 |
| . 48 | 1.616 | . 98 | 2.664 | 6m/2 | 12392. | . 48 | .6219 | . 98 | . 375 | 6 $\pi / 2$ | .000081 |
| . 49 | 1.63217 | . 99 | 2.69127 | 7 $7 / 2$ | 59610. | . 49 | . $613=$ | . 99 | . 372 | 7x/2 | . 000017 |
|  |  |  |  | $8 \pi / 2$ | 286751. |  |  |  |  | $8 \pi / 2$ | . 000003 |
| 0.50 | 1.649 | 1.00 | 2.718 |  |  | 0.50 | 0.607 | 1.00 | . 368 |  |  |

[^23]Common logarithms of numbers and proportional parts

|  | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 3 | 9 | proportional ports |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |  |  |  |  |  |  | 12 | 31 |  |  | 6 |  | 9 |
| 10 | 0000 | 0043 | 0088 | 0128 | 0170 | 0212 | 0253 | 0294 | 0334 | 0374 | 481 |  | 172 | 12 | 5 |  | 337 |
| 11 | 0414 | 0453 | 0492 | 0531 | 0569 | 0607 | . 0645 | 0682 | 0719 | 0755 | 48 | 11 | 1519 | 92 | 3 |  | 3034 |
| 12 | 0792 | 0828 | 0864 | 0899 | 0934 | 0969 | 1004 | 1038 | 1072 | 1106 | 37 | 10 | 14 | 72 | 1 |  | 2831 |
| 13 | 1139 | 1173 | 1206 | 1239 | 1271 | 1303 | 1335 | 1367 | 1399 | 1430 | 36 | 10 | 13 | 6 | 9 | 23 | 2629 |
| 14 | 1461 | 1492 | 1523 | 1553 | 1584 | 1614 | 1644 | 1673 | 1703 | 1732 | 36 | 9 | 12 | 5 | 8 | 21 | 2427 |
| 15 | 1761 | 1790 | 1818 | 1847 | 1875 | 1903 | 1931 | 1959 | 1987 | 2014 | 36 | 8 | 11 | 4 | 7 | 20 | 2225 |
| 16 | 2041 | 2068 | 2095 | 2122 | 2148 | 2175 | 2201 | 2727 | 2253 | 2279 | 35 | 8 |  | 3 | 16 | 18 | 2124 |
| 17. | 2304 | 2330 | 2355 | 2380 | 2405 | 2430 | 2455 | 2480 | 2504 | 2529 | 25 | 7 |  | 2 | 15 | 17 | 2022 |
| 18 | 2553 | 2577 | 2601 | 2625 | 2648 | 2672 | 2695 | 2718 | 2742 | 2765 | 25 | 7 | 9 | 2 | 14 | 16 | 1921 |
| 19 | 2788 | 2810 | 2833 | 2856 | 2878 | 2900 | 2923 | 2945 | 2967 | 2989 | 24 | 7 | 9 | 1 | 13 | 16 | 1820 |
| 20 | 3010 | 3032 | 3054 | 3075 | 3096 | 3118 | 3139 | 3160 | 3181 | 3201 | 24 | 6 | 8 | 1 | 13 | 15 | 1719 |
| 21 | 3222 | 3243 | 3263 | 3284 | 3304 | 3324 | 3345 | 3365 | 3385 | 3404 | 24 | 6 | 8 | 10 | 12 | 14 | 1618 |
| 22 | 3424 | 3444 | 3464 | 3483 | 3502 | 3522 | 3541 | 3560 | 3579 | 3598 | 24 | 6 | 81 | 10 | 12 | 14 | 1517 |
| 23 | 3617 | 3636 | 3655 | 3674 | 3692 | 3711 | 3729 | 3747 | 3766 | 3784 | 24 | 6 | 7 | 9 | 11 | 13 | 1517 |
| 24 | 3802 | 3820 | 3838 | 3856 | 3874 | 3892 | 3909 | 3927 | 3945 | 3962 | 24 | 5 | 7 | 91 | 11 | 12 | 1416 |
| 25 | 3979 | 3997 | 4014 | 4031 | 4048 | 4065 | 4062 | 4099 | 4116 | 4133 | 23 | 5 | 7 | 91 | 10 | 12 | 1415 |
| 26 | 4150 | 4166 | 4183 | 4200 | 4216 | 4232 | 4249 | 4265 | 4281 | 4298 | 23 | 5 | 7 | 81 | 10 |  | 1315 |
| 27 | 4314 | 4330 | 4346 | 4362 | 4378 | 4393 | 4409 | 4425 | 4440 | 4456 | 23 | 5 | 6 | 8 | 9 | 11 | 1314 |
| 28 | 4472 | 4487 | 4502 | 4518 | 4533 | 4548 | 4564 | 4579 | 4594 | 4609 | 23 | 5 | 6 | 8 | 9 | 11 | 1214 |
| 29 | 4624 | 4639 | 4654 | 4669 | 4683 | 4698 | 4713 | 4728 | 4742 | 4757 | 13 | 4 | 6 | 7 | 9 | 10 | 1213 |
| 30 | 4771 | 4786 | 4800 | 4814 | 4829 | 4843 | 4857 | 4871 | 4886 | 4900 | 13 | 4 | 6 | 7 | 9 | 10 | 1113 |
| 31 | 4914 | 4928 | 4942 | 4955 | 4969 | 4983 | 4997 | 5011 | 5024 | 5038 | 13 | 4 | 5 | 7 | 8 |  | 1112 |
| 32 | 5051 | 5065 | 5079 | 5092 | 5105 | 5119 | 5132 | 5145 | 5159 | 5172 | 13 | 4 | 5 | 7 | 8 | 9 | 1112 |
| 33 | 5185 | 5198 | 5211 | 5224 | 5237 | 5250 | 5263 | 5276 | 5289 | 5302 | 13 | 4 | 5 | 6 | 8 | 9 | 1012 |
| 34 | 5315 | 5328 | 5340 | 5353 | 5366 | 5378 | 5391 | 5403 | 5416 | 5428 | 13 | 4 | 5 | 6 | 8 | 9 | 1011 |
| 35 | 5441 | 5453 | 5465 | 5478 | 5490 | 5502 | 5514 | 5527 | 5539 | 5551 | 12 | 4 | 5 | 6 | 7 | 9 |  |
| 36 | 5563 | 5575 | 5587 | 5599 | 5611 | 5623 | 5635 | 5647 | 5658 | 5670 | 12 | 4 | 5 | 6 | 7 | 8 | 1011 |
| 37 | 5682 | 5694 | 5705 | 5717 | 5729 | 5740 | 5752 | 5763 | 5775 | 5786 | 12 | 3 | 5 | 6 | 7 | 8 | 910 |
| 38 | 5798 | 5809 | 5821 | 5832 | 5843 | 5855 | 5866 | 5877 | 5888 | 5899 | 12 | 3 | 5 | 6 | 7 | 8 | 910 |
| 39 | 5911 | 5922 | 5933 | 5944 | 5955 | 5966 | 5977 | 5988 | 5999 | 6010 | 12 | 3 | 4 | 5 | 7 | 8 | 910 |
| 40 | 6021 | 6031 | 6042 | 6053 | 6064 | 6075 | 6085 | 6096 | 6107 | 6117 | 1 | 3 | 4 | 5 | 6 | 7 | 910 |
| 41 | 6128 | 6138 | 8149 | 6160 | 6170 | 6180 | 6191 | 6201 | 6212 | 6222 | 12 | 3 | 4 | 5 | 6 | 7 | 89 |
| 42 | 6232 | 6243 | 6253 | 8263 | 6274 | 6294 | 6294 | 6304 | 6314 | 6325 | 12 | 3 | 4 | 5 | 6 | 7 | 89 |
| 43 | 6335 | 6345 | 6355 | 6365 | 6375 | 6385 | 6395 | 6405 | 6415 | 6425 | 12 | 3 | 4 | 5 | 6 | 7 | 89 |
| 44 | 6435 | 6444 | 6454 | 6464 | 6474 | 6484 | 6493 | 6503 | 6513 | 6522 | 12 | 3 | 4 | 5 | 6 | 7 | 89 |
| 45 | 6532 | 6542 | 6551 | 6561 | 6571 | 6580 | 6590 | 6599 | 6609 | 6618 | 12 | 3 | 4 | 5 | 6 | 7 | 89 |
| 46 | 6628 | 6637 | 6646 | 6656 | 6665 | 6675 | 6684 | 6693 | 6702 | 6712 | 12 | 3 | 4 | 5 | 6 | 7 | 78 |
| 47 | 6721 | 6730 | 6739 | 6749 | 6758 | 8787 | 6776 | 6785 | 6794 | 6803 | 12 | 3 |  | 5 | 5 | 6 | 78 |
| 48 | 6812 | 6821 | 6830 | 6839 | 6848 | 6857 | 6866 | 6875 | 6884 | 6893 | 12 | 3 | 4 | 4 | 5 | 6 | 78 |
| 49 | 6902 | 6911 | 6920 | 6928 | 6937 | 6946 | 6955 | 6964 | 6972 | 6981 | 12 | 3 | 4 | 4 | 5 | 6 | 8 |
| 50 | 6990 | 6998 | 7007 | 7016 | 7024 | 7033 | 7042 | 7050 | 7059 | 7067 | 12 | 3 | 3 | 4 | 5 | 6 | 78 |
| 51 | 7076 | 7084 | 7093 | 7101 | 7110 | 7118 | 7126 | 7135 | 7143 | 7152 | 12 | 3 | 3 | 4 | 5 | 6 | 78 |
| 52 | 7160 | 7168 | 7177 | 7185 | 7193 | 7202 | 7210 | 7218 | 7226 | 7235 | 12 | 2 | 3 | 4 | 5 | 6 | 7 |
| 53 | 7243 | 7251 | 7259 | 7267 | 7275 | 7284 | 7292 | 7300 | 7308 | 7316 | 12 | 2 | 3 | 4 | 5 | 6 | 67 |
| 54 | 7324 | 7332 | 7340 | 7348 | 7356 | 7364 | 7372 | 7380 | 7388 | 7396 | 12 | 2 | 3 | 4 | 5 | 6 | 67 |

Common logarithms of numbers and proportional parts continued

|  | 0 |  |  |  |  |  |  |  |  |  | propertional peris |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | 0 |  |  |  |  |  |  |  |  |  | 12 | 3 | 4 | 5 |  | 7 | 8 | 9 |
| 55 | 7404 | 7412 | 7419 | 7427 | 7435 | 7443 | 7451 | 7459 | 7466 | 7474 | 12 | 2 | 3 | 4 |  | 5 | 6 | 67 |
| 56 | 7482 | 7490 | 7497 | 7505 | 7513 | 7520 | 7528 | 7536 | 7543 | 7551 | 12 | 2 | , |  |  | 5 |  | 67 |
| 57 | 7559 | 7566 | 7574 | 7582 | 7589 | 7597 | 7604 | 7612 | 7619 | 7627 | 12 | 2 | 3 | 4 | 5 | 5 |  | 87 |
| 58 | 7634 | 7642 | 7649 | 7657 | 7664 | 7672 | 7679 | 7686 | 7694 | 7701 | 11 | 2 | 3 | 4 |  | 5 | 6 | 67 |
| 59 | 7709 | 7716 | 7723 | 7731 | 7738 | 7745 | 7752 | 7760 | 7767 | 7774 | 11 | 2 | 3 | 4 | 4 | 5 | 6 | 67 |
| 60 | 7782 | 7789 | 7796 | 7803 | 7810 | 7818 | 7825 | 7832 | 7839 | 7846 | 11 | 2 | 3 | 4 |  | 5 | 6 | 66 |
| 61 | 7853 | 7860 | 7868 | 7875 | 7882 | 7889 | 7896 | 7903 | 7910 | 7917 | 11 | 2 | 3 | 4 |  | 5 |  | 6 |
| 62 | 7924 | 7931 | 7938 | 7945 | 7952 | 7959 | 7966 | 7973 | 7980 | 7987 | 11 | 2 | 3 | , |  | 5 | 6 | 6 |
| 63 | 7993 | 8900 | 8007 | 8014 | 8021 | 8028 | 8035 | 8041 | 8048 | 8055 | 11 | 2 | 3 |  |  | 5 | 5 | 6 |
| 64 | 8062 | 8069 | 8075 | 8082 | 8089 | 8096 | 8102 | 8109 | 8116 | 8122 | 11 | 2 | 3 | 3 | 4 | 5 | 5 | 6 |
| 65 | 8129 | 8136 | 8142 | 8149 | 8156 | 8162 | 8169 | 8176 | 8182 | 8189 | 11 | 2 | 3 | 3 |  | 5 | 5 | 6 |
| 66 | 8195 | 8202 | 8209 | 8215 | 8222 | 8228 | 8235 | 8241 | 8248 | 8254 | 11 | 2 | 3 | 3 | 4 | 5 | 5 | 6 |
| 67 | 8261 | 8267 | 8274 | 8250 | 8287 | 8293 | 8299 | 8306 | 8312 | 8319 | 11 | 2 | 3 | 3 | 4 | 5 | 5 | 6 |
| 68 | 8325 | 8331 | 8338 | 8344 | 8351 | 8357 | 8363 | 8370 | 8376 | 8382 . | 11 | 2 | 3 | , |  | 4 | 5 | 6 |
| 69 | 8388 | 8395 | 8401 | 8407 | 8414 | 8420 | 8426 | 8432 | 8439 | 8445 | 11 | 2 | 2 | 3 | 4 | 4 | 5 | 6 |
| 70 | 8451 | 8457 | 8463 | 8470 | 8476 | 8482 | 8488 | 8494 | 8500 | 8506 | 11 | 2 | 2 | 3 | 4 | 4 | 5 | 6 |
| 71 | 8513 | 8519 | 8525 | 8531 | 8537 | 8543 | 8549 | 8555 | 8561 | 8567 | 11 | 2 | 2 | 3 | 4 | 4 | 5 | 5 |
| 72 | 8573 | 8579 | 8585 | 8591 | 8597 | 8603 | 8609 | 8615 | 8621 | 8627 | 11 | 2 | 2 | 3 | 4 | 4 | 5 | 5 |
| 73 | 8633 | 8639 | 8645 | 8651 | 8657 | 8663 | 8669 | 8675 | 8681 | 8686 | 11 | 2 | 2 | , | 4 | 4 | 5 | 5 |
| 74 | 8692 | 8698 | 8704 | 8710 | 8716 | 8722 | 8727 | 8733 | 8739 | 8745 | 11 | 2 | 2 | 3 | 4 | 4 | 5 | 5 |
| 75 | 8751 | 8756 | 8762 | 8768 | 8774 | 8779 | 8785 | 8791 | 8797 | 8802 | 1 | 2 | 2 | 3 | 3 | 4 | 5 | 5 |
| 76 | 8808 | 8814 | 8820 | 8825 | 8331 | 8837 | 8842 | 8848 | 8854 | 8859 | 11 | 2 | 2 | 3 | 3 | 4 | 5 | 5 |
| 77 | 8855 | 8871 | 8876 | 8882 | 8887 | 8893 | 8899 | 8904 | 8910 | 8915 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 78 | 8921 | 8927 | 8932 | 8938 | 8943 | 8949 | 8954 | 8960 | 8965 | 8971 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 79 | 8976 | 8982 | 8987 | 8993 | 8998 | 9004 | 9009 | . 9015 | 9020 | 9025 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 80 | 9031 | 9036 | 9042 | 9047 | 9053 | 9058 | 9063 | 9069 | 9074 | 9079 | 1 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 81 | 9085 | 9090 | 9096 | 9101 | 9106 | 9112 | 9117 | 9122 | 9128 | 9133 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 82 | 9138 | 9143 | 9149 | 9154 | 9159 | 9165 | 9170 | 9175 | 9180 | 9186 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 83 | 9191 | 9198 | 9201 | 9206 | 9212 | 9217 | 9222 | 9227 | 9232 | 9238 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 84 | 9243 | 9248 | 9253 | 9258 | 9263 | 9269 | 9274 | 9279 | 8284 | 9289 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 85 | 9294 | 9299 | 9304 | 9309 | 9315 | 9320 | 9325 | 9330 | 9335 | 9340 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 86 | 9345 | 9350 | 9355 | 9360 | 9365 | 9370 | 9375 | 9380 | 9385 | 9390 | 11 | 2 | 2 | 3 | 3 | 4 | 4 | 5 |
| 87 | 9395 | 9400 | 9405 | 9410 | 9415 | 9420 | 9425 | 9430 | 9435 | 9440 | 01 | 1 | 2 | 2 | 3 | 3 |  | 4 |
| 88 | 9445 | 9450 | 9455 | 9460 | 9465 | 9469 | 9474 | 9479 | 9484 | 9489 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 89 | 9494 | 9499 | 9504 | 9509 | 9513 | 9518 | 9523 | 9528 | 9533 | 9538 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 90 | 9542 | 9547 | 9552 | 9557 | 9562 | 9566 | 9571 | 9576 | 9581 | 9586 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 91 | 9590 | 9595 | 9600 | 9805 | 9609 | 9614 | 9619 | 9624 | 9628 | 9633 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 92 | 9638 | 9643 | 9647 | 9652 | 9657 | 9661 | 9666 | 9671 | 9875 | 9680 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 93 | 9685 | 9689 | 9694 | 9699 | 9703 | 9708 | 9713 | 9717 | 9722 | 9727 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 94 | 9731 | 9736 | 9741 | 9745 | 9750 | 9754 | 9759 | 9763 | 9768 | 9773 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 95 | 9777 | 9782 | 9786 | 9791 | 9795 | 9800 | 9805 | 9809 | 9814 | 9818 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 96 | 9823 | 9827 | 9832 | 9836 | 9841 | 9845 | 9850 | 9854 | 9859 | 9863 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 97 | 9868 | 9872 | 9877 | 9881 | 9886 | 9890 | 9894 | 9899 | 9903 | 9908 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 98 | 9912 | 9917 | 9921 | 9926 | 9930 | 9934 | 9939 | 9943 | 9948 | 9952 | 01 | 1 | 2 | 2 | 3 | 3 | 4 | 4 |
| 99 | 9956 | 9961 | 9965 | 9969 | 9974 | 9978 | 9983 | 9987 | 9991 | 9996 | 01 | , | 2 | 2 | 3 |  | , | 4 |

306
Natural trigonometric functions
for decimal fractions of a degree

| dag | $\sin$ | cos | Ian | cel |  | deg | sin | cos | Pan | cot |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  |  | $\infty$ | 90.0 | 6.0 | . 10453 | 0.9945 | . 10510 | 9.514 | 84.0 |
| 0.0 | . 00000 | 1.0000 | . 00000 | 573.0 | \% 0.9 | . 1 | . 10628 | . 9943 | . 10687 | $9.357^{\circ}$ | . 9 |
| . 1 | . 00175 | 1.0000 | . 00175 | 573.0 | 8 | . | . 10800 | . 9942 |  | 9.205 |  |
| . 2 | . 00349 | 1.0000 | . 00349 | 286.5 | 8 | .2 | .10800 .10973 | . 9942 | . 10863 | 9.205 | . 8 |
| 3 | .00524 | 1.0000 | .00524 .00698 | 191.0 143.24 | . 6 | . 4 | . 10973 | .9940 | . 11217 | 8.050 |  |
| . 4 | . 00698 | 1.0000 | . 00698 | 143.24 | . 6 | . 5 | . 11147 | . 9938 | . .11217 | 8.915 8.777 | . 6 |
| . 5 | . 00873 | 1.0000 | . 00873 | 114.59 95.49 | . 4 | . 5 | . 11320 | . 9936 | . .11394 | 8.777 | . 4 |
| 8 | . 01347 | 0.9999 .9999 | .01047 .01222 | 95.49 81.85 | . 3 | . 7 | . 11494 | .9934 | . 11747 | 8.513 | 3 |
| . 8 | . 01222 | . 9999 | . 01222 | 81.85 71.62 | . 2 | . 8 | . 11840 | . 9930 | . 11924 | 8.386 | . 2 |
| . 9 | . 01571 | . 9999 | . 01571 | 63.66 | . 1 | . 9 | .12014 | . 9928 | .12101 | 8.264 | . 1 |
| 1.0 | . 01745 | 0.9998 | . 01746 | 57.29 | 89.0 | 7.0 | . 12187 | 0.9925 | . 12278 | 8.144 | 3.0 |
| . 1 | . 01920 | . 9998 | . 01920 | 52.08 | . 9 | . 1 | . 12360 | . 9923 | . 12456 | 8.028 | . 9 |
| . 2 | . 02094 | . 9978 | . 02095 | 47.74 | . 8 | . 2 | . 12533 | . 9921 | . 12633 | 7.916 | 8 |
| . 3 | . 02269 | . 9997 | . 02269 | 44.07 | 7 | . 3 | . $1270 \%$ | . 9919 | . 12810 | 7.806 | 7 |
| . 4 | . 02443 | . 9997 | . 02444 | 40.92 | . 6 | .4 | . 12880 | . 9917 | . 12988 | 7.700 | . 6 |
| . 5 | . 02618 | . 9997 | . 02619 | 38.19 | . 5 | . 5 | . 13053 | . 9914 | . 13165 | 7.596 | . 5 |
| . 6 | . 02792 | . 9996 | . 02793 | 35.80 | . 4 | . 6 | . 13226 | . 9912 | . 13343 | 7.495 | . 4 |
| . 7 | . 02967 | . 9996 | . 02968 | 33.69 | . 3 | . 7 | . 13399 | . 9910 | . 13521 | 7.396 | 3 |
| . 8 | . 03141 | . 9995 | . 03143 | 31.82 | . 2 | 8 | . 13572 | . 9907 | . 13698 | 7.300 | . 2 |
| . 9 | . 03316 | . 9995 | . 03317 | 30.14 | . 1 | . 9 | . 13744 | . 9905 | . 13876 | 7.207 | .1 |
| 2.0 | . 03490 | 0.9994 | . 03492 | 28.64 | 88.0 | 8.0 | . 13917 | 0.9903 | . 14054 | 7.115 | 82.0 |
| . 1 | . 03664 | . 9993 | . 03667 | 27.27 | . 9 | . 1 | . 14990 | . 9900 | . 14232 | 7.026 | . 9 |
| .2 | . 03839 | . 9993 | . 03842 | 26.03 | 8 | .2 | . 14263 | . 9898 | . 14410 | 6.940 | 8 |
| . 3 | . 04013 | . 9992 | . 04016 | 24.90 | . 7 | . 3 | . 14436 | . 9895 | . 14588 | 6.855 | . 7 |
| . 4 | . 04188 | . 9991 | . 04191 | 23.86 | . 6 | . 4 | . 14608 | . 9893 | . 14767 | 6.772 | . 6 |
| . 5 | . 04362 | . 9990 | . 04366 | 22.90 | . 5 | . 5 | . 14781 | . 9890 | . 14945 | 6.691 | . 5 |
| . 6 | . 04536 | . 9990 | . 04541 | 22.02 | . 4 | . 6 | . 14954 | . 9888 | . 15124 | 6.612 | . 4 |
| 7 | . 04711 | . 9989 | . 04716 | 21.20 | . 3 | . 7 | . 15126 | . 9885 | . 15302 | 6.535 | . 3 |
| 8 | . 04885 | . 9988 | . 04891 | 20.45 | . 2 | 8 | . 15299 | . 9882 | . 15481 | 6.460 | .$^{\circ}$ |
| . 9 | . 05059 | . 9987 | . 05066 | 19.74 | . 1 | . 9 | . 15471 | . 9880 | .15660 | 6.386 | . 1 |
| 3.0 | . 05234 | 0.9986 | . 05241 | 19.081 | 87.0 | 9.0 | . 15643 | 0.9877 | . 15838 | 6.314 | 81.0 |
| . 1 | . 05408 | . 9985 | . 05416 | 18.464 | . 9 | . 1 | . 15816 | . 9874 | . 16017 | 6.243 | . 9 |
| .2 | . 05582 | . 9984 | . 05591 | 17.886 | 8 | . 2 | . 15988 | . 9871 | . 16196 | 6.174 | . 8 |
| 3 | . 05756 | . 9983 | . 05768 | 17.343 | . 7 | 3 | . 16160 | . 9869 | . 16376 | 6.107 | 7 |
| . 4 | . 05931 | . 9982 | . 05941 | 16.832 | . 6 | . 4 | . 16333 | . 9868 | . 16555 | 6.041 | . 6 |
| . 5 | . 08105 | . 9981 | . 06116 | 16.350 | . 5 | . 5 | . 16505 | . 9863 | . 16734 | 5.976 | . 5 |
| . 6 | . 06279 | . 9980 | . 08291 | 15.895 | . 4 | 6 | . 16677 | . 9860 | . 16914 | 5.912 | 4 |
| .7 | . 04453 | . 9979 | . 08467 | 15.464 | .3 | 7 | . 16849 | . 9857 | . 17093 | 5.850 5 | 3 |
| . 8 | . 06627 | . 9978 | . 06842 | 15.056 | . 2 | 8 | . 17021 | . 9854 | . 17273 | 5.789 | . 2 |
| . 9 | . 06802 | . 9977 | . 08817 | 14.669 | . 1 | . 9 | . 17193 | . 9851 | . 17453 | 5.730 | . 1 |
| 4.0 | . 06976 | 0.9976 | . 06993 | 14.301 | 18.0 | 10.0 | . 1736 | 0.9848 | . 1763 | 5.671 | 80.0 |
| . 1 | . 07150 | . 9974 | . 07168 | 13.951 | . 9 | . 1 | . 1754 | . 9845 | . 1781 | 5.614 | 9 |
| . 2 | . 07324 | . 9973 | . 07344 | 13.617 | 8 | . 2 | . 1771 | . 9842 | . 1799 | 5.558 | 8 |
| . 3 | . 07498 | . 9972 | . 07519 | 13.300 | . 7 | . 3 | . 1788 | . 9839 | . 1817 | 5.503 | . 7 |
| . 4 | . 07672 | . 9971 | . 07695 | 12.996 | . 6 | 4 | . 1805 | . 9836 | . 1835 | 5.449 | . 6 |
| . 5 | . 07846 | . 9969 | . 07870 | 12.706 | . 5 | . 5 | . 1822 | . 9833 | . 1853 | 5.396 | . 5 |
| . 6 | . 08020 | . 9968 | . 08046 | 12.429 | . 4 | 6 | . 1840 | . 9829 | . 1871 | 5.343 | . 4 |
| . 7 | . 08194 | . 9966 | . 08221 | 12.163 | . 3 | . 7 | . 1857 | . 9826 | . 1890 | 5.292 | . 3 |
| 8 | . 08368 | . 9965 | . 08397 | 11.909 | . 2 | 8 | . 1874 | .9823 | . 1908 | 5.242 5.193 | . 1 |
| . 9 | . 08542 | . 9963 | . 08573 | 11.664 | . 1 | . 9 | .1891 | . 9820 | . 1926 | 5.193 | . |
| 3.0 | . 08716 | 0.9962 | . 08749 | 11.430 | 85.0 | 11.0 | . 1908 | 0.9816 | . 1944 | 5.145 | 79.0 |
| . 1 | '08889 | . 9960 | . 08925 | 11.205 | . 9 | . 1 | . 1925 | . 9813 | . 1962 | 5.097 | . 9 |
| . 2 | . 09063 | . 9959 | . 09101 | 10.988 | 8 | . 2 | . 1942 | . 9810 | . 1980 | 5.050 | 8 |
| . 3 | . 09237 | . 9957 | . 09277 | 10.780 | . 7 | . 3 | . 1959 | . 9806 | . 1998 | 5.005 | 7 |
| . 4 | . 09411 | . 9956 | . 09453 | 10.579 | . 6 | . 4 | . 1977 | . 9803 | . 2016 | 4.959 | . 6 |
| . 5 | . 08585 | . 9954 | . 09629 | 10.385 | . 5 | . 5 | . 1994 | . 9799 | . 2035 | 4.915 | . 5 |
| . 6 | . 09758 | . 9952 | . 09805 | 10.199 | . 4 | . 6 | . 2011 | . 9796 | . 2053 | 4.872 | . 4 |
| . 7 | . 09932 | . 9951 | . 09981 | 10.019 | .3 | 7 | . 2028 | . 9792 | . 2071 | 4.829 | 3 |
| . 8 | . 10108 | . 9949 | . 10158 | 9.845 | . 2 | 8 | . 2045 | . 9789 | . 2089 | 4.787 | . 1 |
| . 9 | .10279 | . 9947 | . 10334 | 9.677 | . 1 | . 9 | . 2062 | . 9785 | . 2107 | 4.745 | 1 |
| 6.0 | . 10453 | 0.9945 | .10510 | 9.514 | 84.0 | 12.0 | . 2079 | 0.9781 | 2126 | 4.705 | 78.0 |
|  | cos | $\sin$ | cot | Sn | deg |  | cos | $\sin$ | cot | \%n | deg |

Nafural trigonomefric tunctions
for decimal fractions of a degree conlinued

| deg | $\sin$ | cos | Pan | cof |  | dog | $\sin$ | cos | fan | cof |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 12.0 | 0.2079 | 0.9781 | 0.2126 | 4.705 | 78.0 | 18.0 | 0.3090 | 0.9511 | 0.3249 | 3.078 | 72.0 |
| . 1 | . 2096 | . 9778 | . 2144 | 4.665 | . 9 | . | . 3107 | . 9505 | . 3269 | 3.060 | . 9 |
| . 2 | . 2113 | . 9774 | . 2162 | 4.625 | . 8 | . 2 | . 3123 | . 9500 | . 3288 | 3.042 | . 8 |
| . 3 | . 2130 | . 9770 | . 2180 | 4.586 | . 7 | . 3 | . 3140 | . 9494 | . 3307 | 3.024 | . 7 |
| .4 | . 2147 | . 9767 | . 2199 | 4.546 | . 6 | . 4 | . 3156 | . 9489 | . 3327 | 3.006 | . 6 |
| . 5 | 2164 | . 9763 | . 2217 | 4.511 | . 5 | . 5 | . 3173 | . 9483 | . 3346 | 2.989 | . 5 |
| . 6 | . 2181 | . 9759 | . 2235 | 4.474 | . 4 | . 6 | . 3190 | . 9478 | . 3365 | 2.971 | . 4 |
| 7 | . 2198 | . 9755 | . 2254 | 4.437 | . 3 | . 7 | . 3205 | . 9472 | . 3385 | 2.954 | . 3 |
| . 8 | . 2215 | . 9751 | .2272 | 4.402 | . 2 | . 8 | . 3223 | . 9466 | . 3404 | 2.937 | . 2 |
| . 9 | .2233 | . 9748 | . 2290 | 4.366 | . 1 | . 9 | . 3239 | . 9461 | . 3424 | 2.921 | .1 |
| 13.0 | 0.2250 | 0.9744 | 0.2309 | 4.331 | 77.0 | 19.0 | 0.3256 | 0.9455 | 0.3443 | 2.904 | 71.0 |
| .1 | . 2267 | . 9740 | . 2327 | 4.297 | . 9 | . 1 | . 3272 | . 9449 | . 3463 | 2.888 | . 9 |
| . 2 | . 2284 | . 9736 | . 2345 | 4.264 | . 8 | .2 | . 3289 | . 9444 | . 3482 | 2.872 | . 8 |
| .3 | . 2300 | . 9732 | . 2364 | 4.230 | . 7 | . 3 | .3305 | . 9438 | . 3502 | 2.856 | . 7 |
| . 4 | . 2317 | . 9728 | . 2382 | 4.198 | . 6 | . 4 | . 3322 | . 9432 | . 3522 | 2.840 | . 6 |
| . 5 | . 2334 | . 9724 | . 2401 | 4.165 | . 5 | . 5 | . 3338 | . 9426 | . 3541 | 2.824 | . 5 |
| . 6 | . 2351 | . 9720 | . 2419 | 4.134 | . 4 | . 6 | . 3355 | . 9421 | . 3561 | 2.808 | . 4 |
| . 7 | . 2368 | . 9715 | . 2438 | 4.102 | .3 | .7 | . 3371 | . 9415 | . 3581 | 2.793 | . 3 |
| . 8 | 2385 | . 9711 | 2456 | 4.078 | . 2 | . 8 | . 3387 | . 9409 | . 3600 | 2.778 | 2 |
| . 9 | . 2402 | . 9707 | . 2475 | 4.041 | . 1 | . 9 | . 3404 | . 9403 | . 3620 | 2.762 | . 1 |
| 14.0 | 0.2419 | 0.9703 | 0.2493 | 4.011 | 76.0 | 20.0 | 0.3420 | 0.9397 | 0.3640 | 2.747 | 70.0 |
| . 1 | 2436 | . 9699 | . 2512 | 3.981 | . 9 | . 1 | . 3437 | . 9391 | . 3859 | 2.733 | . 9 |
| 2 | . 2453 | . 9694 | .2530 | 3.952 | 8 | . 2 | . 3453 | . 9385 | . 3679 | 2.718 | . 8 |
| 3 | . 2470 | . 9690 | . 2549 | 3.923 | . 7 | . 3 | . 3469 | . 9379 | . 3699 | 2.703 | . 7 |
| . 4 | . 2487 | . 9686 | . 2568 | 3.895 | . 6 | . 4 | . 3486 | . 9373 | . 3719 | 2.689 | . 6 |
| . 5 | 2504 | . 9681 | . 2588 | 3.867 | . 5 | . 5 | . 3502 | . 9367 | . 3739 | 2.675 | . 5 |
| . 6 | 2521 | . 9677 | .2605 | 3.839 | . 4 | . 6 | . 3518 | . 9361 | . 3759 | 2.660 | 4 |
| . 7 | 2538 | . 9673 | 2623 | 3.812 | .3 | .7 | . 3535 | . 9354 | . 3779 | 2.646 | . 3 |
| 8 | . 2554 | . 9668 | . 2642 | 3.785 | . 2 | . 8 | . 3551 | . 9348 | . 3799 | 2.633 | . 2 |
| . 9 | .2571 | . 9864 | . 2661 | 3.758 | . 1 | . 9 | 3567 | . 9342 | . 3819 | 2.619 | . 1 |
| 13.0 | 0.2588 | 0.9659 | 0.2679 | 3.732 | 75.0 | 21.0 | 0.3584 | 0.9338 | 0.3839 | 2.805 | 69.0 |
| . 1 | . 2605 | . 9655 | . 2698 | 3.706 | . 9 | . 1 | . 3600 | . 9330 | . 3859 | 2.592 | . 9 |
| . 2 | . 2622 | . 9650 | . 2717 | 3.681 | . 8 | . 2 | . 3616 | . 9323 | . 3879 | 2.578 | . 8 |
| . 3 | . 2639 | . 9646 | . 2736 | 3.655 | . 7 | . 3 | . 3633 | . 9317 | . 3899 | 2.565 | . 7 |
| . 4 | . 2656 | . 9641 | . 2754 | 3.630 | . 6 | . 4 | . 3649 | . 9311 | . 3919 | 2.552 | . 6 |
| . 5 | .2672 | . 9636 | . 2773 | 3.606 | . 5 | .5 | . 3665 | . 9304 | . 3939 | 2.539 | . 5 |
| . 6 | . 2689 | . 9632 | 2792 | 3.582 | . 4 | . 6 | . 3681 | . 9298 | . 3959 | 2.526 | 4 |
| 7 | . 2706 | . 9627 | . 2811 | 3.558 | . 3 | . 7 | 3697 | . 9291 | . 3979 | 2.513 | . 3 |
| . 8 | .2723 | . 9622 | . 2830 | 3.534 | . 2 | . 8 | . 3714 | . 9228 | . 4000 | 2.500 | . 2 |
| . 9 | 2740 | . 9817 | . 2849 | 3.511 | .1 | . 9 | . 3730 | . 9278 | . 4020 | 2.488 | . 1 |
| 16.0 | 0.2756 | 0.9813 | 0.2867 | 3.487 | 74.0 | 22.0 | 0.3746 | 0.9272 | 0.4040 | 2.475 | 68.0 |
| . 1 | . 2773 | . 9608 | . 2886 | 3.465 | . 9 | . 1 | . 3762 | . 9265 | . 4061 | 2.463 | . 9 |
| . 2 | . 2790 | . 9603 | . 2905 | 3.442 | . 8 | . 2 | . 3778 | . 9259 | . 4081 | 2.450 | . 8 |
| . 3 | .2807 | . 9598 | . 2924 | 3.420 | . 7 | . 3 | . 3795 | . 9252 | .4101 | 2.438 | . 7 |
| . 4 | . 2823 | . 9593 | .2943 | 3.398 | . 6 | . 4 | . 3811 | . 9245 | . 4122 | 2.426 | . 6 |
| . 5 | . 2840 | . 9588 | . 2962 | 3.376 | . 5 | . 5 | .3827 | . 9239 | . 4142 | 2.414 | . 5 |
| . 6 | . 2857 | . 9583 | . 2981 | 3.354 | . 4 | . 6 | . 3843 | . 9232 | .4163 | 2.402 | . 4 |
| . 7 | . 2874 | . 9578 | . 3000 | 3.333 | .3 | . 7 | . 3859 | . 9225 | . 4183 | 2.391 | . 3 |
| . 8 | . 2890 | . 9573 | . 3019 | 3.312 | . 2 | . 8 | . 3875 | . 9219 | . 4204 | 2.379 | 2 |
| . 9 | . 2907 | . 9568 | . 3038 | 3.291 | . 1 | . 9 | . 3891 | . 9212 | . 4224 | 2.367 | . 1 |
| 17.0 | 0.2924 | 0.9563 | 0.3057 | 3.271 | 73.0 | 23.0 | 0.3907 | 0.9205 | 0.4245 | 2.356 | 87.0 |
| . 1 | . 2940 | . 9558 | . 3076 | 3.251 | . 9 | . 1 | . 3923 | . 9198 | . 4265 | 2.344 | . 9 |
| .2 | 2957 | . 9553 | . 3096 | 3.230 | . 8 | . 2 | . 3939 | . 9191 | . 4286 | 2.333 | . 8 |
| .3 | . 2974 | . 9548 | . 3115 | 3.211 | 7 | . 3 | . 3955 | . 9184 | . 4307 | 2.322 | 7 |
| . 4 | . 2990 | . 9542 | . 3134 | 3.191 | . 6 | . 4 | . 3971 | . 9178 | . 4327 | 2.311 | . 6 |
| . 5 | . 3007 | . 9537 | . 3153 | 3.172 | . 5 | . 5 | . 3987 | . 9171 | . 4348 | 2.300 | . 5 |
| .6 | . 3024 | . 9532 | . 3172 | 3.152 | .4 | .6 | . 4003 | . 9164 | . 4369 | 2.289 | . 4 |
| . 7 | . 3040 | . 9527 | . 3191 | 3.133 | .3 | . 7 | . 4019 | . 9157 | .4390 | 2.278 | . 3 |
| . 8 | 3057 | . 9521 | . 3211 | 3.115 | . 2 | . 8 | . 4035 | . 9150 | . 4411 | 2.267 | . 2 |
| . 9 | 3074 | . 9516 | .3230 | 3.096 | . 1 | . 9 | . 4051 | . 9143 | . 4431 | 2.257 | . 1 |
| 18.0 | 0.3090 | 0.9511 | 0.3249 | 3.078 | 72.0 | 24.0 | 0.4067 | 0.9135 | 0.4452 | 2.246 | 66.0 |
|  | ces | $\sin$ | col | tan | dog |  | cos | $\sin$ | cot | tan | deg |

for decimal fractions of a degree continued

| deg | $\sin$ | ces | ton | col |  | deg | $\sin$ | cot | Tan | cot |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 24.0 | 0.4067 | 0.9135 | 0.4452 | 2.246 | 66.0 | 30.0 | 0.5000 | 0.8660 | 0.5774 | 1.7321 | 60.0 |
| . 1 | . 4083 | . 9128 | . 4473 | 2.236 | . 9 | . 1 | . 5015 | . 8052 | . 5797 | 1.7251 | . 9 |
| . 2 | . 4099 | . 9121 | . 4494 | 2.225 | 8 | . 2 | . 5030 | 8043 | . 5820 | 1.7182 | . 8 |
| . 3 | . 4115 | . 9114 | . 4515 | 2.215 | . 7 | .3 | . 5045 | . 8034 | . 5844 | 1.7113 | . 7 |
| . 4 | . 4131 | . 9107 | . 4536 | 2.204 | . 6 | . 4 | . 5060 | . 8025 | . 5887 | 1.7045 | . 6 |
| . 5 | . 4147 | . 9100 | . 4557 | 2.194 | . 5 | . 5 | . 5075 | . 8816 | . 5890 | 1.6977 | . 5 |
| . 6 | . 4163 | . 9092 | . 4578 | 2.184 | . 4 | .6 | . 5090 | . 8007 | . 5914 | 1.6909 | . 4 |
| . 7 | . 4179 | .9785 | . 4599 | 2.174 | . 3 | . 7 | . 5105 | 8599 | . 5938 | 1.6842 | 3 |
| 8 | . 4195 | . 9078 | . 4621 | 2.164 | . 2 | 8 | . 5120 | 8590 | . 5961 | 1.6775 | 2 |
| . 9 | . 4210 | . 9070 | . 4642 | 2.154 | . 1 | . 9 | . 5135 | . 8581 | . 5985 | 1.6709 | . 1 |
| 25.0 | 0.4226 | 0.9063 | 0.4663 | 2.145 | 85.0 | 31.0 | 0.5150 | 0.8572 | 0.6009 | 1.6643 | 59.0 |
| . 1 | . 4242 | . 9056 | . 4684 | 2.135 | . 9 | . 1 | . 5165 | . 8563 | . 6032 | 1.6577 | . 9 |
| 2 | . 4258 | . 9048 | . 4706 | 2.125 | . 8 | . 2 | . 5180 | . 8554 | . 6056 | 1.6512 | 8 |
| 3 | . 4274 | . 9041 | . 4727 | 2.116 | . 7 | . 3 | . 5195 | . 8545 | . 6080 | 1.6447 | . 7 |
| . 4 | . 4289 | . 9033 | . 4748 | 2.106 | . 6 | . 4 | . 5210 | . 8536 | . 6104 | 1.6383 | 6 |
| . 5 | . 4305 | . 9026 | . 4770 | 2.097 | . 5 | . 5 | . 5225 | . 8526 | . 6128 | 1.6319 | . 5 |
| . 6 | .4321 | . 9018 | . 4791 | 2.087 | . 4 | . 6 | . 5240 | . 8517 | . 6152 | 1.6255 | . 4 |
| . 7 | . 4337 | . 9011 | . 4813 | 2.078 | .3 | . 7 | . 5255 | .8508 | . 6176 | 1.6191 | . 3 |
| . 8 | . 4352 | . 9003 | . 4834 | 2.069 | 2 | . 8 | . 5270 | . 8499 | . 6200 | 1.6128 | 2 |
| . 9 | . 4368 | . 8996 | . 4856 | 2.059 | . 1 | . 9 | . 5284 | . 8490 | . 6224 | 1.6066 | . 1 |
| 26.0 | 0.4384 | 0.8988 | 0.4877 | 2.050 | 64.0 | 32.0 | 0.5299 | 0.8480 | 0.6249 | 1.6003 | 58.0 |
| .1 | . 4399 | . 8988 | . 4899 | 2.041 | . 9 | . 1 | . 5314 | . 8471 | . 6273 | 1.5941 | . 9 |
| 2 | .4415 | . 8973 | .4921 | 2.032 | 8 | .2 | . 5329 | . 8462 | . 6297 | 1.5880 | 8 |
| 3 | . 4431 | . 8965 | . 4942 | 2.023 | 7 | . 3 | . 5344 | 8453 | . 6322 | 1.5818 | . 7 |
| . 4 | . 4446 | . 8957 | . 4964 | 2.014 | . 6 | . 4 | . 5358 | 8443 | . 6346 | 1.5757 | . 6 |
| . 5 | . 4462 | . 8949 | . 4986 | 2.006 | . 5 | . 5 | . 5373 | . 8434 | . 6371 | 1.5697 | . 5 |
| 6 | . 4478 | . 8942 | . 5008 | 1.997 | 4 | . 6 | . 5388 | . 8425 | . 6395 | 1.5637 | . 4 |
| . 7 | . 4493 | . 8934 | . 5029 | 1.988 | . 3 | . 7 | . 5402 | . 8415 | . 6420 | 1.5577 | . 3 |
| . 8 | . 4509 | . 8926 | . 5051 | 1.980 | . 2 | . 8 | . 5417 | . 8406 | . 6445 | 1.5517 | . 2 |
| . 9 | . 4524 | . 8918 | . 5073 | 1.971 | . 1 | . 9 | . 5432 | . 8396 | . 6469 | 1.5458 | .1 |
| 27.0 | 0.4540 | 0.8910 | 0.5095 | 1.963 | 63.0 | 33.0 | 0.5446 | 0.8387 | 0.6494 | 1.5399 | 57.0 |
| . 1 | . 4555 | . 8902 | . 5117 | 1.954 | . 9 | . 1 | . 5461 | . 8377 | . 6519 | 1.5340 | . 9 |
| 2 | . 4571 | . 8894 | . 5139 | 1.946 | . 8 | . 2 | . 5476 | . 8368 | . 6544 | 1.5282 | . 8 |
| . 3 | . 4586 | . 8888 | . 5161 | 1.937 | . 7 | . 3 | . 5490 | . 8358 | . 6569 | 1.5224 | . 7 |
| . 4 | . 4602 | . 8878 | . 5184 | 1.929 | . 6 | . 4 | . 5505 | -. 8348 | . 6594 | 1.5166 | . 6 |
| . 5 | . 4617 | . 8870 | . 5206 | 1.921 | . 5 | . 5 | . 5519 | . 8339 | . 6619 | 1.5108 | . 5 |
| . 6 | .4633 | . 8882 | . 5228 | 1.913 | . 4 | . 6 | . 5534 | . 8329 | . 6644 | 1.5051 | . 4 |
| . 7 | . 4648 | . 8854 | . 5250 | 1.905 | .3 | . 7 | . 5548 | . 8320 | . 6869 | 1.4994 | 3 |
| . 8 | . 4664 | . 8846 | . 5272 | 1.897 | . 2 | . 8 | . 5563 | . 8310 | . 6694 | 1.4938 | 2 |
| . 9 | . 4679 | . 8838 | . 5295 | 1.869 | . 1 | . 9 | . 5577 | 8300 | . 6720 | 1.4882 | . 1 |
| 28.0 | 0.4695 | 0.8829 | 0.5317 | 1.881 | 62.0 | 34.0 | 0.5592 | 0.8290 | 0.6745 | 1.4826 | 56.0 |
| . 1 | . 4710 | . 8821 | . 5340 | 1.873 | . 9 | . 1 | . 5606 | . 8281 | . 6771 | 1.4770 | . 9 |
| . 2 | . 4726 | . 8813 | . 5362 | 1.865 | . 8 | . 2 | . 5621 | . 8271 | .6796 | 1.4715 | . 8 |
| . 3 | . 4741 | . 8805 | . 5384 | 1.857 | . 7 | . 3 | . 5635 | . 8261 | . 6822 | 1.4659 | . 7 |
| 4 | . 4756 | . 8796 | . 5407 | 1.849 | . 6 | . 4 | . 5650 | . 8251 | . 6847 | 1.4605 | . 6 |
| . 5 | . 4772 | . 8788 | . 5430 | 1.842 | . 5 | . 5 | . 5664 | . 8241 | . 6873 | 1.4550 | . 5 |
| . 6 | . 4787 | . 8780 | . 5452 | 1.834 | .4 | . 6 | . 5678 | . 8231 | . 6899 | 1.4496 | . 4 |
| . 7 | . 4802 | . 8771 | . 5475 | 1.827 | . 3 | . 7 | . 5693 | . 8221 | . 6924 | 1.4442 | . 3 |
| . 8 | . 4818 | . 8763 | . 5498 | 1.819 | . 2 | . 8 | . 5707 | . 8211 | . 6950 | 1.4388 | . 2 |
| . 9 | . 4833 | . 8755 | . 5520 | 1.811 | . 1 | . 9 | . 5721 | . 8202 | . 6976 | 1.4335 | . 1 |
| 29.0 | 0.4848 | 0.8746 | 0.5543 | 1.804 | 61.0 | 35.0 | 0.5736 | 0.8192 | 0.7002 | 1.4281 | 55.0 |
| . 1 | . 4863 | . 8738 | . 5566 | 1.797 | . 9 | . 1 | . 5750 | . 8181 | . 7028 | 1.4229 | . 9 |
| . 2 | . 4879 | . 8729 | . 55889 | 1.789 | . 8 | . 2 | . 5764 | . 8171 | . 7054 | 1.4176 | 8 |
| . 3 | . 4894 | . 8721 | . 5612 | 1.782 | . 7 | 3 | . 5779 | . 8161 | . 7080 | 1.4124 | . 7 |
| . 4 | . 4909 | . 8712 | . 5635 | 1.775 | . 6 | . 4 | . 5793 | . 8151 | . 7107 | 1.4071 | . 6 |
| . 5 | . 4924 | . 8704 | . 5658 | 1.767 | . 5 | . 5 | . 5807 | . 8141 | . 7133 | 1.4019 | . 5 |
| . 6 | . 4939 | . 8695 | . 5881 | 1.760 | 4 | . 6 | . 5821 | . 8131 | . 7159 | 1.3968 | . 4 |
| . 7 | . 4955 | . 8686 | . 5704 | 1.753 | . 3 | . 7 | . 5835 | . 8121 | . 7186 | 1.3916 | . 3 |
| . 8 | . 4970 | . 8678 | . 57727 | 1.746 | . 2 | . 8 | . 5850 | 8111 | . 7212 | 1.3865 | . 2 |
| . 9 | . 4985 | . 8669 | . 5750 | 1.739 | . 1 | . 9 | . 5864 | 8100 | . 7239 | 1.3814 | . 1 |
| 30.0 | 0.5000 | 0.8660 | 0.5774 | 1.732 | 60.0 | 36.0 | 0.5878 | 0.8090 | 0.7265 | 1.3764 | 54.0 |
|  | cos | $\sin$ | cot | Ian | deg |  | cos | $\sin$ | cot | fon | deg |

Natural trigonometric functions
for decimal fractions of a degree continued

| deg | sin | cos | Ian | cot |  | dea | $\sin$ | cos | ton | cot |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 36.0 | 0.5878 | 0.8090 | 0.7265 | 1.3764 | 34.0 | 40.3 | 0.6494 | 0.7604 | 0.8541 | 1.1708 | 49.5 |
| . 1 | . 5892 | . 8080 | 7292 | 1.3713 | . 9 | . 6 | . 6508 | 7593 | . 8571 | 1.1667 | . 4 |
| . 2 | . 5906 | . 8070 | . 7319 | 1.3663 | . 8 | . 7 | . 6521 | . 7581 | . 8601 | 1.1626 | 3 |
| 3 | . 5920 | . 8059 | . 7346 | 1.3613 | . 7 | . 8 | . 6534 | 7570 | 8632 | 1.1585 | . 2 |
| . 4 | . 5934 | . 8049 | . 7373 | 1.3564 | . 6 | . 9 | . 6547 | . 7559 | . 8662 | 1.1544 | . 1 |
| . 5 | . 5948 | . 8039 | . 7400 | 1.3514 | . 5 | 41.0 | 0.6561 | 0.7547 | 0.8693 | 1.1504 | 49.0 |
| . 6 | . 5962 | . 8028 | . 7427 | 1.3465 | .4 | . 1 | . 6574 | . 7536 | . 8724 | 1.1463 | . 9 |
| 7 | . 5976 | . 8018 | . 7454 | 1.3416 | . 3 | . 2 | . 6587 | . 7524 | . 8754 | 1.1423 | 8 |
| 8 | . 5990 | . 8007 | .7481 | 1.3367 | . 2 | . 3 | . 6600 | 7513 | 8785 | 1.1383 | 7 |
| .9 | . 6004 | . 7997 | .7508 | 1.3319 | . 1 | . 4 | . 6613 | . 7501 | 8816 | 1.1343 | . 6 |
| 37.0 | 0.6018 | 0.7986 | 0.7536 | 1.3270 | 53.0 | . 5 | . 6626 | . 7490 | 8847 | 1.1303 | . 5 |
| .1 | . 6032 | . 7976 | . 7563 | 1.3222 | . 9 | . 6 | . 6639 | . 7478 | . 8878 | 1.1263 | 4 |
| 2 | . 6046 | . 7965 | . 7590 | 1.3175 | . 8 | . 7 | . 6652 | . 7466 | 8910 | 1.1224 | . 3 |
| 3 | . 6060 | . 7955 | . 7618 | 1.3127 | . 7 | 8 | . 6665 | 7455 | 8941 | 1.1184 | 2 |
| . 4 | . 6074 | . 7944 | 7646 | 1.3079 | . 6 | . 9 | . 6678 | . 7443 | 8972 | 1.1145 | . 1 |
| . 5 | . 6088 | . 7934 | . 7673 | 1.3032 | . 5 | 42.0 | 0.6691 | 0.7431 | 0.9004 | 1.1108 | 48.0 |
| 6 | . 6101 | .7923 | 7701 | 1.2985 | . 4 | .1 | . 6704 | . 7420 | . 9036 | 1.1057 | . 9 |
| . 7 | . 6115 | 7912 | . 7729 | 1.2938 | . 3 | . 2 | . 6717 | . 7408 | . 9067 | 1.1028 | 8 |
| 8 | . 6129 | . 7902 | .7757 | 1.2892 | . 2 | . 3 | . 6730 | . 7396 | . 9099 | 1.0990 | . 7 |
| .9 | . 6143 | . 7891 | . 7785 | 1.2846 | .1 | . 4 | . 6743 | . 7385 | . 9131 | 1.0951 | . 6 |
| 38.0 | 0.8157 | 0.7880 | 0.7813 | 1.2799 | 32.0 | . 5 | . 6756 | 7373 | . 9163 | 1.0913 | . 5 |
| . 1 | . 6170 | . 7869 | . 7841 | 1.2753 | . 9 | . 6 | . 6769 | . 7361 | . 9195 | 1.0875 | 4 |
| . 2 | . 6184 | . 7859 | . 7869 | 1.2708 | 8 | . 7 | . 6782 | . 7349 | . 9228 | 1.0837 | 3 |
| . 3 | . 6198 | . 7848 | . 7898 | 1.2662 | . 7 | . 8 | . 6794 | . 7337 | . 9260 | 1.0799 | 2 |
| . 4 | .8211 | . 7837 | . 7926 | 1.2617 | . 6 | . 9 | . 6807 | . 7325 | . 9293 | 1.0761 | . 1 |
| . 5 | . 6225 | . 7826 | . 7954 | 1.2572 | . 5 | 43.0 | 0.6820 | 0.7314 | 0.9325 | 1.0724 | 47.0 |
| 6 | . 6239 | . 7815 | . 7983 | 1.2527 | . 4 | .1 | . 6833 | . 7302 | . 9358 | 1.0686 | . 9 |
| . 7 | . 6252 | . 7804 | . 8012 | 1.2482 | 3 | 2 | . 6845 | . 7290 | . 9391 | 1.0649 | . 8 |
| 8 | . 6268 | . 7793 | . 8040 | 1.2437 | . 2 | 3 | . 6858 | . 7278 | . 9424 | 1.0612 | 7 |
| . 9 | . 6280 | . 7782 | . 8069 | 1.2393 | . 1 | . 4 | . 6871 | . 7268 | . 9457 | 1.0575 | . 6 |
| 39.0 | 0.6293 | 0.7771 | 0.8098 | 1.2349 | 51.0 | . 5 | . 6884 | . 7254 | . 9490 | 1.0538 | 5 |
| . 1 | . 8307 | . 7760 | . 8127 | 1.2305 | . 9 | . 6 | . 6896 | . 7242 | . 9523 | 1.0501 | . 4 |
| .2 | . 6320 | . 7749 | .8156 | 1.2281 | 8 | . 7 | . 6909 | 7230 | . 9556 | 1.0464 | 3 |
| .3 | . 6334 | . 7738 | . 8185 | 1.2218 | . 7 | . 8 | . 6921 | . 7218 | . 9590 | 1.0428 | 2 |
| .4 | . 6347 | 7727 | . 8214 | 1.2174 | .6 | . 9 | . 6934 | . 7206 | . 9623 | 1.0392 | . 1 |
| . 5 | . 6361 | . 7716 | . 8243 | 1.2131 | . 5 | 44.0 | 0.6947 | 0.7193 | 0.9657 | 1.0355 | 46.0 |
| . 6 | . 6374 | . 7705 | . 8273 | 1.2688 | 4 | . 1 | . 6959 | . 7181 | . 9691 | 1.0319 | . 9 |
| . 7 | . 6388 | . 7694 | . 8302 | 1.2045 | . 3 | . 2 | . 6972 | . 7169 | . 9725 | 1.0283 | 8 |
| 8 | . 6401 | . 7683 | . 8332 | 1.2002 | . 2 | . 3 | . 6984 | . 7157 | . 9759 | 1.0247 | . 7 |
| . 9 | . 6414 | . 7672 | . 8361 | 1.1960 | . 1 | . 4 | . 6997 | . 7145 | . 9793 | 1.0212 | . 6 |
| 40.0 | 0.6428 | 0.7880 | 0.8391 | 1.1918 | 50.0 | . 5 | . 7009 | . 7133 | . 9827 | 1.0176 | 5 |
| . 1 | . 6441 | . 7649 | . 8421 | 1.1875 | . 9 | . 6 | . 7022 | . 7120 | .9861 | 1.0141 | . 4 |
| . 2 | . 6455 | .7638 | . 8451 | 1.1833 | . 8 | . 7 | . 7034 | .7108 | . 9896 | 1.0105 | 3 |
| 3 | . 6468 | . 7627 | . 8481 | 1.1792 | . 7 | . 8 | . 7046 | . 7096 | . 9930 | 1.0070 | 2 |
| . 4 | . 6481 | . 7615 | 8.811 | 1.1750 | . 6 | . 9 | . 7059 | . 7083 | . 9965 | 1.0035 | . 1 |
| 40.3 | 0.6494 | 0.7604 | 0.8541 | 1.1708 | 49.5 | 45.0 | 0.7071 | 0.7071 | 1.0000 | 1.0000 | 45.0 |
|  | cos | $\sin$ | cof | fan | deg |  | cos | sta | col | tan | deg |

Logarithms of trigonometric functions
for decimal fractions of a degree

| doy | $L \sin$ | 1 cos | 1 tan | 1 col |  | deg | 4 sin | L cos | 1 Ion | 1 col |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | - - | 0.0000 | - - | $\infty$ | 90.0 | 6.0 | 9.0192 | 9.9976 | 9.0216 | 0.9784 | 4.0 |
| . 1 | 7.2419 | 0.0000 | 7.2419 | 2.7581 | . 9 | . 1 | 9.0264 | 9.9975 | 9.0289 | 0.9711 | . 9 |
| . 2 | 7.5429 | 0.0000 | 7.5429 | 2.4571 | . 8 | . 2 | 9.0334 | 9.9975 | 9.0360 | 0.9640 | . 8 |
| . 3 | 7.7190 | 0.0000 | 7.7190 | 2.2810 | . 7 | 3 | 9.0403 | 9.9974 | 9.0430 | 0.9570 | 7 |
| . 4 | 7.8439 | 0.0000 | 7.8439 | 2.1561 | . 6 | . 4 | 9.0472 | 9.9973 | 9.0499 | 0.9501 | . 6 |
| 5 | 7.9408 | 0.0000 | 7.9409 | 2.0591 | . 5 | . 5 | 9.0539 | 9.9972 | 9.0567 | 0.9433 | . 5 |
| . 6 | 8.0200 | 0.0000 | 8.0200 | 1.9800 | . 4 | . 6 | 9.0605 | 9.9971 | 9.0633 | 0.9367 | . 4 |
| . 7 | 8.0870 | 0.0000 | 8.0870 | 1.9130 | 3 | 7 | 9.0670 | 9.9970 | 9.0699 | 0.9301 | 3 |
| . 8 | 8.1450 | 0.0000 | 8.1450 | 1.8550 | . 2 | . 8 | 9.0734 | 9.9969 | 9.0764 | 0.9236 | 2 |
| . 9 | 8.1961 | 9.9999 | 8.1962 | 1.8038 | . 1 | . 9 | 9.0797 | 9.9968 | 9.0828 | 0.9172 | . 1 |
| 1.0 | 8.2419 | 9.9999 | 8.2419 | 1.7581 | 89.0 | 7.0 | 9.0859 | 9.9968 | 9.0891 | 0.9109 | 83.0 |
| . 1 | 8.2832 | 9.9999 | 8.2833 | 1.7167 | . 9 | . 1 | 9.0920 | 9.9967 | 9.0954 | 0.9046 | . 9 |
| 2 | 8.3210 | 9.9999 | 8.3211 | 1.6769 | . 8 | . 2 | 9.0981 | 9.9966 | 9.1015 | 0.8985 | . 8 |
| . 3 | 8.3558 | 9.9999 | 8.3559 | 1.6441 | 7 | . 3 | 9.1040 | 9.9965 | 9.1076 | 0.8924 | . 7 |
| . 4 | 8.3880 | 9.9999 | 8.3881 | 1.6119 | . 6 | . 4 | 9.1099 | 9.9964 | 9.1135 | 0.8885 | . 6 |
| . 5 | 8.4179 | 9.9999 | 8.4181 | 1.5819 | . 5 | . 5 | 9.1157 | 9.9963 | 9.1194 | 0.8806 | . 5 |
| . 6 | 8.4459 | 9.9998 | 8.4461 | 1.5539 | . 4 | . 6 | 9.1214 | 9.9962 | 9.1252 | 0.8748 | . 4 |
| . 7 | 8.4723 | 9.9998 | 8.4725 | 1.5275 | 3 | . 7 | 9.1271 | 9.9961 | 9.1310 | 0.8690 | . 3 |
| . 8 | 8.4971 | 9.9998 | 8.4973 | 1.5027 | . 2 | . 8 | 9.1326 | 9.9960 | 9.1367 | 0.8833 | 2 |
| . 9 | 8.5206 | 9.9998 | 8.5208 | 1.4792 | . 1 | . 9 | 9.1381 | 9.9959 | 9.1423 | 0.8577 | . 1 |
| 2.0 | 8.5428 | 9.9997 | 8.5431 | 1.4569 | 88.0 | 8.0 | 9.1436 | 9.9958 | 9.1478 | 0.8522 | 82.0 |
| . 1 | 8.5640 | 9.9997 | 8.5643 | 1.4357 | . 9 | . 1 | 9.1489 | 9.9956 | 9.1533 | 0.8467 | . 9 |
| . 2 | 8.5842 | 9.9997 | 8.5845 | 1.4155 | . 8 | . 2 | 9.1542 | 9.9955 | 9.1587 | 0.8413 | . 8 |
| . 3 | 8.6035 | 9.9996 | 8.6038 | 1.3962 | . 7 | . 3 | 9.1594 | 9.9954 | 9.1640 | 0.8360 | . 7 |
| . 4 | 8.6220 | 9.9996 | 8.6223 | 1.3777 | . 6 | . 4 | 9.1646 | 9.9953 | 9.1693 | 0.8307 | . 6 |
| . 5 | 8.6397 | 9.9996 | 8.6401 | 1.3599 | . 5 | . 5 | 9.1697 | 9.9952 | 9.1745 | 0.8255 | . 5 |
| . 6 | 8.6567 | 9.9996 | 8.6571 | 1.3429 | . 4 | . 6 | 9.1747 | 9.9951 | 9.1797 | 0.8203 | . 4 |
| . 7 | 8.6731 | 9.9995 | 8.6736 | 1.3264 | . 3 | . 7 | 9.1797 | 9.9950 | 9.1848 | 0.8152 | . 3 |
| . 8 | 8.6889 | 9.9995 | 8.6894 | 1.3106 | . 2 | . 8 | 9.1847 | 9.9949 | 9.1898 | 0.8102 | . 2 |
| . 9 | 8.7041 | 9.9994 | 8.7046 | 1.2954 | . 1 | . 9 | 9.1895 | 9.9947 | 9.1948 | 0.8052 | . 1 |
| 3.0 | 8.7188 | 9.9994 | 8.7194 | 1.2806 | 87.0 | 9.0 | 9.1943 | 9.9946 | 9.1997 | 0.8003 | 81.0 |
| . 1 | 8.7330 | 9.9994 | 8.7337 | 1.2663 | . 9 | . 1 | 9.1991 | 9.9945 | 9.2046 | 0.7954 | . 9 |
| . 2 | 8.7468 | 9.9993 | 8.7475 | 1.2525 | 8 | . 2 | 9.2038 | 9.9944 | 9.2094 | 0.7906 | . 8 |
| . 3 | 8.7802 | 9.9993 | 8.7609 | 1.2391 | . 7 | . 3 | 9.2085 | 9.9943 | 9.2142 | 0.7858 | .7 |
| . 4 | 8.7731 | 9.9992 | 8.7739 | 1.2261 | . 6 | . 4 | 9.2131 | 9.9941 | 9.2189 | 0.7811 | . 6 |
| . 5 | 8.7857 | 9.9992 | 8.7865 | 1.2135 | . 5 | . 5 | 9.2176 | 9.9940 | 9.2236 | 0.7764 | . 5 |
| . 6 | 8.7979 | 9.9991 | 8.7988 | 1.2012 | . 4 | . 6 | 9.2221 | 9.9939 | 9.2282 | 0.7718 | . 4 |
| . 7 | 8.8098 | 9.9991 | 8.8107 | 1.1893 | . 3 | . 7 | 9.2266 | 9.9937 | 9.2328 | 0.7672 | . 3 |
| . 8 | 8.8213 | 9.9990 | 8.8223 | 1.1777 | . 2 | . 8 | 9.2310 | 9.9936 | 9.2374 | 0.7626 | . 2 |
| . 9 | 8.8326 | 9.9990 | 8.8336 | 1.1664 | . 1 | . 9 | 9.2353 | 9.9935 | 9.2419 | 0.7581 | . 1 |
| 4.0 | 8.8436 | 9.9989 | 8.8446 | 1.1554 | 86.0 | 10.0 | 9.2397 | 9.9934 | 9.2463 | 0.7537 | 30.0 |
| . 1 | 8.8543 | 9.9989 | 8.8554 | 1.1446 | . 9 | . 1 | 9.2439 | 9.9932 | 9.2507 | 0.7493 | . 9 |
| . 2 | 8.8647 | 9.9988 | 8.8659 | 1.1341 | . 8 | . 2 | 9.2482 | 9.9931 | 9.2551 | 0.7449 | . 8 |
| . 3 | 8.8749 | 9.9988 | 8.8762 | 1.1238 | . 7 | . 3 | 9.2524 | 9.9929 | 9.2594 | 0.7406 | .7 |
| . 4 | 8.8849 | 9.9987 | 8.8862 | 1.1138 | . 6 | . 4 | 9.2565 | 9.9928 | 9.2637 | 0.7363 | . 6 |
| . 5 | 8.8946 | 9.9987 | 8.8960 | 1.1040 | . 5 | . 5 | 9.2606 | 9.9927 | 9.2680 | 0.7320 | . 5 |
| . 6 | 8.9042 | 9.9986 | 8.9056 | 1.0944 | 4 | . 6 | 9.2647 | 9.9925 | 9.2722 | 0.7278 | . 4 |
| . 7 | 8.9135 | 9.9985 | 8.9150 | 1.0850 | . 3 | . 7 | 9.2687 | 9.9924 | 9.2764 | 0.7236 | .3 |
| . 8 | 8.9226 | 9.9985 | 8.9241 | 1.0759 | . 2 | . 8 | 9.2727 | 9.9922 | 9.2805 | 0.7195 | . 2 |
| . 9 | 8.9315 | 9.9984 | 8.9331 | 1.0669 | . 1 | . 9 | 9.2767 | 9.9921 | 9.2846 | 0.7154 | . 1 |
| 5.0 | 8.9403 | 9.9983 | 8.9420 | 1.0580 | 05.0 | 11.0 | 9.2806 | 9.9919 | 9.2887 | 0.7113 | 79.0 |
| . 1 | 8.9489 | 9.9983 | 8.9506 | 1.0494 | . 9 | . 1 | 9.2845 | 9.9918 | 9.2927 | 0.7073 | . 9 |
| . 2 | 8.9573 | 9.9982 | 8.9591 | 1.0409 | . 8 | . 2 | 9.2883 | 9.9916 | 9.2967 | 0.7033 | . 8 |
| . 3 | 8.9655 | 9.9981 | 8.9674 | 1.0326 | . 7 | . 3 | 9.2921 | 9.9915 | 9.3006 | 0.6994 | . 7 |
| . 4 | 8.9736 | 9.9981 | 8.9756 | 1.0244 | . 6 | . 4 | 9.2959 | 9.9913 | 9.3046 | 0.6954 | . 6 |
| . 5 | 8.9816 | 9.9980 | 8.9836 | 1.0164 | . 5 | . 5 | 9.2997 | 9.9912 | 9.3085 | 0.6915 | . 5 |
| . 6 | 8.9894 | 9.9979 | 8.9915 | 1.0085 | . 4 | . 6 | 9.3034 | 9.9910 | 9.3123 | 0.6877 | . 4 |
| . 7 | 8.9970 | 9.9978 | 8.9992 | 1.0008 | . 3 | . 7 | 9.3070 | 9.9909 | 9.3162 | 0.6838 | . 3 |
| . 8 | 9.0046 | 9.9978 | 9.0068 | 0.9932 | . 2 | . 8 | 9.3107 | 9.9907 | 9.3200 | 0.6800 | . 2 |
| . 9 | 9.0120 | 9.9977 | 9.0143 | 0.9857 | . 1 | . 9 | 9.3143 | 9.9906 | 9.3237 | 0.6763 | .1 |
| 6.0 | 9.0192 | 9.9976 | 9.0216 | 0.9784 | 84.0 | 12.0 | 9.3179 | 9.9904 | 9.3275 | 0.6725 | 78.0 |
|  | L cos | Lsin | L col | Lton | dea |  | L cos | $L \sin$ | L cot | Ltan | dea |

Logarithms of trigonometric functions

## for decimal fractions of a degree continued

| deg | 4 sin | L cos \| | 4 Itan | 4 cot |  | deg | 1 sin | $L \cos$ | 1 ton | L col |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 12.0 | 9.3179 | 9.9904 | 0.3275 | 0.6725 | 7 | 180 | 0.4000 | 0.9782 |  |  |  |
| . 1 |  |  | 9.32 |  | 78.0 | . | 9.49 | 9.978 |  | 0.4882 | 72.0 |
| . |  | 9.9902 | 9.3312 | 0.8688 | . 9 | . 1 | 9.4923 | 9.9780 | 9.5143 | 0.4857 | . 9 |
|  | 9.3250 | 9.9901 | 9.3349 | 0.6651 | . 8 | . 2 | 9.4946 | 9.9777 | 9.5169 | 0.4831 | . 8 |
| . 3 | 9.3284 | 9.9899 | 9.3385 | 0.6615 | . 7 | . 3 | 9.4969 | 9.9775 | 9.5195 | 0.4805 | . 7 |
| . 4 | 9.3319 | 9.9897 | 9.3422 | 0.6578 | . 6 | . 4 | 9.4992 | 9.9772 | 9.5220 | 0.4780 | . 6 |
| . 5 | 9.3353 | 9.9896 | 9.3458 | 0.6542 | . 5 | . 5 | 9.5015 | 9.9770 | 9.5245 | 0.4755 | . 5 |
| . 6 | 9.3387 | 9.9894 | - 9.3493 | 0.6507 | . 4 | . 6 | 9.5037 | 9.9767 | 9.5270 | 0.4730 | . 4 |
| . 7 | 9.3421 | 9.9892 | 9.3529 | 0.6471 | .3 | . 7 | 9.5060 | 9.9764 | 9.5295 | 0.4705 | 3 |
| . 8 | 9.3455 | 9.9891 | 9.3564 | 0.6436 | . 2 | . 8 | 9.5082 | 9.9762 | 9.5320 | 0.4680 | . 2 |
| . 9 | 9.3488 | 9.9889 | 9.3599 | 0.6401 | . 1 | . 9 | 9.5104 | 9.9759 | 9.5345 | 0.4655 | . 1 |
| 13.0 | 9.3521 | 9.9887 | 9.3634 | 0.6366 | 77.0 | 19.0 | 9.5126 | 9.9757 | 9.5370 | 0.4630 | 71.0 |
| . 1 | 9.3554 | 9.9885 | 9.3668 | 0.6332 | . 9 | . 1 | 9.5148 | 9.9754 | 9.5394 | 0.4606 | . 9 |
| . 2 | 9.3586 | 9.9884 | 9.3702 | 0.6298 | 8 | . 2 | 9.5170 | 9.9751 | 9.5419 | 0.4581 | . 8 |
| 3 | 9.3618 | 9.9882 | 9.3736 | 0.6264 | . 7 | . 3 | 9.5192 | 9.9749 | 9.5443 | 0.4557 | 7 |
| . 4 | 93650 | 9.9880 | 9.3770 | 0.6230 | . 6 | . 4 | 9.5213 | 9.9746 | 9.5467 | 0.4533 | 6 |
| . 5 | 9.3682 | 9.9878 | 9.3804 | 0.6196 | . 5 | . 5 | 9.5235 | 9.9743 | 9.5491 | 0.4509 | . 5 |
| 6 | 9.3713 | 9.9876 | 9.3837 | 0.6163 | . 4 | . 6 | 9.5256 | 9.9741 | 9.5516 | 0.4484 | . 4 |
| . 7 | 9.3745 | 9.9875 | 9.3870 | 0.6130 | . 3 | . 7 | 9.5278 | 9.9738 | 9.5539 | 0.4461 | .3 |
| 8 | 9.3775 | 9.9873 | 9.3903 | 0.6097 | 2 | . 8 | 9.5299 | 9.9735 | 9.5563 | 0.4437 | . 2 |
| . 9 | 9.3806 | 9.9871 | 9.3935 | 0.6065 | .1 | . 9 | 9.5320 | 9.9733 | 9.5587 | 0.4413 | . 1 |
| 14.0 | 9.3037 | 9.9669 | 9.3968 | 0.0032 | 76.0 | 20.0 | 9.5341 | 9.9730 | 9.5611 | 0.4389 | 70.0 |
| . 1 | 9.3667 | 9.9867 | 9.4000 | 0.6000 | . 9 | . 1 | 9.5361 | 9.9727 | 95634 | 0.4366 | . 9 |
| . 2 | 9.3897 | 9.9865 | 9.4032 | 0.5968 | . 8 | . 2 | 9.5382 | 9.9724 | 9.5658 | 0.4342 | . 8 |
| 3 | 9.3927 | 9.9863 | 9.4064 | 0.5936 | . 7 | . 3 | 9.5402 | 9.9722 | 9.5481 | 0.4319 | .7 |
| . 4 | 9.3957 | 9.9861 | 9.4095 | 0.5905 | . 6 | . 4 | 9.5423 | 9.9719 | 9.5704 | 0.4296 | . 6 |
| . 5 | 9.3986 | 9.9859 | 9.4127 | 0.5873 | . 5 | . 5 | 9.5443 | 9.9716 | 9.5727 | 0.4273 | . 5 |
| . 6 | 9.4015 | 9.9857 | 9.4158 | 0.5842 | . 4 | . 6 | 9.5463 | 9.9713 | 9.5750 | 0.4250 | . 4 |
| . 7 | 9.4044 | 9.9855 | 9.4189 | 0.5811 | . 3 | . 7 | 9.5484 | 9.9710 | 9.5773 | 0.4227 | . 3 |
| . 8 | 9.4073 | 9.9853 | 9.4220 | 0.5780 | . 2 | . 8 | 9.5504 | 9.9707 | 9.5796 | 0.4204 | . 2 |
| . 9 | 9.4102 | 9.9851 | 9.4250 | 0.5750 | . 1 | . 9 | 9.5523 | 9.9704 | 9.5819 | 0.4181 | . 1 |
| 13.0 | 9.4130 | 9.9849 | 9.4281 | 0.5719 | 75.0 | 21.0 | 9.5543 | 9.9702 | 9.5842 | 0.4158 | 69.0 |
| . 1 | 9.4158 | 9.9847 | 9.4311 | 0.5689 | . 9 | . 1 | 9.5563 | 9.9699 | 9.5864 | 0.4136 | . 9 |
| .2 | 9.4186 | 9.9845 | 9.4341 | 0.5659 | . 8 | . 2 | 9.5583 | 9.9696 | 9.5897 | 0.4113 | . 8 |
| . 3 | 9.4214 | 9.9843 | 9.4371 | 0.5629 | . 7 | .3 | 9.5602 | 9.9693 | 9.5909 | 0.4091 | . 7 |
| . 4 | 9.4242 | 9.9841 | 9.4400 | 0.5600 | .6 | . 4 | 9.5621 | 9.9690 | 9.5932 | 0.4068 | . 6 |
| . 5 | 9.4269 | 9.9839 | 9.4430 | 0.5570 | . 5 | . 5 | 9.5641 | 9.9687 | 9.5954 | 0.4046 | . 5 |
| . 6 | 9.4296 | 9.9837 | 9.4459 | 0.5541 | 4 | . 6 | 9.5660 | 9.9604 | 9.5976 | 0.4024 | . 4 |
| 7 | 9.4323 | 9.9835 | 9.4488 | 0.5512 | . 3 | 7 | 9.5679 | 9.9681 | 9.5998 | 0.4002 | .3 |
| 8 | 9.4350 | 9.9833 | 9.4517 | 0.5483 | .2 | . 8 | 9.5698 | 9.9678 | 9.6020 | 0.3980 | 2 |
| . 9 | 9.4377 | 9.9831 | 9.4546 | 0.5454 | . 1 | . 9 | 9.5717 | 9.9675 | 9.6042 | 0.3958 | .1 |
| 16.0 | 9.4403 | 9.9828 | 9.4575 | 0.5425 | 74.0 | 22.0 | 9.5736 | 9.9672 | 9.6064 | 0.3936 | 68.0 |
| . 1 | 9.4430 | 9.9826 | 9.4603 | 0.5397 | . 9 | . 1 | 9.5754 | 9.9669 | 9.6086 | 0.3914 | . 9 |
| .2 | 9.4456 | 9.9824 | 9.4632 | 0.5368 | . 8 | . 2 | 9.5773 | 9.9666 | 9.6108 | 0.3892 | . 8 |
| . 3 | 9.4482 | 9.9822 | 9.4660 | 0.5340 | . 7 | . 3 | 9.5792 | 9.9662 | 9.6129 | 0.3871 | 7 |
| .4 | 9.4508 | 9.9820 | 9.4688 | 0.5312 | . 6 | . 4 | 9.5810 | 9.9659 | 9.6151 | 0.3849 | . 6 |
| . 5 | $9.4 \leqslant 33$ | 9.9817 | 9.4718 | 0.5284 | . 5 | . 5 | 9.5828 | 9.9656 | 9.6172 | 0.3828 | . 5 |
| . 6 | 9.4559 | 9.9815 | 9.4744 | 0.5256 | .4 | . 6 | 9.5847 | 9.9653 | 9.6194 | 0.3806 | . 4 |
| . 7 | 9.4584 | 9.9813 | 9.4771 | 0.5229 | . 3 | . 7 | 9.5865 | 9.9650 | 9.6215 | 0.3785 | . 3 |
| . 8 | 9.4609 | 9.9811 | 9.4799 | 0.5201 | . 2 | . 8 | 9.5883 | 9.9647 | 9.6236 | 0.3764 | . 2 |
| . 9 | 9.4634 | 9.9808 | 9.4826 | 0.5174 | . 1 | . 9 | 9.5901 | 9.9643 | 9.6257 | 0.3743 | . 1 |
| 17.0 | 9.4659 | 9.9806 | 9.4853 | 0.5147 | 73.0 | 23.0 | 9.5919 | 9.9640 | 9.6279 | 0.3721 | 67.0 |
| . 1 | 9.4684 | 9.9804 | 9.4880 | 0.5120 | . 9 | . 1 | 9.5937 | 9.9637 | 9.6300 | 0.3700 | . 9 |
| . 2 | 9.4709 | 9.9801 | 9.4907 | 0.5093 | . 8 | . 2 | 9.5954 | 9.9634 | 9.6321 | 0.3679 | 8 |
| . 3 | 9.4733 | 9.9799 | 9.4934 | 0.5066 | . 7 | . 3 | 9.5972 | 9.9631 | 9.6341 | 0.3659 | . 7 |
| 4 | 9.4757 | 9.9797 | 9.4961 | 0.5039 | . 6 | . 4 | 9.5990 | 9.9627 | 9.6362 | 0.3638 | 6 |
| . 5 | 9.4781 | 9.9794 | 9.4987 | 0.5013 | .5 | . 5 | 9.6007 | 9.9624 | 9.6383 | 0.3617 | . 5 |
| .6 | 9.4805 | 9.9792 | 9.5014 | 0.4986 | .4 | . 6 | 9.6024 | 9.9621 | 9.6404 | 0.3596 | . 4 |
| . 7 | 9.4829 | 9.9789 | 9.5040 | 0.4960 | 3 | 7 | 9.6042 | 9.9617 | 9.6424 | 0.3576 | . 3 |
| . 8 | 9.4853 | 9.9787 | 9.5066 | 0.4934 | . 2 | . 8 | 9.6059 | 9.9614 | 9.8445 | 0.3555 | . 2 |
| . 9 | 9.4876 | 9.9785 | 9.5092 | 0.4908 | . 1 | . 9 | 9.6076 | 9.9611 | 9.6465 | 0.3535 | . 1 |
| 18.0 | 9.4900 | 9.9782 | 9.5118 | 0.4882 | 72.0 | 24.0 | 9.6093 | 9.9607 | 9.6486 | 0.3514 | 66.0 |
|  | L cos | L sin | L cot | Lion | deg |  | L cos | L sin | L cot | L fon | deg |

Logarithms of trigonometric functions
for decimal fractions of a degree continued

| deg | $L \sin$ | L cos | L Ian | L cot |  | dog | $L \sin$ | L cos | Ltan | 1 col |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 24.0 | 9.6093 | 9.9607 | 9.6486 | 0.3514 | 66.0 | 30.0 | 9.6990 | 9.9375 | 9.7614 | 0.2386 | 60.0 |
| . 1 | 9.6110 | 9.9604 | 9.6506 | 0.3494 | . 9 | . | 9.7003 | 9.9371 | 9.7632 | 0.2368 | . 9 |
| . 2 | 9.6127 | 9.9601 | 9.6527 | 0.3473 | . 8 | 2 | 9.7016 | 9.9367 | 9.7649 | 0.2351 | 8 |
| . 3 | 9.6144 | 9.9597 | 9.6547 | 0.3453 | . 7 | . 3 | 9.7029 | 9.9362 | 9.7667 | 0.2333 | . 7 |
| . 4 | 9.6161 | 9.9594 | 9.6567 | 0.3433 | . 6 | . 4 | 9.7042 | 9.9358 | 9.7684 | 0.2316 | . 6 |
| . 5 | 9.6177 | 9.9590 | 9.6587 | 0.3413 | . 5 | . 5 | 9.7055 | 9.9353 | 9.7701 | 0.2299 | . 5 |
| . 6 | 9.6194 | 9.9587 | 9.6607 | 0.3393 | . 4 | . 6 | 9.7068 | 9.9349 | 9.7719 | 0.2281 | . 4 |
| . 7 | 9.6210 | 9.9583 | 9.6627 | 0.3373 | 3 | 7 | 9.7080 | 9.9344 | 9.7736 | 0.2264 | 3 |
| . 8 | 9.6227 | 9.9580 | 9.6647 | 0.3353 | . 2 | . 8 | 9.7093 | 9.9340 | 9.7753 | 0.2247 | . 2 |
| . 9 | 9.6243 | 9.9576 | 9.6667 | 0.3333 | . 1 | . 9 | 9.7106 | 9.9335 | 9.7771 | 0.2229 | . 1 |
| 25.0 | 9.6259 | 9.9573 | 9.6687 | 0.3313 | 65.0 | 31.0 | 9.7118 | 9.9331 | 9.7788 | 0.2212 | 59.0 |
| . 1 | 9.6276 | 9.9569 | 9.6706 | 0.3294 | . 9 | . 1 | 9.7131 | 9.9326 | 9.7805 | 0.2195 | . 9 |
| . 2 | 9.6292 | 9.9566 | 9.6726 | 0.3274 | . 8 | . 2 | 9.7144 | 9.9322 | 9.7822 | 0.2178 | . 8 |
| . 3 | 9.6338 | 9.9562 | 9.6746 | 0.3254 | . 7 | . 3 | 9.7156 | 9.9317 | 9.7839 | 0.2161 | . 7 |
| 4 | 9.6324 | 9.9558 | 9.6765 | 0.3235 | . 6 | . 4 | 9.7168 | 9.9312 | 9.7856 | 0.2144 | . 6 |
| . 5 | 9.6340 | 9.9555 | 9.6785 | 0.3215 | . 5 | . 5 | 9.7181 | 9.9308 | 9.7873 | 0.2127 | . 5 |
| . 6 | 9.6356 | 9.9551 | 9.6804 | 0.3196 | . 4 | . 6 | 9.7193 | 9.9303 | 9.789 | 0.2110 | 4 |
| 7 | 9.6371 | 9.9548 | 9.6824 | 0.3176 | . 3 | . 7 | 9.7205 | 9.9298 | 9.7907 | 0.2093 | . 3 |
| 8 | 9.6387 | 9.9544 | 9.8843 | 0.3157 | . 2 | . 8 | 9.7218 | 9.9294 | 9.7924 | 0.2076 | 2 |
| . 9 | 9.6403 | 9.9540 | 9.6863 | 0.3137 | . 1 | . 9 | 9.7230 | 9.9289 | 9.7941 | 0.2059 | . 1 |
| 26.0 | 9.6418 | 9.9537 | 9.6882 | 0.3118 | 64.0 | 32.0 | 9.7242 | 9.9284 | 9.7958 | 0.2042 | 58.0 |
| . 1 | 9.6434 | 9.9533 | 9.8901 | 0.3099 | . 9 | . 1 | 9.7254 | 9.9279 | 9.7975 | 0.2025 |  |
| 2 | 9.6449 | 9.9529 | 9.6920 | 0.3080 | . 8 | . 2 | 9.7266 | 9.9275 | 9.7992 | 0.2008 | 8 |
| . 3 | 9.6465 | 9.9525 | 9.6939 | 0.3061 | . 7 | . 3 | 9.7278 | 9.9270 | 9.8008 | 0.1992 | . 7 |
| . 4 | 9.6480 | 9.9522 | 9.6958 | 0.3042 | . 6 | 4 | 9.7290 | 9.9265 | 9.8025 | 0.1975 | . 6 |
| . 5 | 9.6495 | 9.9518 | 9.6977 | 0.3023 | . 5 | . 5 | 9.7302 | 9.9260 | 9.8042 | 0.1958 | . 5 |
| . 6 | 9.6510 | 9.9514 | 9.6996 | 0.3004 | 4 | . 6 | 9.7314 | 9.9255 | 9.8059 | 0.1941 | 4 |
| 7 | 9.6526 | 9.9510 | 9.7015 | 0.2985 | . 3 | . 7 | 9.7326 | 9.9251 | 9.8075 | 0.1925 | . 3 |
| . 8 | 9.6541 | 9.9506 | 9.7034 | 0.2966 | . 2 | . 8 | 9.7338 | 9.9246 | 9.8092 | 0.1908 | . 2 |
| . 9 | 9.6556 | 9.9503 | 9.7053 | 0.2947 | . 1 | . 9 | 9.7349 | 9.9241 | 9.8109 | 0.1891 | . 1 |
| 27.0 | 9.6570 | 9.9499 | 9.7072 | 0.2928 | 63.0 | 33.0 | 9.7361 | 9.9236 | 9.8125 | 0.1875 | 57.0 |
| . 1 | 9.6585 | 9.9495 | 9.7090 | 0.2910 | . 9 | . 1 | 9.7373 | 9.9231 | 9.8142 | 0.1858 | . 9 |
| 2 | 9.6600 | 9.9491 | 9.7109 | 0.2891 | . 8 | . 2 | 9.7384 | 9.9228 | 9.8158 | 0.1842 | 8 |
| . 3 | 9.6615 | 9.9487 | 9.7128 | 0.2872 | . 7 | . 3 | 9.7396 | 9.9221 | 9.8175 | 0.1825 | . 7 |
| . 4 | 9.6629 | 9.9483 | 9.7146 | 0.2854 | . 6 | . 4 | 9.7407 | 9.9216 | 9.8191 | 0.1809 | . 6 |
| . 5 | 9.8644 | 9.9479 | 9.7165 | 0.2835 | . 5 | . 5 | 9.7419 | 9.9211 | 98208 | 0.1792 | . 5 |
| . 6 | 9.6559 | 9.9475 | 9.7183 | 0.2817 | .4 | . 6 | 9.7430 | 9.9206 | 9.8224 | 0.1776 | . 4 |
| . 7 | 9.6673 | 9.9471 | 9.7202 | 0.2798 | 3 | . 7 | 9.7442 | 9.9201 | 9.8241 | 0.1759 | . 3 |
| 8 | 9.6687 | 9.9467 | 9.7220 | 0.2780 | . 2 | . 8 | 9.7453 | 9.9196 | 9.8257 | 0.1743 | . 2 |
| . 9 | 9.6702 | 9.9463 | 9.7238 | 0.2762 | , | . 9 | 9.7464 | 9.9191 | 9.8274 | 0.1726 | , |
| 28.0 | 9.6716 | 9.9459 | 9.7257 | 0.2743 | 62.0 | 34.0 | 9.7478 | 9.9186 | 9.8290 | 0.1710 | 56.0 |
| . 1 | 9.6730 | 9.9455 | 9.7275 | 0.2725 | . 9 | . 1 | 9.7487 | 9.9181 | 9.8306 | 0.1694 | . 9 |
| . 2 | 9.6744 | 9.9451 | 9.7293 | 0.2707 | 8 | . 2 | 9.7498 | 9.9175 | 9.8323 | 0.1677 | . 8 |
| .3 | 9.6759 | 9.9447 | 9.7311 | 0.2689 | . 7 | . 3 | 9.7509 | 9.9170 | 9.8339 | 0.1661 | . 7 |
| . 4 | 9.6773 | 9.9443 | 9.7330 | 0.2670 | . 6 | . 4 | 9.7520 | 9.9165 | 9.8355 | 0.1645 | 6 |
| . 5 | 9.6787 | 9.9439 | 9.7348 | 0.2652 | .5 | . 5 | 9.7531 | 9.9160 | 9.8371 | 0.1629 | . 5 |
| . 6 | 9.6801 | 9.9435 | 9.7366 | 0.2634 | 4 | . 6 | 9.7542 | 9.9155 | 9.8388 | 0.1612 | . 4 |
| . 7 | 9.6814 | 9.9431 | 9.7384 | 0.2616 | . 3 | . 7 | 9.7553 | 9.9149 | 9.8404 | 0.159 | . 3 |
| 8 | 9.6828 | 9.9427 | 9.7402 | 0.2598 | . 2 | . 8 | 9.7564 | 9.9144 | 9.8420 | 0.1580 | . 2 |
| . 9 | 9.6842 | 9.9422 | 9.7420 | 0.2580 | . 1 | . 9 | 9.7575 | 9.9139 | 9.8436 | 0.1564 | . 1 |
| 29.0 | 9.6856 | 9.9418 | 9.7438 | 0.2562 | 61.0 | 35.0 | 9.7586 | 9.9134 | 9.8452 | 0.1548 | 55.0 |
| . 1 | 9.6869 | 9.9414 | 9.7455 | 0.2545 | . 9 | . 1 | 9.7597 | 9.9128 | 9.8468 | 0.1532 | . 9 |
| . 2 | 9.6883 | 9.9410 | 9.7473 | 0.2527 | 8 | . 2 | 9.7807 | 9.9123 | 9.8484 | 0.1516 | . 8 |
| 3 | 9.6896 | 9.9406 | 9.7491 | 0.2509 | . 7 | . 3 | 9.7618 | 9.9118 | 9.8501 | 0.1499 | . 7 |
| . 4 | 9.6910 | 9.9401 | 9.7509 | 0.2491 | . 6 | . 4 | 9.7629 | 9.9112 | 9.8517 | 0.1483 | . 6 |
| . 5 | 9.6923 | 9.9397 | 9.7526 | 0.2474 | . 5 | . 5 | 9.7640 | 9.9107 | 9.8533 | 0.1467 | . 5 |
| . 6 | 9.6937 | 9.9393 | 9.7544 | 0.2456 | 4 | . 6 | 9.7650 | 9.901 | 9.8549 | 0.1451 | 4 |
| 7 | 9.6950 | 9.9388 | 9.7562 | 0.2438 | .3 | . 7 | 9.7661 | 9.9006 | 9.8555 | 0.1435 | . 3 |
| . 8 | 9.6963 | 9.9384 | 9.7579 | 0.2421 | ${ }^{2}$ | 8 | 9.7671 | 9.9091 9.9085 | 9.8581 9.8597 | 0.1419 | . 2 |
| . 9 | 9.6977 | 9.9380 | 9.7597 | 0.2403 | . 1 | . 9 | 9.7682 | 9.9085 | 9.8597 | 0.1403 | . 1 |
| 30.0 | 9.6990 | 9.9375 | 9.7614 | 0.2386 | 60.0 | 36.0 | 9.7692 | 9.9080 | 9.8613 | 0.1387 | 54.0 |
|  | Leos | L sin | Leot | Lton | dog |  | Leos | L sin | L cot | Lian | do |

Logarithms of trigonometric functions

## for decimal fractions of a degree continued

| deg | $L \sin$ | 1 cos | $L$ tan | 1 cot |  | deg | $L \sin$ | 1 cos | LIan | 1 col |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 36.0 | 9.7692 | 9.9080 | 9.8613 | 0.1387 | 54.0 | 40.3 | 9.8125 | 9.8810 | 9.9315 | 0.0685 | 49.3 |
| . 1 | 9.7703 | 9.9074 | 9.8629 | 0.1371 | . 9 | . 6 | 9.8134 | 9.8804 | 9.9330 | 0.6670 | . 4 |
| . 2 | 9.7713 | 9.9069 | 9.8644 | 0.1356 | . 8 | . 7 | 9.8143 | 9.8797 | 9.9346 | . 0.0654 | . 3 |
| . 3 | 9.7723 | 9.9063 | 9.8860 | 0.1340 | 7 | . 8 | 9.8152 | 9.8791 | 9.9361 | 0.0639 | . 2 |
| . 4 | 9.7734 | 9.9057 | 9.8676 | 0.1324 | . 6 | . 9 | 9.8161 | 9.8784 | 9.9376 | 0.0624 | . 1 |
| . 5 | 9.7744 | 9.9052 | 9.8692 | 0.1308 | . 5 | 41.0 | 9.8169 | 9.8778 | 9.9392 | 0.0608 | 49.0 |
| 6 | 9.7754 | 9.9046 | 9.8708 | 0.1292 | . 4 | . 1 | 9.8178 | 9.8771 | 9.9407 | 0.0593 | . 9 |
| . 7 | 9.7764 | 9.9041 | 9.8724 | 0.1276 | . 3 | . 2 | 9.8187 | 9.8765 | 9.9422 | 0.0578 | . 8 |
| . 8 | 9.7774 | 9.9035 | 9.8740 | 0.1260 | . 2 | . 3 | 9.8195 | 9.8758 | 9.9438 | 0.0562 | 7 |
| . 9 | 9.7785 | 9.9029 | 9.8755 | 0.1245 | . 1 | . 4 | 9.8204 | 9.8751 | 9.9453 | 0.0547 | . 6 |
| 37.0 | 9.7795 | 9.9023 | 9.8771 | 0.1229 | 53.0 | . 5 | 9.8213 | 9.8745 | 9.9468 | 0.0532 | . 5 |
| . 1 | 9.7805 | 9.9018 | 9.8787 | 0.1213 | . 9 | . 6 | 9.8221 | 9.8738 | 9.9483 | 0.0517 | . 4 |
| 2 | 9.7815 | 9.9012 | 9.6303 | 0.1197 | . 8 | 7 | 9.8230 | 9.8731 | 9.9499 | 0.0501 | 3 |
| . 3 | 9.7825 | 9.9006 | 9.8818 | 0.1182 | . 7 | . 8 | 9.8238 | 9.8724 | 9.9514 | 0.0486 | 2 |
| .4 | 9.7835 | 9.9000 | 9.8834 | 0.1166 | . 6 | . 9 | 9.8247 | 9.8718 | 9.9529 | 0.0471 | . 1 |
| . 5 | 9.7844 | 9.8995 | 9.8850 | 0.1150 | . 5 | 42.0 | 9.8255 | 9.8711 | 9.9544 | 0.0456 | 48.0 |
| . 6 | 9.7854 | 9.8989 | 9.8865 | 0.1135 | . 4 | . 1 | 9.8264 | 9.8704 | 9.9560 | 0.0440 | . 9 |
| . 7 | 9.7864 | 9.8983 | 9.8881 | 0.1119 | . 3 | . 2 | 9.8272 | 9.8697 | 9.9575 | 0.0425 | . 8 |
| 8 | 9.7874 | 9.8977 | 9.8897 | 0.1103 | . 2 | 3 | 9.8280 | 9.8690 | 9.9590 | 0.0410 | 7 |
| . 9 | 9.7884 | 9.8971 | 9.8912 | 0.1088 | . 1 | . 4 | 9.8289 | 9.8683 | 9.9605 | 0.0395 | 6 |
| 38.0 | 9.7893 | 9.8965 | 9.8928 | 0.1072 | 52.0 | . 5 | 9.8297 | 9.8676 | 9.9621 | 0.0379 | . 5 |
| . 1 | 9.7903 | 9.8959 | 9.8944 | 0.1056 | . 9 | . 6 | 9.8305 | 9.8669 | 9.9636 | 0.0354 | . 4 |
| . 2 | 9.7913 | 9.8753 | 9.8959 | 0.1041 | . 8 | . 7 | 9.8313 | 9.8662 | 9.9651 | 0.0349 | . 3 |
| .3 | 9.7922 | 9.8947 | 9.8975 | 0.1025 | 7 | 8 | 9.8322 | 9.8655 | 9.9688 | 0.0334 | . 2 |
| . 4 | 9.7932 | 9.8941 | 9.8990 | 0.1010 | . 6 | . 9 | 9.8330 | 9.8648 | 9.9681 | 0.0319 | . 1 |
| . 5 | 9.7941 | 9.8935 | 9.9006 | 0.0994 | . 5 | 43.0 | 9.8338 | 9.8641 | 9.9697 | 0.0303 | 47.0 |
| . 6 | 9.7951 | 9.8929 | 9.9022 | 0.0978 | . 4 | . 1 | 9.8346 | 9.8634 | 9.9712 | 0.0288 | . 9 |
| . 7 | 9.7960 | 9.8923 | 9.9037 | 0.0963 | 3 | . 2 | 9.8354 | 9.8627 | 9.9727 | 0.0273 | . 8 |
| . 8 | 9.7970 | 9.8917 | 9.9053 | 0.0947 | 2 | . 3 | 9.8362 | 9.8620 | 9.9742 | 0.0258 | . 7 |
| . 9 | 9.7979 | 9.8911 | 9.9068 | 0.0932 | . 1 | . 4 | 9.8370 | 9.8613 | 9.9757 | 0.0243 | . 6 |
| 39.0 | 9.7989 | 9.8905 | 9.9084 | 0.0916 | 51.0 | . 5 | 9.8378 | 9.8606 | 9.9772 | 0.0228 | . 5 |
| . 1 | 9.7998 | 9.8899 | 9.9099 | 0.0901 | . 9 | . 6 | 9.8386 | 9.8598 | 9.9788 | 0.0212 | . 4 |
| 2 | 9.8007 | 9.8893 | 9.9115 | 0.0885 | . 8 | . 7 | 9.8394 | 9.8591 | 9.9803 | 0.0197 | . 3 |
| . 3 | 9.8017 | 9.8887 | 9.9130 | 0.0870 | . 7 | . 8 | 9.8402 | 9.8584 | 9.9818 | 0.0182 | 2 |
| . 4 | 9.8026 | 9.8880 | 9.9146 | 0.0854 | . 6 | . 9 | 9.8410 | 9.8577 | 9.9833 | 0.0167 | . 1 |
| . 5 | 9.8035 | 9.8874 | 9.9161 | 0.0839 | . 5 | 44.0 | 9.8418 | 9.8569 | 9.9848 | 0.0152 |  |
| .6 | 9.8044 | 9.8868 | 9.9176 | 0.0824 | .4 | . 1 | 9.8426 | 9.8562 | 9.9864 | 0.0136 | . 9 |
| 2 | 9.8053 | 9.8362 | 9.9192 | 0.0808 | . 3 | . 2 | 9.8433 | 9.8555 | 9.9879 | 0.0121 | . 8 |
| . 8 | 9.8063 | 9.8855 | 9.9207 | 0.0793 | . 2 | 3 | 9.8441 | 9.8547 | 9.9894 | 0.0106 | 7 |
| .9 | 9.8072 | 9.8849 | 9.9223 | 0.0777 | . 1 | . 4 | 9.8449 | 9.8540 | 9.9909 | 0.0091 | . 6 |
| 40.0 | 9.8081 | 9.8843 | 9.9238 | 0.0762 | 50.0 | . 5 | 9.8457 | 9.8532 | 9.9924 | 0.0076 | . 5 |
| . 1 | 9.8090 | 9.8836 | 9.9254 | 0.0746 | . 9 | . 6 | 9.8464 | 9.8525 | 9.9939 | 0.0061 | . 4 |
| . 2 | 9.8099 | 9.8830 | 9.9269 | 0.0731 | 8 | 7 | 9.8472 | 9.8517 | 9.9955 | 0.0045 | . 3 |
| . 3 | 9.8108 | 9.8823 | 9.9284 | 0.0716 | 7 | . 8 | 9.8480 | 9.8510 | 9.9970 | 0.0030 | 2 |
| . 4 | 9.8117 | 9.8817 | 9.9300 | 0.0700 | . 6 | . 9 | 9.8487 | 9.8502 | 9.9985 | 0.0015 | .1 |
| 40.5 | 9.8125 | 9.8810 | 9.9315 | 0.0685 | 49.5 | 45.0 | 9.8495 | 9.8495 | 0.0000 | 0.0000 | 48.0 |
|  | $1 . \cos$ | $L$ sin | L cot | Lian | deg |  | L ces | $1 . \sin$ | L col | 6 tan | des |

Nafural logarithms


Nafural logarithms of $10^{+n}$

| $\boldsymbol{n}$ | $\mathbf{1}$ | $\mathbf{2}$ | $\mathbf{3}$ | $\mathbf{4}$ | $\mathbf{5}$ | $\mathbf{5}$ | $\mathbf{6}$ | $\mathbf{7}$ | $\mathbf{7}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\log _{0} 10^{n}$ | 2.3026 | 4.6052 | 6.9078 | 9.2103 | 11.5129 | 13.8155 | 16.1181 | 18.4207 | $\mathbf{2 0 . 7 2 3 3}$ |

## MATHEMATICAL TABLES <br> 315

Nafural logarithms continued

|  |  |  |  |  |  |  |  |  |  |  | mean differences |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | 2 |  |  |  |  |  |  |  | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 5.5 | 1.7047 | 7068 | 7084 | 7102 | 7120 | 7138 | 7156 | 7174 | 7192 | 7210 | 2 | 4 | 5 | 7 | 9 | 11 | 13 | 1416 |
| 5.6 | 1.7228 | 7246 | 7263 | 7281 | 7299 | 7317 | 7334 | 7352 | 7370 | 7387 | 2 | 4 | 5 | 7 | 9 | 11 | 12 | 1416 |
| 5.7 | 1.7405 | 7422 | 7440 | 7457 | 7475 | 7492 | 7509 | 7527 | 7544 | 7561 | 2 | 3 | 5 | 7 | 9 | 10 | 12 | 1416 |
| 5.8 | 1.7579 | 7596 | 7613 | 7630 | 7647 | 7664 | 7681 | 7699 | 7716 | 7733 | 2 |  | 5 | 7 | 9 | 10 | 12 | 1415 |
| 5.9 | 1.7750 | 7766 | 7783 | 7800 | 7817 | 7834 | 7851 | 7867 | 7884 | 7901 | 2 | 3 | 5 | 7 | 8 | 10 | 12 | 1315 |
| 6.0 | 1.7918 | 7934 | 7951 | 7967 | 7984 | 8001 | 8017 | 8034 | 8050 | 8066 | 2 | , | 5 | 7 | 8 | 10 | 12 | 1315 |
| 6.1 | 1.8083 | 8099 | 8116 | 8132 | 8148 | 8165 | 8181 | 8197 | 8213 | 8229 | 2 | 3 | 5 | 6 | 8 | 10 | 11 | 1315 |
| 6.2 | 1.8245 | 8262 | 8278 | 8294 | 8310 | 8326 | 8342 | 8358 | 8374 | 8390 | 2 | 3 | 5 | 6 | 8 | 10 | 11 | 1314 |
| 6.3 | 1.8405 | 8421 | 8437 | 8453 | 8469 | 8485 | 8500 | 8516 | 8532 | 8547 | 2 | 3 | 5 | 6 | 8 | 9 | 11 | 1314 |
| 6.4 | 1.8563 | 8579 | 8594 | 8610 | 8625 | 8641 | 8656 | 8672 | 8687 | 8703 | 2 | 3 | 5 | 6 | 8 | 9 | 11 | 1214 |
| 6.5 | 1.8718 | 8733 | 8749 | 8764 | 8779 | 8795 | 8810 | 8825 | 8840 | 8856 | 2 | 3 | 5 | 6 | 8 | 9 | 11 | 1214 |
| 6.6 | 1.8871 | 8886 | 8901 | 8916 | 8931 | 8946 | 8961 | 8976 | 8991 | 9006 | 2 | 3 | 5 | 6 | 8 | 9 |  | 1214 |
| 6.7 | 1.9021 | 9036 | 9051 | 9066 | -9081 | 9095 | 9110 | 9125 | 9140 | 9155 | 1 | 3 | 4 | 6 | 7 | 9 | 10 | 1213 |
| 6.8 | 1.9169 | 9184 | 9199 | 9213 | 9228 | 9242 | 9257 | 9272 | 9286 | 9301 | 1 | 3 | 4 | 6 | 7 | 9 | 10 | $12 \quad 13$ |
| 6.9 | 1.9315 | 9330 | 934 | 9359 | 9373 | 9387 | 9402 | 9416 | 9430 | 9445 | i | 3 | 4 | 6 | 7 | 9 | 10 | $12 \quad 13$ |
| 7.0 | 1.9459 | 9473 | 9488 | 9502 | 9516 | 9530 | 9544 | 9559 | 9573 | 9587 |  | , | 4 | 6 | 7 | 9 | 10 | 1113 |
| 7.1 | 1.9801 | 9615 | 9629 | 9643 | 9657 | 9671 | 9685 | 9699 | 9713 | 9727 |  | 3 | 4 | 6 | 7 | 8 |  | 1113 |
| 7.2 | 1.9741 | 9755 | 9769 | 9782 | 9796 | 9810 | 9824 | 9838 | 9851 | 9865 |  | 3 | 4 | 6 | 7 | 8 | 10 | If 12 |
| 7.3 | 1.9879 | 9892 | 9906 | 9920 | 9933 | 9947 | 9961 | 9974 | 9988 | 2.0001 | 1 | 3 | 4 | 5 | 7 | 8 | 10 | 1112 |
| 7.4 | 2.0015 | 0028 | 0042 | 0055 | 0069 | 0082 | 0096 | 0109 | 0122 | 0136 | 1 | 3 | 4 | 5 | 7 | 8 | 9 | 1112 |
| 7.5 | 2.0149 | 0162 | 0176 | 0189 | 0202 | 0215 | 0229 | 0242 | 0255 | 0268 |  | 3 | 4 | S | 7 | 8 | 91 | 1112 |
| 7.6 | 2.0281 | 0285 | 0308 | 0321 | 0334 | 0347 | 0360 | 0373 | 0386 | 0399 | 1 | 3 | 4 | 5 | 7 | 8 | 91 | $10 \quad 12$ |
| 7.7 | 2.0412 | 0425 | 0438 | 0451 | 0464 | 0477 | 0490 | 0503 | 0516 | 0.528 | 1 | 3 | 4 | 5 | 6 | 8 | , | 1012 |
| 7.8 | 2.0541 | 0554 | 0567 | 0580 | 0592 | 0605 | 0818 | 0631 | 0643 | 0656 | 1 | 3 | 4 | 5 | 6 | 8 | 9 | 1011 |
| 7.9 | 2.0669 | 0681 | 0694 | 0707 | 0719 | 0732 | 0744 | 0757 | 0769 | 0782 | 1 | 3 | 4 | 5 | 6 | 8 | 9 | 1011 |
| 8.0 | 2.0794 | 0807 | 0819 | 0832 | 0844 | 0857 | 0869 | 0882 | 0894 | 0906 |  | 3 | 4 | 5 | 6 | 7 | 910 | 1011 |
| 8.1 | 2.0919 | 0931 | 0943 | 0956 | 0968 | 0980 | 0892 | 1005 | 1017 | 1029 | 1 | 2 | 4 | 5 | 6 | 7 |  | 1011 |
| 8.2 | 2.1041 | 1054 | 1066 | 1078 | 1090 | 1102 | 1114 | 1126 | 1138 | 1150 | 1 | 2 | 4 | 5 | 6 | 7 | 9 | 1011 |
| 8.3 | 2.1163 | 1175 | 1187 | 1199 | 1211 | 1223 | 1235 | 1247 | 1258 | 1270 | 1 | 2 | 4 | 5 | 6 | 7 | 8 | 1011 |
| 8.4 | 2.1282 | 1294 | 1306 | 1318 | 1330 | 1342 | 1353 | 1365 | 1377 | 1389 | 1 | 2 | 4 | 5 | 6 | 7 | 8 | 911 |
| E. ${ }^{\text {e }}$ | 2.1401 | 1412 | 1424 | 1436 | 1448 | 1459 | 1471 | 1483 | 1494 | 1506 |  | 2 | 4 | S | 6 | 7 |  |  |
| 8.6 | 2.1518 | 1529 | 1541 | 1552 | 1564 | 1576 | 1587 | 1599 | 1610 | 1622 | 1 | 2 | 3 | 5 | 6 | 7 | 8 | 910 |
| 8.7 | 2.1633 | 1645 | 1656 | 1688 | 1679 | 1691 | 1702 | 1713 | 1725 | 1736 | 1 | 2 | 3 | 5 | 6 | 7 | 8 | 910 |
| 8.8 | 2.1748 | 1759 | 1770 | 1782 | 1793 | 1804 | 1815 | 1827 | 1838 | 1849 | 1 | 2 | 3 | 5 | 6 | 7 | 8 | 910 |
| 8.9 | 2.1861 | 1872 | 1883 | 1894 | 1905 | 1917 | 1928 | 1939 | 1950 | 1961 | 1 | 2 | 3 | 4 | 6 | 7 | 8 | 910 |
| 9.0 | 2.1972 | 1983 | 1994 | 2006 | 2017 | 2028 | 2039 | 2050 | 2061 | 2072 | 1 | 2 | 3 | 4 | 6 | 7 | 8 | 910 |
| 9.1 | 2.2083 | 2094 | 2105 | 2116 | 2127 | 2138 | 2148 | 2159 | 2170 | 2181 | 1 | 2 | 3 | 4 | 5 | 7 | 8 | 910 |
| 9.2 | 2.2192 | 2203 | 2214 | 2225 | 2235 | 2246 | 2257 | 2268 | 2279 | 2289 | 1 | 2 | 3 | 4 | 5 | 6 | 8 | 910 |
| 9.3 | 2.2300 | 2311 | 2322 | 2332 | 2343 | 2354 | 2364 | 2375 | 2386 | 2396 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 910 |
| 9.4 | 2.2407 | 2418 | 2428 | 2439 | 2450 | 2460 | 2471 | 2481 | 2492 | 2502 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 810 |
| 9.5 | 2.2513 | 2523 | 2534 | 2544 | 2555 | 2565 | 2576 | 2586 | 2597 | 2607 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 9.6 | 2.2618 | 2628 | 2638 | 2649 | 2659 | 2670 | 2680 | 2690 | 2701 | 2711 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 9.7 | 2.2721 | 2732 | 2742 | 2752 | 2762 | 2773 | 2783 | 2793 | 2803 | 2814 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 9.8 | 2.2824 | 2834 | 2844 | 2854 | 2865 | 2875 | 2885 | 2895 | 2905 | 2915 | , | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 9.9 | 2.2925 | 2935 | 2946 | 2956 | 2966 | 2976 | 2986 | 2996 | 3006 | 3016 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 89 |
| 10.0 | 2.3026 |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |

Nafural logarithms of $10^{-n}$


Hyperbolic sines [sinh $\left.x=1 / 2\left(e^{x}-e^{-x}\right)\right]$

| X | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | * | 9 | $\begin{aligned} & \text { dvef } \\ & \text { diff } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | 0.0000 | 0.0100 | 0.0200 | 0.0300 | 0.0400 | 0.0500 | 0.0600 | 0.0701 | 0.0801 | 0.0901 | 100 |
| . 1 | 0.1002 | 0.1102 | 0.1203 | 0.1304 | 0.1405 | 0.1506 | 0.1607 | 0.1708 | 0.1810 | 0.1911 | 101 |
| . 2 | 0.2013 | 0.2115 | 0.2218 | 0.2320 | 0.2423 | 0.2526 | 0.2629 | 0.2733 | 0.2837 | 0.2941 | 103 |
| . 3 | 0.3045 | 0.3150 | 0.3255 | 0.3360 | 0.3466 | 0.3572 | 0.3678 | 0.3785 | 0.3892 | 0.4000 | 106 |
| .4 | 0.4108 | 0.4216 | 0.4325 | 0.4434 | 0.4543 | 0.4653 | 0.4764 | 0.4875 | 0.4986 | 0.5098 | 110 |
| 0.5 | 0.5211 | 0.5324 | 0.5438 | 0.5552 | 0.5666 | 0.5782 | 0.5897 | 0.6014 | 0.6131 | 0.6248 | 116 |
| . 6 | 0.6367 | 0.6485 | 0.6605 | 0.6725 | 0.6846 | 0.6967 | 0.7090 | 0.7213 | 0.7336 | 0.7461 | 122 |
| . 7 | 0.7586 | 0.7712 | 0.7838 | 0.7966 | 0.8094 | 0.8223 | 0.8353 | 0.8484 | 0.8615 | 0.8748 | 130 |
| 8 | 0.8881 | 0.9015 | 0.9150 | 0.9286 | 0.9423 | 0.9561 | 0.9700 | 0.9840 | 0.9981 | 1.012 | 138 |
| . 9 | 1.027 | 1.041 | 1.055 | 1.070 | 1.085 | 1.099 | 1.114 | 1.129 | 1.145 | 1.160 | 15 |
| 1.0 | 1.175 | 1.191 | 1.206 | 1.222 | 1.238 | 1.254 | 1.270 | 1.286 | 1.303 | 1.319 | 16 |
| . 1 | 1.336 | 1.352 | 1.369 | 1.388 | 1.403 | 1.421 | 1.438 | 1.456 | 1.474 | 1.491 | 17 |
| . 2 | 1.509 | 1.528 | 1.546 | 1.564 | 1.583 | 1.602 | 1.621 | 1.640 | 1.659 | 1.679 | 19 |
| . 3 | 1.698 | 1.718 | 1.738 | 1.758 | 1.779 | 1.799 | 1.820 | 1.841 | 1.862 | 1.883 | 21 |
| . 4 | 1.904 | 1.926 | 1.948 | 1.970 | 1.992 | 2.014 | 2.037 | 2.060 | 2.083 | 2.106 | 22 |
| 1.5 | 2.129 | 2.153 | 2.177 | 2.201 | 2.225 | 2.250 | 2.274 | 2.299 | 2.324 | 2.350 | 25 |
| . 6 | 2.376 | 2.401 | 2428 | 2.454 | 2.481 | 2.507 | 2.535 | 2.562 | 2.590 | 2.617 | 27 |
| . 7 | 2.646 | 2.674 | 2.703 | 2.732 | 2.761 | 2.790 | 2.820 | 2.850 | 2.881 | 2.911 | 30 |
| . 8 | 2.942 | 2.973 | 3.005 | 3.037 | 3.069 | 3.101 | 3.134 | 3.167 | 3.200 | 3.234 | 33 |
| . 9 | 3.268 | 3.303 | 3.337 | 3.372 | 3.408 | 3.443 | 3.479 | 3.516 | 3.552 | 3.589 | 36 |
| 2.0 | 3.627 | 3.665 | 3.703 | 3.741 | 3.780 | 3.820 | 3.859 | 3.899 | 3.940 | 3.981 | 39 |
| . 1 | 4.022 | 4.064 | 4.106 | 4.148 | 4.191 | 4.234 | 4.278 | 4.322 | 4.367 | 4.412 | 44 |
| .2 | 4.457 | 4.503 | 4.549 | 4.596 | 4.643 | 4.691 | 4.739 | 4.788 | 4.837 | 4.887 | 48 |
| 3 | 4.937 | 4.988 | 5.039 | 5.090 | 5.142 | 5.195 | 5.248 | 5.302 | 5.356 | 5.411 | 53 |
| . 4 | 5.466 | 5.522 | 5.578 | 5.635 | 5.693 | 5.751 | 5.810 | 5.869 | 5.929 | 5.989 | 58 |
| 2.5 | 6.050 | 6.112 | 6.174 | 6.237 | 6.300 | 6.365 | 6.429 | 6.495 | 6.561 | 6.627 | 64 |
| . 6 | 6.695 | 6.763 | 6.831 | 6.901 | 6.971 | 7.042 | 7.113 | 7.185 | 7.258 | 7.332 | 71 |
| . 7 | 7.408 | 7.481 | 7.557 | 7.634 | 7.711 | 7.789 | 7.868 | 7.948 | 8.028 | 8.110 | 79 |
| . 8 | 8.192 | 8.275 | 8.359 | 8.443 | 8.529 | 8.615 | 8.702 | 8.790 | 8.879 | 8.969 | 87 |
| . 9 | 9.060 | 9.151 | 9.244 | 9.337 | 9.431 | 9.527 | 9.623 | 9.720 | 9.819 | 9.918 | 96 |
| 3.0 | 10.02 | 10.12 | 10.22 | 10.32 | 10.43 | 10.53 | 10.64 | 10.75 | 10.86 | 10.97 | 11 |
| . 1 | 11.08 | 11.19 | 11.30 | 11.42 | 11.53 | 11.65 | 11.76 | 11.88 | 12.00 | 12.12 | 12 |
| . 2 | 12.25 | 12.37 | 12.49 | 12.62 | 12.75 | 12.88 | 13.01 | 13.14 | 13.27 | 13.40 | 13 |
| . 3 | 13.54 | 13.67 | 13.81 | 13.95 | 14.09 | 14.23 | 14.38 | 14.52 | 14.67 | 14.82 | 14 |
| 4 | 14.97 | 15.12 | 15.27 | 15.42 | 15.58 | 15.73 | 15.89 | 16.05 | 16.21 | 16.38 | 16 |
| 3.3 | 16.54 | 16.71 | 16.88 | 17.05 | 17.22 | 17.39 | 17.57 | 17.74 | 17.92 | 18.10 | 17 |
| .6 | 18.29 | 18.47 | 18.66 | 18.84 | 19.03 | 19.22 | 19.42 | 19.61 | 19.81 | 20.01 | 19 |
| 7 | 20.21 | 20.41 | 20.62 | 20.83 | 21.04 | 21.25 | 21.46 | 21.68 | 21.90 | 22.12 | 21 |
| . 8 | 22.34 | 22.56 | 22.79 | 23.02 | 23.25 | 23.49 | 23.72 | 23.96 | 24.20 | 24.45 | 24 |
| . 9 | 24.69 | 24.94 | 25.19 | 25.44 | 25.70 | 25.96 | 26.22 | 26.48 | 26.75 | 27.02 | 26 |
| 4.0 | 27.29 | 27.56 | 27.84 | 28.12 | 28.40 | 28.69 | 28.98 | 29.27 | 29.56 | 29.86 | 29 |
| . 1 | 30.16 | 30.47 | 30.77 | 31.08 | 31.39 | 31.71 | 32.03 | 32.35 | 32.68 | 33.00 | 32 |
| .2 | 33.34 | 33.67 | 34,01 | 34.35 | 34.70 | 35.05 | 35.40 | 35.75 | 36.11 | 36.48 | 35 |
| .3 | 36.84 | 37.21 | 37.59 | 37.97 | 38.35 | 38.73 | 39.12 | 39.52 | 39.91 | 40.31 | 39 |
| .4 | 40.72 | 41.13 | 41.54 | 41.96 | 42.38 | 42.81 | 43.24 | 43.67 | 44.11 | 44.56 | 43 |
| 4.5 | 45.00 | 45.46 | 45.91 | 46.37 | 46.84 | 47.31 | 47.79 | 48.27 | 48.75 | 49.24 | 47 |
| . 6 | 49.74 | 50.24 | 50.74 | 51.25 | 51.77 | 52.29 | 52.81 | 53.34 | 53.88 | 54.42 | 52 |
| . 7 | 54.97 | 55.52 | 56.08 | 56.64 | 57.21 | 57.79 | 58.37 | 58.96 | 59.55 | 60.15 | 58 |
| . 8 | 60.75 | 61.36 | 61.98 | 62.60 | 63.23 | 63.87 | 64.51 | 65.16 | 65.81 | 66.47 | 64 |
| . 9 | 67.14 | 67.82 | 68.50 | 69.19 | 69.88 | 70.58 | 71.29 | 72.01 | 72.73 | 73.46 | 71 |
| 3.0 | 74.20 |  |  |  |  |  |  |  |  |  |  |

$H x>5, \sinh x=1 / 2 \operatorname{lof}$ and $\log _{n} \sinh x=10.43431 x+0.690-1$, correct to four signifizant figures.

Hyperbolic cosines [cosh $x=1 / 2\left(e^{x}+e^{-x}\right)$ ]

| $\mathbf{x}$ | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | $\begin{aligned} & \text { dro } \\ & \text { diff } \end{aligned}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | 1.000 | 1.000 | 1.000 | 1.000 | 1.001 | 1.001 | 1.002 | 1.002 | 1.003 | 1.004 | 1 |
| . 1 | 1.005 | 1.006 | 1.007 | 1.008 | 1.010 | 1.011 | 1.013 | 1.014 | 1.016 | 1.018 | 2 |
| . 2 | 1.020 | 1.022 | 1.024 | 1.027 | 1.029 | 1.031 | 1.034 | 1.037 | 1.039 | 1.042 | 3 |
| . 3 | 1.045 | 1.048 | 1.052 | 1.055 | 1.058 | 1.062 | 1.066 | 1.069 | 1.073 | 1.077 | 4 |
| 4 | - 1.081 | 1.085 | 1.090 | 1.094 | 1.098 | 1.103 | 1.108 | 1.112 | 1.117 | 1.122 | 5 |
| 0.5 | 1.128 | 1.133 | 1.138 | 1.144 | 1.149 | 1.155 | 1.161 | 1.167 | 1.173 | 1.179 | 6 |
| . 6 | 1.185 | 1.192 | 1.198 | 1.205 | 1.212 | 1.219 | 1.226 | 1.233 | 1.240 | 1.248 | 7 |
| . 7 | 1.255 | 1.263 | 1.271 | 1.278 | 1.287 | 1.295 | 1.303 | 1.311 | 1.320 | 1.329 | 8 |
| . 8 | 1.337 | 1.346 | 1.355 | 1.365 | 1.374 | 1.384 | 1.393 | 1.403 | 1.413 | 1.423 | 10 |
| . 9 | 1.433 | 1.443 | 1.454 | 1.465 | 1.475 | 1.486 | 1.497 | 1.509 | 1.520 | 1.531 | 11 |
| 2.0 | 1.543 | 1.555 | 1.567 | 1.579 | 1.591 | 1.604 | 1.616 | 1.629 | 1,642 | 1.655 | 13 |
| .1 | 1.669 | 1.682 | 1.696 | 1.709 | 1.723 | 1.737 | 1.752 | 1.766 | 1.781 | 1.796 | 14 |
| . 2 | 1.811 | 1.826 | 1.841 | 1.857 | 1.872 | 1.888 | 1.905 | 1.921 | 1.937 | 1.954 | 16 |
| .3 | 1.971 | 1.988 | 2.005 | 2.023 | 2040 | 2.058 | 2.076 | 2.095 | 2.113 | 2.132 | 18 |
| . 4 | 2.151 | 2.170 | 2.189 | 2.209 | 2.229 | 2.249 | 2.269 | 2290 | 2.310 | 2.331 | 20 |
| 8.5 | 2352 | 2.374 | 2395 | 2.417 | 2.439 | 2.462 | 2.484 | 2.507 | 2.530 | 2.554 | 23 |
| . 6 | 2.577 | 2.601 | 2.625 | 2.650 | 2.675 | 2.700 | 2.725 | 2.750 | -2.776 | 2.802 | 25 |
| . 7 | 2.828 | 2.855 | 2882 | 2.909 | 2.936 | $2.964^{\circ}$ | 2.992 | 3.021 | 3.049 | 3.078 | 28 |
| . 8 | 3.107 | 3.137 | 3.167 | 3.197 | 3.228 | 3.259 | 3.290 | 3.321 | 3.353 | 3.385 | 31 |
| . 9 | 3.418 | 3.451 | 3.484 | 3.517 | 3.551 | 3.585 | 3.620 | 3.655 | 3.690 | 3.726 | 34 |
| 2.0 | 3.762 | 3.799 | 3.835 | 3.873 | 3.910 | 3.948 | 3.987 | 4.026 | 4.065 | 4.104 | 38 |
| . 1 | 4.144 | 4.185 | 4.226 | 4.267 | 4.309 | 4.351 | 4.393 | 4.436 | 4.480 | 4.524 | 42 |
| . 2 | 4.568 | 4.613 | 4.658 | 4.704 | 4.750 | 4.797 | 4.844 | 4.891 | 4.939 | 4.988 | 47 |
| . 3 | 5.037 | 5.087 | 5.137 | 5.188 | 5.239 | 5.290 | 5.343 | 5.395 | 5.449 | 5.503 | 52 |
| . 4 | 5.557 | 5.612 | 5.667 | 5.723 | 5.780 | 5.837 | 5.895 | 5.954 | 6.013 | 6.072 | 58 |
| 2.5 | 6.132 | 8.193 | 6.255 | 6.317 | 6.379 | 6.443 | 6.507 | 6.571 | 6.636 | 6.702 | 64 |
| . 6 | 6.769 | 6.836 | 6.904 | 6.973 | 7.042 | 7.112 | 7.183 | 7.255 | 7.327 | 7.400 | 70 |
| . 7 | 7.473 | 7.548 | 7.623 | 7.699 | 7.776 | 7.853 | 7.932 | 8.011 | 8.091 | 8.171 | 78 |
| . 8 | 8.253 | 8.335 | 8.418 | 8.502 | 8.587 | 8.673 | 8.759 | 8.847 | 8.935 | 9.024 | 86 |
| . 9 | 9.115 | 9.206 | 9.298 | 9.391 | 9.484 | 9.579 | 9.675 | 9.772 | 9.869 | 9.968 | 95 |
| 3.0 | 10.07 . | 10.17 | 10.27 | 10.37 | 10.48 | 10.58 | 10.69 | 10.79 | 10.90 | 11.01 | 11 |
| . 1 | 11.12 | 11.23 | 11.35 | 11.46 | 11.57 | 11.69 | 11.81 | 11.92 | 12.04 | 12.16 | 12 |
| . 2 | 12.29 | 12.41 | 12.53 | 12.66 | 12.79 | 12.91 | 13.04 | 13.17 | 13.31 | 13.44 | 13 |
| 3 | 13.57 | 13.71 | 13.85 | 13.99 | 14.13 | 14.27 | 14.41 | 14.56 | 14.70 | 14.85 | 14 |
| . 4 | 15.00 | 15.15 | 15.30 | 15.45 | 15.61 | 15.77 | 15.92 | 16.08 | 16.25 | 16.41 | 16 |
| 3.5 | 16.57 | 16.74 | 16.91 | 17.08 | 17.25 | 17.42 | 17.60 | 17.77 | 17.95 | 18.13 | 17 |
| 6 | 18.31 | 18.50 | 18.68 | 18.87 | 19.06 | 19.25 | 19.44 | 19.64 | 19.84 | 20.03 | 19 |
| . 7 | 20.24 | 20.44 | 20.64 | 20.85 | 21.06 | 21.27 | 21.49 | 21.70 | 21.92 | 22.14 | 21 |
| . 8 | 22.36 | 22.59 | 22.81 | 23.04 | 23.27 | 23.51 | 23.74 | 23.98 | 24.22 | 24.47 | 23 |
| . 9 | 24.71 | 24.96 | 25.21 | 25.46 | 25.72 | 25.98 | 28.24 | 26.50 | 26.77 | 27.04 | 26 |
| 4.0 | 27.31 | 27.58 | 27.86 | 28.14 | 28.42 | 28.71 | 29.00 | 29.29 | 29.58 | 29.88 | 29 |
| . 1 | 30.18 | 30.48 | 30.79 | 31.10 | 31.41 | 31.72 | 32.04 | 32.37 | 32.69 | 33.02 | 32 |
| . 2 | 33.35 | 33.69 | 34.02 | 34.37 | 34.71 | 35.06 | 35.41 | 35.77 | 36.13 | 36.49 | 35 |
| .3 | 36.86 | 37.23 | 37.60 | 37.98 | 38.36 | 38.75 | 39.13 | 39.53 | 39.93 | 40.33 | 39 |
| . 4 | 40.73 | 41.14 | 41.55 | 41.97 | 42.39 | 42.82 | 43.25 | 43.68 | 44.12 | 44.57 | 43 |
| 4.5 | 45.01 | 45.47 | 45.92 | 46.38 | 46.85 | 47.32 | 47.80 | 48.28 | 48.76 | 49.25 | 47 |
| 6 | 49.75 | 50.25 | 50,75 | 51.26 | 51.78 | 52.30 | 52.82 | 53.35 | 53.89 | 54.43 | 52 |
| 7 | 54.98 | 55.53 | 56.09 | 56.65 | 57.22 | 57.80 | 58.38 | 58.96 | 59.56 | 60.15 | 58 |
| . 8 | 60.76 | 61.37 | 61.99 | 62.61 | 63.24 | 63.87 | 64.52 | 65.16 | 65.82 | 66.48 | 64 |
| . 9 | 67.15 | 67.82 | 68.50 | 69.19 | 69.89 | 70.59 | 71.30 | 72.02 | 72.74 | 73.47 | 71 |
| 5.0 | 74.21 |  |  |  |  |  |  |  |  |  |  |

If $x>5, \cosh x=1 / 2$ lasi, ond logio cosh $x=10.43431 x+0.6990-1$, correct to four significant figures.

Hyperbolic tangents [tanh $\left.x=\left(e^{x}-e^{-x}\right) /\left(e^{x}+e^{-x}\right)=\sinh x / \cosh x\right]$

| X | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | diff |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | . 0000 | . 0100 | . 0200 | . 0300 | . 0400 | . 0500 | . 0599 | . 0699 | . 0798 | . 0898 | 100 |
| . 1 | . 0997 | . 1096 | . 1194 | . 1293 | . 1391 | . 1489 | . 1587 | . 1684 | . 1781 | . 1878 | 98 |
| . 2 | . 1974 | . 2070 | . 2165 | . 2260 | . 2355 | . 2449 | . 2543 | . 2636 | 2729 | 2821 | 94 |
| . 3 | . 2913 | . 3004 | . 3095 | . 3185 | . 3275 | . 3364 | 3452 | 3540 | . 3627 | . 3714 | 89 |
| . 4 | 3800 | . 3885 | . 3969 | . 4053 | . 4136 | . 4219 | . 4301 | . 4382 | . 4462 | . 4542 | 82 |
| 0.5 | . 4621 | . 4700 | . 4777 | . 4854 | . 4930 | . 5005 | . 5080 | . 5154 | . 5227 | . 5299 | 75 |
| . 6 | . 5370 | . 5441 | . 5511 | . 5581 | . 5649 | . 5717 | . 5784 | . 5850 | . 5915 | . 5980 | 67 |
| . 7 | . 6044 | . 6107 | . 6169 | . 8231 | . 6291 | . 6352 | . 6411 | . 6469 | . 6527 | . 6584 | 60 |
| . 8 | . 6640 | . 6896 | . 6751 | . 6805 | . 6858 | . 6911 | . 6963 | . 7014 | . 7064 | . 7114 | 52 |
| . 9 | . 7163 | . 7211 | . 7259 | . 7306 | . 7352 | . 7398 | . 7443 | . 7487 | . 7531 | . 7574 | 45 |
| 1.0 | . 7616 | . 7858 | . 7699 | . 7739 | . 7779 | . 7818 | . 7857 | . 7895 | . 7932 | . 7969 | 39 |
| . 1 | . 8005 | . 8041 | . 8076 | . 8110 | . 8144 | . 8178 | . 8210 | . 8243 | . 8275 | . 8306 | 33 |
| 2 | . 8337 | . 8367 | . 8397 | . 8426 | . 8455 | . 8483 | . 8511 | . 8538 | . 8565 | . 8591 | 28 |
| . 3 | . 8617 | . 8643 | . 8668 | . 8693 | . 8717 | . 8741 | . 8764 | . 8787 | . 8810 | . 8832 | 24 |
| . 4 | . 8854 | . 8875 | . 8898 | 8917 | . 8937 | . 8957 | . 8977 | . 8996 | . 9015 | . 9033 | 20 |
| 1.5 | . 9052 | . 9069 | . 9087 | . 9104 | . 9121 | . 9138 | . 9154 | . 9170 | . 9186 | . 9202 | 17 |
| . 6 | . 9217 | . 9232 | . 9246 | . 9261 | . 9275 | . 9289 | . 9302 | . 9316 | . 9329 | . 9342 | 14 |
| . 7 | . 9354 | . 9367 | . 9379 | . 9391 | . 9402 | . 9414 | . 9425 | . 9436 | . 9447 | . 9458 | 11 |
| . 8 | . 9468 | . 9478 | . 9488 | . 9498 | . 9508 | . 9518 | . 9527 | . 9536 | . 9545 | . 9554 | 9 |
| . 9 | . 9562 | . 9571 | . 9579 | . 9587 | . 9595 | . 9603 | . 9611 | . 9619 | . 9626 | . 9633 | 8 |
| 2.0 | . 9640 | . 9647 | . 9654 | . 9661 | . 9868 | . 9674 | . 9880 | . 9687 | . 9693 | . 9699 | 6 |
| . 1 | . 9705 | . 9710 | . 9716 | . 9722 | . 9727 | . 9732 | . 9738 | . 9743 | . 9748 | . 9753 | 5 |
| . 2 | . 9757 | . 9762 | . 9767 | . 9771 | . 9776 | . 9780 | . 9785 | . 9789 | . 9793 | . 9797 | 4 |
| . 3 | . 9801 | . 9805 | . 9809 | . 9812 | . 9816 | . 9820 | . 9823 | . 9827 | . 9830 | . 9834 | 4 |
| . 4 | . 9837 | . 9840 | . 9843 | . 9846 | . 9849 | . 9852 | . 9855 | . 9858 | . 9861 | . 9863 | 3 |
| 2.5 | . 9886 | . 9869 | . 9871 | . 9874 | . 9876 | . 9879 | . 9881 | . 9884 | . 9888 | . 9888 | 2 |
| . 6 | . 9890 | . 9892 | . 9895 | . 9897 | . 9899 | . 9901 | . 9903 | . 9905 | . 9906 | . 9908 | 2 |
| . 7 | . 9910 | . 9912 | . 9914 | . 9915 | . 9917 | . 9919 | . 9920 | . 9922 | . 9923 | . 9925 | 2 |
| . 8 | . 9926 | . 9928 | . 9929 | . 9931 | . 9932 | . 9933 | . 9935 | . 9936 | . 9937 | . 9938 | 1 |
| . 9 | . 9940 | . 9941 | . 9942 | . 9943 | . 9944 | . 9945 | . 9946 | . 9947 | . 9949 | . 9950 | 1 |
| 3.0 | . 9951 | . 9959 | . 9967 | . 9973 | . 9978 | . 9982 | . 9985 | . 9988 | . 9990 | . 9992 | 4 |
| 4.0 | . 99993 | . 9995 | . 9996 | . 9996 | . 9997 | . 9998 | . 9998 | . 9998 | . 9999 | . 9999 | 1 |

If $x>5$, tanh $x=1.0000$ to four decimal ploces.
Multiples of 0.4343 [ $\left.0.43429448=\log _{10} \mathrm{e}\right]$

| X | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | 0.0000 | 0.0434 | 0.0869 | 0.1303 | 0.1737 | 0.2171 | 0.2606 | 0.3040 | 0.3474 | 0.3909 |
| 1.0 | 0.4343 | 0.4777 | 0.5212 | 0.5646 | 0.6080 | 0.6514 | 0.6949 | 0.7383 | 0.7817 | 0.8252 |
| 2.0 | 0.8686 | 0.9120 | 0.9554 | 0.9989 | 1.0423 | 1.0857 | 1.1292 | 1.1726 | 1.2160 | 1.2595 |
| 3.0 | 1.3029 | 1.3463 | 1.3897 | 1.4332 | 1.4768 | 1.5200 | 1.5635 | 1.6069 | 1.6503 | 1.6937 |
| 4.0 | 1.7372 | 1.7806 | 1.8240 | 1.8675 | 1.9109 | 1.9543 | 1.9978 | 2.0412 | 2.0846 | 2.1280 |
| 3.0 | 2.1715 | 2.2149 | 2.2583 | 2.3018 | 2.3452 | 23886 | 2.4320 | 2.4755 | 2.5189 | 2.5623 |
| 6.0 | 2.6058 | 2.6492 | 2.6926 | 2.7361 | 2.7795 | 2.8229 | 2.8663 | 2.9098 | 2.9532 | 2.9960 |
| 7.0 | 3.0401 | 3.0835 | 3.1269 | 3.1703 | 3.2138 | 3.2572 | 3.3006 | 3.3441 | 3.3875 | 3.4309 |
| 8.0 | 3.4744 | 3.5178 | 3.5612 | 3.6046 | 3.6481 | 3.6915 | 3.7349 | 3.7784 | 3.8218 | 3.8652 |
| 9.0 | 3.9087 | 3.9521 | 3.9955 | 4.0389 | 4.0824 | 4.1258 | 4.1692 | 4.2127 | 4.2561 | 4.2995 |

Multiples of $2.3026\left[2.3025851=1 / 0.4343=\log _{e} 10\right]$

| X | 0 | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | - |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0.0 | 0.0000 | 0.2303 | 0.4605 | 0.6908 | 0.9210 | 1.1513 | 1.3816 | 1.6118 | 1.8421 | 2.0723 |
| 1.0 | 2.3026 | 2.5328 | 2.7631 | 2.9934 | 3.2236 | 3.4539 | 3.6841 | 3.9144 | 4.1447 | 4.3749 |
| 2.0 | 4.6052 | 4.8354 | 5.0657 | 5.2959 | 5.5262 | 5.7565 | 5.9867 | 6.2170 | 6.4472 | 6.6775 |
| 3.0 | 6.9078 | 7.1380 | 7.3683 | 7.5985 | 7.8288 | 8.0590 | 8.2893 | 8.5196 | 8.7498 | 8.9801 |
| 4.0 | 9.2103 | 9.4406 | 9.6709 | 9.9011 | 10.131 | 10.362 | 10.592 | 10.822 | 11.052 | 11.283 |
| 3.0 | 11.513 | 11.743 | 11.973 | 12.204 | 12.434 | 12.664 | 12.894 | 13.125 | 13.355 | 13.585 |
| 6.0 | 13.816 | 14.046 | 14.276 | 14.506 | 14.737 | 14.967 | 15.197 | 15.427 | 15.658 | 15.888 |
| 7.0 | 16.118 | 16.348 | 16.579 | 16.809 | 17.039 | 17.269 | 17.500 | 17.730 | 17.980 | 18.190 |
| 8.0 | 18.421 | 18.651 | 18.881 | 19.111 | 19.342 | 19.572 | 19.802 | 20.032 | 20.263 | 20.493 |
| 9.0 | 20.723 | 20.954 | 21.184 | 21.414 | 21.644 | 21.875 | 22.105 | 22.335 | 22.565 | 22.796 |

Table 1- $J_{0}(z)$
Bessel functions

| 2 | 0 | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 1.0000 | 0.9975 | 0.9900 | 0.9776 | 0.9604 | 0.9385 | 0.9120 | 0.8812 | 0.8463 | 0.8075 |
| 1 | 0.7652 | 0.7196 | 0.6711 | 0.6201 | 0.5669 | 0.5118 | 0.4554 | 0.3980 | 0.3400 | 0.2818 |
| 2 | 0.2239 | 0.1666 | 0.1104 | 0.0555 | 0.0025 | -0.0484 | -0.0968 | -0.1424 | -0.1850 | -0.2243 |
| 3 | -0.2601 | -0.2921 | -0.3202 | -0.344 | -0.3643 | -0.3801 | -0.3918 | -0.3992 | -0.4026 | -0.4018 |
| 4 | -0.3971 | -0.3887 | -0.3766 | -0.3610 | -0.3423 | -0.3205 | -0.2961 | -0.2693 | -0.2404 | -0.2097 |
| 5 | -0.1776 | -0.1443 | -0.1103 | -0.0758 | -0.0412 | -0.0068 | +0.0270 | 0.0599 | 0.0917 | 0.1220 |
| 6 | 0.1506 | 0.1773 | 0.2017 | 0.2238 | 0.2433 | 0.2601 | 0.2740 | 0.2851 | 0.2931 | 0.2981 |
| 7 | 0.3001 | 0.2991 | 0.2951 | 0.2882 | 0.2786 | 0.2663 | 0.2516 | 0.2346 | 0.2154 | 0.1944 |
| 8 | 0.1717 | 0.1475 | 0.1222 | 0.0960 | 0.0692 | 0.0419 | 0.0146 | -0.0125 | -0.0392 | -0.0653 |
| 9 | -0.0903 | -0.1142 | -0.1367 | -0.1577 | -0.1768 | -0.1939 | -0.2090 | -0.2218 | -0.2323 | -0.2403 |
| 10 | -0.2459 | -0.2490 | -0.2496 | -0.2477 | -0.2434 | -0.2366 | -0.2276 | -0.2164 | -0.2032 | -0.1881 |
| 11 | -0.1712 | -0.1528 | -0.1330 | -0.1121 | -0.0902 | -0.0677 | -0.0446 | $-0.0213$ | +0.0020 | 0.0250 |
| 12 | 0.0477 | 0.0697 | 0.0908 | 0.1108 | 0.1296 | 0.1469 | 0.1626 | 0.1766 | 0.1887 | 0.1988 |
| 13 | 0.2069 | 0.2129 | 0.2167 | 0.2183 | 0.2177 | 0.2150 | 0.2101 | 0.2032 | 0.1943 | 0.1836 |
| 14 | 0.1711 | 0.1570 | 0.1414 | 0.1245 | 0.1065 | 0.0875 | 0.0679 | 0.0476 | 0.0271 | 0.0064 |
| 15 | -0.0142 | -0.0346 | -0.0544 | -0.0736 | -0.0919 | -0.1092 | -0.1253 | -0.1401 | -0.1533 | -0.1650 |

Table II- $\mathrm{J}_{1}(\mathrm{z})$ continued Bessel functions

| 2 | 0 | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.0000 | 0.0499 | 0.0995 | 0.1483 | 0.1960 | 0.2423 | 0.2867 | 0.3290 | 0.3688 | 0.4059 |
| 1 | 0.4401 | 0.4709 | 0.4983 | 0.5220 | 0.5419 | 0.5579 | 0.5699 | 0.5778 | 0.5815 | 0.5812 |
| 2 | 0.5767 | 0.5683 | 0.5560 | 0.5399 | 0.5202 | 0.4971 | 0.4708 | 0.4416 | 0.4097 | 0.3754 |
| 3 | 0.3391 | 0.3009 | 0.2613 | 0.2207 | 0.1792 | 0.1374 | 0.0955 | 0.0538 | 0.0128 | -0.0272 |
| 4 | $-0.0660$ | -0.1033 | -0.1386 | -0.1719 | -0.2028 | -0.2311 | -0.2566 | -0.2791 | -0.2985 | -0.3147 |
| 5 | -0.3276 | -0.3371 | -0.3432 | -0.3460 | -0.3453 | -0.3414 | -0.3343 | -0.3241 | -0.3110 | -0.2951 |
| 6 | -0.2767 | -0.2559 | -0.2329 | -0.2081 | -0.1816 | -0.1538 | -0.1250 | -0.0953 | -0.0652 | -0.0349 |
| 7 | $-0.0047$ | +0.0252 | 0.0543 | 0.0826 | 0.1096 | 0.1352 | 0.1592 | 0.1813 | 0.2014 | 0.2192 |
| 8 | 0.2346 | 0.2476 | 0.2580 | 0.2657 | 0.2708 | 0.2731 | 0.2728 | 0.2697 | 0.2641 | 0.2559 |
| 9 | 0.2453 | 0.2324 | 0.2174 | 0.2004 | 0.1816 | 0.1613 | 0.1395 | 0.1166 | 0.0928 | 0.0684 |
| 10 | 0.0435 | 0.0184 | -0.0066 | -0.0313 | -0.0555 | -0.0789 | -0.1012 | -0.1224 | -0.1422 | $-0.1603$ |
| 11 | -0.1768 | -0.1913 | -0.2039 | -0.2143 | -0.2225 | -0.2284 | -0.2320 | -0.2333 | -0.2323 | -0.2290 |
| 12 | -0.2234 | -0.2157 | -0.2060 | -0.1943 | -0.1807 | -0.1655 | -0.1487 | -0.1307 | -0.1114 | -0.0912 |
| 13 | -0.0703 | -0.0489 | -0.0271 | -0.0052 | +0.0166 | 0.0380 | 0.0590 | 0.0791 | 0.0984 | 0.1165 |
| 14 | 0.1334 | 0.1488 | 0.1626 | 0.1747 | 0.1850 | 0.1934 | 0.1999 | 0.2043 | 0.2066 | 0.2069 |
| 15 | 0.2051 | 0.2013 | 0.1955 | 0.1879 | 0.1784 | 0.1672 | 0.1544 | 0.1402 | 0.1247 | 0.1080 |

Table III- $\mathrm{J}_{2}(\mathrm{z})$

| $z$ | 0 | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.0000 | 0.0012 | 0.0050 | 0.0112 | 0.0197 | 0.0306 | 0.0437 | 0.0588 | 0.0758 | 0.0946 |
| 1 | 0.1149 | 0.1366 | 0.1593 | 0.1830 | 0.2074 | 0.2321 | 0.2570 | 0.2817 | 0.3061 | 0.3299 |
| 2 | 0.3528 | 0.3746 | 0.3951 | 0.4139 | 0.4310 | 0.4461 | 0.4590 | 0.4696 | 0.4777 | 0.4832 |
| 3 | 0.4861 | 0.4862 | 0.4835 | 0.4780 | 0.4697 | 0.4586 | 0.4448 | 0.4283 | 0.4093 | 0.3879 |
| 4 | 0.3641 | 0.3383 | 0.3105 | 0.2811 | 0.2501 | 0.2178 | 0.1846 | 0.1506 | 0.1161 | 0.0813 |

Table IV- $\mathrm{J}_{3}(\mathrm{z})$

| $z$ | 0 | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.0000 | 0.0000 | 0.0002 | 0.0006 | 0.0013 | 0.0026 | 0.0044 | 0.0069 | 0.0102 | 0.0144 |
| 1 | 0.0196 | 0.0257 | 0.0329 | 0.0411 | 0.0505 | 0.0610 | 0.0725 | 0.0851 | 0.0988 | 0.1134 |
| 2 | 0.1289 | 0.1453 | 0.1623 | 0.1800 | 0.1981 | 0.2166 | 0.2353 | 0.2540 | 0.2727 | 0.2911 |
| 3 | 0.3091 | 0.3264 | 0.3431 | 0.3588 | 0.3734 | 0.3868 | 0.3988 | 0.4092 | 0.4180 | 0.4250 |
| 4 | 0.4302 | 0.4333 | 0.4344 | 0.4333 | 0.4301 | 0.4247 | 0.4171 | 0.4072 | 0.3952 | 0.3811 |

Table V - $\mathrm{d}_{4}(\mathrm{z})$

| $\underline{2}$ | 0 | 0.1 | 0.2 | 0.3 | 0.4 | 0.5 | 0.6 | 0.7 | 0.8 | 0.9 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0.0000 | 0.0000 | 0.0000 | 0.0000 | 0.0001 | 0.0002 | 0.0003 | 0.0006 | 0.0010 | 0.0016 |
| 1 | 0.0025 | 0.0036 | 0.0050 | 0.0068 | 0.0091 | 0.0118 | 0.0150 | 0.0188 | 0.0232 | 0.0283 |
| 2 | 0.0340 | 0.0405 | 0.0476 | 0.0556 | 0.0643 | 0.0738 | 0.0840 | 0.0950 | 0.1067 | 0.1190 |
| 3 | 0.1320 | 0.1456 | 0.1597 | 0.1743 | 0.1891 | 0.2044 | 0.2198 | 0.2353 | 0.2507 | 0.2661 |
| 4 | 0.2811 | 0.2958 | 0.3100 | 0.3236 | 0.3365 | 0.3484 | 0.3594 | 0.3693 | 0.3780 | 0.3853 |


| $p$ | Jp(1) | Jp(2) | dp(3) | Jp(4) | Jp(5) | Jp(6) | Jp(7) | Jp(8) | Jp(9) | Jp(10) | dp(11) | Jp(12) | Jp(13) | Jp(14) |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\begin{aligned} & 0 \\ & 0.5 \end{aligned}$ | +.7652 +.6714 | +.2239 +.5130 | -.2601 +.06501 | -.3971 -.3019 | $\begin{aligned} & -.1776 \\ & -.3422 \end{aligned}$ | $\begin{aligned} & +.1506 \\ & -.09102 \end{aligned}$ | +.3001 +.1981 | +.1717 +.2791 | $\begin{aligned} & -.09033 \\ & +.1096 \end{aligned}$ | $\begin{aligned} & -.2459 \\ & -.1373 \end{aligned}$ | $\begin{aligned} & -.1712 \\ & -.2406 \end{aligned}$ | $\begin{aligned} & +.04769 \\ & -.1236 \end{aligned}$ | $\begin{aligned} & +.2069 \\ & +.09298 \end{aligned}$ | $\begin{aligned} & +.1711 \\ & +.2112 \end{aligned}$ |
| $\begin{aligned} & 1.0 \\ & 1.5 \end{aligned}$ | +.4401 +.2403 | +.5767 +.4913 | +.3391 +.4777 | -.06604 +.1853 | -.3276 -.1697 | -.2767 -.3279 | -.024683 -.1991 | +.2346 +.07593 | +2453 +.2545 | +.04347 +.1980 | $\begin{aligned} & -.1768 \\ & -.02293 \end{aligned}$ | -.2234 -.2047 | $\begin{aligned} & -.07032 \\ & -.1937 \end{aligned}$ | $\begin{aligned} & +.1334 \\ & -.01407 \end{aligned}$ |
| 2.0 2.5 | +.1149 +.04950 | +.3528 +.2239 | +.4861 +.4127 | + 3641 +.4409 | +.04657 +.2404 | -.2429 -.07295 | -.3014 -.2834 | -.1130 -.2506 | +.1448 -.02477 | +.2546 +.1967 | $\begin{aligned} & +.1390 \\ & +.2343 \end{aligned}$ | $\begin{array}{r} -.08493 \\ +.07242 \end{array}$ | -.2177 -.1377 | $\begin{array}{r} -.1520 \\ -.2143 \end{array}$ |
| $\begin{aligned} & 3.0 \\ & 3.5 \end{aligned}$ | +.01956 +.07186 | +.1289 +.06852 | +.3091 +.2101 | +.4302 +.3658 | +.3648 +.4100 | +.1148 +.2671 | -.1676 -.023403 | -.2911 | -.1809 -.2683 | $+.05838$ <br> $-.09965$ | +.2273 +.1294 | $\begin{aligned} & +.1951 \\ & +.2348 \end{aligned}$ | $\begin{aligned} & +.023320 \\ & +.1407 \end{aligned}$ | $\begin{aligned} & -.1768 \\ & -.06245 \end{aligned}$ |
| $\begin{aligned} & 4.0 \\ & 4.5 \end{aligned}$ | +.022477 +.03807 | $\begin{array}{r} +.03400 \\ +.01589 \end{array}$ | +.1320 .+ .07760 | +.2811 +.1993 | +.3912 +.3337 | +3576 +.3846 | $\begin{aligned} & +.1578 \\ & +.2800 \end{aligned}$ | -.1054 +.04712 | -.2655 -.1839 | -.2196 -.2664 | $\begin{aligned} & -.01504 \\ & -.1519 \end{aligned}$ | $\begin{aligned} & +.1825 \\ & +.06457 \end{aligned}$ | +.2193 +.2134 | $\begin{aligned} & +.07624 \\ & +.1830 \end{aligned}$ |
| $\begin{aligned} & 5.0 \\ & 5.5 \end{aligned}$ | $\begin{aligned} & +.032498 \\ & +.0474 \end{aligned}$ | +.097040 +.02973 | +.04303 +.02266 | +.1321 +.08261 | +.2611 +.1906 | +.3621 +.3098 | +.3479 +.3634 | +.1858 +.2856 | -.05504 +.08439 | -.2341 -.1401 | $\begin{aligned} & -.2383 \\ & -.2538 \end{aligned}$ | $\begin{aligned} & -.07347 \\ & -.1864 \end{aligned}$ | $\begin{aligned} & +.1316 \\ & +.087055 \end{aligned}$ | $\begin{aligned} & +.2204 \\ & +.1801 \end{aligned}$ |
| $\begin{aligned} & 6.0 \\ & 6.5 \end{aligned}$ | $\begin{aligned} & +.042094 \\ & +.056 \end{aligned}$ | $\begin{aligned} & +.0^{2} 1202 \\ & +.0^{3} 467 \end{aligned}$ | $\begin{aligned} & +.01139 \\ & +.035493 \end{aligned}$ | $\begin{aligned} & +.04909 \\ & +.02787 \end{aligned}$ | $\begin{aligned} & +.1310 \\ & +.08558 \end{aligned}$ | +.2458 +.1833 | +.3392 +.2911 | +3376 +.3456 | +.2043 +.2870 | -.01446 +.1123 | -.2016 -.1018 | -.2437 -.2354 | $\begin{aligned} & -.1180 \\ & -.2075 \end{aligned}$ | $\begin{array}{r} +.08117 \\ -.04151 \end{array}$ |
| $\begin{aligned} & 7.0 \\ & 7.5 \end{aligned}$ | $+.051502$ | $\underline{+.031749}$ | $+.022547$ | $+.01518$ | $+.05338$ | $\begin{aligned} & +.1296 \\ & +.08741 \end{aligned}$ | +.2336 +.1772 | +.3206 +.2759 | +3275 +.3302 | $\begin{aligned} & +.2167 \\ & +.2861 \end{aligned}$ | $\begin{aligned} & +.01838 \\ & +.1334 \end{aligned}$ | $\begin{aligned} & -.1703 \\ & -.06865 \end{aligned}$ | -.2406 -.2145 | $\begin{aligned} & -.1508 \\ & -.2187 \end{aligned}$ |
| $\begin{aligned} & 8.0 \\ & 8.5 \end{aligned}$ | $\underline{+.079422}$ | +.042218 | $+.0^{2} 4934$ | $+.024029$ | +. 01841 | $\begin{aligned} & +.05653 \\ & +.03520 \end{aligned}$ | $\begin{aligned} & +.1280 \\ & +.08854 \end{aligned}$ | +.2235 +.1718 | +.3051 +.2633 | +.3179 +.3169 | +.2250 +.2838 | $\begin{aligned} & +.04510 \\ & +.1496 \end{aligned}$ | $\begin{aligned} & -.1410 \\ & -.04006 \end{aligned}$ | $\begin{aligned} & -.2320 \\ & -.1928 \end{aligned}$ |
| $\begin{aligned} & 9.0 \\ & 9.5 \end{aligned}$ | $+.085249$ | +.02492 | +.048440 | $+.029386$ | $+.023520$ | $\begin{aligned} & +.02117 \\ & +.01232 \end{aligned}$ | $\begin{aligned} & +.05892 \\ & +.03785 \end{aligned}$ | $\begin{aligned} & +.1263 \\ & +.08921 \end{aligned}$ | $\begin{aligned} & +.2149 \\ & +.1672 \end{aligned}$ | $\begin{aligned} & +.2919 \\ & +.2526 \end{aligned}$ | $\begin{aligned} & +.3089 \\ & +.3051 \end{aligned}$ | +.2304 +.2806 | $\begin{aligned} & +.06698 \\ & +.1621 \end{aligned}$ | $\begin{aligned} & -.1143 \\ & -.01541 \end{aligned}$ |
| 10.0 | +.02631 | +.002515 | +.041293 | $+.011950$ | +.021468 | $+0^{2} 6964$ | +.02354 | +. 06077 | $+.1247$ | +. 2075 | +. 2804 | $+.3005$ | +. 2338 | +.08501 |

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[^0]:    From "Rodlo," Moy, 1944 (compliled by John M. Borsi)
    The table gives the name and defining equalion for each unit in six systems and shows factors for the conversion of all units from one systom into any other.
    Column 3, "equation," of the table lists the relationships of the physical quantities Involved. Consider, as an example, column 5,
    1 esu $=N$ omu. The conversion factor in this column can be opplied in any of the following ways:

[^1]:    Seproduced from "Treaty Serles No. 948, Telecommunication-General Radio Regulations ICairo Rovision, 1933) and final Radio Protocal (Cairo Revision, 1938) annexed to the Telecommunication Convention Madrid, 1932 Between the United States of America and Other Powers," Appendix I, pp. 234, 235 and 236, United States Government Printing Offics, Washington, D. C. References rofor to this publicotion.

[^2]:    1 With regord to tolerances for mobile stations, an attempt shall be mode to achieve, so far as
    possible, the figures specified for fixed stations.
    I A transmitter, the hamonic intensity of which is not above the figures specified but which
    nevertheless couses interference, must be subjected to speciol measures intended to eliminate
    such interforence.

    * See footnote under Frequency Tolerances, Treaty Series No. 948, Telecommunication

[^3]:    ${ }^{1}$ These waves are used only in special cases, such as standard frequency omissions.
    : Objects is used here in the optical sense of the word.
    *See Footnote under Frequency Tolerances, Treaty Series No. 948, Telecommunication.

[^4]:    * As measured with a cup anemometer, these being the average maximum for a period of five minutes.

[^5]:    * See following page.

[^6]:    * Formulas and chart (Fig. I) derived from aquations and tables in Bureau of Siandards Circular No. 74.

[^7]:    - Nominal bare diometer plus moximum additions.

    For additional data on copper wire, see poges 35, 36, and 126.

[^8]:    - $\phi$ is negative for $\Delta f$ positive, and vice versa.

[^9]:    * Carrent copacity af 1000 amperes per square inch. For other current densilies, multiply by lcurrent donsind/1000.
    $\dagger$ hierloyer insulation is usually Kroft poper.
    See olso poge 60.

[^10]:    * Sections on Electrodes, Characteristics, and Application Notes prepared by I. E. Lempert, Allon B. Dumont Laboratories, lnc.

[^11]:    - All potentials are with respect to the cathode except when otherwise indicated.

[^12]:    Consultation with applicable service laboratory's electron tube group is recommended before application In equipment.
    $\$$ Diode Pentode.

[^13]:    - Compiled by Edward J. Content, consulting engineer.

[^14]:    Noie: These curves show the desirable ratio of the reverberation time for varlous frequencles to the reverberation fime for 512 gycles. The desirable reverberation time far ony frequency between 60 and 8000 cycles moy be found by multiplying the reverberation time at 512 cycles (from Rg. 21 by the number in the vertical scale which correspands to the frequency chosen.

[^15]:    - The noise-reduction coelficient is the overage of the coefficients of frequencles from 256
    to 2048 cycles inclusive, given to the nearest 5 percent. This average coelficient is recommended for use in comparing materials for noise-quieting purposas as in offices, hospitals, banks, corridors, atc.

[^16]:    Note: Thls table for use for medium and long-wave propagation with Norton's, van der Pol's, Eckersley's, or other developments of Sommerfeld propagation formulos.

    * For more exact methods of computation see Terman, F. E., Radia Engineers' Handbook. Sec. 10; or Norton, K. A., The Colculotion of Ground wave Field Intensities Over o Finitely Conducting Spherical Eorth. Proc. I.R.E., vol. 29, p. 623 (December, 19411.

[^17]:    * See Burrows, C. R., Radio Propagation over Plone Eorth-Field Strength Curves. Bell System Tech. Jour, val. 16 Ulanuary 1937.
    \$937. Norton, K. A., The Effect of Frequency on the Signal Ronge of on Ulira-High Frequency Rodio Station. FCC Mimeo Repart 48466 (March 20, 19411.

[^18]:    * See olso section on Wire Telephony-Noise and Noise Meosuroment.

[^19]:    - Bosed on Mesny, R., Radio-Electricité Gënérale.

[^20]:    * for information on the eflect of some practical current distributions on fiold intonsitios see Gihring, H. E. and Brown, G. H, General Considerations of Tower Antennas for Broadcast Use. Proc. IR.E., vol. 23, p. 311 Upril, 19354.

[^21]:    

[^22]:    * For more complete information see Harper, A. E. Rhombic Antenna Design. D. Van Nostrand Co. (1941).

[^23]:    * Noter Do not interpolato in thls column.
    

